

Conduction and Convection Heat Transfer
Prof. S.K. Som
Prof. Suman Chakraborty
Department of Mechanical Engineering
Indian Institute of Technology – Kharagpur

Lecture 52
Viscous Dissipation – II

So, this phi is the viscous dissipation function which let me write the expression of this in an index notation. This we have derived earlier but let me just write it.

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The image shows a handwritten derivation of the viscous dissipation function Φ in index notation on a chalkboard. The derivation starts with the general expression for Φ as a sum of squares of velocity gradients. The terms are arranged in two rows. The first row contains three terms: $\left(\frac{\partial u_1}{\partial x_1} - \frac{\partial u_2}{\partial x_2}\right)^2$, $\left(\frac{\partial u_1}{\partial x_1} - \frac{\partial u_3}{\partial x_3}\right)^2$, and $\left(\frac{\partial u_2}{\partial x_2} - \frac{\partial u_3}{\partial x_3}\right)^2$. The second row contains three terms: $\left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1}\right)^2$, $\left(\frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1}\right)^2$, and $\left(\frac{\partial u_2}{\partial x_3} + \frac{\partial u_3}{\partial x_2}\right)^2$. The final result is $\Phi = \left(\frac{du_1}{dy}\right)^2$, where all other terms are crossed out with diagonal lines.

So, phi - so this we have derived earlier. So, can you recall that what are the assumptions under which this is valid? This is not a general, for any general situation this is not the expression for viscous dissipation. For what kind of fluids? Yes. Homogenous, Isotropic, Newtonian and Stokesian fluid because we had used lamda is equal to minus two third mu in this derivation.

So, Stokes hypothesis has to be valid. So, in the usual index location u_1 means u , u_2 means v , u_3 means w , x_1 means x , x_2 means y , x_3 means z . So now let us see, first of all u is a function of y only and there is no other velocity component. So u_1 , terms with u_1 , with derivative with respect to x_2 will only be there, right? So only $\text{del } u_1, \text{del } x_2$, that term will be there. All other terms will be zero.

So, all these terms will be zero. So, this will be there. This will be zero, this will be zero and this will be zero. So, this becomes this for this particular problem. Now let us come to the energy equation. This is zero because it is steady flow T is not a function of x , so this is zero and this is zero because v is zero. So, if there is a hole in the plate, then you will have the dT/dy term.

So not that the left-hand side always become zero. You have to be careful about the situation and then put that understanding into the mathematical terms because T is not a function of x , this is zero. So, you are left with $k d^2T/dy^2$ plus $\mu \phi$ is equal to zero. What is this ϕ ? ϕ is $u \dots du/dy$ square, so this is u_1 square by H square. So d^2T/dy^2 plus μu_1 square by $k H$ square is equal to zero.

This is our governing differential equation. Now for the boundary conditions we have to assume one of the plates to be hotter than the other. Let us take the example that T_1 is greater than T_0 . If the two plates are at the same temperature, then it is not an interesting problem for heat transfer. But even then, we will see that if the two plates are at the same temperature there is some interest in the heat transfer scenario because of the existence of this term.

So, there is a heat source. Now, let us first complete the solution of this problem.

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The image shows a chalkboard with the following handwritten work:

$$\frac{dT}{dy} + \frac{\mu u_1^2}{kH^2} y = C_3$$

$$T + \frac{\mu u_1^2}{kH^2} \frac{y^2}{2} = C_3 y + C_4$$

bc's

$$(1) \text{ At } y=0, T=T_0 \Rightarrow C_4 = T_0$$

$$(2) \text{ At } y=H, T=T_1 \Rightarrow T_1 + \frac{\mu u_1^2}{kH^2} \frac{H^2}{2} = C_3 H + T_0$$

$$\Rightarrow C_3 = \frac{T_1 - T_0}{H} + \frac{\mu u_1^2}{2kH}$$

So, let us integrate it once, dT/dy plus μu_1 square by kH square y is equal to c_1 , c_1 c_2 we have used, so c_3 , T plus μu_1 square by $K H$ square, y square by 2 is equal to $c_3 y$ plus c_4 . What are the boundary conditions? At y is equal to zero, T equal to T_0 and at y equal to H , T

equal to T_1 . So, the first boundary condition tells, that c_4 is equal to T_0 . Right, at y equal to zero, T is equal to T_0 , so c_4 is equal to T_0 . At y equal to H , T equal to T_1 .

So, at y equal to H , you have T_1 plus μu_1^2 by $k H^2$ into H^2 by 2 is equal to $c_3 H$, plus c_4 is T_0 . So, you have c_3 is equal to $T_1 - T_0$ by H plus μu_1^2 by $2k H$. So, the solution is T plus.

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So

$$T + \frac{\mu u_1^2}{k H^2} \frac{y^2}{2} = (T_1 - T_0) \frac{y}{H} + \frac{\mu u_1^2}{2k} \frac{y}{H} + T_0$$

$$\Rightarrow T - T_0 = (T_1 - T_0) \frac{y}{H} + \frac{\mu u_1^2}{2k} \left(\frac{y}{H} - \frac{y^2}{H^2} \right)$$

$$\frac{dT}{dy} = \frac{(T_1 - T_0)}{H} + \frac{\mu u_1^2}{2k} \left(\frac{1}{H} - \frac{2y}{H^2} \right)$$

So, the solution is T plus μu_1^2 square by $k H^2$, y square by 2 is equal to $c_3 y$ plus c_4 . So, we can write, T minus T_0 is equal to T_1 minus T_0 multiplied by y by H plus μu_1^2 square by $2k$ multiplied by y by H minus y square by H^2 . So, you can see that the temperature distribution is because of two components. This component is because of what? This is because of heat conduction. You can see this is a linear temperature profile.

This is because of heat conduction and this is because of viscous dissipation. So, the net change in temperature is due to a combination and here a linear combination because the governing equation is linear, is because of a combination of heat transfer due to conduction plus heat transfer due to viscous dissipation. Now what is the matter of our interest is to calculate the heat flux.

So, what is our intuition. Intuition is that this top plate is at a higher temperature than the bottom plate, so heat has a natural tendency to flow from the top plate to the bottom plate. That is the common intuition. Now let us see, let us find out what is the heat flux, so dT/dy . dT/dy is T_1 minus T_0 by H plus μu_1^2 square by $2k$ into one by H minus $2y$ by H^2 .

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$$\left. \frac{dT}{dy} \right|_{y=H} = \frac{T_1 - T_0}{H} - \frac{\mu u_1^2}{2kH}$$

$$q''_H = -k \left. \frac{dT}{dy} \right|_{y=H} = -k \left[\frac{T_1 - T_0}{H} - \frac{\mu u_1^2}{2kH} \right]$$

$$= -k \frac{(T_1 - T_0)}{H} \left[1 - \frac{\mu u_1^2}{2kH(T_1 - T_0)} \right]$$

$$q''_H = -k \frac{(T_1 - T_0)}{H} \left[1 - \frac{\mu u_1^2}{2k(T_1 - T_0)} \frac{1}{k} \right]$$

$$= -k \frac{(T_1 - T_0)}{H} \left[1 - \frac{1}{2c_p(T_1 - T_0)} \frac{\mu c_p u_1^2}{k} \right]$$

So, let us find out what is dT/dy at the top wall. So, what is dT/dy at y equal to H ? First term is due to conduction and the second term arises due to viscous dissipation present within the fluid. One very important thing to note from this equation is that when the two plates are at same temperature or T_1 is equal to T_0 , then also one can obtain a non-zero temperature gradient at the top wall.

So, the viscous dissipation which acts as a source of thermal energy causes this temperature gradient. Now, the heat flux at the top wall can be obtained by using the above temperature gradient in the following way. q''_H is equal to minus k dT/dy at y is equal to H is equal to minus k multiplied by T_1 minus T_0 by H minus μu_1^2 square divided by $2kH$. Rearranging different terms.

We obtain the heat flux at the top wall as q''_H is equal to minus k multiplied by T_1 minus T_0 divided by H multiplied by one minus μ multiplied by M_1 square divided by $2k$ multiplied by T_1 minus T_0 is equal to minus k multiplied by T_1 minus T_0 by H multiplied by one minus c_p multiplied by T_1 minus T_0 multiplied by μc_p by k .

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Eckert no, $Ec = \frac{U_1^2}{c_p(T_1 - T_0)}$
 $q''_{H1} = -k \frac{(T_1 - T_0)}{H} \left[1 - \frac{1}{2} Ec Pr \right]$
 If $Ec Pr < 2 \Rightarrow q''_{H1} < 0$
 If $Ec Pr > 2 \Rightarrow q''_{H1} > 0$
 If $Ec Pr = 2 \Rightarrow q''_{H1} = 0$

The above representation is of prime importance from the view point of understanding the physical process in terms of non-dimensional numbers which are the combination of different physical parameters, combined effect of which govern the heat transfer process. The right-hand side of this equation can be represented in terms Prandtl number defined as Pr is equal to $\frac{\mu c_p}{k}$ and Eckert number defined as Ec is equal to $\frac{u_1^2}{c_p(T_1 - T_0)}$.

Prandtl number, Pr represents the ratio of momentum diffusion and thermal diffusion, while the Eckert number (Ec) is basically the ratio of kinetic energy associated with the fluid velocity and the thermal energy associated with the temperature difference. So, the heat flux at the top wall can be written as q''_{H1} is equal to $-k \frac{(T_1 - T_0)}{H}$ multiplied by $1 - \frac{1}{2} Ec Pr$.

Now we have expressed the heat flux in terms of temperatures at the two walls and the two important non-dimensional numbers which are Ec and Pr . A closer look into this equation reveals that the direction of heat flux is solely governed by the following two terms $T_1 - T_0$ and $1 - \frac{1}{2} Ec Pr$. Relative sign of these two terms decides the direction of heat flux.

So, for the case $T_1 > T_0$, the direction in which the heat through the top plate will depend on the magnitude of the dimensionless group $Ec Pr$ that is a, if $Ec Pr < 2$ implies $q''_{H1} < 0$ and the heat will flow from the upper to the lower plate as per intuition. If on the other hand $Ec Pr > 2$ implies $q''_{H1} > 0$ then

interestingly the heat flow from the lower plate to the upper plate even though the upper plate is at a higher temperature as compared to the lower plate.

So, in this particular case there is no heat transfer between the top plate and the bottom plate because that will violate the second law of thermodynamics. So, in this case due to viscous dissipation, locally, the temperature near the upper plate becomes more than T_1 and hence heat transfer takes place from the lower plate to the upper plate.

Here the viscous dissipation basically acts as a heat source. So as if there is a heater sitting in lower plate which is making the local temperature more than T_1 . This is something which is non-intuitive. Finally, if $EcPr$ equal to two implies q_H is equal to zero and the upper plates acts as an insulator. Here despite a temperature gradient being present between the two plates, there is no heat transfer.

This is because the temperature difference created due to the viscous dissipation is exactly nullified by the conduction between the two plates. So, there is zero net heat flux. So, the product (and not the Eckert number alone) determines the strength of viscous dissipation in heat transfer. Based on the value of this product, the direction of heat transfer between two plates will change.

This kind of analysis is very important in determining the heat transfer in bearings. In bearings, the temperature of the shaft and its outer casing is very important as the viscosity of the lube oil is a function of temperature (although we have considered the viscosity to be a constant for the present case). Top wall towards the bottom wall, at the top wall, that is something very little because of the natural temperature difference between the top plate and the bottom plate.

However, when Eckert and Prandtl number becomes greater than two, then despite the top plane greater than, I think the temperature greater than that of the bottom plate there is actually the heat transfer of the bottom to the top. This is something which is non-infinite. So actually, there is no direct heat transfer between the bottom plate and the top plate because that will violate the same wall boundary linings.

So, what is happening is that because of viscous dissipation locally here the temperature is becoming greater than T_1 . Viscous dissipation essentially is in line with heat source. So, I think there is a heater setting here that heater is making the local temperature greater than T_1 . So, there is a heat transfer from T_1 to T_0 . But this is something which is not included. The third case is interesting that despite our temperature gradient being created by a natural boundary condition there is no heat transfer.

So why it is happening like that? It is happening like that because this heat transfer because of the temperature difference created by viscous dissipation is exactly nullified by the heat transfer because of the heat conduction between these two. So, there is zero net heat flux. So that you can also see mathematically. This is the heat transfer; this term is equationally what the heat transfer through conduction.

This term is the representation of the heat transfer in viscous dissipation and you can see that when they are exactly nullified each other the heat will pass this. That is what is the physically founded. So, the very important message is that, the Eckert number into Prandtl number is done by the Eckert number, but the Eckert number into Prandtl number that decides the strength of viscous dissipation in heat transfer.

And based on the value one can have interesting upper edge inversions of the direction of the flux. So (()) 48:36) what is happening is, there is local rise in temperature because of viscosity and that creates the (()) (48:46). So, this kind of analysis is very important for analyzing heat transfer in bearings. So, one can understand the temperature distribution, the heat flux and so on in bearings.

And why temperature bearings temperature in the gap between the shaft and the outer casing is important. It is important because the entire functionality of the oil, which is say a lube oil, it depends strongly on the viscosity of the oil. And the viscosity of the oil is a strong function of temperature. So here, for simplicity we have assumed that μ is a constant. But in practice μ is not a constant.

Viscosity of the oil, if it varies the temperature of the oil, it can vary thus dramatically and that will dramatically alter the performance of the bearing. So, in the practical engineering scenario also this appears to be very important. So, to summarize the discussion so far, in

convection, we started with derivation of the energy equations. Of course, before that we had a preliminary discussion on fluid mechanics and then we had derivation of energy equation.

And we analyzed cases of flow over flat plate for different Prandtl numbers and internal flow that is flow through parallel plate channel and circular pipe. In boundary conditions like constant temperature and constant velocity plus and we also have considered the examples where viscous dissipation is important as has been discussed today. So, we will be having some (()) (50:34) problems for this portion of the scenes of nature that we had, these projects will be uploaded.

And this will be discussed extensively in the tutorial sessions. As I told you the type of problems that you will face for this part of the discussion is not dramatically different from a derivation that we have done. These derivations all are actually problems. So, in some cases boundary conditions may be changing, some cases one additional term may come in, some case velocity profile will change.

Please try to (()) (51:15) but the entire (()) (51:18) these problems is what exactly what we have discussed in the class for this portion. So, we will stop this lecture now.