

**Conduction and Convection Heat Transfer**  
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**Lecture - 34**  
**Review of Fluid Mechanics – II**

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Final volume  
 $= \Delta x(1 + \dot{\epsilon}_x \Delta t) \Delta y(1 + \dot{\epsilon}_y \Delta t) \Delta z(1 + \dot{\epsilon}_z \Delta t)$   
 $= \Delta x \Delta y \Delta z (1 + \dot{\epsilon}_x \Delta t + \dot{\epsilon}_y \Delta t + \dot{\epsilon}_z \Delta t)$   
 $= \Delta x \Delta y \Delta z (1 + \dot{\epsilon}_x \Delta t + \dot{\epsilon}_y \Delta t + \dot{\epsilon}_z \Delta t)$

Final vol - initial vol  
 $\frac{\text{Final vol} - \text{initial vol}}{\text{initial vol} \times \text{time}} = \dot{\epsilon}_x + \dot{\epsilon}_y + \dot{\epsilon}_z$   
 Rate of volumetric strain  
 $= \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z}$

So, this what does it indicate, this is rate of volumetric strain so it is clear from here that this term, which is put in the square bracket represents the rate of volumetric strain. For what kind of situation, you do not have any change in volume of fluid element, when the flow is incompressible.

Remember there is a very important distinction between incompressible fluid and incompressible flow. Incompressible fluid is a fluid where or rather a compressible fluid is a fluid where density is a significant function of pressure, whereas when we were talking about incompressible flow, incompressible flow means there is no change in volume of a fluid element.

So, it is purely a kinematic parameter whereas when you are talking about incompressible fluid you are trying to see whether density is a strong function of pressure or not. If density of the fluid is a strong function of pressure it is compressible, otherwise it is incompressible

fluid. On the other hand, incompressible flow means there is no change in volume of a fluid element. So incompressible flow is a kinematic parameter, it is a kinematic constant, right.

We have to keep this in mind. So, this if it is incompressible flow, this term must be zero for incompressible flow. So, this does not follow from conservation of mass, right. We have shown that this follows from pure kinematic constraints. So many times, there are many misunderstandings, one misunderstanding is that divergence of velocity, which is the term in the square bracket is zero, which follows from the conservation of mass from incompressible flow.

It is not true, it is following purely from kinematic constraints. We can relate that with conservation of mass in some way or the other by coupling that with the continuity equation that is fine. But fundamentally it has nothing to do with the conservation of mass. It has something to do with change in volume of a fluid element, which is a purely kinematic parameter.

Now if this is zero for incompressible flow, then these also must be zero for incompressible flow, because sum total has to be zero, zero for incompressible flow, right, because sum total has to be zero. So, if the term in the square box is zero this curly term should also be zero.

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The image shows a green chalkboard with handwritten mathematical equations. At the top, the continuity equation is written as  $\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{v}) = 0$ . Below this, the velocity vector is defined as  $\vec{v} = (u, v, w)$ . The next line shows the expanded form of the continuity equation:  $\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x}(\rho u) + \frac{\partial}{\partial y}(\rho v) + \frac{\partial}{\partial z}(\rho w) = 0$ . The final line shows the expansion of the terms in parentheses, with a curly bracket under the first part and a square box around the second part. The curly bracketed part is  $\left( \frac{\partial \rho}{\partial t} + u \frac{\partial \rho}{\partial x} + v \frac{\partial \rho}{\partial y} + w \frac{\partial \rho}{\partial z} \right)$  and the boxed part is  $\rho \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right)$ . A blue arrow points from the boxed part to the right. Below the curly bracketed part, there is a handwritten note: "0 for incompressible flow".

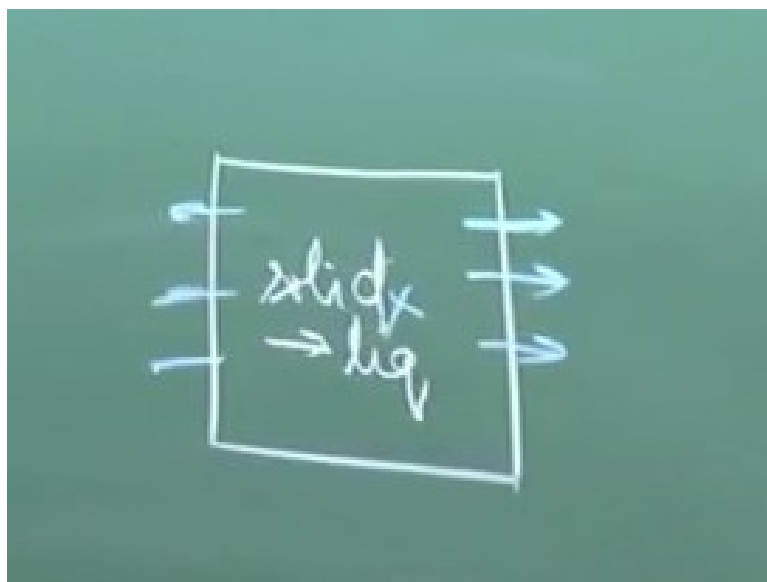
So, what can we tell from this, see there are some many misunderstandings propagated by standard books, even very good books. And one of the misunderstandings is that, in books it is written assuming incompressible flow we take density as constant. This is completely

wrong, you can see from here that for incompressible flow this term has to be zero, that does not mean that density has to be constant.

I will give you an example, when density is not a constant but still it is incompressible flow, I will give you an example, but first you can appreciate mathematically. That this equal to zero has to be true for incompressible flow that does not mean that  $\rho$  has to be constant. But it is true that  $\rho$  is equal to constant is a special case of this, because if  $\rho$  is equal to constant this is trivially satisfied, right.

So,  $\rho$  is equal to constant is a special type of incompressible flow, so that is a constant density incompressible flow. But you can also have variable density incompressible flow. So, let us take an example.

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Let us say that you have a domain in which some solid phase is converted into liquid phase, right. Let us say that it is some metal where the solid metal when is converted into liquid it is expanding. So, when it is converted into liquid and it is expanding and this is the control volume what will happen, first this volume could accommodate the solid. When it is expanding to become liquid this volume cannot accommodate that much of liquid.

So, liquid will flow from different boundaries, okay. Because it cannot be adjusted with in this volume, so what it should do, it should escape from the volume, okay. So, what does it mean, it means that see look at this term,  $\frac{d\rho}{dt}$ , what is this, this is rate of change of

density at a given location due to change in time. You have what  $\frac{\partial \rho}{\partial t}$ , because at a given instant of time within the control volume.

The solid has changed to liquid, which are of different density. So, you have this term, then you have these flow terms  $u \frac{\partial \rho}{\partial x}$ ,  $v \frac{\partial \rho}{\partial y}$ , and  $w \frac{\partial \rho}{\partial z}$ . These are advective components of the change in density. So, it says that for incompressible flow the sum total has to be zero, but this is possibly balanced by these two, these three terms. If this is plus this is minus so that, if this is plus A this is minus A so that sum total is zero.

So here neither of solid nor of liquid can be thought of as compressible phases, right. These are the normally incompressible phases. But here you have a density change. So, it is possible that you have incompressible flow, but with variable density. So, we should keep in mind that we should not have the prejudice that incompressible flow means constant density flow. Incompressible flow may also mean variable density flow.

But a special case of incompressible flow is constant density flow. Finally, we will give a mathematical perspective to this constant.

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The image shows a chalkboard with handwritten mathematical equations. At the top, it says  $\frac{D\rho}{Dt} = 0$  for incompressible flow. Below this, a larger equation is written:  $\left( \frac{\partial \rho}{\partial t} + u \frac{\partial \rho}{\partial x} + v \frac{\partial \rho}{\partial y} + w \frac{\partial \rho}{\partial z} \right) + \rho \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = 0$ . The first term in parentheses is circled and labeled "for incompressible flow". The second term in parentheses is boxed and labeled "for incompressible flow".

So, if you write this particular term in short hand notation you can write that the total derivative this is the total derivative, right, you in your earlier course on fluid mechanics you have learnt that total derivative of velocity is acceleration, right. In place of rho if you write u then that is acceleration along x, right. So, this is just  $D/Dt$  operator operating along rho, this is the unsteady component.

And this is the advective component and the component due to flow, okay. So, sum total of these two is zero for incompressible flow. So, we have discussed about conservation of mass, next we will discuss about conservation of momentum.

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Conservation of momentum.

R.T.T

$$\frac{dN}{dt}\bigg|_{\text{sys}} = \frac{d}{dt} \int_{\text{cv}} \rho n dV + \int_{\text{cs}} \rho n (\vec{V}_r \cdot \hat{\eta}) dA$$

$N = \int dm \vec{V} \rightarrow n = \vec{V}$

So, we will start with the Reynolds transport theorem, so if you recall this  $N$  is the extensive property of the system,  $n$  is  $N$  per unit mass, and  $V_r$  is the velocity of the fluid relative to the control volume, and  $\eta$  is the unit vector outward normal to the control surface. So, for linear momentum conservation  $N$  is what  $MV$ , right, or you can write integrate of  $dmV$ , because every mass of the fluid element may have a different velocity.

But loosely  $MV$  is fine, I mean, because I mean there are many things which we write rigorously, there are many things which we do not write very rigorously, still it is okay if we mean there are same kind of thing and as I said go ahead with the conceptual understanding.

So, this means that what is small  $n$ ,  $V$ , so you can see that this becomes a vector equation, because this  $V$  will have its own components.

So, let us write the different components of the vector equation. So, let us write the  $x$  component, now we will make the two important assumptions that we made for working out the details of the continuity equation, that stationary control volume and non-deformable control volume.

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(1) Stationary CV  
(2) Non deformable CV

$$\sum F_{x, CV} = \int_{CV} \frac{\partial (\rho u)}{\partial t} dV + \int_{CS} (\rho u \vec{v}) \cdot \hat{n} dA$$

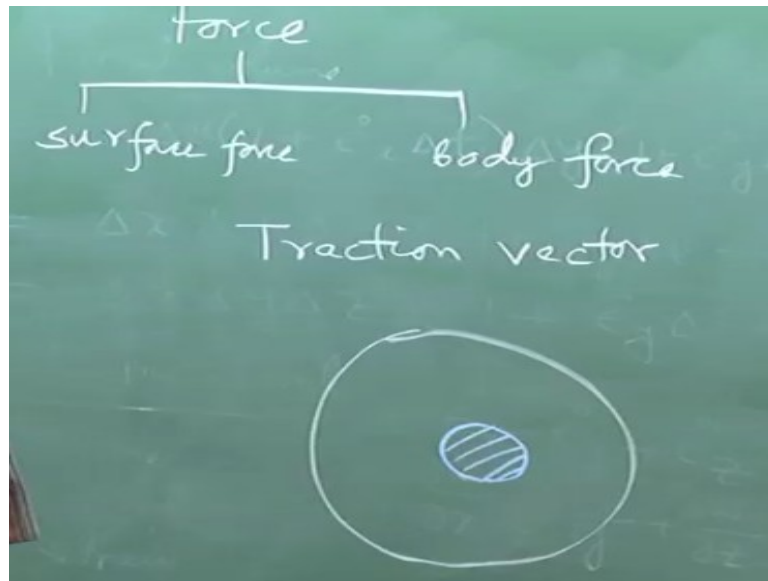
So non-deformable control volume means we can take this derivative within the integral, because volume is not a function of time, and stationary control volume means  $V$  relative and  $V$  at the same. Not only that, what is the left-hand side, this is the, see this is why we apply the Reynolds transport theorem. For this, you can directly apply the Newton's second law, but not for this term, because this refers to a control mass system, this is rate change of linear momentum of the system.

So, this is the resultant force along  $x$  acting on the system. And because while deriving the Reynolds transport theorem we have taken a limit as  $\Delta T$  tends to zero. The control volume almost coincides with the system. Therefore, the force on the system and force on the control volume are the same. So, this is resultant force acting on the control volume.

So, we can write resultant force along  $x$  on the control volume is equal to, so this term you can use the Divergence theorem, to convert the area integral to the volume integral. So, this will be integral of Divergence theorem, by Divergence theorem. So right hand side we have been able to express all the terms in terms of volume integral, left hand side also we will be attempting to do so.

So, the first important conceptual thing to understand is how to represent forces in terms of different integrals. So, forces in continual mechanics when we say continual mechanics, it can be mechanics of solid or mechanics of fluid whatever. You will have two types of forces.

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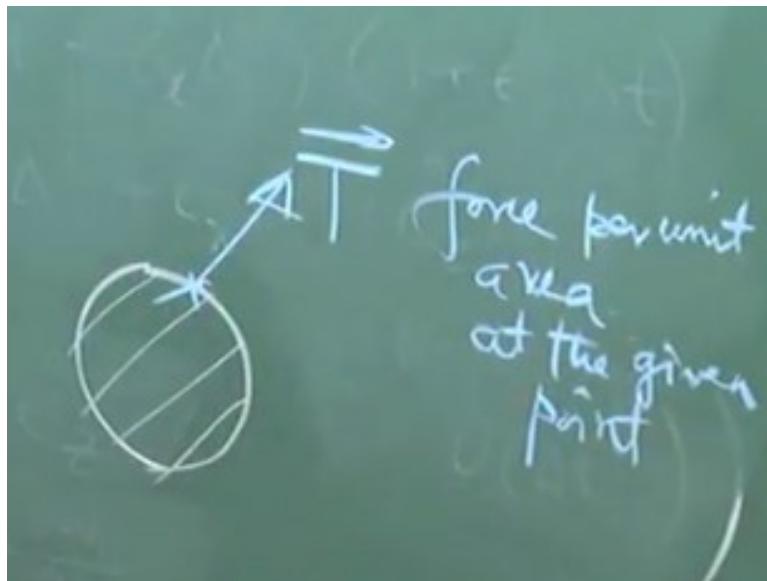


One is surface force another is body force. By the name it is clear that surface force will act on the surface element, and volume body force will act on the volume elements. So, we will define something called as traction vector. So, what is a traction vector, let us say that you have an element like this may be a solid or it may be a fluid whatever. Now you take a small volume out from it.

So, if you take a small volume you have to represent the force exerted by the remaining part of the volume on this one. This is just like drawing a free body diagram, if you isolate one part of the system, then you have to show all the forces exerted by the other parts on that part. That is the concept of a free body diagram. So, this is just like a free body diagram, let us say you have isolated this part and you are interested to draw the free body diagram of that.

So, if you do that, let us say that  $T$  is a force per unit area, at the given point.

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This is a force. This is not actually any external applied force. This force is because of the interaction between the shaded volume with the remaining part of the volume. This is like action reaction type of force, okay. Now, this is not the total force. This force per unit area, okay. So, if this is force per unit area that means the total force will be dependent on the area that is chosen here. Let us say, you choose an area.

If you choose an area  $dA$ , it may be a differential force  $dT$  whatever. Now, this differential force  $dT$  will depend also on the orientation of this area, right. Because at a given point, if you choose the same area, but with different orientation, you will get different force. So, this force at a given point does not only depend on the location of the point, but also on the orientation of the area chosen to calculate the force.

To specify that, you use a superscript eta with  $T$ . This eta is a unit vector normal to the surface  $dA$ , okay. So basically, this is a vector because it has its components, but it is a special type of vector where it is actually something more general than a vector in a sense that it also depends on orientation of the area to calculate the force. So, you can write the traction vector using this notation, but in reality, it is very difficult to deal with surfaces, which are arbitrarily oriented.

So, to systematize the entire behaviour we intend to take surfaces, which are either having normal along  $x$  or normal along  $y$ , or normal along  $z$ . So, we consider those surfaces, then we can represent the behaviour of any arbitrary surface as a function of the force on those

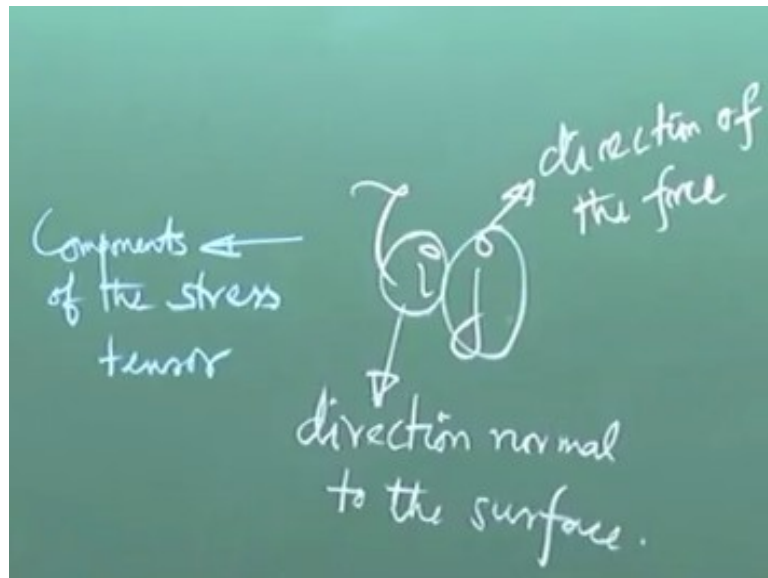


surfaces. That we will show later on. But we will first take an example. Let us say this is x axis. From now onwards, we will try the patient use a notation, which is called as index notation.

So, in index notation we will write x axis as  $x_1$ , y axis as  $x_2$  and z axis as  $x_3$ . So, we will use one index  $i$ ,  $x_i$ ,  $i = 1$  means x axis,  $i = 2$  means y axis,  $i = 3$  means z axis, okay. So now this has how many surfaces. This volume, this has six surfaces. The speciality of these surfaces is that their normals are either along x or along y, or along z. So, for this surface, this traction vector is alternatively replaced by a notation of  $\tau_{ij}$ .

It is represented by a notation  $\tau_{ij}$ , what does this  $i$  indicate, I will write it separately,  $\tau_{ij}$ .

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Very important in terms of the notation  $\tau_{ij}$ . It is again force per unit area, what. So just like traction vector or traction vector on special surfaces, which are normal either along x or along y or along z, so this  $i$  represents the direction normal to the surface and  $j$  represents direction of the force. So, you can see that this  $\tau_{ij}$ , which are called as components of the stress tensor, is not a vector.

This is the first thing that we understand. So, this is something a little bit more general than a vector. Why this is more general than a vector? A vector for its specification requires how many indices? It requires one index for its specification. Let us say a force  $f$ . if you write a force  $f$ , if you write  $f$  with subscript  $i$ , then  $i = 1$  means x component,  $i = 2$  means y component,  $i = 3$  means z component.

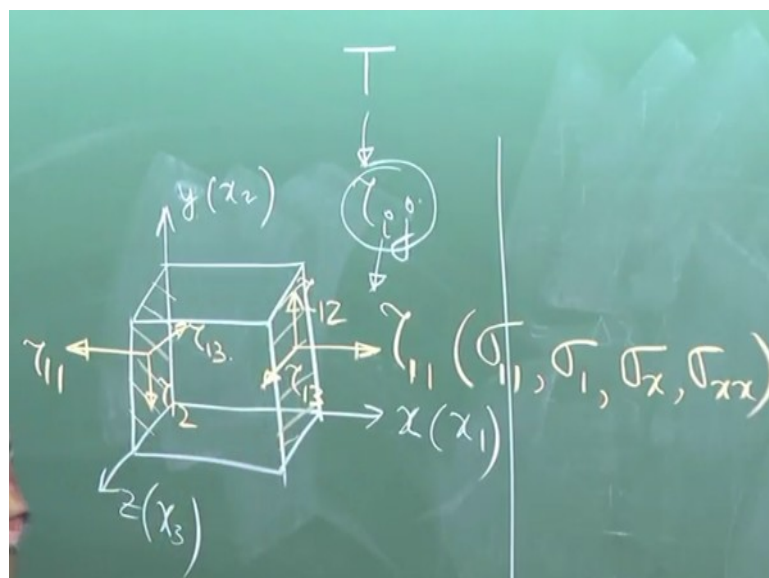
But here you require two indices for its specification, why? Because it acts like a vector in a sense that it has its own components, but this individual components also depend on the orientation of the area chosen to calculate the force. So, direction normal to the surface that appears as another index. So, it requires two indices for its specification that makes it a second order tensor.

So, it is very difficult to define at this level what is a tensor and I do not want to get into that abstract mathematical discussions here. It is not very important, but we have to understand that at least the tensor is something which is somewhat more general than a vector. So, but vector is a special type of tensor. So, this example  $\tau_{ij}$ , this is called as second order tensor because it requires two indices for its specification.

A vector requires one index for its specification so vector is a first order tensor. A scalar requires no index for its specification. So, scalar is a zeroic order tensor. So, in this way you can also have higher and higher order tensors. In the equations that we are deriving, we will come across another tensor, which is a fourth order tensor, which require four indices for its specification, but we will discuss it at an appropriate time.

But I mean just you will be having two indices does not mean that it is a second order tensor, so there are several other properties, a second order tensor maps a vector on to a vector and we will show that how this maps a vector on to a vector.

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Now, what we will do is we will try to draw the various components of the stress tensor on this element, so let us take this surface. I will just write tau, you have to tell what will be the indices. What will be the first index, first index is normal to the surface, which is under consideration. This surface has normal in what direction? 1, x1 right, so the first index is 1, okay. The second index is what?

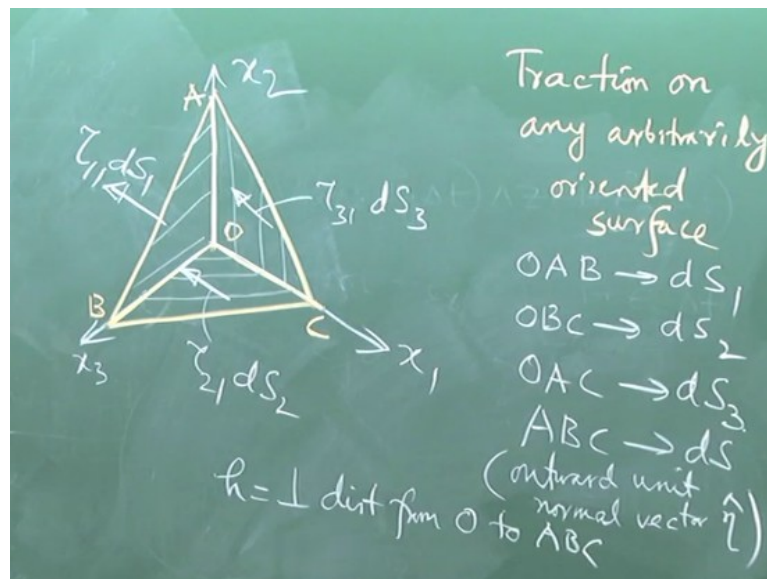
The second index is the direction in which this force is acting, so what is that? That is also 1, so this is tau-11. In books, correspondingly it is also sometimes. You can see this is the normal component of stress, so in books, sometimes to write it as normal component of stress, it is sigma-11, or because 11 is repeating sigma-1 or sigma-x or sigma-xx. These are different notations used in different books, okay. Then think about this.

This is tau12, direction normal to the surface is 1 and direction of the force is 2. Similarly, this is tau13, right. So, let us now consider this surface. There is a sine convention that if the outward normal of the surface is along a positive direction, then the force will also be shown in the positive direction. Otherwise if the outward normal of the surface is in the negative direction, we will show the force in a negative direction.

So here the outward normal to the surface is negative direction, so we will show the force in the negative direction, so this is tau11, this is tau12 and this is tau13. Similarly, for other four surfaces, you can write these things, I do not want to make this figure clumsy by doing the same thing. I mean these two examples should be good enough and you should practice to draw these or represent these forces in other four surfaces also.

So that you get a good grasp on this representation. Now, this is so far so good, so this tau-i is a notation we can use for those surfaces, which have normal either along x, or along y, or along z, but in reality, any surface, any arbitrary surface will not have normal along x, or along y, or along z, so how will we represent that.

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So, to understand that this is the traction on any arbitrary surface, arbitrary oriented surface rather. So, let us that we have a volume taken, which is like this. How many surfaces does this volume have? Four surfaces. Can you tell that why have we chosen this type of surface? So, in this volume, there are three surfaces, which have normals along  $x_1$ ,  $x_2$ , and  $x_3$ , but the fourth one is not.

The surface ABC is having arbitrary orientation and the other surfaces have orientation with normal along  $x_1$ ,  $x_2$ ,  $x_3$ , so for other surfaces, we can represent it by the tau-ij notation, but not the surface ABC and by writing equation of the equilibrium for the entire volume, we will be able to represent the force on an arbitrarily oriented surface in terms of the force on the other surfaces.

That is the motivation of taking these type of element. So, let us try to represent the force along  $x$ . Let us try to represent the force along  $x$ . So, for this surface, OAB let us say that the surface OAB is  $ds_1$ . What is the force on this? First of all, how do we represent it? What is the direction normal to this surface? Negative of  $x_1$ , right. So, the force along  $x_1$  will be shown along negative of  $x_1$  by sine convention, so this is  $\tau_{11}$ , right.

First one is direction normal and second one is it is the action of the force into this is per unit area, so  $\tau_{11}$  into  $ds_1$ . Let us say that we consider OBC as  $ds_2$ . This is the surface OBC, so what is its outward normal, negative of  $x_2$ . Negative of  $x_2$  is its outward normal. So, the

force along  $x_1$ , this is along negative  $x_1$ . This is  $\tau_{21} ds_2$ , but along negative  $x_1$ . The third say OAC, let us say the area is  $ds_3$ .

So similarly, for this also the force is  $\tau_{31} ds_3$ . Let us say the ABC is  $ds$  and ABC let us say as outward unit normal vector of  $\eta$ .

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$\vec{f}_b = \text{body force/volume}$   
 $\sum F_x = (\Delta m) a_x$   
 $-\tau_{11} ds_1 - \tau_{21} ds_2 - \tau_{31} ds_3 + T_1^\eta ds + \rho \frac{1}{3} x ds \times h = \rho \frac{1}{3} ds \times h \times a_x$   
 $\hat{\eta} = \eta_1 \hat{i} + \eta_2 \hat{j} + \eta_3 \hat{k}$   
 $ds_1 = ds \eta_1, ds_2 = ds \eta_2, ds_3 = ds \eta_3$

Let us write the equation of motion for this element, so we can write resultant force along  $x$  is equal to the mass, let us say  $\Delta m$  is the mass times acceleration along  $x$ . So, what is the resultant force along  $x$ . Minus  $\tau_{11} ds_1$ , minus because it is along negative  $x$ , minus  $\tau_{21} ds_2$ , minus  $\tau_{31} ds_3$ , so these three surfaces we have represented. The fourth surface ABC, how will we represent that.

How will we represent the force on ABC, we will use the traction vector notation because it is an arbitrarily oriented surface. So plus,  $T$  with subscript 1 because this is the component of the force and  $\eta$  superscript, which is the direction normal into  $ds$ ,  $ds$  is the area of ABC. This is the surface force, then there is also a force, which is a volumetric force on a body force.