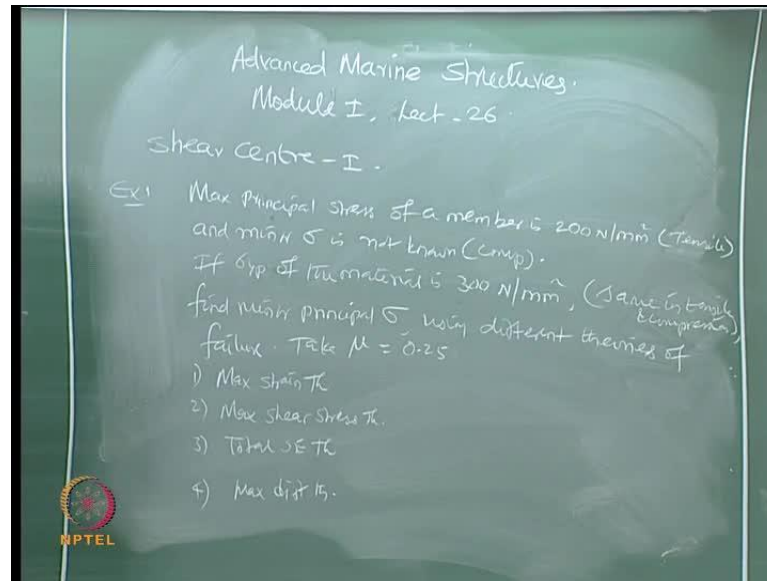


**Advanced Marine Structures**  
**Prof. Dr. Srinivasan Chandrasekaran**  
**Department of Ocean Engineering**  
**Indian Institute of Technology Madras**

**Lecture - 26**  
**Shear centre-I**

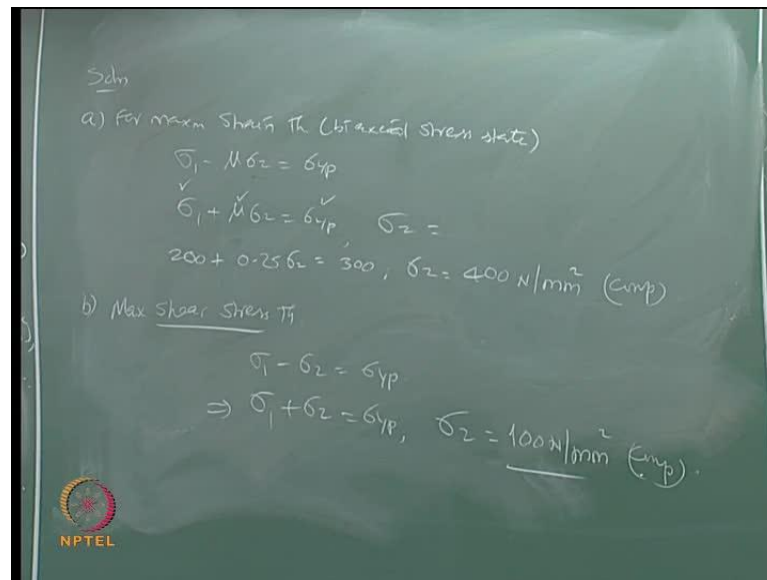
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So we will discuss now one another important topic which is, shear centre. In the last lecture we discussed about, or last few lectures discussed about plastic analysis and design. And we also discussed about theories of failure as discussed by, as given by 5 different theories. For understanding let us go quickly to numerical examples, these are examples of different theories. Then we will move on to the shear centre which is an important topic as far as thin asymmetric sections are concerned. We will tell you why it is important and that is how that is predominantly important as a designer for marine structures, let us see that. But before that let us do one design example using different theories. Let us say the maximum principal stress of a member is given as 200 newton per mm square tensile and the minor is maximum, and the minor sigma is not known, but the nature is compulsory. If  $\sigma_{yp}$  of the material is 300 newton per mm square same in tensile and compression, find the minor principal stress using different theories of failure. Now take  $\mu$  for the material as 0.25, so following theory should be used; Maximum strain theory, Maximum shear stress theory, Total strain energy theory and Maximum distortion.

Read the problem think it for few minutes. Let us see how we will solve this problem. You have the governing equation of all these theories I gave you in the previous lectures please turn them back and be ready with the equations, and read the problem what is given and what is asked how we handle this problem.

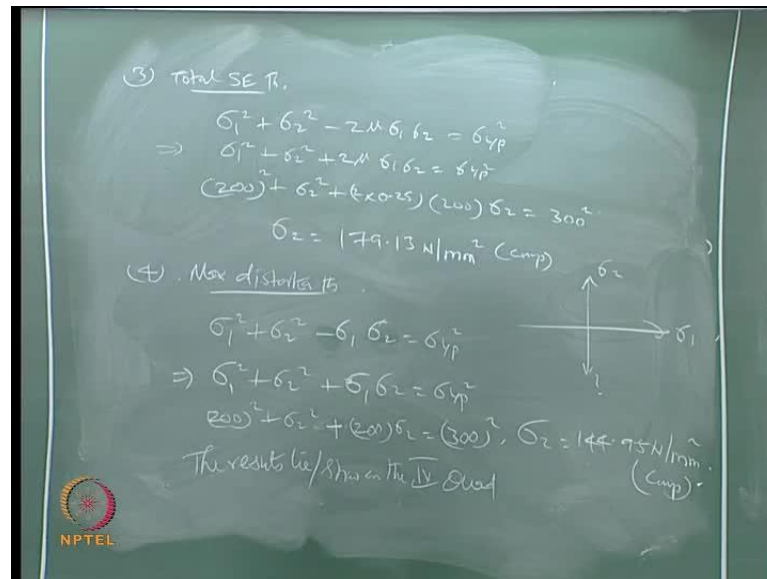
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So let us say for maximum strain theory in a biaxial stress state, see I have deliberately made these 2 stresses of different nature. Because 1 is tensile other is compression, so I am looking for quadrants of 2 and 4; so it can give me a good difference. I deliberately made this as a choice for a problem. So let us see how they are vary, so for the maximum strain theory for the given biaxial stress strain what is the controlling equation? Can you give me the equation? Is this the equation?  $\sigma_1 - \mu \sigma_2 = \sigma_{yp}$  and already we know  $\sigma_2$  is the compression that is the indication given in the problem. So, I should say  $\sigma_1 + \mu \sigma_2 = \sigma_{yp}$ , now is  $\sigma_{yp}$  because this is minus of minus and you know  $\sigma_1$ , you know  $\sigma_{yp}$ , you know  $\mu$ , can you find  $\sigma_2$ ?

So, this is 200 plus 0.25 of  $\sigma_2$ , is 3, right? Which gives me,  $\sigma_2$  plus 400 that is 1. Answer is compression. Maximum shear stress theory, According to this theory what is the control equation?  $\sigma_1 - \sigma_2 = \sigma_{yp}$ , is it not? So for my problem  $\sigma_2$  being compressive, which gives me  $\sigma_2$  as simply 100 compressions, is it not? Take away this.

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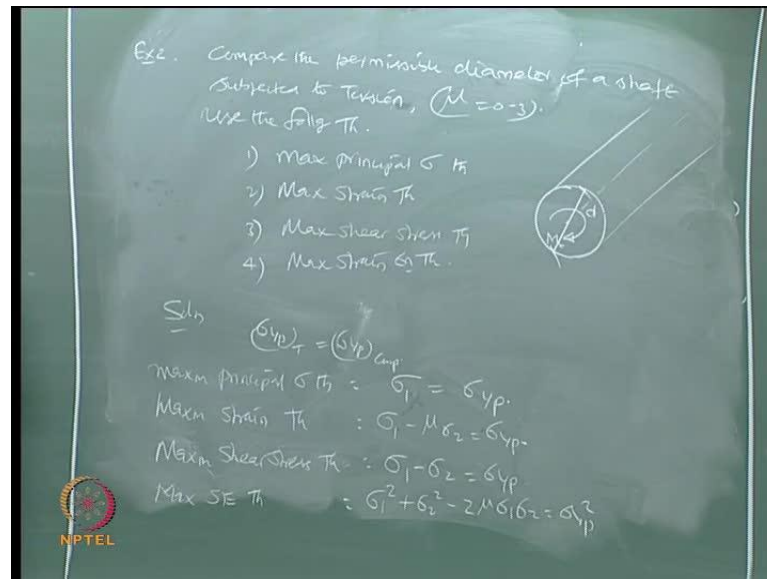


Now total strain energy theory, what is the control equation?  $\sigma_1^2 + \sigma_2^2 - 2\mu \sigma_1 \sigma_2 = \sigma_{yp}^2$ , am I right? You know everything solve the quadratic and get me  $\sigma_2$ . So, for  $\sigma_2$  being compressive  $\sigma_1^2 + \sigma_2^2 + 2\mu \sigma_1 \sigma_2 = \sigma_{yp}^2$ . Now what is  $\sigma_1$  value? 200 then you find  $\sigma_2$ ; solve the quadratic and get this is 179.13.

Look at the fourth 1, maximum distortion theory. According to this theory the control equation is  $\sigma_1^2 + \sigma_2^2 - \sigma_1 \sigma_2 = \sigma_{yp}^2$ . Ya minus  $\sigma_1 \sigma_2$  good is  $\sigma_{yp}^2$ , is it not? So,  $\sigma_1^2 + \sigma_2^2 + \sigma_1 \sigma_2 = \sigma_{yp}^2$ . So, 200 square plus  $\sigma_2^2$  square, so here plus 200 of  $\sigma_2$  is, 300 square which gives me  $\sigma_2$  as 144.95 newton per meter square compression; and if we see the discrepancy since  $\sigma_2$  is compressive all these values are in which quadrant of your stress theory?  $\sigma_2$  is compressive.

$\sigma_2$  is compressive,  $\sigma_1$  is tensile, which quadrant? This is strain  $\sigma_1$   $\sigma_2$   $\sigma_2$  is negative, so I am in the fourth quadrant. So, as I expected the variation between the values are significantly high. So, designer it's very large. Second example, this is more alarming I will show you an example now.

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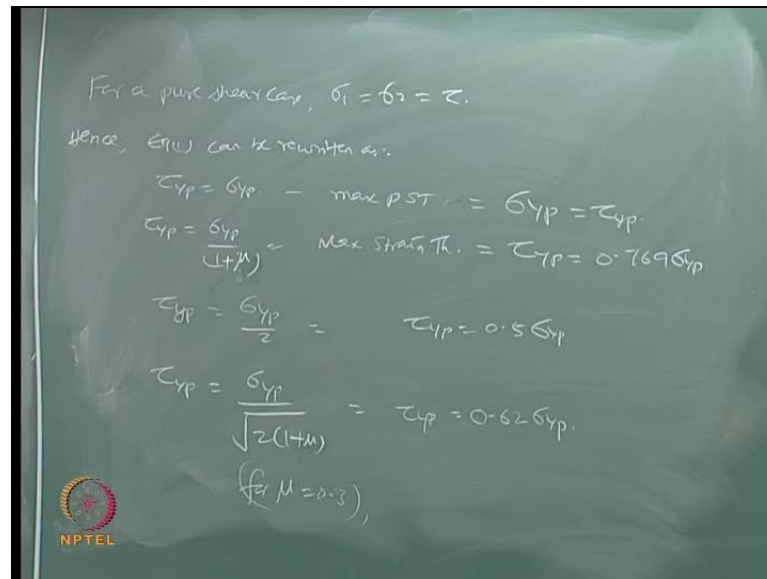


Compare the permissible diameter of a shaft. I have a shaft, I want to check the diameter of the shaft, subjected to torsion. Taking  $\mu$  as point you use the following theories; 1. Maximum principle stress theory, 2. Maximum strain theory, 3. Maximum shear stress theory, 4. Maximum strain energy theory. So, I have a shaft whose diameter is  $d$ . The shaft is subjected to torsion at twisting moment  $M$  at twisting moment,  $M$  t. Let say this diameter of my shaft. I want to estimate the diameter of the shaft, the design problem based on the following things; so all should give me the same value, more or less similar values.

Let us see what happens when we use the different theories. Let us say for example take up the maximum principal stress theory. Before that let us say  $\sigma_y$  in tension is same as  $\sigma_y$  in compression. So the maximum principle stress theory say  $\sigma_1$  is equal to  $\sigma_y$ , is it not? That exact irrespective of other status stresses.

When the maximum principal stresses reaches yield point failure has started, that is what the theory says. According to maximum strain theory what is the equation? Maximum strain theory, is this the equation? Maximum strain theory, then maximum shear stress theory? There is stress caused in this equation? Is it right or wrong? Then according to the maximum strain energy theory, ok?

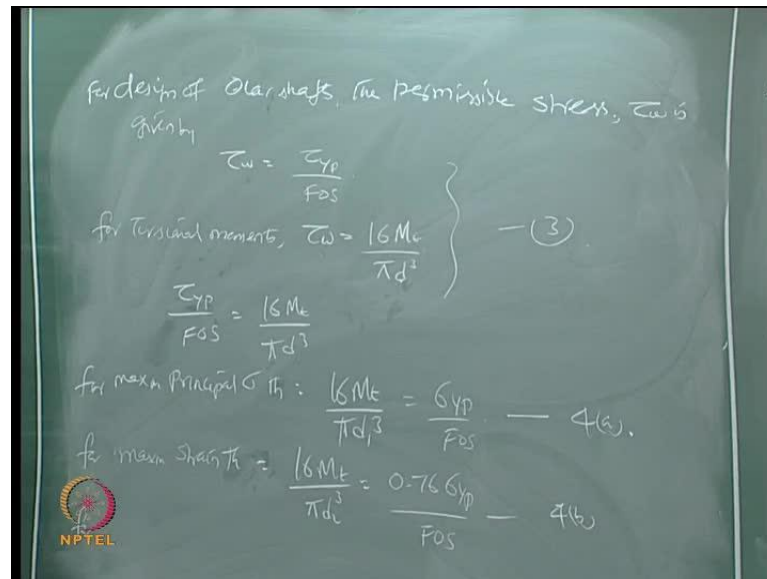
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Now for a pure shear case as in this problem  $\sigma_1$  equal  $\sigma_2$  will be equal to  $\tau$ , the shear stress, pure shear case hence the equations 1 can be rewritten as  $\tau_{yp}$  is  $\sigma_{yp}$  this for the maximum principal stress theory,  $\tau_{yp}$  is  $\sigma_{yp}$  by 1 minus  $\mu$  that is for the maximum strain theory. Now, in this case they are of different nature so it will become 1 plus. Therefore different nature it will become 1 plus.

So for maximum shear stress theory  $\tau_{yp}$  will be  $\sigma_{yp}$  by 2, and for this case  $\tau_{yp}$  will be  $\sigma_{yp}$  by root of 2 of 1 plus  $\mu$ , is that ok? Again different nature, when different nature so plus 1; root because I am talking about squares. Now I have the value of  $\mu$ , I have the value of  $\mu$  can you just tell me what are these values, equivalently I am substituting for  $\mu$  because  $\mu$  already have has 0.3, can you tell me these value? So this is going to be simply  $\sigma_{yp}$  no change in this that is  $\tau_{yp}$ , and in this case  $\tau_{yp}$  will be equal to 0.769, and in this case  $\tau_{yp}$  is equal 0.5, and in this case  $\tau_{yp}$  is equal to how much? Point, remove this.

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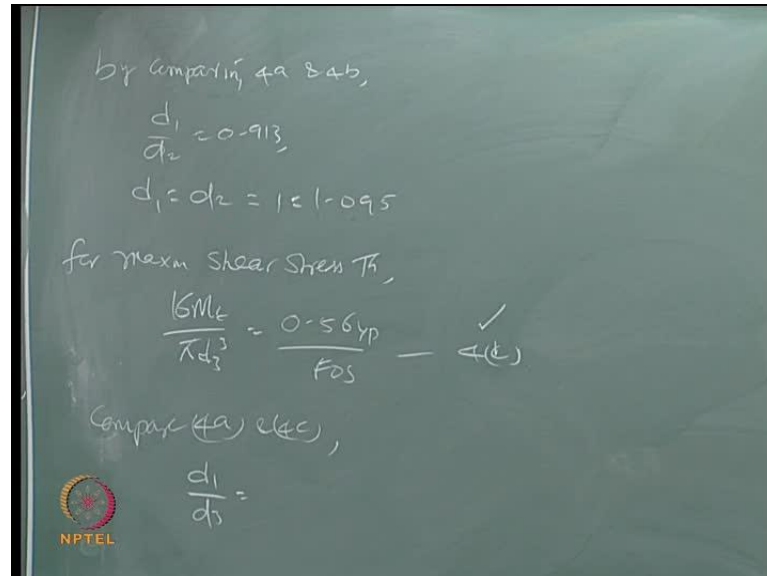
In the case of design of circular shaft, the permissible stress which I say  $\tau_w$  is given by  $\tau_w = \frac{\tau_{yp}}{FOS}$  by some factor, is it not? For torsional moments  $\tau_w$  is also equal to shear stress by torsion, for torsional moments  $\tau_w$  is also equal to  $\frac{16 M_t}{\pi d^3}$ .

How do you get this? How do you get this? What is the control equation by bending? What is the control equation for torsion? What is the control equation for bending?  $M = \sigma y I$  by a stress by  $y$  is  $e$  by  $r$  control equation for torsion?  $\tau = \frac{T r}{J}$  is stress by  $r$  max is it not? So,  $J$  is polar moment of inertia. For the circular shaft what is polar moment of inertia, polar moment of inertia for a circular shaft, what is moment of inertia for a circular shaft?  $J = \frac{\pi d^4}{32}$ , what is polar moment of inertia?  $J = \frac{\pi d^4}{32}$  half of that, I am talking about  $y$  also which is  $d/2$ . I get 16, is that clear. So, now let us find out this equation, I call this as equation number, let me call this equation number let me call this equation number; you missed out some number in between this is 3 that can be 2 or whatever may be. Now I have  $\tau_w = \frac{\tau_{yp}}{FOS} = \frac{16 M_t}{\pi d^3}$  or simply  $\frac{16 M_t}{\pi d^3} = \frac{\tau_{yp}}{FOS}$ . Now this  $\tau_w$  is different for different theories, for principle stress it is  $\sigma_{yp}$  directly, for maximum strain it is 0.769, for the other theory 0.5 and the fourth one 0.62. I will keep on substituting and find keep on different diameter and compare; can you give me what is the diameter for the first theory?

So, let us write down the equation first  $\frac{16 M_t}{\pi d^3} = \frac{\sigma_{yp}}{FOS}$  for maximum principle stress theory,  $\frac{16 M_t}{\pi d^3} = \frac{\sigma_{yp}}{FOS}$  for me to understanding;  $d^3 = \frac{16 M_t FOS}{\pi \sigma_{yp}}$  because

equation 4 a, is it ok? For maximum strain theory  $16 M \text{ by pie } d^2 \text{ cube is } 0.76 M$ , I am talking  $0.76 \sigma_{yp}$  by factor, 4 b.

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For let us compare these 2 then you compare 4 a and 4 b get me the relationship between  $d_1$  and  $d_2$  by comparing 4 a and 4 b. See, say if I bring this multiply these 2 equations can be equated, is not? So, you have got a ratio of  $d_1$  by  $d_2$ , is it not?

Give me the ratio. So,  $d_1$  by  $d_2$  will become 0.91 that it is  $d_1$  is to  $d_2$ , 1 is to 1.909 is it right? 1.909. The diameter suggested by the maximum strain theory for this problem is about 1 point about 10 percent more than the diameter suggested by the maximum principle that is why what we are meaning. The diameter recommended by the maximum strain theory is about 10 percent more, about 10 percent more than the diameter suggested by the maximum principle stress theory.

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for max shear stress Th,

$$\frac{16Mc}{\pi d_3^3} = \frac{0.5 \sigma_{yp}}{FOS} \quad \text{--- (4c)}$$

Compare (4a) & (4c),

$$\frac{d_1}{d_3} =$$

$d_1 = d_3 = 1.26$

NPTEL

Now let us do for the third case, for the third theory maximum shear stress theory. Now let us say  $16 M t$  by  $\pi d^3$  cube and this was 0.5, I call this 4 c. Now compare 4a and 4 c and comparing this now can you give me the ratio between  $d_1$  by  $d_3$ . So this says  $d_1$  is to  $d_3$  is 1 is to the proportion is 1.26, is very high. It is about 26 percent higher when you use this theory for design. I will rub this I will rewrite here.

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for total strain energy Th,

$$\frac{16Mc}{\pi d_4^3} = \frac{0.62 \sigma_{yp}}{FOS} \quad \text{--- (4d)}$$

Compare (4d) with (4a),

$$d_1 = d_4 = 1.17$$

$$d_1 = d_2 = d_3 = d_4 = 1.09 = 1.26 = 1.17$$

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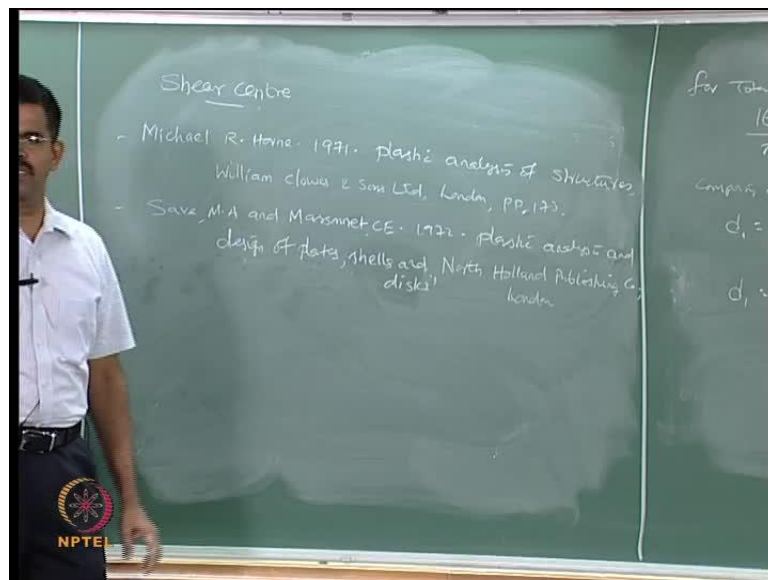
For strain energy theory please do not mind me shortened here, you can write it for total strain energy theory  $16 M t$  by  $\pi d^4$  cube is 0.62 is it? I call this as 4 d.



Now, comparing equation 4 d with 4 a we get  $d_1$  is to  $d_4$  and 1 is to 17 percent. I should say  $d_1, d_2, d_3, d_4$  are in the relationship of 1 is to 1.09, 1.26, and 11.17. So, that is amazing, simple theory simple problem give different dimensions for the design. This is where the design is getting deviated by different design engineers by following the same analysis and design, for example plastic design. So, if we chose any wrong theory applicable to your problem you will land up in a wrong diameter, simple example.

Ok, so you have to be very careful in understanding the failure behavior based on the theories. Let us talk about shear centre, any questions here? Are you understanding the importance of this problem, we have demonstrated how the diameter selection can be chosen can be varying by using different theories on a simple problem like this. So, even there exist uncertainties on the theory suggested by the literature. Therefore, our original argument of limit state design or ultimate living states being probabilistic non deterministic is all justified; because we cannot land up in a single unique answer even the theory suggest different solutions as we see for this example. So, it is not that simple. Your design is always a close form answer in unique number, no. Let us move on to the next topic which we are interested now to discuss which is very closely relevant to marine structures is shear centre.

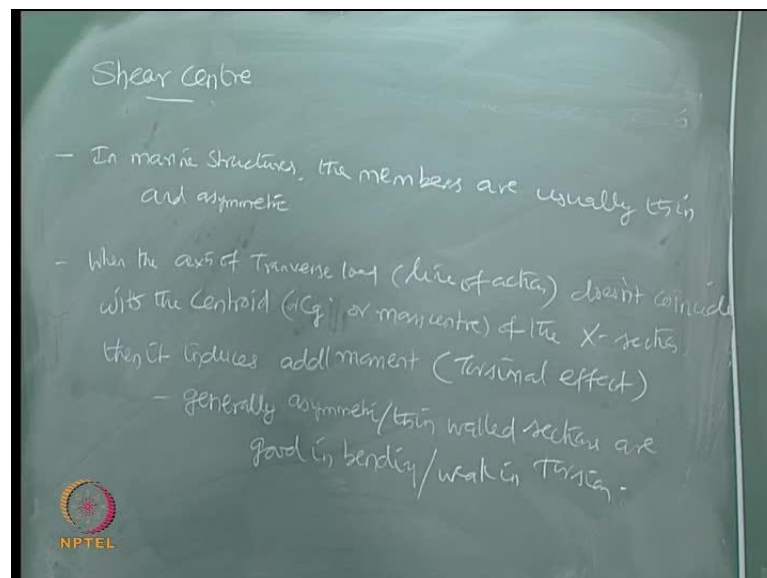
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I can give a very interesting reference for the plastic analysis of structures, please read this book if you have time; Michael R Horne 1971 "Plastic analysis of structures" all the

relevance of this theory is all discussed by this author, William Clowes and Sons limited, London, pp173. A good book for plastic analysis of structures. One more book is there, it is slightly of a higher order but still "Save, M.A and Marsnet" there is double n, "Plates and shells, North Holland publishing, I think this is design of plates shells and discuss if you remember correctly. North Holland and publishing city London, these two reference are very good for plastic analysis and design. You can go through them, of course these examples are not applicable directly to marine structures, but you can still find the members which are designed using in theory and discrepant of the different theory are earlier discussed here so you can read it.

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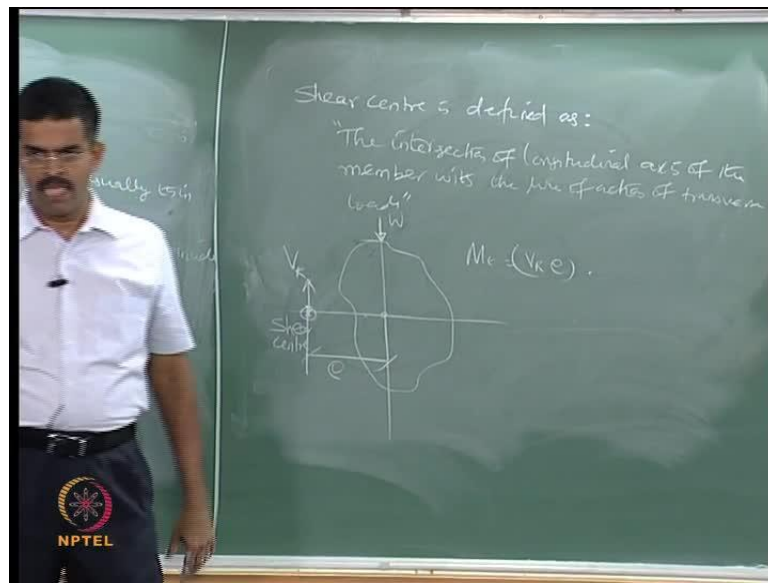


Now the question comes, what is shear centre, how it is relevant in main structures? Now the most important factor in marine structure, the members are usually thin and asymmetric, why thin? Because we are talking about  $y$  and  $c$  we do not want to increase the payload, we do not want to increase the weight during installation etc we say thin. Thin doesn't mean that it is very very thin, the thickness of the material in comparison to diameter is very small. Thus  $d$  by  $t$  ratio is very small, not thin means that we are using a paper it will be like a paper, not like that. The thickness of the member compared to this diameter is very small, because we want large diameter to  $y$  and  $c$  effects that is different and we want to for storage for blasting there are daily applications seen in the previous lectures of this module. So, we understand why we are talking about large diameter structures. We have a specific choice of material or member which has thin cross

sections, means thickness of the material of the member is less. We have asymmetric cross sections, why, because we are working on different geometric shapes which can effectively disperse the wave loads, ok, so asymmetric.

Now, when the axis of transverse loads axis means the line of action, I should say the line of action of transverse load does not coincide with the centroid or centroid gravity or mass centre of the cross section then it induces additional movement and this movement will create torsional effect in the section, that is the problem. And generally asymmetric section, thin sections are good in bending but very weak in torsion. Now the difference between the line of action of transverse force or transverse loads to that of c g is what we call, actually is a shear centre, ok. We will see this now here in a classical definition of shear centre.

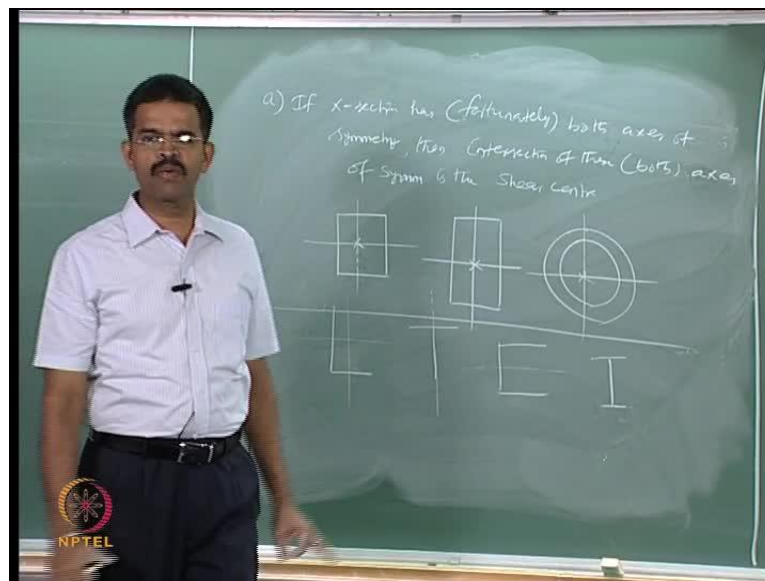
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Shear centre is defined as the intersection of longitudinal axis of a member with the line of action of transverse loads. Let us have a section, some cross section, asymmetric. This is my c in section. This is my w, then what 2 is nothing but, the self feeds geometrically mass will axis this point. Whereas this point is what we call as shear centre because this is the point where my line of action of later loads selected. The difference between these 2 is what we call as e, it may lie in the same section, it may lie outside the section also; I will show you. It may lie somewhere here also, the shear centre may lie here also or may lie outside also.

So, if I say this is  $W$  and this is the  $V$  ray  $r$ , because it is a reaction of all the forces, transverse force, then additional movement comes is nothing but  $V$   $r$  into  $e$ . And of course, we know that  $V$   $r$  will be equal to  $W$  for static equilibrium, is it not? They should match. Then we also said that  $W$  into, so our problem is for a given section what to be the value of  $e$ . So, what is the offset of the shear centre from the centroid for a given section which is asymmetric? Now the question obviously comes if this is symmetric what will happen, if a section has both types of symmetry?

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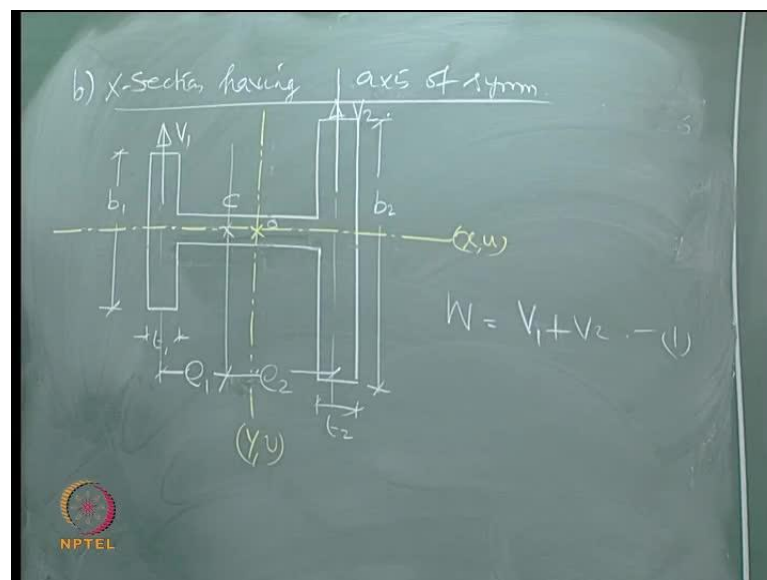


If the cross section has, I should put a word here fortunately both axis of symmetry or what I mean to say is the geometric shape of the cross section is symmetric at both the principle axis. The movement you say axis is principle axis, then intersection of these, these means both, these axis of symmetry is the shear centre. So, we have no problem at all.

For example, let us say a square, we have 2 axis symmetry, this itself is the shear centre. This is the geometric centre or the mass centre, no torsion. Rectangle can identify 2 angles axis so no problem, circular no problem, angular no problem. Now the problems are L T, channel, I with unequal planes is it not, so all of them have only 1 axis symmetry. For example, in this case, is there any axis symmetry? In this case, is there any axis of symmetry? No, axis you have own vertical axis symmetry is it not; In this case you have one horizontal axis symmetry.

In this case we have both of the axis if they are equal. If it is not equal then, in this case it is symmetric to both the axis. So if you have sections which is has got both the axis symmetry, fortunately, it is in a geometric shape. We have absolutely no problem. Shear centre will become coincide with that of geometric centre and the mass centre or centroid, so we have no difficulty of invoking an additional movement which will cause torsion in the cross section, there is no difficulty. So, for sections where there are 2 axis of symmetry we need not have to bother about the shear centre. So, we will talk about b case where sections have 1 axis of symmetry. The shear Centre will lie on that axis but, where it will lie, on this axis, is it here? Here? Where? So we have to locate the shear centre but, the shear centre will lie on that axis of symmetry itself, one axis.

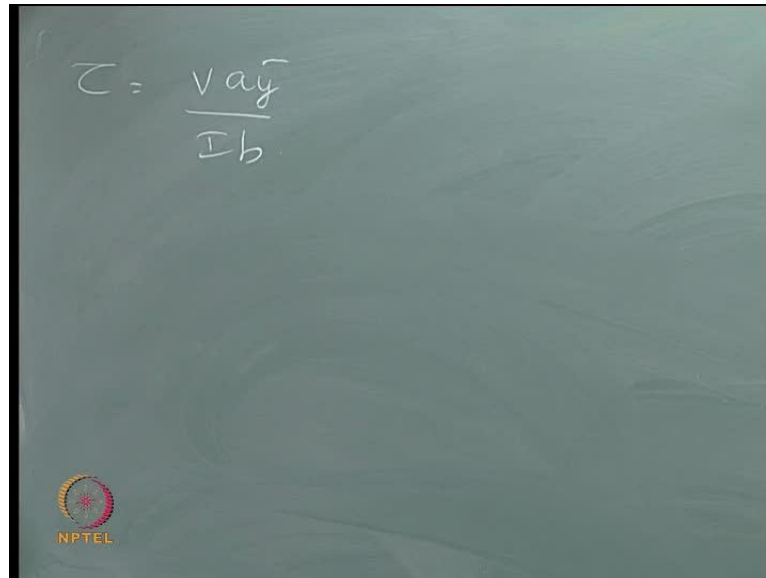
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So, we will take up an example where we have got sections. I should say cross sections having 1 axis of symmetry, which is the case what we discussed. How to compute the shear centre for this? So, we draw an example and try to derive this. The shape may be symmetric but, the thicknesses are different. So let us say this is  $b_1$  and this is  $b_2$ , this is  $t_1$ , this is  $t_2$  and we have somewhere the  $c$  here and this is my principle axis. I call this  $x$  and  $y$ , you call this  $x$  and  $y$ , and this is my geometric centroid  $c$  and I call this as  $e_1$  and this as  $e_2$ .

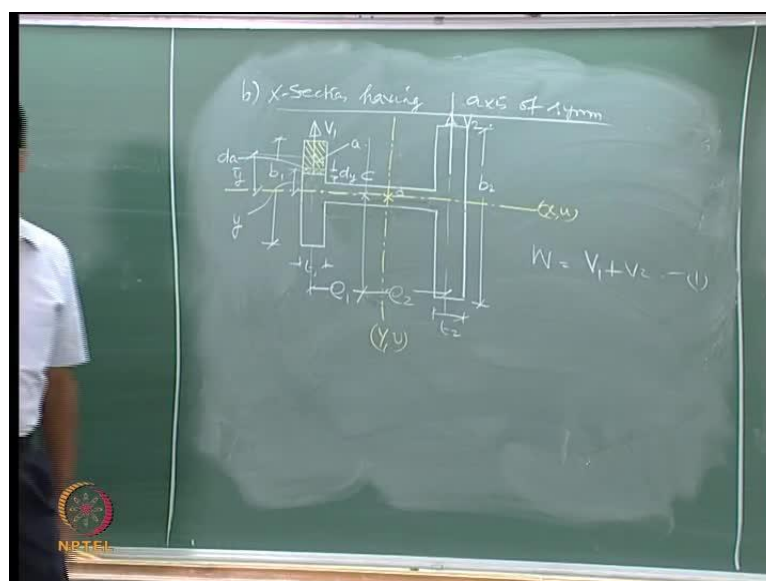
Let us say, the shear resistance of this flange is  $V_1$  and of this flange is  $V_2$  and I am neglecting the web. So, I should say that total load acting on the system will be now resisted only by  $V_1$  and  $V_2$ , this is equation number 1.

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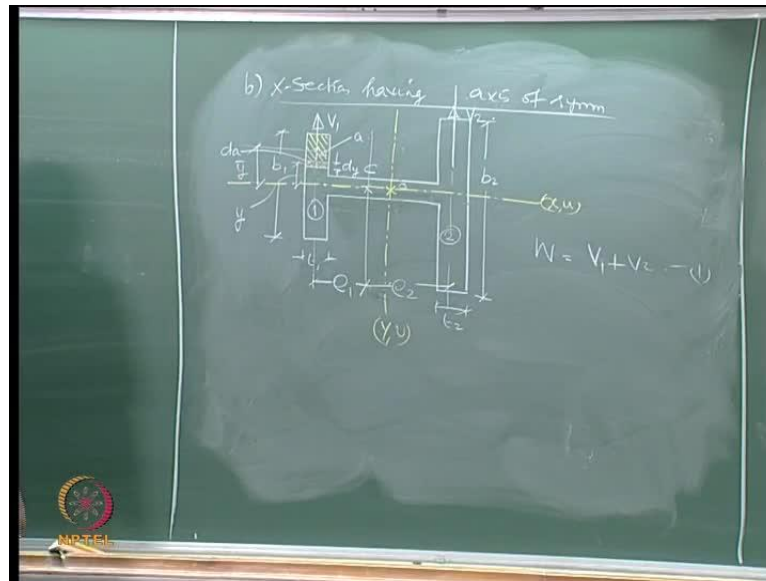
We know shear stress is given by a general equation  $V a \bar{y}$  by  $I b$ . So, let us take the piece 1 here and mark a strip of area above this. Let us consider this strip which is at a distance  $y$ , let us call this as  $d y$ .

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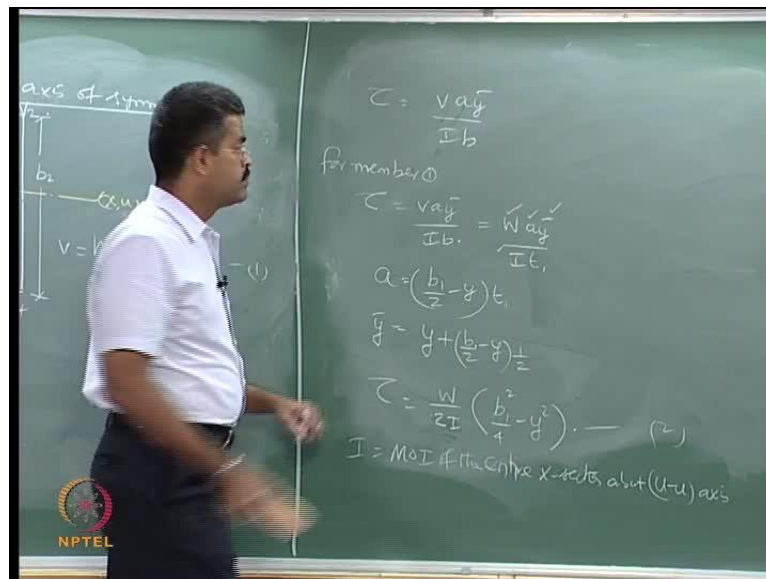


And area about this is what we call as  $a$ , and area of this strip alone is what we call as  $d$ . So,  $V$  is nothing but, the shear force acting at the section at that time,  $a$  is the area of above the level of contagion,  $y$  bar the centroid of the area respectively line of configuration,  $I$  is the movement of inertia of whole section and  $b$  is the width of the section under consumption. So, if I say the centroid of this area which I know, this we measured from herethis I should say  $y$  bar as per the equation.

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Let us expand this for member 1; I call this just member for 1. This is my 1, tow is V a y bar by I b. This total V which can be W, I should say W a y bar by I b and in my case the breath of the section ((recognition) which is t 1, I can say t 1. And what is areaof this piece?It is b one by 2 minus y, eliminating the thickness, into t 1 that is the area is it not? And of course, y bar if the distance of that from here is the axis symmetry is it not? So y bar can be written as, we already seen this is y, I can say y plus 1 by 2 minus y half, half of it b 1 by 2 minus y eliminate the thickness is very small, d y is very small, half half of that.

That is what my distance of the centroid from here. This of the shear centre, is it not? Substitute here and get tow. So, we know V that is w, we know a, we know y bar, I we written as I, t 1 is we written as t 1. So, give me an explanation for tow for member 1. So W by 2 I b 1 square by 4 minus y square, I call it equation number 2, in simply we can find out. In this case of course, I is movement of inertiaof the entire cross section remember that, about in my case going to be u-u axis. Ok, not that part alone whole section. So, I want to find V 1, I know tow, remove this because I want, I am interested in this V 1.

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$$V_1 = \int y dA$$

$$= \frac{W}{2I} \int_{-b/2}^{b/2} \left( \frac{b^2}{4} - y^2 \right) t_1 dy$$

$$= \frac{W t_1}{I} \left( \frac{b^3}{12} \right)$$

$$V_1 = \frac{W}{I} I_1 \quad \left\{ I_1 = \frac{b_1^3}{12} \right\} \quad (3)$$

So, V1 is going to be tow d a entire, is it not? So, W by 2 I integral tow value was b1 square by 4 minus y square is that right? And d a is this area which is t 1, and we are looking for the whole member so from this point we should say minus b 1 by 2 t 2 plus b



1 by or 0 to by 1 by 2 by or work force integral will get me this quickly. You get W by I, t  
 1 b1 cube by tow is simplified. I can always say this equation as W by t, sorry W by I  
 into I 1, where I 1 is the movement of inertia of this equation alone about this axis which  
 is t 1b 1 cube by 12 is itok? Call equation 3 I think.

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$$V = \frac{W}{2I} \int_{-b/2}^{b/2} (b^2 - y^2) dy$$

$$= \frac{W}{I} \left( \frac{b^3}{12} \right)$$

$$V_1 = \frac{W}{I} I_1 \quad \left\{ I_1 = \left( \frac{b_1^3}{12} \right) \right\} \quad \text{--- (3)}$$

$$V_2 = \frac{W}{I} I_2 \quad \left\{ \text{When } I_2 = \left( \frac{b_2^3}{12} \right) \right\} \quad \text{--- (4)}$$

So similarly, I can always say V 2, please do it instead of b 1 you will get b 2, do the  
 same axis again it will be W by I of I 2 where I 2 is t 2 b cube by 12. Now total V, I am  
 neglecting the V by this web, only flanges.

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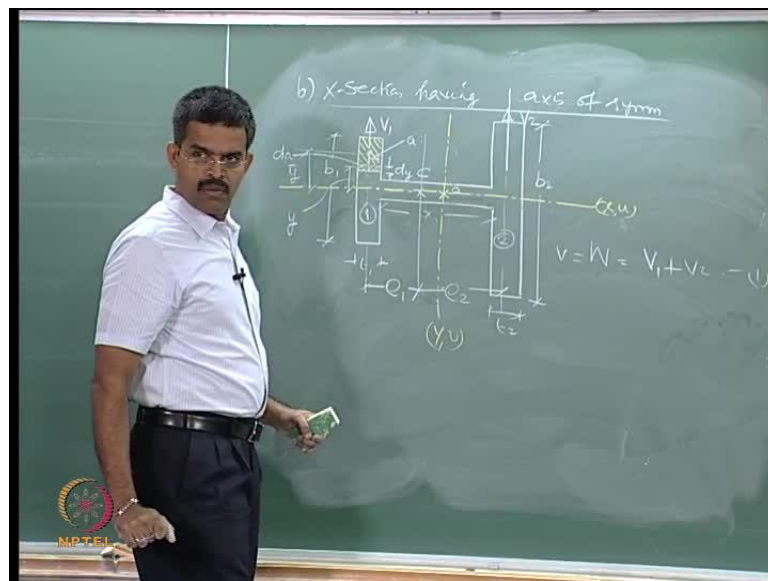
$$V = V_1 + V_2$$

$$= \frac{W}{I} (I_1 + I_2) = W$$

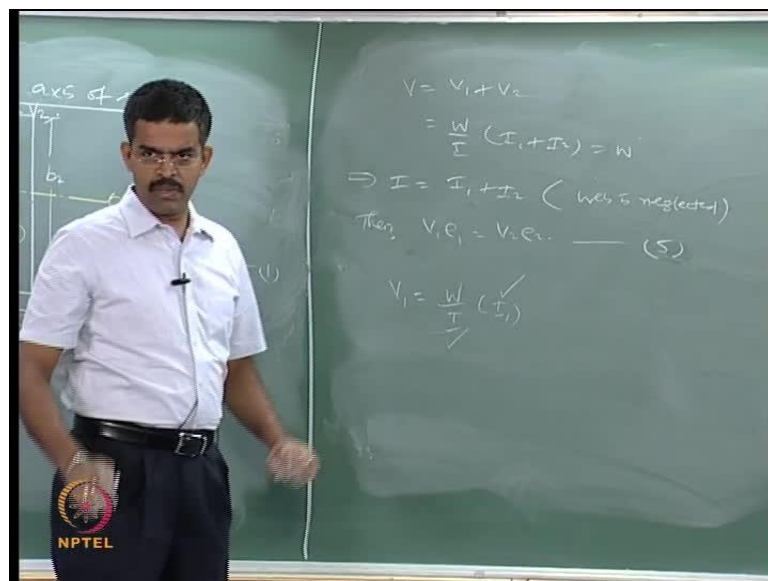
$$\Rightarrow I = I_1 + I_2 \quad (\text{W is neglected})$$

So, total  $V$  as we see from equation 1 is  $V_1$  plus  $V_2$ , which is  $W$  by  $I$  of  $I_1$  plus  $I_2$  is that ok, which is also equal to  $W$ . So, what does it mean, it implies that the total movement of inertia is only sum of  $I_1$  plus  $I_2$ ; web is neglected. Remember that because web also has movement of inertia which will be this dimension,  $e_1$  plus  $e_2$  minus  $t$  by 2 this dimension. If I know this is  $x$  by  $12$ , the thickness is very very small; you can neglect that is why getting this relationship.

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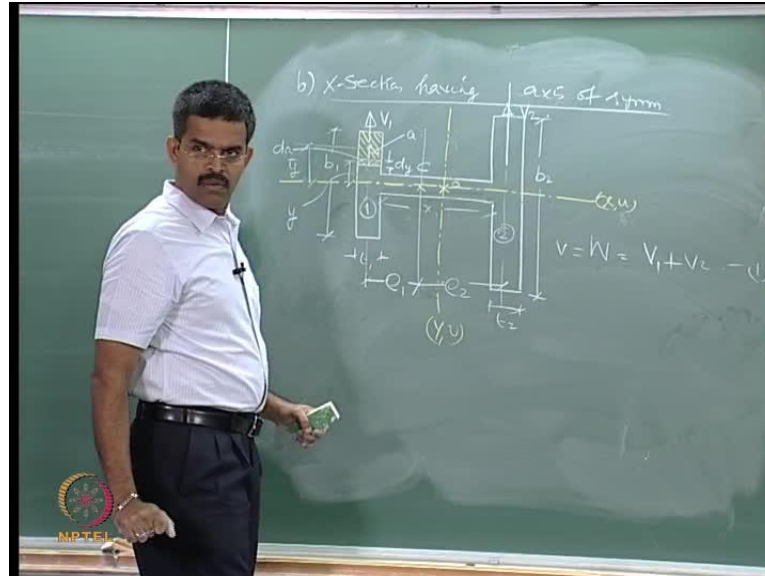


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Then taking movement about this point  $c$   $V_1$  into  $e_1$  is  $V_2$  into  $e_2$ . So, you know the relationship between  $e_1$  and  $e_2$ .

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For a given section of course, this dimension is known to you therefore, if  $e_1$  is known we can find  $e_2$ ;  $V_1$  and  $V_2$  are already known to me. How  $V_1$  is nothing but, what is  $V_1$ ?  $W$  by  $I$  of  $I_1$ ,  $I$  is known to me,  $W$  is given to me I can find  $e_1$ . Similarly  $V_2$ , so I can get the ratio of  $V_1$  and  $V_2$ , I can express it and find  $e_1$ . So I can locate the shear centre in the given.

This is one example where the sections are symmetric about 1 axis; it is not symmetric about the other one because the thicknesses are different. So in that case how will you locate the shear centre? So there are 2 things we answered in this lecture, 1 few design examples understanding that how the selection of diameter for a given simple example can vary when you apply different theory. The second is what is the shear centre, what is its important in geometrical design for marine structures, when the sections have fortunately 2 axis of symmetry then we have no problem at all, shear centre will be going inside the centre of the section, then there will be a problem and inducing movement which is torsion which is generally thin asymmetry section as we select to choose for marine structure are very good in bending but, they are very weak in torsion. So, whenever a section is subjected to additional is torsion, we must locate the shear centre and check for its shear stress exceeding permissible limits.

So, this is one of the important aspects of design in marine structural members. In the next example, next class, we will take a few more examples case centre and try to solve some more sections and then we are more want to the design check for members and as suggested by different course, ok!.