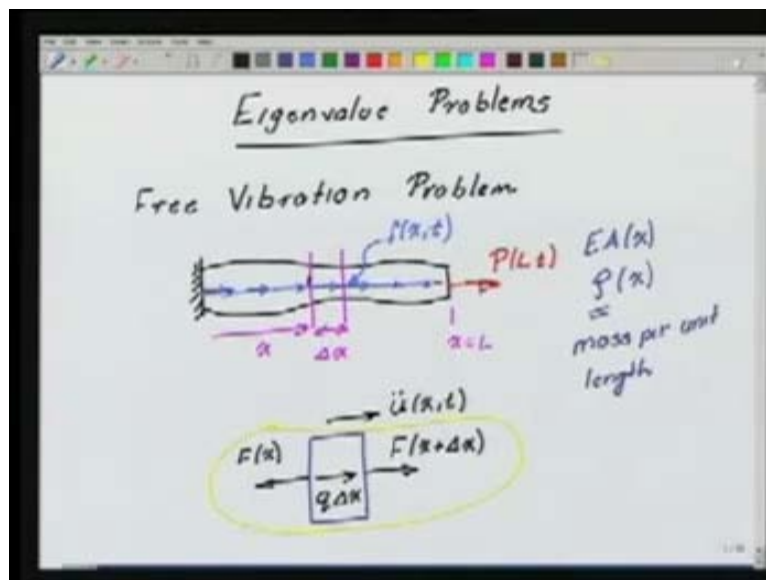


Finite Element Method
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Module – 11 Lecture - 02

Till now we have discussed the static problem in 1 and 2 dimensions and with a little bit of introduction to what happens in 3 dimensions. Let us now move on to a different class of problems.

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So what we are going to look at need not necessarily be from the same type of analysis. We will look at Eigen value problems and under this big class of Eigen value problems, we are going to look at to start up with specific example of the free vibration problem. What do we mean by the free vibration problem? In order to get to the free vibration problem, let us do a review of what do we mean by it in the case of a deformable structure.

Let us again go back to our 1 dimensional example problem of a bar, possibly of a variable cross section but with the central line of the symmetry with an end load P which is a function of where it is being applied and time and a distributed load in the interior. The loading is changing with time that is the end load which is applied on the bar is changing with time as well as the

distributed load on the body. It could be due to any temporal disturbances arising from the environment in which the body exists. That is not the major issue but issue is we now like to set up the equation of motion. Till now we had been only talking about equation of equilibrium or from the conservation of linear momentum. We did not have the inertia part. Now we have to build in the inertia part. Go again to a section at a distance x from the end. This is at the end x equal to l and take the section of size Δx . We will assume that the material is such that the E and the area are functions of x . The density is also a function of x and this density is mass per unit length. This is what we mean by this density mass per unit length.

When we have this thing given to us, these quantities, we would now like to pose, what is the equation of motion corresponding to this small piece that we have cut of? Let us take that small piece and look at the force system acting on it. So the interaction of this piece with its neighbor on this side is this actual force F at $x + \Delta x$. Interaction of this piece on this side with its remaining neighbor is the force F at x . The effect of the distributed load is $q \Delta x$ and we have to remember that this piece has an acceleration given by the second derivative of u with respect to time, u double dot as, when I write dots, it means derivative with respect to time. This is essentially under this force system, this piece is going to move. So how do I write the equation of motion? This all must have done in the standard dynamics classes that I will have in the actual direction.

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The image shows a whiteboard with handwritten mathematical equations. The top equation is $F(x+\Delta x) - F(x) + f(x,t)\Delta x = \rho\Delta x\ddot{u}(x,t)$, with a bracket underneath labeled "Resultant force". Below this, the equation $\frac{\partial}{\partial x} \left(EA \frac{\partial u}{\partial x} \right) \Delta x + f \Delta x = \rho \ddot{u} \Delta x$ is written. At the bottom, the final differential equation $\frac{\partial}{\partial x} \left(EA \frac{\partial u}{\partial x} \right) + f = \rho \ddot{u}, 0 < x < L$ is boxed in blue.

If I look at force balance, $F(x + \Delta x)$ minus $F(x)$, plus f as the function of x and t , Δx , this will be resultant force acting on this piece in the positive x direction. This will be equal to $\rho \Delta x$ which is the mass of this piece into the acceleration of this piece. The resultant force is equal to the mass into acceleration. So this is very easy for us to do, because we have already done this part in the static case. So this part will give me $\frac{\partial}{\partial x}$ of $EA \frac{\partial u}{\partial x}$ divided by Δx into Δx plus f , I am dropping the x and t part. It is understood now is equal to $\rho \ddot{u} \Delta x$. In the standard way, we remove this part and we are left with the differential equation $\frac{\partial}{\partial x}$ of $EA \frac{\partial u}{\partial x}$ plus f is equal to $\rho \ddot{u}$ for all x lying from 0 to L . This is for the dynamic case, the equation of motion for this actual bar member. We would like to do the free vibration analysis. What does the free vibration analysis mean? In the case of the free vibration analysis, this f is 0. The equation of motion is $\frac{\partial}{\partial x}$ of $EA \frac{\partial u}{\partial x}$ is equal to $\rho \ddot{u}$. How is the vibration happening? It is due to the initial conditions, initial perturbations or disturbances given to the body. We are interested in this analysis. What do we do next in the case of the free vibration analysis?

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$$u(x,t) = \underbrace{U(x) T(t)}_{\text{separation of variables}}$$
$$\frac{d}{dx} \left(EA \frac{dU}{dx} \right) T = \rho U \frac{d^2 T}{dt^2}$$
$$\Rightarrow \frac{\frac{d}{dx} \left(EA \frac{dU}{dx} \right)}{\rho U} = \frac{\frac{d^2 T}{dt^2}}{T} = \underbrace{-\omega^2}$$

In this case, we can do a partition or a separation of variable approach and write U is a function of x , a pure function of x and pure function of t , of x and t . So this is a partition or separation of variable approach that we have used. So we will use the separation variable approach and once we use it, the differential equation will now become d/dx of EA into dU divided by dx . This is the big U into T is equal to ρU into d^2T divided by dt squared, so this is what the differential equation will become.

Let us take a simple case first or in turn have to take it. Let us now take this and see that here is a time part, here is the time part, this is the space part, and this is the space part. That is the function of x and t . So let us bring the 2 parts together. So we have implies d/dx of EA dU divided by dx divided by ρU . This is equal to d^2T divided by dt squared divided by T . Since, this is a pure function of x . This is a pure function of t . The 2 can only be equal, when they both are constant. We cannot have a function of x being equal to function of time for all x and time except when they both equal to constant and that constant, we know from our basic vibration analysis is given in terms of minus omega squared with the argument that really in terms of time, the solution is going to remain bounded and oscillatory in nature. It will not blow up so because of that this function has to be minus omega squared. All that we can learn in our vibrations classes.

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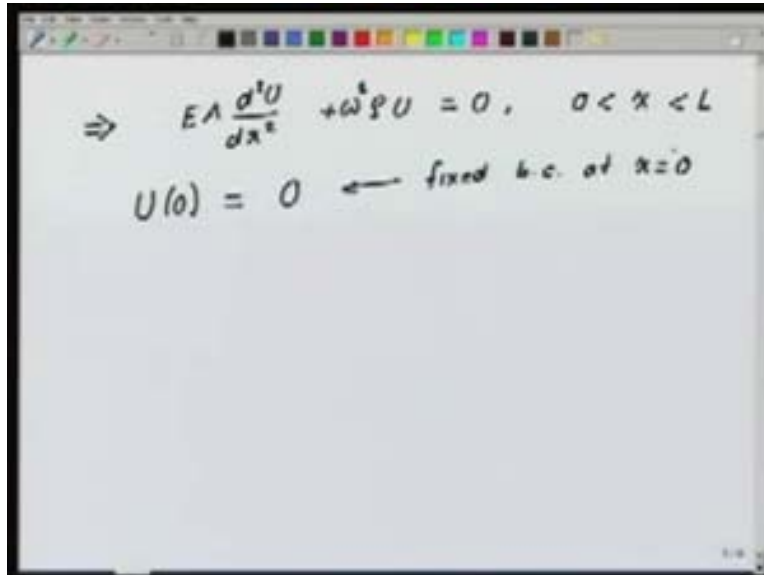
$$\Rightarrow \frac{d}{dx} (EA \frac{dU}{dx}) + \rho \omega^2 U = 0, 0 < x < L$$
$$\frac{d^2 T}{dt^2} + \omega^2 T = 0, t_0 < t < t_f$$

$\omega \rightarrow$ natural frequency
 $U \rightarrow$ mode
 $\rho \rightarrow$ Constant, $EA \rightarrow$ Constant

Let us go and look at this problem as the 2 separate problems that implies I will have $\frac{d}{dx}$ of EA , $\frac{dU}{dx}$ divided by dx plus $\rho \omega^2 U$ is equal to 0, $0 < x < L$. And I will have $\frac{d^2 T}{dt^2}$ plus $\omega^2 T$ is equal to 0, for $t_0 < t < t_f$ so initial time final time, that is why we have this. We can go on forever also. That is not the issue. We are interested in looking at this problem. Why? Because this problem gives us what is the value of this ω ? Corresponding to the ω , what is the value of this U ? So where ω forms the natural frequencies and U is the corresponding mode that is corresponding to this ω , what is the corresponding shape deflected shape of the bar as a function of x ? How do we solve it?

Let us take a simple case of the bar with ρ is constant. It is not a function of x and so is EA that is, I have a uniform cross section and it does not change neither does the material change. If it take that special case, then as an example, we will get implies, if I look at the problem corresponding to the U , I will get, $EA \frac{d^2 U}{dx^2}$ plus ρU is equal to 0 in $0 < x < L$.

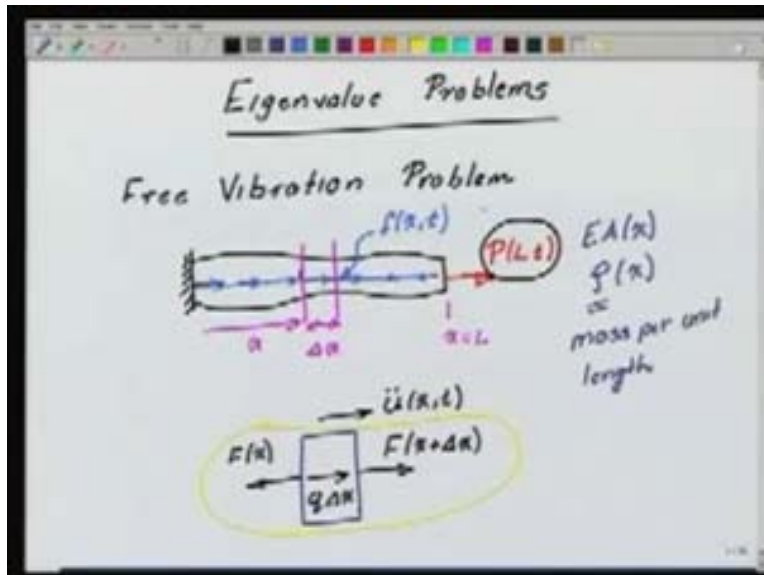
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The image shows a whiteboard with handwritten mathematical equations. The top equation is $\Rightarrow EA \frac{d^2 U}{dx^2} + \omega^2 \rho U = 0, \quad 0 < x < L$. Below it is $U(0) = 0 \leftarrow \text{fixed b.c. at } x=0$. The whiteboard has a toolbar at the top with various drawing tools.

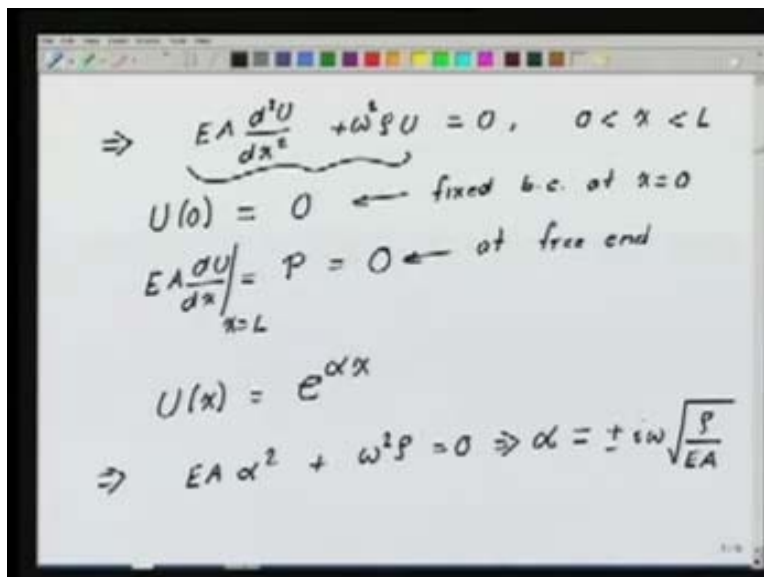
It will be into omega squared, omega squared rho U will be 0. How do I find omega and how do I find the corresponding U's? As for as U is concerned, U at 0 for all times the displacement small u as the function of x and t was equal to 0, at x equal to 0. So U of 0 is equal to 0 from the fixed boundary condition at 0. For this particular problem, it need not be the same for all problems, boundary condition at x equal to 0.

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Further, we also have in the case of the free vibration problem, we are also going to force our this P as the function of L and t as the function of time to be equal to 0, because there is no external forcing function acting on the body. So for the free vibration problem, F will also be 0, P will also be 0. So in our case, what we will get? We will show it later on.

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EA dU divided by dx is equal to P is equal to 0. This is at point x equal to L. This is equal to P which is equal to 0. So this is at free end. The force at the free end will be given in terms of EA dU divided by dx because that is the actual force. This is equal to P which is equal to 0.

As far as this is concerned, this representation of the solution, in this case the solution can be represented as e to the power of alpha x implies we will get EA alpha squared plus omega squared rho is equal to 0 implies alpha is equal to plus minus i omega root of rho by EA, if I do this, alpha squared is equal to minus omega squared of rho divided by EA in the root of the minus will be plus minus i omega root of rho by EA.

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$$U(x) = A_1 \cos \alpha x + A_2 \sin \alpha x$$

$$U(0) = 0 \Rightarrow A_1 = 0$$

$$EA \left. \frac{dU}{dx} \right|_L = 0 \Rightarrow A_2 \alpha \cos \alpha L = 0$$

$$\cos \alpha L = 0$$

$$\Rightarrow \alpha L = \frac{(2n-1)\pi}{2}, \quad n=1, 2, \dots$$

$$\Rightarrow \alpha_n = \frac{(2n-1)\pi}{2L} \quad \leftarrow \text{infinite natural frequencies}$$

This will give the representation of U of x becomes equal to, we can write it, A₁ cosine alpha x plus A₂ sine alpha x because the alpha came out to be imaginary. So it is in terms of this. Now U at 0 equal to 0 implies the A₁ is equal to 0, because cos alpha x becomes 1. This term becomes 0. So I am left with A₂. We will have EA dU divided by dx at L is equal to 0 implies A₂ alpha cos alpha L is equal to 0. We are interested in the free vibration analysis that is we want the non trivial modes and the corresponding frequencies. Now if I say, this is equal to 0 which means A₂ equal to 0. We will end up getting a trivial solution. In order to not get a trivial solution, we want cos alpha L equal to 0 implies alpha L is equal to (2n-1) pi by 2, n is equal to 1, 2 and so on. It implies we will now write alpha_n is equal to (2n-1) pi by 2L, we have so called infinite modes. In

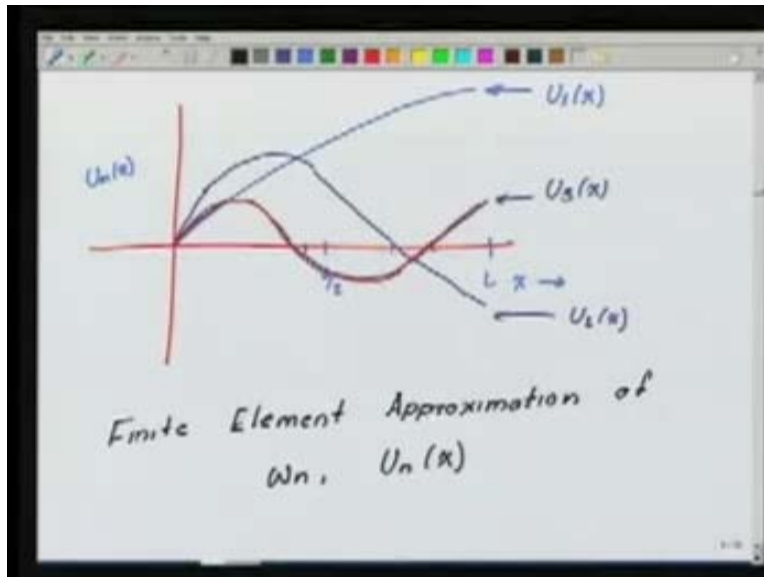
the case of a continuous system, we will have infinite natural frequencies. This is the very important point that we have to keep in mind before we progress to how to do the finite element approximation. So we have these infinite natural frequencies, for n going from 1 to infinity, 1, 2, 3 up to infinity and given this α_n , we have the corresponding mode.

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The image shows a whiteboard with handwritten mathematical derivations. At the top, it states $\alpha_n = \omega_n \sqrt{\frac{\rho}{EA}}$. Below this, an arrow points to the expression $\omega_n = \frac{(2n-1)\pi}{2L} \sqrt{\frac{EA}{\rho}}$, which is circled in blue. A bracket under this expression is labeled "nth natural frequency". Below the circled expression, the mode shape is given as $U_n(x) = \sin \alpha_n x$, which is boxed in red. A bracket under this expression is labeled "Oscillatory" in red, and another bracket to the right is labeled "nth mode" in red.

We will have α_n is equal to ω_n root of ρ over EA implies ω_n is equal to $(2n-1)\pi$ by $2L$ into root of EA by ρ . This is what we call as the natural frequency and by putting this value of α_n , this expression for α_n in the expression for the corresponding U see remember, we had this U here, we will get, U what we know as the corresponding mode for the natural frequency, this will be equal to $\sin \alpha_n x$, n th natural frequency, n th mode. This is very important thing to keep in mind, before we go for the approximation. Before we do the approximation of anything, we should not do it blindly, we should know what exactly the nature of the functions is, we are trying to approximate. All of the entities that we are going to approximate, nature not that we have to know the exact function. We should be able to say what is the nature? The nature is that these are oscillatory in nature and not only oscillatory as the n rises the oscillatory nature becomes more and more pronounce.

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For example for this problem if I plot, this U_n s as the function of x , this is x equal to L , this is L by 2. So here I will plot the $U_n(x)$. U_1 is essentially a sin of $\alpha_1 x$ which is 0 at x equal to 0, at x equal to L , it becomes 1 and we get something like this as $U_1, U_1(x)$. If I look at U_2 what will happen is, U_2 will do the following. It will go up come, this is $U_2(x)$. Similarly, if I look at $U_3(x)$, it will do the following. Let me reiterate that $U_3(x)$ is this function which is given by the corresponding choice. As the α is increasing, the n is increasing, this function the mode becomes more and more oscillatory in nature and this way, I can continue. This picture we have to have in mind because we have to see how well is the finite element solution doing as far as this problem is concerned.

We have these solutions for this then how do we go and construct a finite element solution? How to construct a finite element approximation of ω_n and U_n as the function of x ? In order to do this, we will go back to the differential equation that we had created as far as the spatial part is concerned, this differential equation. Let us come to that differential equation and write it down.

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The whiteboard shows the following steps:

$$\int_0^L w \left(\frac{d}{dx} \left(EA \frac{dU}{dx} \right) + \omega^2 \rho U \right) dx = 0, \quad 0 < x < L$$

$$\Downarrow$$

$$\int_0^L \left(\frac{d}{dx} \left(EA \frac{dU}{dx} \right) + \omega^2 \rho U \right) w dx = 0$$

$$\Downarrow$$

$$- \int_0^L EA \frac{dU}{dx} \frac{dw}{dx} dx + \omega^2 \int_0^L \rho U w dx = \left(EA \frac{dU}{dx} w \right) \Big|_0^L$$

We will have $\frac{d}{dx}$ of $EA \frac{dU}{dx}$ divided by dx is equal to ρU , $0 < x < L$. This is equal to minus $\omega^2 \rho U$ or I will erase it and rewrite it, here for the convenience plus $\omega^2 \rho U$ is equal to 0. In the standard finite element setting that we have been following till now over and over again. Take this differential equation multiply it with a weight function w and let me have w and the ω very clearly written of, integrated from 0 to L , then we will get, this will give me integral 0 to L , $\frac{d}{dx}$ of $EA \frac{dU}{dx}$ plus $\omega^2 \rho U$ whole thing into $w dx$ is equal to 0. This is the weighted residual form for this given differential equation, where ω is an unknown and U is an unknown. We have to solve for both. First we have to set up that problem. We see here again, we have the same problem that second derivative of u is sitting, w is sitting by itself. So do an integration by parts once for this quantity and this integration by parts will give me, integral of 0 to L , dx minus of this into dw divided by dx into dx plus integral 0 to L , I can take the ω^2 out, $\rho U w dx$, this will be equal to $EA \frac{dU}{dx}$ divided by dx into w evaluated at 0 and L .

From what we have done earlier, this part corresponds to the specified force on the boundary. This part has to be made 0 wherever the U is specified to be 0 on the boundary. At x is equal to 0, U is equal to 0, big U is equal to 0. So w also has to be 0, at x equal to L for the free vibration problem $EA \frac{dU}{dx}$ is equal to P is equal to 0. So that, this term is 0 at both the boundary. So I can knock this of. So this becomes equal to 0.

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The image shows a whiteboard with the following content:

Weak Form :

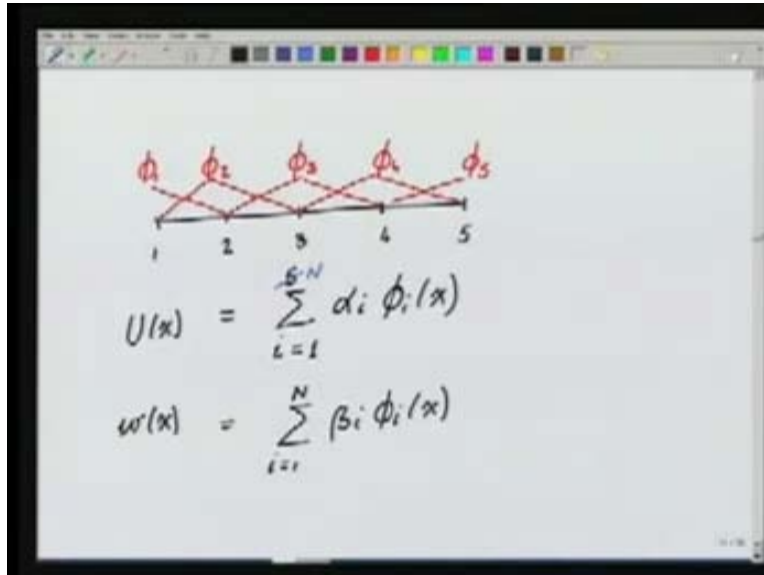
$$\int_0^L EA \frac{dU}{dx} \frac{dw}{dx} dx = \omega^2 \int_0^L \rho U w dx$$

Finite Element Approximation

\int
 C^0

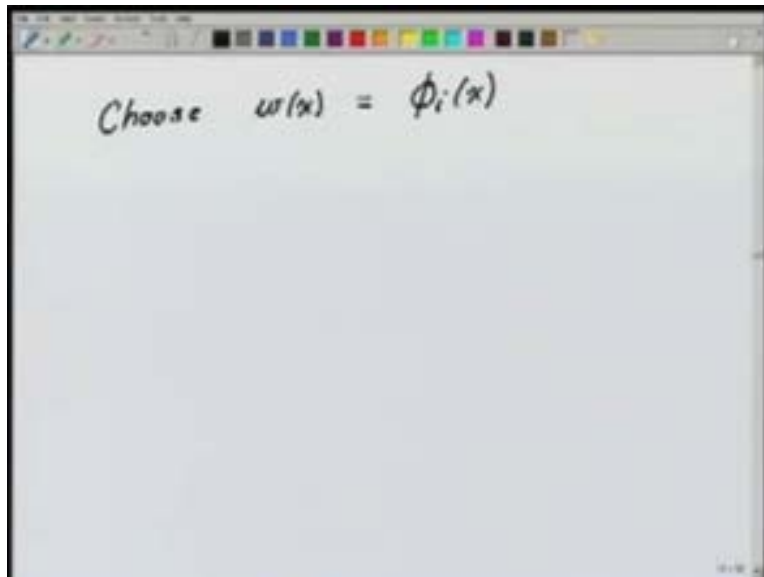
If I now write this weak form, as we have obtained here, this is the weak form, we will write it as integral of 0 to L, EA dU divided by dx into dw divided by dx is equal to omega squared integral 0 to L, rho U w dx. This is the weak form of the problem that we want to solve. Once we have this weak form then what do we do? We have to now create a finite element approximation. How do we do it? We see that again in this case, we want dU divided by dx to be defined. Similarly, dw divided by dx is to be defined. So we want functions U and w for which the first derivative is defined. That is the functions need to be only continuous, so as far as the approximation will go to the C⁰ approximation.

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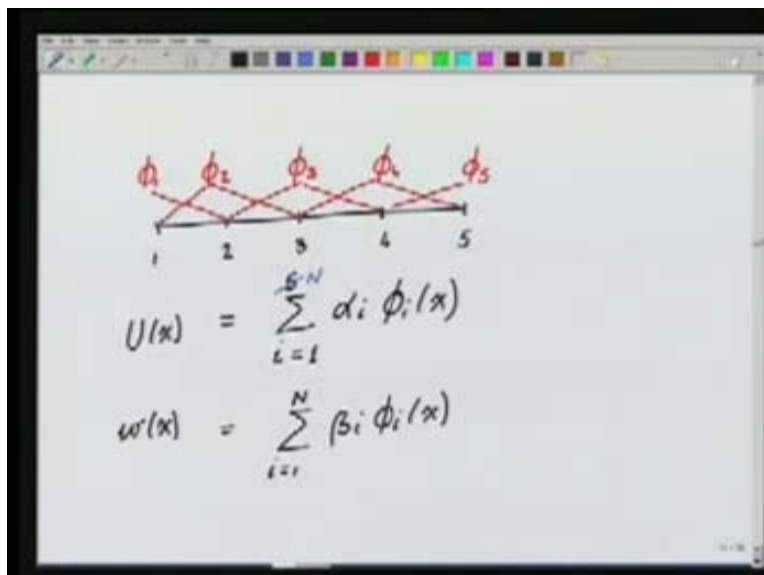
Now we will construct. So let us say that we have a bar, bar is broken into some number of elements. Let us this is 1, 2 as an example 3, 4, 5, I will go for the simplest possible representation of the basis functions which corresponds to the C^0 approximation which we have already done. We have this add functions. Everything else is exactly like what we did in the static problem, at all that we set up. Here we will have our ϕ_1 . This is basis function ϕ_2 . This is basis function ϕ_3 . This is ϕ_4 and this is ϕ_5 . We will write U as the function of x is equal to sum, i going from 1 to 5, here $\alpha_i \phi_i$. If I have, instead of 5, I generalize it. If I have N , then I will replace it by N . As many number of elements that we need, so here we had 4 elements 4 plus 1 is 5 degrees of freedom and if we have N elements and N plus 1 degrees of freedom. That is not a problem. So this is the representation that we do for the U of x . Similarly, the way we have done, the w of x will also be equal to sum i is equal to 1 to N . I will generalize it $\beta_i \phi_i$ of x .

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So we substitute this representation, so choose w as the function of x to be equal to ϕ_i of x because this choice is allowed from what we had argued earlier. Remember that in the charge for the w .

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The w has to satisfy the essential conditions wherever they are specified. In this problem for example β_1 has to be equal to 0 that is in the representation of w will actually start from i equal to 2 to N because the part multiplying ϕ_1 is 0, the w has to be 0 at 0.

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Choose $w(x) = \phi_i(x)$

$$\Rightarrow \int_0^L EA \left(\sum_{j=1}^N \alpha_j \phi_j' \right) \phi_i' dx = \omega^2 \int_0^L \rho \left(\sum_{j=1}^N \alpha_j \phi_j \right) \phi_i dx$$

$$\Rightarrow \sum_{j=1}^N \alpha_j \left(\int_0^L EA \phi_j' \phi_i' dx \right) = \omega^2 \sum_{j=1}^N \alpha_j \left(\int_0^L \rho \phi_j \phi_i dx \right)$$

K_{ij} M_{ij}

$[K] \rightarrow$ STIFFNESS MATRIX $[M] \rightarrow$ MASS MATRIX

So given all those things, we choose $w(x)$ is equal to $\phi_i(x)$ put it in the weak form that we have obtained, we will get implies integral 0 to L EA into sigma i is equal to 1 to N α_j . Instead of i , I will put it j , just again not to confuse $\alpha_j \phi_j$ prime into the dw divided by dx which is ϕ_i prime. Prime means derivative with respect to x is equal to ω^2 integral 0 to L ρ into sigma j is equal to 1 to n $\alpha_j \phi_j \phi_i dx$. This will imply that we will have sum j is equal to 1 to N , α_j into integral 0 to L $EA \phi_j$ prime ϕ_i prime dx . This is equal to ω^2 sum 0 to L $\rho \phi_j \phi_i dx$. This I will put again in the brackets. When w is equal to ϕ_i that gives me the i th equation as we have done earlier. For the i th ρ this corresponds to the ij th entry corresponding to α_j here also corresponds to the ij th entry corresponding to α_j . From basic dynamics we will know that this part which comes due to the EA is called the stiffness part, the part coming due to the ρ . I will have the ρ setting here, the part coming due to the ρ is called the mass part so this is essentially component K_{ij} of the so called stiffness matrix this is the component M_{ij} where M , K is called the stiffness matrix and M is called the mass matrix.

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$$[K]_{N \times N} \{\alpha\} = \omega^2 [M]_{N \times N} \{\alpha\}$$

Generalized Eigenvalue problem

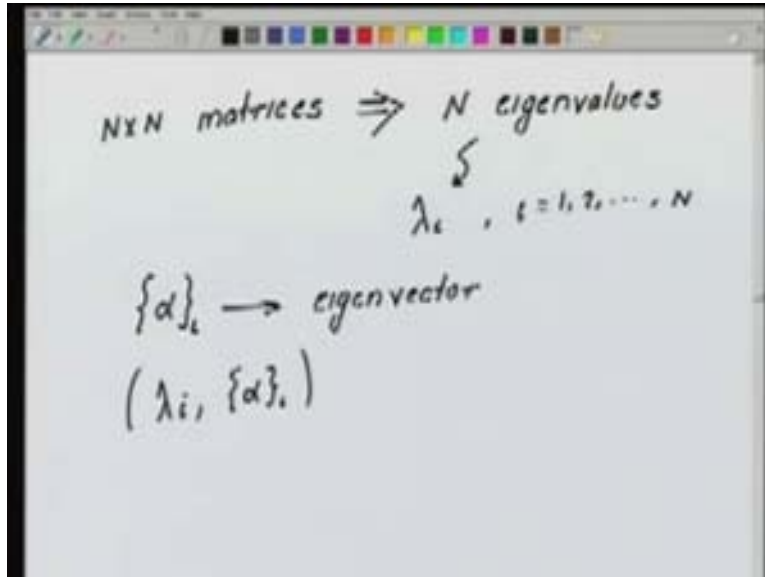
$$[K] = [K]^T \quad [M] = [M]^T$$

$\lambda \rightarrow \text{real, positive}$
 $\Rightarrow \omega^2 > 0, \omega > 0$

We have here the problem $K \alpha$ is equal to $\omega^2 M \alpha$ where we have given what are the entries of the matrix K entries of the matrix M . This problem that we have written is called a generalized Eigen value problem because in general the standard Eigen value problem is in the form $K \alpha$ is equal to $\lambda \alpha$. This I can write as quantity λ , ω^2 is replaced with λ look whether things are consistent or not. If I look at how we have done things? K is equal to K transpose that is K is symmetric. Similarly, M will be equal to M transpose that is M is also symmetric, the stiffness and the mass matrix in this case are symmetric. The mass matrix is diagonal not necessarily, but all we can say is mass matrix is symmetric. If these two things are symmetric then we say that the Eigen value λ is going to be real for this problem which means that our ω^2 is going to be a real number if it is a complex then we are in trouble because it makes no sense so this tells success looking at it that yes it is going to give us a real Eigen value. This matrix K will turn out to be K and M or going to turn out to be symmetric positive definite. They will turn out after imposing the boundary conditions to be symmetric positive definite so in this case your λ will not only be real but it will also be positive. Everything is consistent implies $\omega^2 > 0$, implies ω is greater is a real number which is greater than 0. It is a real positive number so this is the crux of what we have done and we see from the actual solution that exact solution that we had done ω was a number. It was a positive number so what are seen that here in the parallel

problem that we have created omega or the omega or the lambda turns out to be positive numbers. Here the problem is an N by N system. M and N or N by N matrices where n is a number of unknowns in the problem. In the unknowns in the problem means the number of basis function which have been used to represent the solution u as the function of x.

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We will have N by N problem which will imply we have N Eigen values which are given by λ_{i_1} , i going from 1 to N and corresponding to λ_{i_1} we will get a vector α_{i_1} . This in the matrix problem sense is the corresponding Eigen vector so corresponding to the i th Eigen value we have the corresponding ith Eigen vector α_{i_1} or we can talk of the Eigen pair λ_{i_1} and α_{i_1} from this Eigen value problem.

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$$[K]_{N \times N} \{ \alpha \} = \omega^2 [M]_{N \times N} \{ \alpha \}$$

Generalized Eigenvalue problem

$$[K] = [K]^T \quad [M] = [M]^T$$

$\lambda \rightarrow$ real, positive
 $\Rightarrow \omega^2 > 0, \omega > 0$

Solve for the Eigen values of this, I get λ_i , put the λ_i back into this expression and we will get the corresponding α_i .

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$N \times N$ matrices $\Rightarrow N$ eigenvalues
 $\lambda_i, i = 1, 2, \dots, N$

$\{ \alpha \}_i \rightarrow$ eigenvector
 $(\lambda_i, \{ \alpha \}_i)$

$U_i(x) = \sum_{i=1}^N \alpha_i \phi_i(x)$

Given this α_i , a corresponding mode the i th mode $U_i(x)$ will be equal to $\sum_{i=1}^N \alpha_i \phi_i(x)$. So given this Eigen vector α_i , we can construct the Eigen function or the mode shape which is N by $U_i(x)$. So this essentially is the crux of what we are doing with the

finite element method here, some more features of what we get is that here if I look at the exact solution, the exact Eigen values or the Eigen vectors corresponding to what we have done here.

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The image shows a whiteboard with handwritten mathematical notes. At the top, the equation $d_n = \omega_n \sqrt{\frac{P}{EA}}$ is written. Below it, an arrow points to the equation $\omega_n = \frac{(2n-1)\pi}{2L} \sqrt{\frac{EA}{P}}$, which is circled. To the left of this equation, a bracket indicates it is the n^{th} natural frequency. To the right, the inequality $\omega_1 < \omega_2 < \omega_3$ is written, with a note that these are distinct eigenvalues. Below the circled equation, the mode shape is given as $U_n(x) = \sin \alpha_n x$, which is boxed in red. A bracket below this indicates it is oscillatory. To the right of the boxed equation, a note says ω_n^{th} mode.

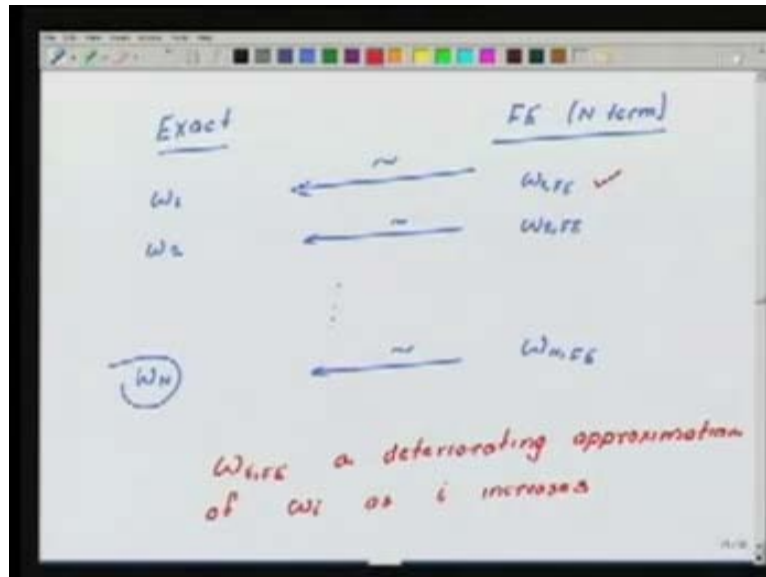
We can arrange this ω_n 's in the sequence such that ω_1 is less than ω_2 is less than ω_3 . We have in this problem, we have so called distinct Eigen values in fact these are the Eigen values of the continuous problem or the distinct natural frequencies.

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The image shows handwritten notes on a whiteboard. At the top, it says "N x N matrices \Rightarrow N eigenvalues". Below this, a set of eigenvalues is written as $\lambda_i, i = 1, 2, \dots, N$. A bracket groups these eigenvalues with the word "distinct". To the left, a set of eigenvectors is written as $\{\alpha\}_i \rightarrow$ eigenvector. Below this, a pair of eigenvalue and eigenvector is written as $(\lambda_i, \{\alpha\}_i)$. To the right, the frequency $\omega_i = \sqrt{\lambda_i}$ is written, with a bracket underneath it labeled "distinct". At the bottom, the solution $u_i(x) = \sum_{i=1}^N \alpha_i \phi_i(x)$ is written and circled in blue.

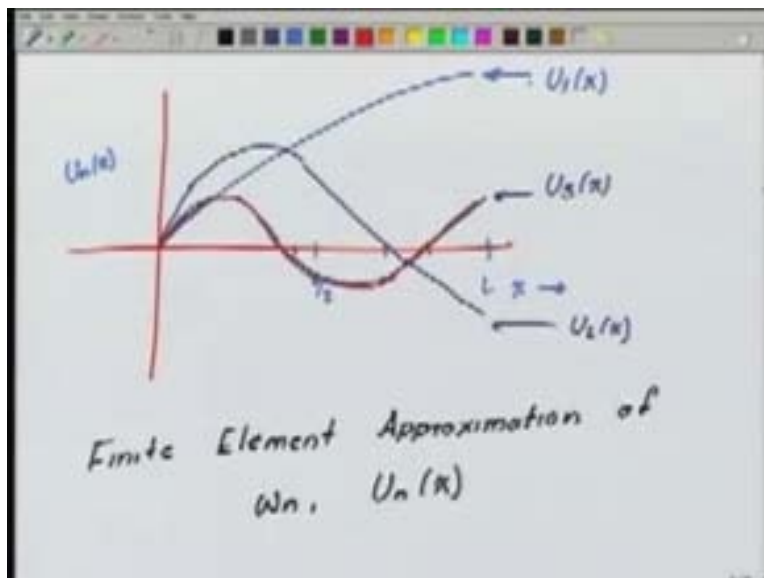
In the case of the problem that we have posed the matrix problem, we will get the λ_i s are also distinct that is you have a different value of λ_i . Given this λ_i your ω_i is equal to root of λ_i which is also distinct. These are distinct, in this case these are some things that we have to keep in mind that some features of the solution which may not always be true. Here we are talking of distinct Eigen values that is, an Eigen value is not repeated if the Eigen value is repeated which means corresponding to that Eigen value we have multiple Eigen vectors. That is the completely different situation which has to be handled in a different way, I mean we have to talk about it differently we do not handled it differently. Here we are only concentrating on problems where the Eigen values are distinct so I have this representation of $u_i(x)$ and now pretty much we have the solution.

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Here is the exact and here is the FE where from N term. The exact ω_1 is this, ω_1 FE is this. The first Eigen value corresponding to the finite element solution this is an approximation of the exact one. Here is the exact one ω_2 , $\omega_{2,FE}$ is the approximation of the second Eigen frequency and so on we continue till ω_n so the nth ω that we get from the finite element solution or n th Eigen values the square root of that is an approximation of the nth natural frequency because they are distinct so this is an approximation of the nth natural frequency. We will get here, this value will be very good depending on the mesh and as we keep on going further down that is go to the higher frequencies. The ω , as I go higher and higher $\omega_{i,FE}$ will be a more and more inferior approximation of the actual ω_i . This will be a deteriorating approximation of ω_i . As i increases, this is very important that as the i increases the ω_i that we get from the finite element solution becomes worse and worse approximation of the actual ω_i . So in most of the applications we are only interested in the first few natural frequencies, so in order to get those first few natural frequencies properly we have to use a mesh which is sufficiently fine and we see that here if I look at again I go back to what we had drawn, the Eigen functions have particular shapes.

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The whole business of how well is the approximation doing depends on what is the corresponding Eigen function and how well can the finite element solution approximate the Eigen function?

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Approximation of ω_i depends on how well $U_i(x)$ is approximated.

$\omega_{i,FE} \geq \omega_i$

There is a generalized result here which says that the approximation of ω_i , the quality of it depends on how well $u_i(x)$ is approximated by our finite element representation. Further we will see another feature of it. I will stop this lecture with this particular feature that we will see that ω_i finite element is greater than equal to the actual ω_i . This is the very important result, the frequency obtained from the finite element solution in this case is greater than equal to the frequency that we get from the exact solution as compared to the exact frequencies.

With this we are going to stop this lecture. In the next lecture we will continue a little bit further with these Eigen value problems and look at some more features of it.