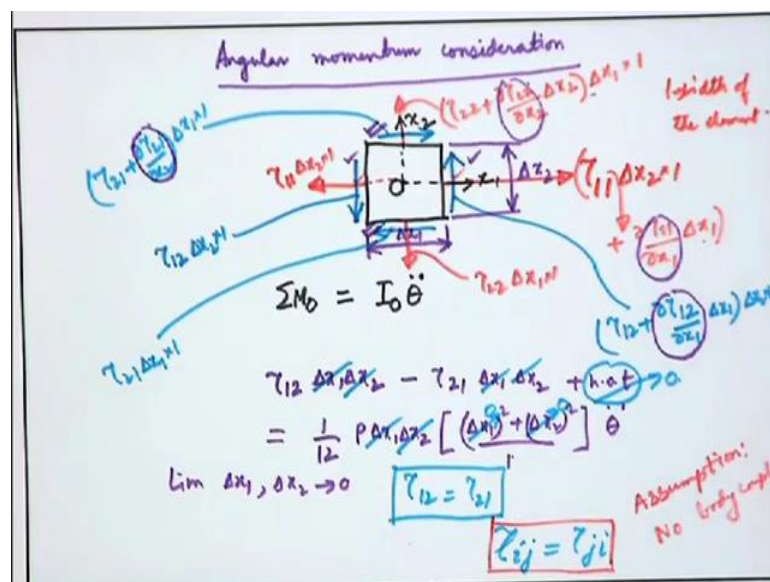


Introduction to Fluid Mechanics
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Lecture – 52
Navier - stokes equation- Part-II

Now so far we have considered only linear momentum conservation. So, we will take little bit of distraction from that, and we will now think of angular momentum consideration. The linear momentum consideration is very common; angular momentum consideration in fluid mechanics is not always so obvious but we need to carefully look into it.

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So, angular momentum consideration; so let us take a now angular momentum just like linear momentum refers to translation, angular momentum refers to rotation. Now let us say that we are interested about rotation with respect to z axis, now we can decouple the rotation in terms of rotation with respect to x axis, rotation with respect to y axis and z axis; x means x 1, y means x 2 and z means x 3. So, we can isolate these effects and consider 1 at a time.

So, we can consider for example, rotation with respect to the z axis or x 3 axis, then it is important that we consider only the forces which take place which are there in the x y plane or x 1, x 2 plane. So, let us say this is the x 1 axis and this is the x 2 axis; and let us

say that we give dimensions to this such that this is Δx_1 and Δx_2 . Now we are interested to have rotation with respect to or an equation for rotation with respect to x_3 axis an axis that passes through O. So, for rotational motion we can write resultant moment of all forces with respect to an axis which is normal to this plane and passing through O is equal to the moment of inertia with respect to the same axis times the angular acceleration ok.

So, now resultant moment of forces. So, let us identify the forces here, let us write the normal forces and the tangential forces. So, this force this is τ_{11} what is the first index? the first index is 1 because the direction normal is 1 the second index is also 1. So, $\tau_{11} \Delta x_2 \Delta x_1$, let us say Δx_1 is the width of the element. If this is $\tau_{11} \Delta x_2 \Delta x_1$, essentially, just a matter of nomenclature, if we call this as τ_{11} , this will be $\tau_{11} \Delta x_2 \Delta x_1$, plus $\Delta \tau_{11} \Delta x_1 \Delta x_2$; because the difference in x coordinate between the 2 phases is Δx_1 . Interestingly we will see that this increment or this higher order effect will not matter for calculating the moments, this will matter for calculation of forces, but not calculation of moments. These are the normal forces; so let us write the normal forces also along this. So, this is $\tau_{22} \Delta x_1 \Delta x_2$, this is $\tau_{22} \Delta x_1 \Delta x_2$ plus $\Delta \tau_{22} \Delta x_1 \Delta x_2$. These are the normal forces now this forces I have written just for completeness, but this forces actually do not contribute to any moment, why because all this forces pass through this point O therefore, this do not contribute to the moment with respect to an axis passing through O.

Then let us draw the tangential forces. So, first let us show their proper directions and then we will write the force. So, I am showing the forces in the proper direction as per the sign convention we discussed earlier, and then I am writing the description of the forces. So, this force what is this? This is τ_{12} sorry τ_{21} direction normal is 1. So, $\tau_{21} \Delta x_2 \Delta x_1$, what is the area $\Delta x_2 \Delta x_1$, what is this? This is $\tau_{21} \Delta x_2 \Delta x_1$ plus $\Delta \tau_{21} \Delta x_2 \Delta x_1$; what is this? This is τ_{21} why it is τ_{21} because its normal direction is 2 minus 2 actually and force acting along 1

So, $\tau_{21} \Delta x_2 \Delta x_1$, and this is $\tau_{21} \Delta x_2 \Delta x_1$ plus $\Delta \tau_{21} \Delta x_2 \Delta x_1$. So, now, we are essentially interested about the moment of forces and as I told for calculating the moment of the forces this incremental changes will not essentially matter this changes will not essentially matter. So, essentially for calculating the moments these 2 forces are like almost equal and opposite they are actually not equal there is an

incremental change, but in the limit they are almost equal opposite and anti parallel. So, they will form a couple.

So, the couple moment of these 2 forces what is that clockwise or anticlockwise you see this force is upward and this force is downward, so this is creating an anticlockwise moment, so this is a positive moment. So, this is $\tau_{12} \text{ into } \delta x_1 \text{ into } \delta x_2$. So, $\tau_{12} \text{ into } \delta x_2$ was the force times δx_1 is the arm of the force, arm of the moment. So, $\tau_{12} \text{ into } \delta x_1 \text{ into } \delta x_2$ plus there will be higher order effect because of this we are neglecting that, then let us consider these 2 forces these 2 forces will create a clockwise moment.

So, this is $\tau_{21} \text{ into } \delta x_1 \text{ into } \delta x_2$; if you are interested you write the higher order terms also, but I am writing symbolically as higher order terms; this if there is any body force if it passes through the point O that will also not create any moment. So, even if there is a body force it is still that will not give rise to a moment that is equal to the moment of inertia. Moment of inertia is for this element is $\frac{1}{12} \text{ into } m \rho \delta x_1 \text{ into } \delta x_2, \text{ into } \delta x_1^2 \text{ plus } \delta x_2^2 \text{ by } 12$ oh sorry 12 I have already written that into θ double dot.

Now, take the limit as $\delta x_1, \delta x_2$ tends to 0 that is this entire element shrinks to a point. If you take this limit then what happens first of all these terms will be canceled then if you take the limit then this higher order term will tend to 0 and this terms will tend to 0 therefore, you are left with τ_{12} is equal to τ_{21} . So, from angular momentum consideration, whatever we get we can generalize this and write τ_{ij} is equal to τ_{ji} . This is a very powerful relationship, but we have to keep in mind that we have made one very important assumption while deriving this; what is the assumption? The assumption that we have made is that we have neglected anybody couple; see if there is a body force it is still because its moment with respect to o is 0 until and unless it is asymmetrically distributed, but if there is a body couple then because of the body couple there could be an additional couple moment. So, the assumption is let me write it and assumption no body couple.

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$$\rho \Delta x_1 \Delta x_2 \Delta x_3 \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = \left[\frac{\partial \tau_{ji}}{\partial x_j} + \rho b_i \right] \Delta x_1 \Delta x_2 \Delta x_3 + \text{h.o.t.}$$

Lim $\Delta x_1, \Delta x_2, \Delta x_3 \rightarrow 0$

$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = \frac{\partial \tau_{ji}}{\partial x_j} + \rho b_i$$

Cauchy Eq/
Navier eq

$\rho, u_1, u_2, u_3, \tau_{ji} \rightarrow 9$

No of unknowns

No of independent equations

continuity
x-mom (i=1)
x₂-mom (i=2)
x₃-mom (i=3)

13

4

do not match

$\tau_{ij} = \tau_{ji} \rightarrow 13 \rightarrow 10$

So, let us revisit the number of equations and the number of unknowns, I will bring the previous page. So, what you see here is that we had 13 number of unknowns and 4 number of independent equations. Now we have proven that τ_{ij} is equal to τ_{ji} ; with this proof the number of unknowns they get reduced, from 13 the unknowns become what. See out of this 9 τ_{ij} now 6 are only independent, because τ_{ij} is equal to τ_{ji} . So, τ_{12} is equal to τ_{21} , τ_{13} is equal to τ_{31} , and τ_{23} is equal to τ_{32} . So, we can say that out of this 9 unknowns we have only 6 as independent unknowns. So, from 13 we have reduced to 10, but we still have a gap between 10 equations sorry 10 unknowns and 4 independent equations.

stress, is it deformation or rotation? So, I am asking this question in this chat; out of deformation and rotation which will give rise to stress out of deformation and rotation which will give rise to stress please answer this question in the chat. So, I have got already got five answers, and all these answers are correct that it is deformation that gives rise to stress.

But it is also true that when the fluid is not under motion or under deformation, then also there is a stress. What physically contributes to the stress when the fluid is at rest this is my next question. So, we have now understood that when the fluid is under deformation there is a stress because of the deformation, but even if the fluid is not under deformation or under motion, there could be a source of stress what is that source of stress that is my question please give answer in the chat. So, my question is that if the fluid is under even if the fluid is under rest, it is subjected to a stress what is the source of that stress that is not deformation, but what is the source of that stress.

I have got one answer which is viscosity this is not correct, because viscosity will give rise to stress when the fluid is under deformation or motion. So, the other answer is weight that is also not correct. So, it is of course, it is it may be related to weight no doubt about it, but it is a fundamental fluid property which is not weight. Body force is another answer which is also not correct. So, I have not yet got a correct answer in the chat another answer is mass which is also not correct, gravitation force this is also not correct. So, I am looking for the correct answer and anyway let me discuss about the correct answer. So, the correct answer. So, I have not yet got correct answers, so many answers I have got in the chat, but none of these are correct. So, one answer is very close to the correct answer which is the most recent answer, any non tangential forces. So, non tangential force, but you have to be specific what is the physical origin of a non tangential force, when the fluid is at rest? It is the pressure of the fluid. So, when the fluid is at rest the force is due to pressure the stress is due to or stress is due to pressure this is the normal stress.

When the fluid is deforming, still then the pressure is present; stress is due to pressure plus stress is due to deformation. The deformation can be linear or angular linear will give rise to volumetric deformation and shear deformation, angular deformation will give rise to shear deformation. So, what is the expression for the volumetric deformation? The expression for the volumetric deformation is divergence of the velocity vector, that is $\text{div } \mathbf{v}$

$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y}$ this is the rate of volumetric strain this is the quantification of the rate of volumetric strain. So, this is $\frac{\partial u_1}{\partial x_1} + \frac{\partial u_2}{\partial x_2} + \frac{\partial u_3}{\partial x_3}$ in terms of the index notation. So, as per the index notation this is as good as $\frac{\partial u_k}{\partial x_k}$. Remember this k index is arbitrary you can put as l m n whatever, because it is a repeated index it is a dummy index because there will be summation over this it will become number instead of the index.

Instead of the symbolic index it will become the number. So, and the shear deformation is related to $\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$ and. In fact, this shear deformation and also linear deformation why? Because if i and j are the same then it is $\frac{\partial u_i}{\partial x_i}$. So, that becomes that represents linear deformation. So, this is a representative of both shear and linear deformation, because there is no guarantee that i and j are the same are different if i and j are different it will represent shear deformation, if i and j are the same it will represent linear deformation. So, let us write summarily that τ_{ij} is a function of what.

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τ_{ij} is a function of p , $\frac{\partial u_k}{\partial x_k}$, $\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$

$\frac{\partial u_k}{\partial x_k}$ Influence Normal stress

$\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$ Influence shear & trans.

Kronecker delta δ_{ij} $\delta_{ij} = 1$ if $j = i$
 $= 0$ otherwise

For a Newtonian fluid, relationship between τ_{ij} and rate of deformation is linear. If the fluid is homogeneous and isotropic, that can be represented by a point-independent and direction-invariant fluid property called as viscosity.

$\tau_{ij} = -p \delta_{ij} + \lambda \frac{\partial u_k}{\partial x_k} \delta_{ij} + \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$

λ volumetric dilation coeff

μ viscosity.

τ_{ij} (stress)

$\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$ (rate of deformation)

So, τ_{ij} is a function of pressure $\frac{\partial u_k}{\partial x_k}$, and $\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$. Out of these 2 will be giving rise to normal stress, influence normal stress and these influence shear stress. So, now, we can write therefore, τ_{ij} as a function of this is like τ_{ij} is related to this. Now you can write, so to isolate the normal and shear component we use a notation which is τ_{ij} sorry which is the kronecker delta notation δ_{ij} . δ_{ij} is again a second order tensor is equal to 1 if j is equal to i , and is equal to 0

otherwise. So, if we use δ_{ij} you make sure that it is only the j equal to i that is the normal component, and that is non 0 and the shear component will be 0. So, these if they are multiplied by δ_{ij} they will give you a correct picture of the normal stress, you do not have to specify separately normal stress you just multiply this with δ_{ij} .

Because δ_{ij} is equal to one only if j equal to i . So, that will only have normal component; therefore, τ_{ij} will be a function of these multiplied by δ_{ij} plus a function of these. Now question is that what kind of function. The function can be any arbitrary function, but the dependence on the deformation if it is linear then it is called as a Newtonian fluid. So, for a Newtonian fluid relationship between τ_{ij} and rate of deformation is linear.

If the liquid is homogeneous and isotropic that can be represented by a homogeneous means it is independent of position, position independent and isotropic means direction invariant fluid property called as viscosity. So, let us write the expression for τ_{ij} , this is minus $P \delta_{ij}$ why δ_{ij} with a normal component I have already described; why minus the minus the reason is see positive sense of pressure is compressive in nature and positive sense of normal stress is tensile in nature. So, to adjust for that there is a negative sign.

Now, it is also linearly related to the volumetric deformation. So, we put a dilation coefficient λ , this is volumetric dilation coefficient plus μ . So, this λ is volumetric dilation coefficient, and this μ is viscosity this is the generalized Newton's law of viscosity. Now most of you again I presume that most of you are teachers in various colleges and you teach your students Newton's law of viscosity through a very simple relationship $\tau_{ij} = \mu \frac{d u_j}{d x_i} - \mu \frac{d u_i}{d x_j}$. Now suddenly you see this expression and you may have a doubt that what is then that $\tau_{ij} = \mu \frac{d u_j}{d x_i} - \mu \frac{d u_i}{d x_j}$, I mean how it relates to this $\mu \frac{d u_j}{d x_i} - \mu \frac{d u_i}{d x_j}$. So, you look into this expression and we will soon clarify that $\tau_{ij} = \mu \frac{d u_j}{d x_i} - \mu \frac{d u_i}{d x_j}$ is a special case of these. So, when we say $\tau_{ij} = \mu \frac{d u_j}{d x_i} - \mu \frac{d u_i}{d x_j}$ first of all there is if I get back to the expression that I have written, I have missed δ_{ij} here. So, please write this δ_{ij} .

Then $\tau_{ij} = \mu \frac{d u_j}{d x_i} - \mu \frac{d u_i}{d x_j}$ when we say we are considering only this portion this portion. So, if you write this part, this is this will imply that τ_{xy} just use the x y notation i equal to x and j is equal to y is equal to $\mu \frac{d u}{d y} - \mu \frac{d v}{d x}$. If you

are considering a unidirectional flow, then there is v and there is u and no w . So, that makes τ_{xy} viscous is $\mu \frac{du}{dy}$, but you have to keep in mind that there could also be a stress we will relate this volumetric dilation with the viscous effect soon, but that is an implicit thing that will come into the picture later if there is a volumetric deformation then there is an additional stress that may not be explicitly viscous stress, but there will be an additional stress.

So, τ_{ij} equal to $\mu \frac{du_i}{dx_j}$ is an exclusive viscous stress representation when it is unidirectional flow, where the velocity component is along x which varies along y . So, τ_{ij} there is nothing wrong with τ_{ij} equal to $\mu \frac{du_i}{dx_j}$, but that is a very special case, but this is something which is a more general case. So, let us go ahead with this representation of τ_{ij} , I am taking a fresh page and writing it again.

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$$\tau_{ij} = -p \delta_{ij} + \lambda \frac{\partial u_k}{\partial x_k} \delta_{ij} + \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

Ex: $\tau_{11} = -p + \lambda \frac{\partial u_k}{\partial x_k} + 2\mu \left(\frac{\partial u_1}{\partial x_1} \right)$ ✓
 $\tau_{22} = -p + \lambda \frac{\partial u_k}{\partial x_k} + 2\mu \left(\frac{\partial u_2}{\partial x_2} \right)$ ✓
 $\tau_{33} = -p + \lambda \frac{\partial u_k}{\partial x_k} + 2\mu \left(\frac{\partial u_3}{\partial x_3} \right)$ ✓

$$-p_m = -p + \left(\lambda + \frac{2\mu}{3} \right) \frac{\partial u_k}{\partial x_k}$$

below $p_m = p$?
 (1) $\frac{\partial u_k}{\partial x_k} = 0 \rightarrow$ incompressible flow
 (2) dilute, monoatomic gas

$\tau_{ij} \rightarrow$ expressed in terms of p, μ , velocity related parameters.

$p \rightarrow$ Thermodynamic pressure
 $p_m \rightarrow$ Mechanical pressure
 $= -\frac{(\tau_{11} + \tau_{22} + \tau_{33})}{3}$

Stokes Hypothesis
 $\lambda + \frac{2\mu}{3} = 0$
 Stokesian fluid
 $\lambda = -\frac{2}{3}\mu$

So, τ_{ij} now very interesting is the normal stress. So, take an example τ_{11} what is δ_{11} ? δ_{11} is 1 because δ_{ij} is 1 when i is equal to j . So, minus P plus $\lambda \frac{\partial u_k}{\partial x_k} \delta_{ij}$ plus $2\mu \frac{\partial u_1}{\partial x_1}$; then τ_{22} this is minus P plus $\lambda \frac{\partial u_k}{\partial x_k} \delta_{ij}$ plus $2\mu \frac{\partial u_2}{\partial x_2}$; τ_{33} minus P plus $\lambda \frac{\partial u_k}{\partial x_k} \delta_{ij}$ plus $2\mu \frac{\partial u_3}{\partial x_3}$.

Now, this P what we write here is called as thermodynamic pressure, why do we call it thermodynamic pressure? We call it thermodynamic pressure because of the following reason, we call it thermodynamic pressure because this pressure will be related to the

thermodynamic equation of state for example, for an ideal gas P is equal to $\rho r t$. So that means, the equation of state relates basically pressure density and temperature and the thermodynamic pressure is that pressure which appears in the equation of state, but there is also another terminology which is called as mechanical pressure, which is defined as minus of τ_{11} , plus τ_{22} , plus τ_{33} by 3. So, this is arithmetic average of the normal components of the stress. So, let us find the mechanical pressure. So, what we will do is we will add this and divide by 3. So, that will become minus P_m is equal to minus P plus $\lambda \frac{\Delta u}{\Delta x}$ will become λ , plus this will be $2\mu \left(\frac{\Delta u_1}{\Delta x_1} + \frac{\Delta u_2}{\Delta x_2} + \frac{\Delta u_3}{\Delta x_3} \right)$. So, that is $2\mu \left(\frac{\Delta u}{\Delta x} \right)$. So, that becomes 2μ then divide by 3 becomes $\frac{2\mu}{3} \left(\frac{\Delta u}{\Delta x} \right)$.

So, this is an expression that relates mechanical pressure with thermodynamic pressure. So, we get a very important fundamental concept that mechanical pressure may not be equal to thermodynamic pressure right, but there are common cases when mechanical pressure is identically equal to thermodynamic pressure. So, let us write that. So, first question is why mechanical pressure and thermodynamic pressure are not the same.

So, mechanical pressure will usually reflect the translational degrees of freedom of molecules whereas, the thermodynamic pressure will reflect all sorts of degrees of freedom translational vibrational, rotational also these to represent different philosophical aspects or different conceptual aspects of the quantity pressure. Now when P_m and P are identical equal, case one which is a very straightforward case that you have $\frac{\Delta u}{\Delta x} = 0$. What is this? This is incompressible flow. So, for incompressible flow P_m and P are identically equal, the second possible scenario is dilute mono atomic gas. So, for dilute mono atomic gas you have only translational degree of freedom and no other degrees of freedom and therefore, they are identically equal this term will be 0.

Otherwise it is not possible to say that they are equal, but Stokes made an hypothesis which is called as Stokes hypothesis, that $\lambda + \frac{2}{3}\mu = 0$. If this is 0 then mechanical pressure and thermodynamic pressure are the same, and we have seen that this need not always be the case, but we will figure out that when is such a case true. First of all the fluids which obey this scenario or obey this relationship these are called as Stokesian fluids. λ is equal to minus two-third of μ ; now let us discuss little bit about Stokes hypothesis.

Now, we have to keep in mind that every fluid when subjected to a change will take some time to adapt to the change, and this time is known as relaxation time. So, how will it adapt to the change? It will adapt to the change by arriving at new thermodynamic properties. Now this time that the fluid takes to adapt to the change it is known as the relaxation time, but there can be a very interesting situation when the time over which the change on the fluid is imposed is shorter than the relaxation time, then the fluid cannot adapt to the change and then a new change by the time has come.

So, then we can say that it is a very rapid process; let us take an example let us consider that there is a bubble which is rapidly expanding contracting expanding contracting like this. So, if you have a bubble which is rapidly expanding and contracting then what is happening is that that the time scale of imposition of the change is very rapid, it is very fast and if its relaxation time is not that fast, then the fluid cannot relax to a new thermodynamic stage before a new change has taken place. And then it cannot equilibrate its properties so that, it can convert all degrees of freedom to the translational degrees of freedom instantaneously.

So, if it cannot convert all degrees of freedom to the translational degrees of freedom instantaneously, then the equality of mechanical pressure and thermodynamic pressure will not take place. Therefore, for very rapid changes where the time scale of imposition of the change is faster than the relaxation time of the fluid, then the stokes hypothesis will no more be valid.

So, it is very important to understand that stokes hypothesis is not a law, it is a hypothesis. So, it does not have a proof, but there are physical arguments that for most of the practical scenarios the time scale of imposition of the disturbance is significantly larger as compared to relaxation time scale. So, that fluid relaxes almost instantaneously, so that it attains an equilibration of mechanical and thermodynamic properties, and then we can say that of course, the stokes hypothesis is a very general hypothesis, there are exceptional scenarios when the change is so fast that the stokes hypothesis is not valid, otherwise the stokes hypothesis will be valid.

So, now when the stokes hypothesis is valid, so let us come to this equation now. So, λ is equal to $-\frac{2}{3}\mu$. So, let us come to this equation which we have just written. So, now, in place of this λ you write $-\frac{2}{3}\mu$, then the τ_{ij} we are able to write

in terms of how many parameters; expressed in terms of P mu and velocity related parameters. So, we have reduced the number of unknown parameter see tau i j, there were 6 unknowns and we have reduced the number of unknowns. So, now, what we will do is we will complete the derivation. So, we had a equation on tau i j for Newtonian homogeneous isotropic Newtonian and Stokesian fluid, we will substitute that in the Navier's equation.

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Navier eq:

$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = \frac{\partial \tau_{ji}}{\partial x_j} + \rho b_i$$

Angular momentum: $\tau_{ij} = \tau_{ji} \rightarrow$ in the absence of body couples

$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = \frac{\partial \tau_{ij}}{\partial x_j} + \rho b_i$$

Stokes hypothesis $\lambda = -\frac{2}{3}\mu$

$$\tau_{ij} = -p \delta_{ij} + \lambda \frac{\partial u_k}{\partial x_k} \delta_{ij} + \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \rightarrow \text{homogeneous, isotropic, Newtonian fluid.}$$

$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_i} \left(\lambda \frac{\partial u_k}{\partial x_k} \right) + \frac{\partial}{\partial x_j} \left(\mu \frac{\partial u_i}{\partial x_j} \right)$$

$$\mu \frac{\partial}{\partial x_j} \left(\frac{\partial u_i}{\partial x_j} \right) = \mu \frac{\partial}{\partial x_i} \left(\frac{\partial u_j}{\partial x_j} \right) = \mu \frac{\partial}{\partial x_i} \left(\frac{\partial u_k}{\partial x_k} \right)$$

So, let us write the Navier's equation. So, this was the Navier's equation, now from angular momentum consideration tau i j is equal to tau j i. So, then this will become then we will use the expression for tau i j here, minus P delta i j plus lambda del u k del x k delta i j, plus mu del u i del x j, plus del u j del x i; and stokes hypothesis lambda is equal to minus two- third mu.

So, angular momentum this in the absence of body tuples, so I am writing the assumptions and this is for homogeneous isotropic Newtonian fluid. So, we will substitute this equation here. So, if we substitute this equation here we will get rho del u i del t, plus u j del u i del x j is equal to del tau i j del x j. So, first this term del del x j of minus p delta i j. So, delta i j will become j when j is equal to i so; that means, it will be minus del p del x i you follow it, it is minus del x j of p into delta i j, but delta i j is equal to 1 only if j is equal to i. So, that is how this j has become i; very similarly the next term is del del x i of lambda del u k del x k.

Then the next couple of terms plus $\frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j})$, but μ if it is homogeneous it can be taken in and out of the derivative as per your wish, plus $\frac{\partial}{\partial x_i} (\lambda \frac{\partial u_k}{\partial x_k})$. So, now, we will manipulate this last term see μ is a constant. So, we can we have taken it to be function of independent of position. So, this we can write $\mu \frac{\partial}{\partial x_j} (\frac{\partial u_j}{\partial x_i})$. So, this is as good as $\mu \frac{\partial}{\partial x_i} (\frac{\partial u_j}{\partial x_j})$. Why we can write this? We can write this with an assumption of the continuity of the second order derivative. So, we it does not matter whether we differentiate with respect to x_j first or with respect to x_i first; and this is as good as $\mu \frac{\partial}{\partial x_i} (\frac{\partial u_k}{\partial x_k})$ see this j is repeated index therefore, it is a dummy index does not matter whether you instead of j we write k l m n whatever. So, $\mu \frac{\partial}{\partial x_i} (\frac{\partial u_k}{\partial x_k})$.

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$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = - \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_i} (\lambda \frac{\partial u_k}{\partial x_k}) + \frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j}) + \frac{\partial}{\partial x_i} (\mu \frac{\partial u_k}{\partial x_k}) + \rho b_i$$

$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = - \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j}) + \frac{\partial}{\partial x_i} (\lambda + \frac{2}{3} \mu) \frac{\partial u_k}{\partial x_k} + \rho b_i$$

$$\rho \left[\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right] = - \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j}) + \frac{\partial}{\partial x_i} (\frac{\lambda}{3} + \frac{2}{3} \mu) \frac{\partial u_k}{\partial x_k} + \rho b_i$$

Navier Stokes eq.

- Homogeneous fluid
- Isotropic fluid
- Newtonian fluid
- Stationary fluid
- No body couple

i=1
i=2
i=3.

So, now we can write $\rho \frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j}$ is equal to minus $\frac{\partial p}{\partial x_i}$ plus $\frac{\partial}{\partial x_i} (\lambda \frac{\partial u_k}{\partial x_k})$, plus $\frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j})$ plus the last term because μ is a constant we can take the μ also inside as per our wish constant means independent of position. So, the last term we can write $\frac{\partial}{\partial x_i} (\mu \frac{\partial u_k}{\partial x_k})$, again we can bring the μ into the derivative.

So, now we can combine these two terms and write the equation; $\rho \frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j}$ is equal to minus $\frac{\partial p}{\partial x_i}$, we are writing this term first plus $\frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j})$, plus $\frac{\partial}{\partial x_i} (\lambda + \frac{2}{3} \mu) \frac{\partial u_k}{\partial x_k}$. So that means, you have $\rho \frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j}$ is equal to minus

$\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} (\mu \frac{\partial u_i}{\partial x_j} - \lambda \frac{\partial u_k}{\partial x_k})$ what is $\lambda + \mu$ λ is minus two-third of μ . So, $\frac{\partial}{\partial x_i} (\mu \frac{\partial u_k}{\partial x_k})$, this equation is known as Navier stokes equation.

So, what are the assumptions that we have taken for this equation? Homogeneous fluid, Isotropic fluid, Newtonian fluid, Stokesian fluid and nobody couple. So, these are the assumptions that we have made, where does the body couple come into the picture see we often Navier stokes equation we say momentum equation or momentum conservation equation, we do not say linear or angular because the fact that we have used τ_{ij} is equal to τ_{ji} , oh there is a body force in this equation which we have not written ρb_i yes plus ρb_i . So, add the body force in all the equation. So, we have assumed that τ_{ij} is equal to τ_{ji} . So, that is true if there is no body couple.

Now, we have derived the Navier stokes equation and this has essentially 3 components; i equal to 1, i equal to 2, i equal to 3 plus you have the continuity equation.

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Continuity equation: $\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x_j} (\rho u_j) = 0$

No of independent equations	No of unknowns
continuity x_1 -mom x_2 -mom x_3 -mom	ρ u_1 u_2 u_3 p
} 4	} 5

Special case: ρ given \Rightarrow 4 eqs & 4 unknowns \Rightarrow match

If ρ is not given $\rightarrow \rho = \rho(p, T) \rightarrow$ Eq of state (add eq.)

If T is not a constant, require another governing eq for $T \rightarrow$ Energy equation

So, I am writing the continuity equation in the index notation, this is the continuity equation in the index notation. So, now, let us check number of equations, independent equations and number of unknowns; first let us write the unknowns. So, now, if you look at this equation I am bringing. So, first in the continuity equation what are the unknowns you see?

Let us list the unknowns ρ , u_1 , u_2 , u_3 ; in the Navier Stokes equation you have again ρ , u_1 , u_2 , u_3 there is a additional unknown which is pressure. So, P how many unknowns are there 1, 2, 3, 4, 5, 5 unknowns what are the independent equations the continuity? X_1 momentum, x_2 momentum, and x_3 momentum. So, 4 equations, still there are number of unknowns number of equations do not close each other, in a special case they will close each other what is the special case when ρ is known.

So, let us write this special case; ρ is given, it may be a constant it may be a function of other variables, but if it is given then you have four equations and four unknowns that matches. So, you do not require any more equation. If ρ is not given then what will ρ depend on? If ρ is not given you may assume that ρ is a function of pressure and temperature, this is equation of state additional equation. So, you may believe that with this additional equation you close the equations with unknowns, but a very interesting thing with this additional equation you have got an additional parameter which is temperature.

So, for an isothermal case it is good enough because there the temperature is constant it is no more a variable, but if T is not a constant you require another governing equation for temperature, and then the system will close and that equation is called as energy equation. We will discuss about this energy equation in one of our upcoming courses on convective heat transfer on convection where we will discuss about this energy equation. In this brief purview of the dynamics of fluids flow we do not have the scope of discussing the energy equation, but I am just showing you that how these fluid flow equations are coming in the context of the energy equation or in other words how the energy equation is coming in the context of the fluid flow equations.

Now, a final note, so we have seen that Navier Stokes equation, now when what are the characteristics or what are the specialties of the Navier Stokes equation. See the Navier Stokes equation is a second order partial differential equation, and it is a non-linear partial differential equation why non-linear? It is non-linear because of this term $u_j \frac{\partial u_i}{\partial x_j}$ and it is a as coupled system of partial differential equations. So, that is one of the big challenges. So, you can have a linear system of equations if the left hand side of the Navier Stokes equation becomes 0, and we will see in the next lecture that there are certain special cases like fully developed flow and all this we will discuss in the next

lecture, when the left hand side of the Navier stokes equation becomes 0, and then that equation is called as stokes equation.

So, remember that I mean stokes was a brilliant mathematician. So, he contributed several things in fluid mechanics. So, there is something which is called as a stokes law. Stokes law is the law that describes the drag force of a slowly moving sphere in a viscous liquid. So, that is stokes law, then stokes equation stokes equation is the Navier stokes equation with this left hand side equal to 0, and stokes hypothesis that we have discussed today that is λ is equal to minus two- third μ .

So, we need not confuse all this stokes law, stokes equation and stokes hypothesis and. So, until and unless the left hand side is 0 it is a system of coupled non-linear partial differential equations. Not only that you have pressure as a variable although you do not have an explicit governing equation for pressure. So, all this makes the solution of the Navier stokes equation very challenging until and unless you encounter very simple special cases, otherwise analytical solution may be difficult and we have to go for computational techniques. The computational techniques that solve the Navier stokes equation coupled with the continuity equation, these are known as computational fluid dynamics or c f d technique.