

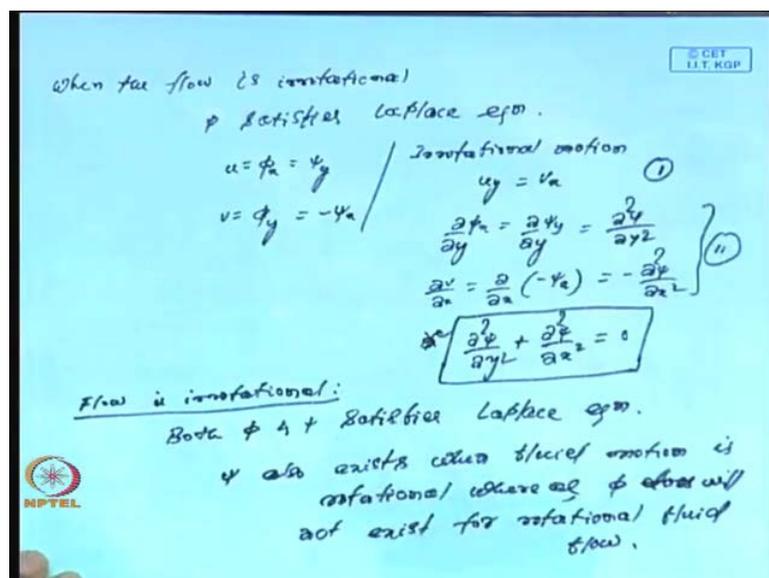
Marine Hydrodynamics
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Lecture - 04
Worked Examples on Various Types of Flow

Good afternoon, so today we will it is the 4th lecture in the series and we will be talking about Various Types of a Flow examples. In the last couple of lectures, in the last three lectures rather we have already talked about conservation of mass, then we talked about stream lines, stream function, velocity potentials, rotational flow, potential flow. Now, within this much of information and then we have discussed about relation between the velocity potentials and the string functions.

Now, with this background let us show clear our basics let, so spend some time on working out some more examples and that will give us a very good understanding about the flow characteristics. So, this we can start with this few examples before that I will give you one one of the relation that the stream function satisfies particularly when the flow is irrotational.

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When the flow is irrotational our solution extreme function psi satisfies the Laplace equation satisfy Laplace equation. So, it is very easy we know that we have the phi x is equal to psi y, further we know that we have phi y is minus psi x. If we now for a

irrotational flow or a irrotational motion, motion we have we have u_y is equal to v_x that is in two dimensional, for two dimensional flow. So, if u_y is v_x , so now, substitute $\frac{\partial}{\partial y}$ by $\frac{\partial}{\partial y}$ this is same as $\frac{\partial}{\partial y}$ u is ϕ_x here other I will say u is ϕ_x and v is equal to ϕ_y .

So, $\frac{\partial}{\partial y} u$ is ϕ_x , which is same as $\frac{\partial}{\partial y} \psi$ by and again further we have $\frac{\partial}{\partial y} v_x$ is $\frac{\partial v}{\partial x}$, $\frac{\partial}{\partial x} v$ is this minus ψ_x . So, this is equal to minus $\frac{\partial^2 \psi}{\partial x^2}$ and here also this is also equal to $\frac{\partial^2 \psi}{\partial y^2}$. Now, from 1 and if we substitute for this 2, in 1 then what will happen from 1 and 2 we will easily get this is $\frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0$ and v_x is as as minus $\frac{\partial^2 \psi}{\partial x^2}$.

So, this will be plus $\frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0$, which is nothing but the Laplace equation. So, this itself is the and this only possible when the flow is irrotational. So, that is very clear that when the flow is irrotational flow is a irrotational we have both ϕ and ψ satisfies Laplace equation. On the other hand on the other hand we have ϕ ψ also exist when fluid is irrotational fluid motion is rotational.

Whereas, whereas ϕ does not exist ψ will not exist for rotational flow rotational fluid flow this is one of the very important characteristic of this a stream lines and stream function. Now, I will a give you another example I will go to another example where we will talk about how to find the flow characteristics.

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$\vec{q} = i(Ax - By) + j(Bx - Cz) + k(Cy - Ax)$
 A, B, C are non zero constants.
 Characterize the flow
 $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0 + 0 + 0 = 0$
 Fluid motion is possible, fluid is incompressible.
vorticity vector:
 $\Omega_x = \frac{\partial w}{\partial y} - \frac{\partial v}{\partial z} = C - C = 0 \neq 0$
 $\Omega_y = \frac{\partial u}{\partial z} - \frac{\partial w}{\partial x} = A - A = 0 \neq 0$
 $\Omega_z = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} = B - B = 0 \neq 0$
 Flow is rotational. $\vec{\Omega}$ exists.
 Egn. of vorticity
 $\frac{d}{dx} = \frac{dy}{dy} = \frac{dz}{dz}$

Particularly suppose I have been given a velocity vector $\vec{q} = i(Ax - By) + j(Bx - Cz) + k(Cy - Ax)$. Then A, B, C are non zero constants they are non zero constants. So, what will happen to the flow consider characterize the flow characterize the flow. So, the first thing is that to check whether the flow is irrotational, flow fluid is a there is a fluid motion is possible or not. So, we will say that since it is a independent of v, v, w, ρ .

So, you have to check that whether $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z}$ what happened to this, if we do that, so you have $\frac{\partial u}{\partial x}$ you have $\frac{\partial u}{\partial x}$ is A minus B . So, it will be $\frac{\partial u}{\partial x}$ is A and $\frac{\partial v}{\partial y}$ plus 0 $\frac{\partial w}{\partial z}$ this is sorry a z minus sorry this is Az minus By , so this is Az minus By , so this is 0 plus 0 here also 0 here also 0 , so this is 0 . So, fluid motion is possible in fluid motion is possible and the flow is in compressible fluid is in compressible in compressible.

Now, again we will check what what will happen to flow, vorticity vector because if we can check that we can find the what are the vorticity vector in this case what are the vorticity vectors vorticity vector $\vec{\omega}$. So, in that case we have ω_x is equal to $\frac{\partial w}{\partial y} - \frac{\partial v}{\partial z}$ $\frac{\partial w}{\partial y} - \frac{\partial v}{\partial z}$. So, $\frac{\partial w}{\partial y}$ this is C minus $\frac{\partial v}{\partial z}$ plus C this is $2C$ and we have ω_y is equal to $\frac{\partial u}{\partial z} - \frac{\partial w}{\partial x}$.

So, that will give us $A \frac{du}{dz} - \frac{dw}{dx} = A$ minus minus plus A , that is $2A$. And similarly we have ω_z is equal to $\frac{dv}{dx} - \frac{du}{dy}$. So, you have $\frac{dv}{dx} - \frac{du}{dy} = B - B$ that is plus B that is $2B$. So, since ω is non zero, non zero not equal to 0. So, flow zero flow is rotational in this case the flow is rotational and once the flow is rotational we can have a vorticity vector So, $\bar{\omega}$ exists $\bar{\omega}$ exist.

And once $\bar{\omega}$ exist then we can find what are the vorticity vectors what are the equations of vortex line if you want to find the equation of vortex lines; that means, $\frac{dx}{\omega_x} = \frac{dy}{\omega_y} = \frac{dz}{\omega_z}$.

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$$\frac{dx}{C} = \frac{dy}{A} = \frac{dz}{B}$$

$$\frac{dx}{C} = \frac{dy}{A} \quad \Rightarrow \quad Ax - Cy = k_1 \text{ (say)} \quad (i)$$

$$\frac{dy}{A} = \frac{dz}{B} \quad \Rightarrow \quad By - Az = k_2 \quad (ii)$$
 Hence the vortex lines are the intersection of (i) & (ii).

Ex: $\phi(x, y, z) = (x-t)(y-t)$
 $u = \frac{\partial \phi}{\partial x} = y-t, \quad v = \frac{\partial \phi}{\partial y} = x-t$
 $\frac{dx}{u} = \frac{dy}{v} \quad (\text{Eqs. of streamlines})$
 $(y-t) dx = (x-t) dy \Rightarrow (x-t)^2 + (y-t)^2 = \text{const}$

To do that that gives us $\frac{dx}{\omega_x} = \frac{dy}{\omega_y} = \frac{dz}{\omega_z}$ is C is $\frac{dy}{\omega_y}$, ω_y is A and this is $\frac{dz}{\omega_z}$ ω_z is B , this gives us my full length of first two members, then it will give us $\frac{dx}{C} = \frac{dy}{A}$; that gives us $Ax - Cy = \text{constant}$ call this constant as k_1 say. And similarly from the last two equations we can always get we have $\frac{dy}{A} = \frac{dz}{B}$ and which implies $By - Az = \text{constant}$ this gives B by minus Az is equal to another constant let this constant be k_2 , so one equation is this the other equation is this.

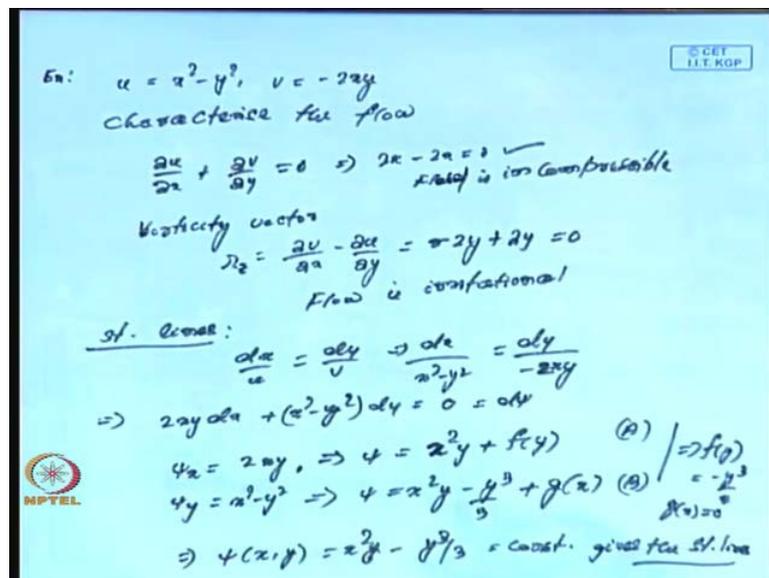
So, the vortex and hence the vortex lines are the hence the vortex lines are the are the intersection of 1 and 2 if I call this as 1 and this I call it as 2 intersection of 1 and 2, so this is the way we will find the vortex lines. Now, I will go to another example this is a

very simple example I will concentrate mainly today on various type of example in the velocity potential. Suppose we have been given $\phi = x^2 - y^2$ this is x minus t into y minus t , if ϕ exist then what will happen to u , u is $\frac{\partial \phi}{\partial x}$ u is $\frac{\partial \phi}{\partial x}$.

Now, it will give us y minus t and v is $\frac{\partial \phi}{\partial y}$ that will give us x minus t . So, so about u v and we can if we will once u v is known then what is the equation of the stream lines that is $\frac{dx}{u} = \frac{dy}{v}$ these are the equation of the stream lines these are the equation of the stream lines. Then what will happen this will give us $\int \frac{dx}{x^2 - y^2} = \int \frac{dy}{-2xy}$ $\frac{dx}{x^2 - y^2} = \frac{dy}{-2xy}$ and that gives us $x^2 - y^2 = \text{constant}$.

So, these are the equation of stream lines these are the equation of stream lines. So, and here the centre of this circle is constant for various values of constant, we get a stream line and here the centre is just the time period t . Now, with this understanding I will go to a very one more example on this flow characteristic, to understand the flow characteristic.

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And we just say that suppose the, you have been given the velocity vector u component of the velocity vector u is equal to x^2 . This is another example u is equal to x^2 minus y^2 and v is equal to $-2xy$, first thing is that characterize the flow, the question comes to characterize the flow. We have to emphasize on few things,

one of the things is that is the first of all whether the flow is incompressible or compressible or whether there is possible fluid motion.

And once the fluid motion exists of a particular nature we need to know whether what are the stream lines, because that will give us the flow direction. Then once you know the stream lines, then we need to know whether the flow is irrotational and if the flow is rotational, then you find the velocity potential and if the flow is rotational, then better to find out what is the vorticity vector, because that will give us the direction of the flow in the angular motion of the flow, so these are the things.

So, in this case, so to understand this first, when we characterize the flow here the question is itself and first of all we have to say whether there is fluid motion. So, for that we need to know what is the $\text{div } \mathbf{u} = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y}$ first thing is to check that whether this is a two dimensional motion, the problem is two dimensional in nature. So, flow is in two dimensional so, you have this, so you have $\frac{\partial u}{\partial x}$ and that is $\frac{\partial u}{\partial x}$ which implies $\frac{\partial u}{\partial x} = 2x$ and $\frac{\partial v}{\partial y} = -2y$ and $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$.

And since this is 0, so there is a there is fluid motion possible and flow is incompressible that itself. So, that itself clarifies that the motion of an incompressible fluid exists there is a flow of an incompressible fluid, incompressible that fluid is incompressible, other we will call it fluid is incompressible. Now, with this, now whether to check whether the flow is what about the vorticity vector, because that will give us the vorticity vector, if you look at the vorticity vector that $\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$. And this is equal to $\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} = -2y - (-2x) = -2y + 2x$.

So, this is $-2y + 2x = 0$, so $\frac{\partial v}{\partial x} = \frac{\partial u}{\partial y}$; that means, $\omega_z = 0$, so it says that flow is irrotational. Once these two things are known that flow is incompressible and irrotational; obviously, the next question is come to now to stream lines, to know the stream lines we have to first look at the equation of the stream lines. So, equation of the stream lines are $\frac{dx}{u} = \frac{dy}{v}$ and that gives us $\frac{dx}{x^2 - y^2} = \frac{dy}{-2xy}$ if this is.

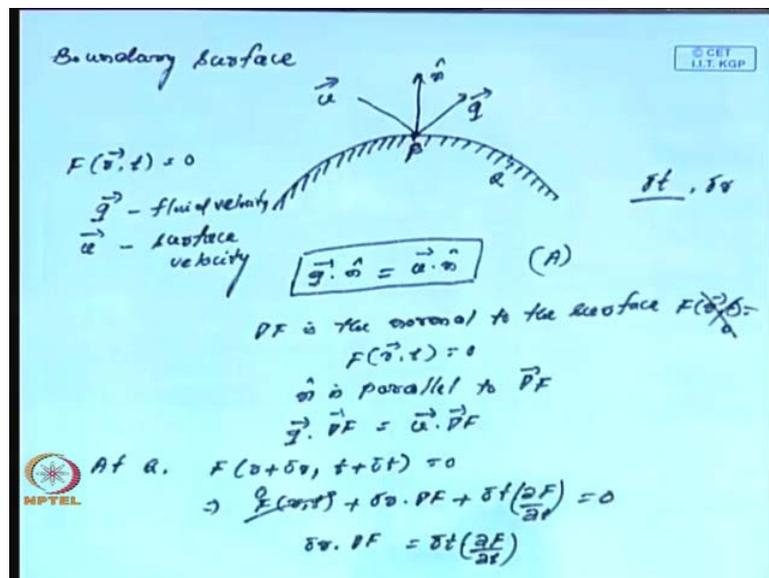
And then so, which simplifies to I will just simplify that will give you $2xy \frac{dx}{x^2 - y^2} + x^2 - y^2 \frac{dy}{-2xy} = 0$. Since, we are looking for stream lines, then here if I call this as $d\psi$ then what will happen to my ψ , my $\psi_x = 2xy$ and my $\psi_y = x^2 - y^2$, if $\psi_x = 2xy$ which implies means $\psi = x^2 - y^2$. So,

this will be $x^2 y + f(y)$ where x is an arbitrary constant. On the other hand if ψ is $x^2 y - y^3$, then my ψ is equal to this is $x^2 y - y^3 + 3$ plus some other function of x , because this is a derivative with respect to y .

So, that will give us another function. So, now, if we compare A and B because A gives an representation of ψ B also gives an representation of ψ . And here that will give us that from this two we get $f(y) = -y^3 + 3$ and also what will be $g(x) = 0$ which implies my $\psi(x, y) = x^2 y - y^3 + 3$ and this $\psi(x, y)$ is $\psi(x, y)$ then this is called constant gives us stream line, these are the stream lines.

So, this is better to go by this way, because otherwise if you go by try to directly sum there is a little complexity you will get this answer, but there will be little complex here better go by $\psi_x = 2xy$ go for ψ and $\psi_y = x^2 - 3y^2$ which will go for and this will give us the stream line in a very easy manner. Now, with this I will go for now already we have now worked out a few examples, with this now let us concentrate on few one of thing is that what is the boundary surface.

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Let us know what is a boundary surface and what are the conditions satisfied on a rigid boundary or a moving boundary. If let me say that this is a boundary surface, so what will happen let me take a point P, so there is I consider the top side is the boundary it is with the fluid. So, let me take a point P here, once I have a point k p, let me draw two

things let me call this surface as F is a function of \vec{r} and t , where there is fluid is flowing and f is this is a P is a point here.

And f is a F is the boundary surface, this surface I call it this is satisfied F of \vec{r} and t where the fluid velocity is \vec{q} and the surface velocity this is the fluid velocity and let \vec{u} is the surface velocity. So, now, at this point, so there will be a fluid velocity that is \vec{q} and let the surface velocity be \vec{u} . So, fluid is moving with velocity \vec{q} , the surface itself is moving with the velocity \vec{u} .

Let \vec{n} be the unit normal from this point in the outer direction. So, if this as to be a boundary surface if this surface has to be a boundary surface what is the normal component velocity of the fluid that is $\vec{q} \cdot \vec{n}$ as to be the same as normal component of the velocity of the surface. So, this is the normal component, so that will be $\vec{u} \cdot \vec{n}$. So, at a boundary surface this condition has to be satisfied, because this is the normal component of velocity of the fluid particular the point this the normal component of the surface and if now further, F is the surface this surface is F .

So, if I call this surface this is by 0 , so what will happen if this is the surface given by F of \vec{r} and t is 0 , then what is the what is $\text{grad } F$ $\text{grad } f$ will is the normal to the surface F of \vec{r} and t is 0 F of \vec{r} and t is 0 , if this is the normal to the surface. Now we have \vec{n} is parallel to $\text{grad } F$, because we have \vec{n} is the unit normal and $\text{grad } F$ is normal to the surface, so the both are parallel. So, because of this we can always say from this a from a we can always say that we have $\vec{q} \cdot \text{grad } F$ is same as $\vec{u} \cdot \text{grad } f$. So, this two has to be same, once this two are same.

Now, let the point because this is a moving surface, so the point let us say that after time Δt the points move towards p moves to q . And here and p moves to q , then the new point at the point q , at q the surface becomes F of $\vec{r} + \Delta \vec{r}$ and $t + \Delta t$ because assuming that there is a shift of the time with the change in time Δt , let there be a change in the position is $\Delta \vec{r}$. So, at q again we we have f of because this is the surface.

So, F of $\vec{r} + \Delta \vec{r}$ and $t + \Delta t$ is 0 and if that is 0 , then that itself is gives us F of \vec{r} and t , which is same as rather which implies F of \vec{r} and t plus $\Delta \vec{r} \cdot \text{grad } F$ plus $\Delta t \cdot \frac{\partial F}{\partial t}$ this is a cross relation dot product this is a multiplication this is 0 . Now and already we have F of \vec{r} and t is 0 , if F of \vec{r} and t is 0 , then this term will give us zero and this is

zero. So, we have got $\frac{d\vec{r}}{dt} \cdot \vec{\nabla} F$ is $\frac{dF}{dt}$ into $\vec{\nabla} F$ by $\frac{d\vec{r}}{dt}$ rather put it is not dot product this is just multiplication.

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The image shows a handwritten derivation on a blue background. At the top right, there is a small logo for 'SCET I.I.T. RGP'. The derivation starts with the equation $\frac{d\vec{r}}{dt} \cdot \vec{\nabla} F = -\frac{\partial F}{\partial t}$. Below this, it shows the limit process: $\lim_{\delta t \rightarrow 0} \frac{\delta \vec{r}}{\delta t} \cdot \vec{\nabla} F = -\frac{\partial F}{\partial t}$. This leads to $\frac{d\vec{r}}{dt} \cdot \vec{\nabla} F = -\frac{\partial F}{\partial t} \Rightarrow \vec{u} \cdot \vec{\nabla} F = -\frac{\partial F}{\partial t}$, where $\vec{u} = \frac{d\vec{r}}{dt}$. Further steps show $\vec{q} \cdot \vec{\nabla} F = -\frac{\partial F}{\partial t}$, $\frac{\partial F}{\partial t} + \vec{q} \cdot \vec{\nabla} F = 0$, and $\frac{DF}{Dt} = 0$, labeled as 'Material Derivative'. A note says 'If the surface be Comol rigid then'. At the bottom, it shows $\frac{\partial F}{\partial t} = 0$ and $u \frac{\partial F}{\partial x} + v \frac{\partial F}{\partial y} + w \frac{\partial F}{\partial z} = 0$.

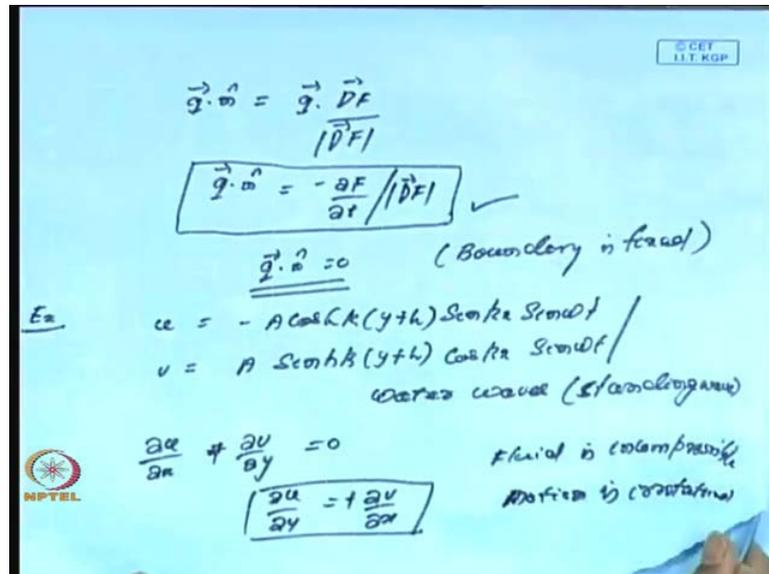
Now, with this now if we go back to this then we have, now take the limit if we look at the limit. So, we have $\frac{d\vec{r}}{dt}$ by $\frac{d\vec{r}}{dt}$, we have $\frac{d\vec{r}}{dt} \cdot \vec{\nabla} F$ is equal to equal to minus $\frac{\partial F}{\partial t}$ and that is, so limit. So, if you take the limit $\frac{d\vec{r}}{dt}$ tends to 0 $\frac{d\vec{r}}{dt}$ by $\frac{d\vec{r}}{dt}$ that is nothing but into $\vec{\nabla} F$ that is again same as minus $\frac{\partial F}{\partial t}$. And this is what, look at this, this is nothing but we have already talked about this is $\frac{d\vec{r}}{dt} \cdot \vec{\nabla} F$ this is minus $\frac{\partial F}{\partial t}$.

And which is same as and since \vec{r} is a position vector of the surface then we have $\frac{d\vec{r}}{dt}$ give us \vec{u} , we have $\vec{u} \cdot \vec{\nabla} F$ is minus $\frac{\partial F}{\partial t}$, now this is very important result. So, now, we have already seen that, so which is same as $\vec{u} \cdot \vec{\nabla} F$ is already we have seen in the we have $\vec{u} \cdot \vec{\nabla} F$ can be written as $\vec{q} \cdot \vec{\nabla} F$. So, which implies $\vec{q} \cdot \vec{\nabla} F$ equal to minus $\frac{\partial F}{\partial t}$, so which implies $\frac{\partial F}{\partial t} + \vec{q} \cdot \vec{\nabla} F$ equal to 0 which nothing but which is same as $\frac{DF}{Dt}$ is equal to 0.

That means, if f is the surface it is a boundary surface, then we have on the boundary surface $\frac{df}{dt}$ zero this is very important result which is always used particularly, when you have a boundary surface. Now, which is same as already $\frac{df}{dt}$ is the where the $\frac{df}{dt}$, we know this is the material derivative. And again if I simplify this if the

surface becomes rigid which gives if the surface is becomes rigid, then we have $\frac{df}{dt}$ is 0 and this becomes $u \frac{df}{dx} + v \frac{df}{dy} + w \frac{df}{dz}$.

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And which gives me hence the normal component of the velocity for the boundary becomes hence $\vec{q} \cdot \hat{n}$ becomes $\vec{q} \cdot \frac{\vec{\nabla} F}{|\vec{\nabla} F|}$ and that is and this $\vec{q} \cdot \frac{\vec{\nabla} F}{|\vec{\nabla} F|}$ becomes $\frac{df}{dt}$ is 0 $\vec{q} \cdot \frac{\vec{\nabla} F}{|\vec{\nabla} F|}$ becomes $-\frac{\partial F}{\partial t} / |\vec{\nabla} F|$. So, you have two things that is the normal component of velocity on the boundary surface and again we have already shown that $\vec{q} \cdot \hat{n}$ on the boundary surface if this is 0 then $\vec{q} \cdot \hat{n}$ is 0 if the boundary is fixed.

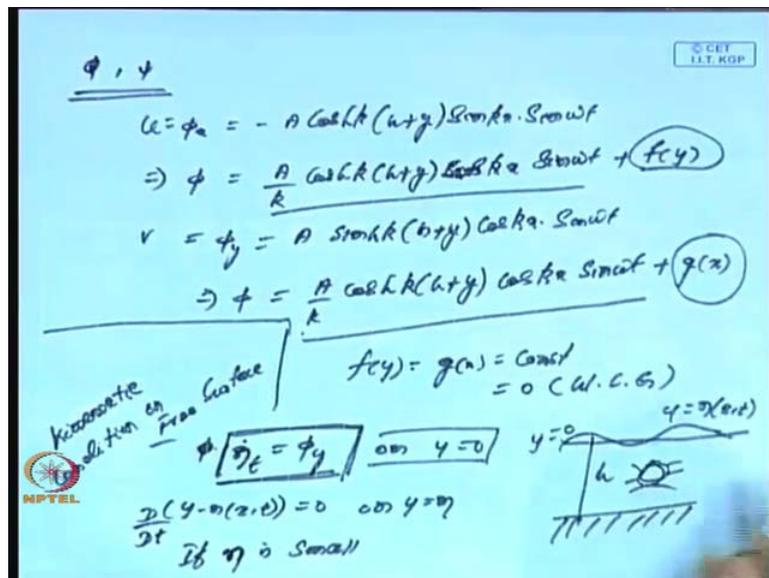
So, this is for fixed boundary this is another important result that how to get the boundary surface and if you know the boundary surface and if the boundary is fixed then you have $\vec{q} \cdot \hat{n}$ is 0 that is... Now, with this I will go to very, very nice example. Now suppose I have been given the velocity vector u is equal to $-A \cos kx \sin(\omega t + h) \sin ky \sin kz$ and my v is a \sin hyperbolic k into y plus h into $\cos kx \sin(\omega t + h) \sin ky \sin kz$ if these are the u v .

So, basically these u v represents these are standing wave in water for a basically standing wave in water particularly in a it is a program related to water waves. Basically it is a standing wave, it can be easily seen that here $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$

equal to 0 it can be easily check this. Further you can find easily that your u v $\text{del } u$ by $\text{del } y$ equal to minus $\text{del } v$ by $\text{del } x$ sorry plus $\text{del } v$ by $\text{del } x$.

So, here it shows the fluid is incompressible here the fluid is incompressible and again our flow is irrotational motion is irrotational our fluid motion is irrotational. Once these two things are satisfied then we can easily find out that. So, what is our velocity potential u or velocity potential, because we have to find our velocity potential, our stream lines and a , because already flow is rotational, so there is vorticity vector will be 0.

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So, we will have to only find what is velocity potential ϕ and what are the stream lines and accordingly stream functions. So, if we have to find ϕ we have been already given, we have ϕ you have been given u is nothing but ϕ_x is minus $a \cos$ hyperbolic k into h plus $y \sin kx \sin \omega t$. And which implies ϕ is equal to we have $\phi_x = a$ by $k a$ by $k \cos$ hyperbolic k into h plus y into $\cos kx \sin \omega t$ plus f of y again from this is u is equal to ϕ_x .

Further from we have been given v v is equal to ϕ_y which is already given to us that is a \sin hyperbolic k into h plus y into $\cos kx \sin \omega t$. And which implies my ϕ becomes I have to take the derivative with y becomes a by k since \cos hyperbolic k into h plus y into $\cos kx \sin \omega t$. And which implies plus here it can be constant of g of x because expression for ϕ this part is same as this part. So, my $f(y)$ and $g(x)$ has to be 0

then there has to be a constant is equal to constant on that constant can be taken as 0 without loss of generality.

So, we have got what is ϕ and once we know ϕ then easily you can and this very important, because this ϕ there is a relation in water waves that is ϕ_t , if we ϕ_t plus $g t$ or you can say η_t is ϕ_y that is on y is equal to 0. And where η is the free surface suppose this is the depth of water, this is what if this is the mean free surface that is y is equal to 0 and here it is the depth of water. And then we can say this is y is equal to η_x t you have just derived that on this surface it can be easily shown that in linear equation d by $d t$.

Because y is equal to η_y minus $\eta_x t$ 0, this is the boundary surface, because it is a surface of the fluid. So, on this surface it is 0 just now we have derived boundary surface d by $d t$ this is 0 and that will be that is satisfied on y is equal to η . And assuming that η is small, if η is small if you can say that η is small, then it can be proved that this condition becomes η_t is equal to ϕ_y is satisfied on y is equal to 0. And that is the again the call the kinematic condition kinematic condition on the free surface on free surface of water.

I will come to this little towards the end when I will come in detail water waves and because what I say that this once we know the velocity potential ϕ we can know what exactly is our free surface and this free surface because. We know η_t is a ϕ_y already have been given ϕ_y and then we can as if we know what is η and that is again on y is equal to 0. So, this way knowing the velocity potential we can always come to the know what the free surface is.

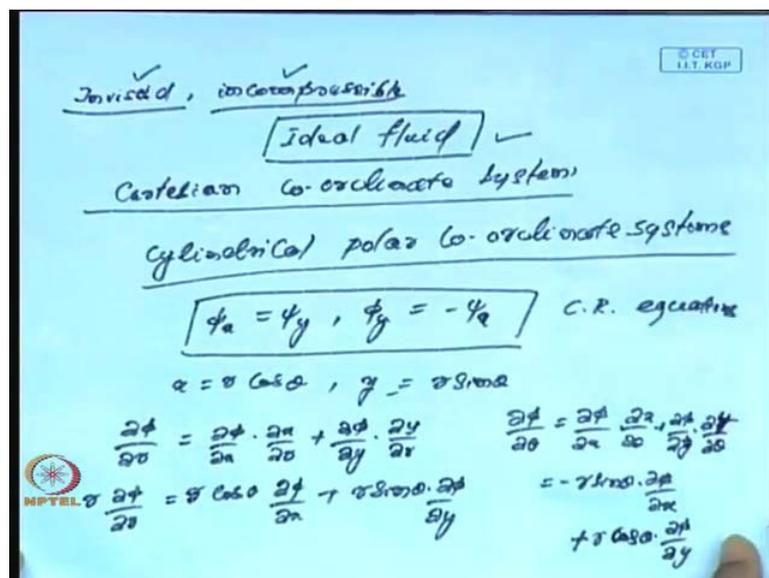
Once we know the velocity potential and the free surface then if i have any structure here then I can always calculate the pressure that is happening on this basically the fluid on this structure what is the pressure. So, I can always calculate the pressure, so this is that is why it is very important to know the velocity potential and the stream like the stream functions is important, because it gives us the stream lines will give us the flow direction similarly the velocity potential for potential flow.

Velocity potentials are very important, because this velocity potential gives us that we utilize this to get the free surface elevation. Sometimes we use this even if to calculate the force because we can calculate the pressure by using the Bernoulli's equation I will

come to those things in detail later. Because there is an equation for an irrotational flow equation of motion later to Bernoulli's equation and that from there we can calculate the pressure at each point.

And once when we know pressure at each point of the fluid, basically what we call this we will call this as hydrodynamic pressure. And once we know the pressure we can always calculate the force, so so I will come to those equations of motion in my next lectures and once we know this then after knowing the equation of motion Bernoulli's equation and other things then at the towards the end I will come in detail about water waves. So, now, already you have been over of when the fluid is incompressible and the motion is irrotational.

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And also another aspect we have not yet emphasized that is when the fluid is inviscid. So, if the fluid is inviscid and incompressible this is another point to be noted I will come to this later because I have not yet talked about viscous fluid. So, when the fluid is inviscid and incompressible we will say that in practice there are very few situations. Where we will come across a fluid which is very 100 percent inviscid or 100 percent incompressible, but for a simplicity of the modeling many fluids we can always under this assumption of incompressibility.

And assume the fluid is inviscid large number of problems can be handled and this is that is why this kind of fluid is called a ideal fluid fluid, where the fluid is in viscid and

incompressible we call them as ideal fluid. Now, already, in fact most of the derivation we have done as of date as of now is based on Cartesian coordinate system Cartesian coordinate system without going to detail I will just say what happen if we are looking at polar cylindrical polar coordinate system.

This equation motion I will not go to the details it can be derived from the mini text book you can go through this and independently it can be derived. But, however, I will just say that what will happen suppose we have a very important result that is the relation between ϕ_x is equal to ψ_y and ϕ_y is equal to minus ψ_x . Let me say what will happen to these equations what we call the co ceriman equations and that is that is the relation between the stream function and velocity potential.

If what will happen if it is a suppose I say x is $r \cos \theta$ and my y is $r \sin \theta$, if I take this what happened to the flow, what happened to this ϕ_x is ψ_y and ϕ_y 's. So, it can be easily because the derivation what is $\frac{\partial \phi}{\partial r}$; that means, what is the corresponding forming $r \theta$ coordinate if you look at $\frac{\partial \phi}{\partial r}$. Basically this is done in this way $\frac{\partial \phi}{\partial x}$ into $\frac{\partial x}{\partial r}$ because plus $\frac{\partial \phi}{\partial y}$ into $\frac{\partial y}{\partial r}$ and that gives us $\frac{\partial \phi}{\partial r}$, $\frac{\partial x}{\partial r}$ is $\cos \theta$ $\frac{\partial \phi}{\partial r}$ $\frac{\partial x}{\partial r}$ is $\cos \theta$ into $\frac{\partial \phi}{\partial x}$.

So, I can always rise $r \frac{\partial \phi}{\partial r}$ is equal to $r \cos \theta \frac{\partial \phi}{\partial x}$ plus $\frac{\partial \phi}{\partial y} \frac{\partial y}{\partial r}$ is $\sin \theta$. So, it can be $r \sin \theta$ into $\frac{\partial \phi}{\partial y}$. Similarly if you look at what happened to $\frac{\partial \phi}{\partial \theta}$ because I am $\frac{\partial \phi}{\partial \theta}$, $\frac{\partial \phi}{\partial \theta}$ is same as $\frac{\partial \phi}{\partial x} \frac{\partial x}{\partial \theta}$ plus $\frac{\partial \phi}{\partial y} \frac{\partial y}{\partial \theta}$. And this gives us $\frac{\partial \phi}{\partial x} \frac{\partial x}{\partial \theta}$, $\frac{\partial x}{\partial \theta}$ is minus $r \sin \theta$ into $\frac{\partial \phi}{\partial x}$ then if I $\frac{\partial r}{\partial \theta}$ sorry this I this is x this is y so $\frac{\partial y}{\partial \theta}$.

So, this is minus $r \sin \theta \frac{\partial \phi}{\partial x}$ plus $\frac{\partial \phi}{\partial y} \frac{\partial y}{\partial \theta}$ is $r \cos \theta$ plus $r \cos \theta$ into $\frac{\partial \phi}{\partial y}$. So, once we have now we know that $\frac{\partial \phi}{\partial r}$ is $r \cos \theta \frac{\partial \phi}{\partial x}$ plus $r \sin \theta \frac{\partial \phi}{\partial y}$. On the other hand you have $\frac{\partial \phi}{\partial \theta}$ is minus $r \sin \theta \frac{\partial \phi}{\partial x}$ into $r \cos \theta$ plus $r \cos \theta \frac{\partial \phi}{\partial y}$, now in a similar manner what will happen to.

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Handwritten mathematical derivations for the gradient of a scalar field ψ in polar coordinates. The derivations show the relationship between the gradient components in Cartesian and polar coordinates.

$$\frac{\partial \psi}{\partial x} = \frac{\partial \psi}{\partial r} \cos \theta - \frac{\partial \psi}{\partial \theta} \sin \theta$$

$$\frac{\partial \psi}{\partial y} = \frac{\partial \psi}{\partial r} \sin \theta + \frac{\partial \psi}{\partial \theta} \cos \theta$$

From these, the following relations are derived:

$$\frac{\partial \psi}{\partial r} = \frac{\partial \psi}{\partial x} \cos \theta + \frac{\partial \psi}{\partial y} \sin \theta$$

$$\frac{\partial \psi}{\partial \theta} = -\frac{\partial \psi}{\partial x} \sin \theta + \frac{\partial \psi}{\partial y} \cos \theta$$

The final boxed results are:

$$\frac{\partial \psi}{\partial r} = \frac{\partial \psi}{\partial x} \cos \theta + \frac{\partial \psi}{\partial y} \sin \theta$$

$$\frac{\partial \psi}{\partial \theta} = -\frac{\partial \psi}{\partial x} \sin \theta + \frac{\partial \psi}{\partial y} \cos \theta$$

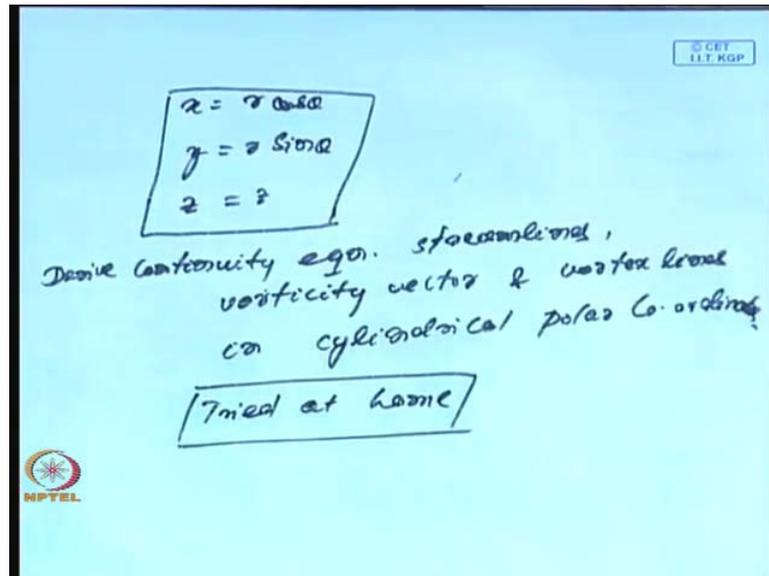
Now, if I go back to $r \frac{\partial \psi}{\partial r}$ which is nothing but $r \cos \theta \frac{\partial \psi}{\partial x} + r \sin \theta \frac{\partial \psi}{\partial y}$. Now, what I will do $\frac{\partial \psi}{\partial x}$ is $\frac{\partial \psi}{\partial y}$. So, I can always write $r \cos \theta \frac{\partial \psi}{\partial y}$, then what is then $\frac{\partial \psi}{\partial y}$ $\frac{\partial \psi}{\partial x}$ by $\frac{\partial \psi}{\partial y}$ is minus $\frac{\partial \psi}{\partial x}$. So, this is minus $r \sin \theta$ into $\frac{\partial \psi}{\partial x}$ now if I relate it $r \cos \theta$ if I go back to my previous place $\frac{\partial \psi}{\partial \theta}$ is minus $r \sin \theta \frac{\partial \psi}{\partial x}$.

This I can call it as $\frac{\partial \psi}{\partial \theta}$ because in the previous place I have seen $\frac{\partial \psi}{\partial \theta}$. So, if I replace ψ by θ because I know my $\frac{\partial \psi}{\partial \theta}$ $\frac{\partial \psi}{\partial \theta}$ will be minus $r \sin \theta \frac{\partial \psi}{\partial x} + r \cos \theta \frac{\partial \psi}{\partial y}$. So, that will give me $r \frac{\partial \psi}{\partial r}$ or so, hence I get in a similar manner I can easily get if I proceed in the same manner, I will get it my $\frac{\partial \psi}{\partial \theta}$ I will get minus $r \sin \theta \frac{\partial \psi}{\partial x} + r \cos \theta \frac{\partial \psi}{\partial y}$.

And that will give you minus $r \sin \theta$ minus $r \sin \theta \frac{\partial \psi}{\partial y} + r \cos \theta$, $r \cos \theta \frac{\partial \psi}{\partial y}$ is minus $r \sin \theta \frac{\partial \psi}{\partial x}$. And that I can call it minus $r \sin \theta$ times $r \sin \theta \frac{\partial \psi}{\partial y} + r \cos \theta \frac{\partial \psi}{\partial x}$ and I have if I go back to my $\frac{\partial \psi}{\partial r}$ $r \cos \theta \frac{\partial \psi}{\partial x} + r \sin \theta \frac{\partial \psi}{\partial y}$ that is nothing but minus $\frac{\partial \psi}{\partial \theta}$ minus $r \frac{\partial \psi}{\partial r}$. Hence from this two I will get I will get my $\frac{\partial \psi}{\partial r}$ this is one relation; that means, giving me $\frac{\partial \psi}{\partial \theta}$ is minus $r \frac{\partial \psi}{\partial r}$.

And here from here I get which is same as $r \frac{\partial \phi}{\partial r}$. So, these two equations are the corresponding equations of the shear equation equation in the cylindrical polar coordinate. In a similar manner I will we can always say that what will happen to the equation of continuity in cylindrical polar coordinate.

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We just take x is equal to $r \cos \theta$ in the cylindrical we will x is a $r \cos \theta$ y is a $r \sin \theta$. And z is equal to z if you take this we can similar manner we can always derive what are the continuity equation. Rather these thing I leave it as a homework continuity equation and derive, derive the continuity equation, you can derive the vortex stream lines, vortex line, vorticity vector vorticity vector and still vortex vortex lines in cylindrical polar coordinate.

So, this can be done and I am not going to the details of this this I everyone it is can be tried at home. It is very interesting to do these exercise with these I think I will stop today and next class we will not talk more about this, because we will rather use this in the coming classes we will concentrate on the particularly. In the next class I will talk to we have not yet talked about the motion because forces what are the forces acting on a fluid.

And what about the fluid motion it is something called equation of motion we will come to that and that is based on the law of conservation of momentum. Basically the Newton's second law we will come to this in the next class.

Thank you.