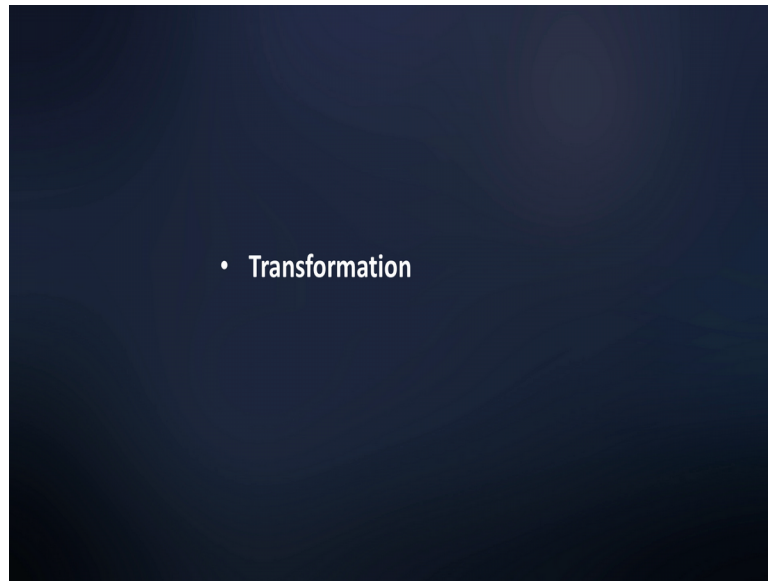


Computer Methods of Analysis of Offshore Structures
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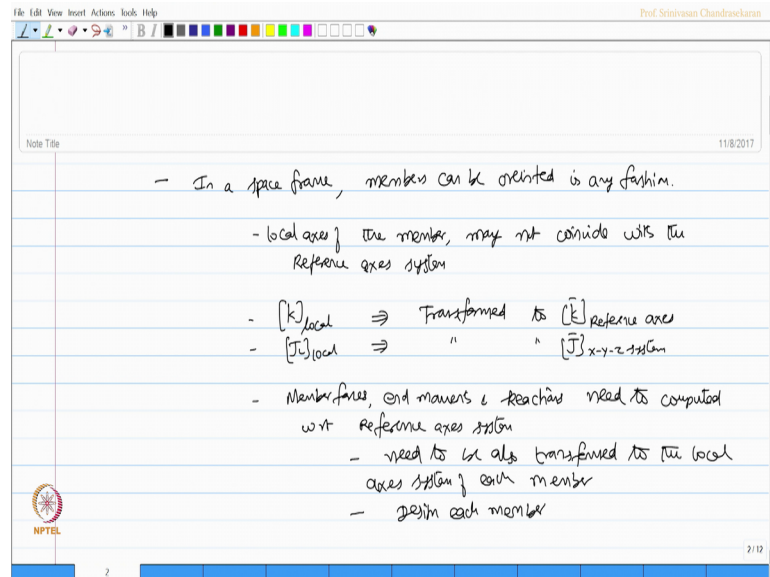
Module - 01
Lecture - 21
3d structures: Transformation matrix

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Friends, welcome to the 21st lecture in module 1, we have been discussing about the computer methods of structural analysis applied to offshore structures, we have been taking examples of beam element truss element with planar and non planar members involved in a structural system on a 2 dimensional analysis. In the last lecture, we introduced the 3 dimensional analysis and the member stiffness matrix of size 12 by 12, we derived the stiffness matrix for the member which is on a space frame.

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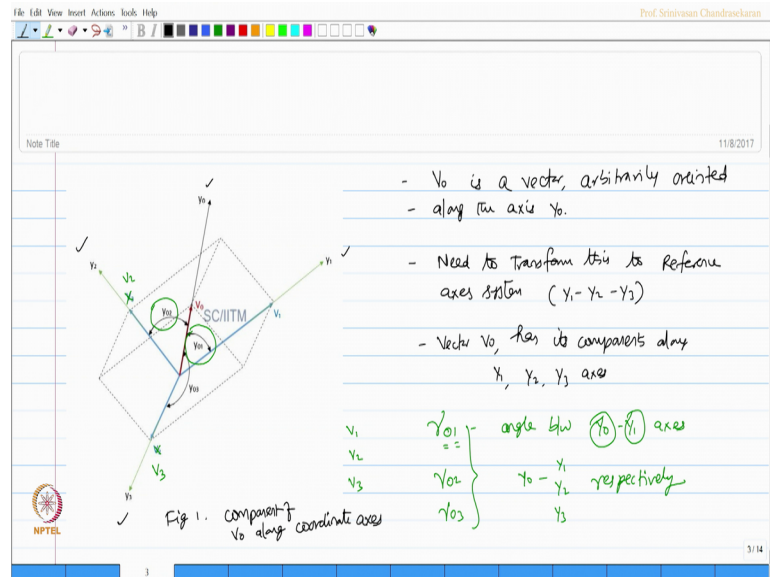


Now, in this lecture we are going to discuss about how do we obtain or derive the transformation matrix. Now the 1st question comes, why transformation matrix is important. Very simple reason we all know that, in a space frame members can be oriented in any fashion to be very specific; the local axes of the member may not coincide with the reference axes system. In such situation whatever stiffness matrix we derived in the last lecture they are considered to be local and they need to be transformed which is with respect to the reference axes system.

Secondly, whatever load which is applied on the local need to be also transformed to get J bar with reference to the x-y-z system. Most importantly, the member forces end moments and reactions need to be computed with respect to the reference axes system, but need to be also transformed to the local axes system of each member the question is why this is required this is required because to design the member .

So, now the question is in a space frame members oriented in different formats and alignments, how **do** we transform them to match the alignment with the top the reference axes system. So, that is what we are going to discuss today in this lecture which we call as a transformation matrix.

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One can see here, V_0 is a vector which is arbitrarily oriented; this V_0 vector is oriented along the axis Y_0 . Now I want to transform this, this to the reference axes system for example, we have considered the reference axes here has Y_1, Y_2 and Y_3 axis which also has an orientation with reference to X, Y, Z plane, but let us say I want to transformed this to Y_1, Y_2 axis we can see here Y_1, Y_2 and Y_3 axis are different from that of the Y_0 axis along which the vector or the member is oriented.

Now, let us defined the angles V_0 is a vector which is oriented with respect to some set of orthogonal axis Y_1, Y_2, Y_3 . Now vector V_0 has its components, along Y_1, Y_2 and Y_3 axis. To know it is component we should like to know the inclination of this vector or position of this vector with reference to these 3 axis.

Let us, look at this symbol this is γ_{01} . So, for example, γ_{01} is the angle between the Y_0 and Y_1 axis. So, the first let us stands for Y_0 and the second subscript stands for Y_1 , with that algorithm please look at this angle γ_{02} this is an inclination with respect to Y_0 and Y_2 . Similarly, Y_0 and Y_3 will give γ_{03} . So, $\gamma_{01}, \gamma_{02}, \gamma_{03}$ are now the angles between the Y_0 axis and Y_1, Y_2 or Y_3 respectively.

Now, the corresponding components of them along Y_1, Y_2, Y_3 are marked in blue color and they are V_1, V_2 and V_3 . So this should be V_3 , this is V_3 and this is V_2 along V_1, V_2 and V_3 . So one can easily find the following relationship valid V_1, V_2 and V_3 will be

$V_0 \cos \gamma_{01}$, $V_0 \cos \gamma_{02}$ and $V_0 \cos \gamma_{03}$, I call this equation number as 1.

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The image shows a digital whiteboard with the following content:

$$\left. \begin{aligned} V_1 &= V_0 \cos \gamma_{01} \\ V_2 &= V_0 \cos \gamma_{02} \\ V_3 &= V_0 \cos \gamma_{03} \end{aligned} \right\} \quad (1)$$

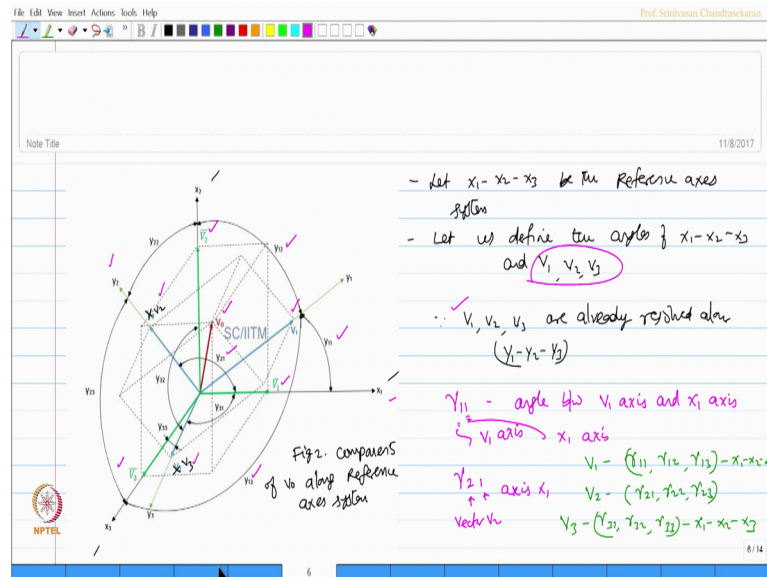
whereas $(\gamma_{01}, \gamma_{02}, \gamma_{03})$ are defined as angles b/w
the vector V_0 (or the axis Y_0 , to which vector is aligned)
and the coordinate axes Y_1, Y_2, Y_3 respectively

In Eqn, $\cos(\gamma_{01}), \cos(\gamma_{02}), \cos(\gamma_{03})$ are called
Direction Cosines

Whereas γ_{01} , γ_{02} , and γ_{03} are defined as angles between the vector V_0 or the axis Y_0 to which vector is placed and the coordinate axes Y_1 , Y_2 and Y_3 respectively, now the above equation 1 the terms $\cos \gamma_{01}$, $\cos \gamma_{02}$, $\cos \gamma_{03}$ are called direction cosines.

Now we have resolved the vector V_0 along Y_1 , Y_2 , Y_3 which are some set of coordinate axes. Let us, now resolved this or transform this to a standard reference axes system which is xyz. So, let us now draw a figure where the standard reference axes let X_1 , X_2 and X_3 be the reference axes as shown in the figure.

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So, let us name the figures this will be figure 1 which has component of vector V_0 along coordinate axes; let this be figure 2, where we are looking at components of V_0 along reference axes ok.

Let X_1, X_2, X_3 be the reference axis; obviously, this vector as a mistake this was V_3 and this was V_2 , I borrowed this figure from the previous. Now let us define the angles, now let us define the angles of X_1, X_2 and X_3 and the vector V_0 or vector V_1, V_2 and V_3 because vectors V_1, V_2, V_3 are already resolved along Y_1, Y_2, Y_3 is it not we already have this we can see here, V_1 we have V_1 we have, V_2 we have along Y_2 and V_3 we have along Y_3 .

So, now I am interested in knowing, what is the angle between this vectors and the reference axes system X_1, X_2, X_3 . So, let us mark those angles let us say γ_{11} is the angle between the V_1 axis and X_1 axis. So, the first subscript stands for V_1 axis the second subscript stands for the X_1 axis. So, you can see here γ_{11} . Similarly, the angle between V_1 and X_2 should be γ_{12} .

Similarly, the angle between V_1 and X_3 there is this angle it will be γ_{13} by that logic, I can now find or defined angles between the vector V_2 with that of X_1, X_2 and X_3 like for example, V_2 vector is here, X_1 is here the angle between V_2 and X_1 is here. So, the first subscript in γ_{21} this stands for vector V_2 and this stands for axis X_1 and so on.

So, this figure explains or defines all angles between the respective vectors V_1 , V_2 and V_3 which were resolved along some coordinate axes system Y_1 , Y_2 , Y_3 when a push in vector V_0 is placed in space. Now I am transforming that vector or those vectors V_1 , V_2 and V_3 which were initially transformed through direction cosines along some reference axes Y_1 , Y_2 , Y_3 are now being transformed to the standard set of reference axes X_1 , X_2 , X_3 , so you will like to do this.

So, now we can still write a statement that V_1 vector makes an angle γ_{11} , γ_{12} and γ_{13} with reference to X_1 , X_2 , X_3 . Similarly, V_2 vector makes an angle γ_{21} , γ_{22} , γ_{23} with reference to X_1 , X_2 , X_3 . Similarly, V_3 makes an angle γ_{31} , γ_{32} , γ_{33} with reference to X_1 , X_2 , X_3 is that clear, so this figure is clear.

Now, let us write this connecting equation or the matrix which connects the transformed components of these vectors. So, what is the transformed component of this vector along X_1 , X_2 , X_3 that is very clear here the transformed components are V_1 bar, V_2 bar and V_3 bar. So, now I want to connect V_1 bar, V_2 bar and V_3 bar with that of V_1 , V_2 , V_3 . Because V_1 , V_2 , V_3 we already know, we can see from this previous equation V_1 , V_2 and V_3 are known provided the direction cosines are measurable and the vector is known to us ok.

So, now having now V_1 , V_2 , V_3 we are interested in finding out the components of those vectors resolved or transformed along the standard reference axes system X_1 , X_2 , X_3 which I call as V_1 bar, V_2 bar and V_3 bar.

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Note Title 11/8/2017

Find connecting components along $(x_1-x_2-x_3)$ system, for the known components of the vector V_0 along $(y_1-y_2-y_3)$ are given below:

$$\begin{cases} \bar{V}_1 = V_1 \cos \gamma_{11} + V_2 \cos \gamma_{21} + V_3 \cos \gamma_{31} \\ \bar{V}_2 = V_1 \cos \gamma_{12} + V_2 \cos \gamma_{22} + V_3 \cos \gamma_{32} \\ \bar{V}_3 = V_1 \cos \gamma_{13} + V_2 \cos \gamma_{23} + V_3 \cos \gamma_{33} \end{cases} \quad (2)$$

Let $C_{ij} = \cos \gamma_{ij}$ where i represents $y_1-y_2-y_3$ axes system
 j represents $x_1-x_2-x_3$ axes system

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So, now we can write the governing equation which connects these 2 as below. So, the set of equations connecting components along X_1, X_2, X_3 system for the known components of the vector V_0 along Y_1, Y_2, Y_3 are given below.

So, I should say V_1 bar, V_2 bar, V_3 bar or simply $V_1 \cos \gamma_{11}$ plus $V_2 \cos \gamma_{21}$ plus $V_3 \cos \gamma_{31}$, because these angles are defined with respect to X_1 axis where V_1 is the component similarly with reference to X_2 axis I can still write $V_1 \cos \gamma_{12}$ plus $V_2 \cos \gamma_{22}$ plus $V_3 \cos \gamma_{32}$, because these are all with reference to axis 2 that is X_2 . Similarly, with reference to axis X_3 , 13, 23 and 33 I call this as equation set 2.

Now, let us replace this cos term by a letter C. So, let C_{ij} is actually equal to $\cos \gamma_{ij}$ where, i represents the Y_1, Y_2, Y_3 axes system and j represents X_1, X_2, X_3 axis system; by that logic I can convert equation 2 in much a simpler form as given here.

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$$\begin{bmatrix} \bar{V}_1 \\ \bar{V}_2 \\ \bar{V}_3 \end{bmatrix} = \begin{bmatrix} C_{11} & C_{21} & C_{31} \\ C_{12} & C_{22} & C_{32} \\ C_{13} & C_{23} & C_{33} \end{bmatrix} \begin{bmatrix} V_1 \\ V_2 \\ V_3 \end{bmatrix} \quad (3)$$

$$\begin{bmatrix} \bar{V} \end{bmatrix} = [C_s]^T [V] \quad (4)$$

where $[C_s] = \begin{bmatrix} C_{11} & C_{12} & C_{13} \\ C_{21} & C_{22} & C_{23} \\ C_{31} & C_{32} & C_{33} \end{bmatrix} \quad (5)$

So, I can now write V_1 bar, V_2 bar and V_3 bar in a matrix form as $C_{11}, C_{21}, C_{31}, C_{12}, C_{22}, C_{32}$ and $C_{31}, C_{32}, C_{13}, C_{23}$ and C_{33} . Connecting these two V_1, V_2 and V_3 call equation number 3 I can now write a comprehensive equation of this as saying V bar is expressed as C of the space transpose of that of V where C_s matrix refers to $C_{11}, C_{12}, C_{13}, C_{21}, C_{22}, C_{23}, C_{31}, C_{32}, C_{33}$.

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It also verifies that

$$[C_s]^T = [C_s]^{-1} \quad (5)$$

Friends, interestingly it also verifies that you can verify this individually verifies that C space frame transpose matrix is as same as C space frame inverse I call this as equation number 4 5.

So, now we have very comprehensive equation which is equation 4, \bar{V} the component of the vector, V_0 along reference axes system X_1, X_2, X_3 is simply given by the transformation matrix for the direction cosine matrix; the direction cosine matrix C_s , if I know the comprehensive of this vector along any predefined coordinate axes system Y_1, Y_2, Y_3 . Please understand here one important statement this refers to a reference axes system, this refers to some coordinate axes system.

The important point to note that is both of them do not refer to the axis of the original vector V_0 is it not that is very important. Having said this instead of a vector let us considered as a member. So, I want to now find the member transformation matrix.