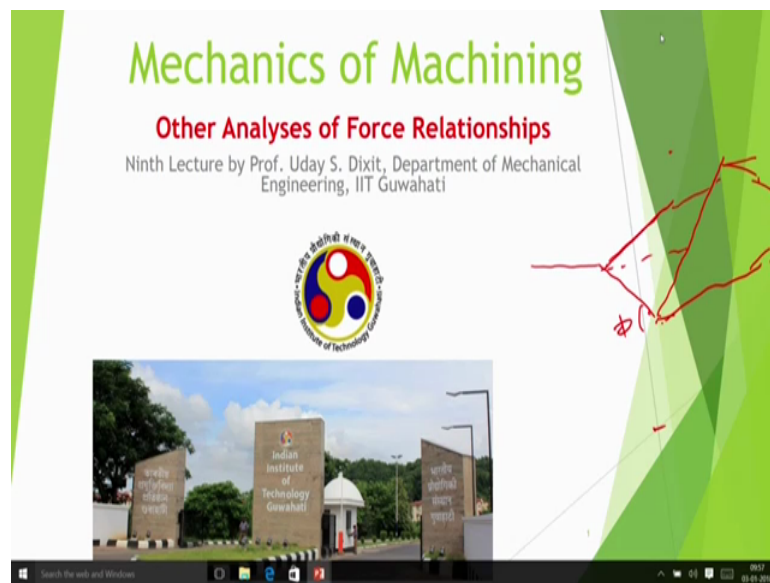


**Mechanics of Machining**  
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**Department of Mechanical Engineering**  
**Indian Institute of Technology, Guwahati**

**Lecture - 09**  
**Other Analysis for Force Relationships**

Dear students, welcome to the course on Mechanics of Machining, this is 9th lecture. Today I will be talking about various other models of finding out the cutting forces in orthogonal machining. Till now we have discussed about Merchant's force analysis, fused single shear plane model, Ernst and Merchant together they published many papers and Merchant also has published papers. So, that is very popularly known as single shear plane model. Even I discussed about slip line analysis of Lee and Shaffer. Lee and Shaffer slip line solution also assumes type of single shear plane.

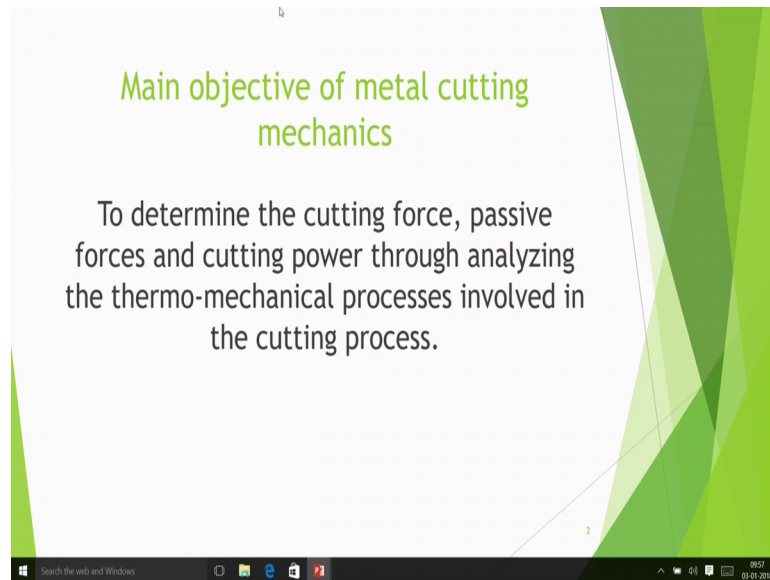
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But although it is a slip line, but here we established this type of relation there was  $\phi$  and then there was slip line. This is considered as a slip line and then there was a free surface, this was the tool like that and it is. And after that he deduced the relation between the shear angle, rake angle and friction angle which is different from merchant's relation and therefore, the power etcetera will also be different.

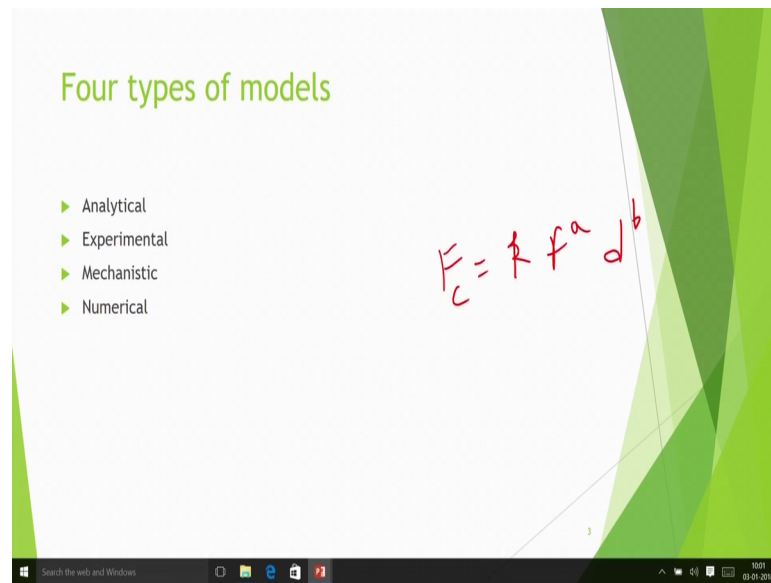
But nevertheless it is also assuming a existence of a shear plane like this. Then after that there are other models we will discuss those things today in the lecture and also we may give some exposure to the finite element method of finding out the cutting forces.

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So, if we go to this one that what is the main objective of metal cutting mechanics? The main objective is to determine the cutting force, passive forces means which do not participate like first force which does not produce any power and cutting power through analyzing the thermo mechanical processes involved in the cutting process. So, we have to do temperature analysis also and we have to do stress analysis also. Till now we are only talking about stress analysis, we are yet not talking about the temperature analysis.

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Then, we can say totally there can be four type of models, one is the analytical one like merchants analysis you can say it provides analytical model, it will usually give a closed form solution even, Lee staffers model also gives closed form solution. Then, there can be experimental or empirical model which is based on the number of experiments. We conduct number of experiments and we try to set one relation. In the last lecture also I wrote something like this, cutting force is equal to  $k$  times  $F$  to the power  $a$  and  $d$  to the power  $b$  here  $F$  is the feed and  $d$  is the depth of cut. So, such type of relations can be fitted by conducting number of experiments and then finding out some relation.

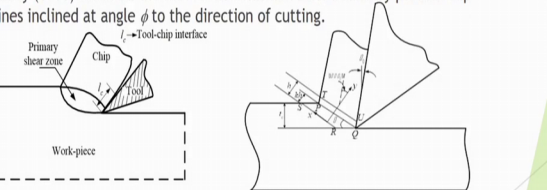
Then, third type of model is mechanistic model. In the mechanistic model we actually it is some type of we can say semi analytical model, semi experimental model. How do we do that? Suppose we obtain some type of relation that it should be like this, but the coefficients of that relation we can obtain by experimentation. Means, there is some physical basis, but it is not completely only by physics of that. So, that type of model is called mechanistic model ok.

Like I can say with my physical reasoning that the cutting force should be proportional to the chip cross sectional area. So, maybe it is proportional to  $F$  into  $d$ , but what is that proportionality constant that can come by some other methods. And then, if there is a even number of cutting edges are there are the cutting edges in the helical form, that type of thing also can be done by transformation process we can find out the aggregate effect;

Then the fourth type of model is the numerical model. In that we use some governing equations which govern the behavior of metal cutting. So, those governing equations we can take like continuity equation, equilibrium equations these equations have to be solved not analytically, but by means of numerical methods. For example, we can use finite difference method or we can use finite element method and other type of techniques also, so such types of methods are called Numerical Models.

## Slip line field solution of Oxley

- ▶ Palmer and Oxley (1959) studied the deformation pattern during low-speed machining.
- ▶ They recorded the paths of individual grains on the side surface of the workpiece using a cine film camera.
- ▶ They concluded that there is a thin shear zone in machining.
- ▶ Oxley (1961) assumed the narrow deformation zone bound by parallel slip lines inclined at angle  $\phi$  to the direction of cutting.



I have not shown small curvature near S. Slip line meets at angle of  $45^\circ$  at free surface.

So, they conducted lot of experiments and published paper in 1959 those experiments were of course conducted at low speed machining, and they recorded the paths of individual grains on the side surface of the work piece using a cinema film camera by that they recorded that how the grains are deforming and then they concluded that there is a thin shear zone in machining.



So, Oxley assumed the narrow deformation zone bound by parallel slip lines inclined at an angle  $\phi$  to the direction of cutting. So, he assumed that although this zone is like this, but to a first approximation I can consider it like zone bounded between two parallel slip lines and these are inclined at angle  $\phi$ . So, it is just like a shear plane angle. So, here this (Refer Time: 07:39) is one bound and may be  $T_u$  is another one. In fact, they provided some small curvature near S so that they it ensures that the slip line meets at angle of 45 degree at free surface at this one.

So, they there is a small curvature type of thing. Why it should meet at angle of 45 degree at the free surface? Because free surface will be divide of any stress and it will be one of the principle stress plane, one of the principle plane. So, the shear plane must make 45 degree angle from this. So, since the slip line is the gives the directions of maximum shear stress. So therefore, it must meet at the free surface at 45 degree angle.

So, that is why they provided some small tilting type of thing, but that is very small and in this figure I have not depicted I have depicted two straight lines. So, may be that this side total zone is  $h$  and this may be something like  $k$  times  $h$  and this may be  $1$  minus  $k$  times  $h$  like you that. So, this type of assumption they took.

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**Relevant equations: Generalized Hencky equations**

$$p + 2k\psi + \int \frac{\partial k}{\partial s_\beta} ds_\alpha - 2 \int \psi \frac{\partial k}{\partial s_\alpha} ds_\alpha = \text{Constant along } \alpha\text{-line}$$

$$p - 2k\psi + \int \frac{\partial k}{\partial s_\alpha} ds_\beta + 2 \int \psi \frac{\partial k}{\partial s_\beta} ds_\beta = \text{Constant along } \beta\text{-line}$$

Neglecting the work hardening term along the slip lines

$$p + 2k\psi + \int \frac{\partial k}{\partial s_\beta} ds_\alpha = \text{Constant along } \alpha\text{-line}$$

$$p - 2k\psi + \int \frac{\partial k}{\partial s_\alpha} ds_\beta = \text{Constant along } \beta\text{-line}$$

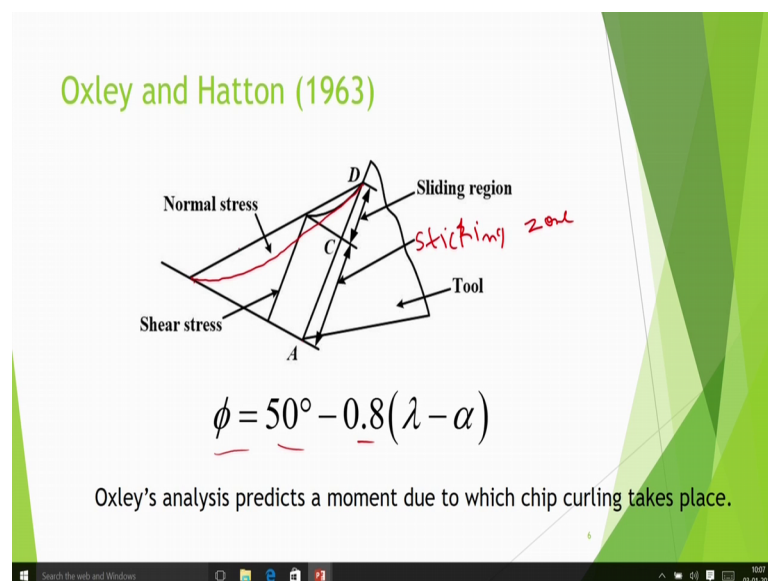
And then after that they I am not going to describe the complete details about that, but they considered the strain hardening also. Usually this slip line method is used for plane strain problems and material is rigid plastic, there is no hardening. But, it can be in

calculated, it makes the analysis cumbersome. So, they in calculated the hardening also and relations are like that  $p + 2k \sin \alpha$  and this portion. If  $k$  is constant throughout then this portion will be 0 and you will get usual Hencky's equation.

So, but these equation are given here. There one equation is constant along alpha line another is constant along beta line. Now if you neglect the work hardening term along the slip line; that means, across this slip line there is work hardening there is no doubt about that. Here, there is  $a$ ; that means, if you go from  $S_r$  to  $T_u$  naturally material will get strain hardened, but assumption is that along  $S_r$ .

There is no hardening if we assume that then you will get this type of relation  $p + 2k \sin \alpha$  is equal to  $\frac{d\sigma}{dS_\alpha}$  because  $\frac{dk}{dS_\alpha}$  here will be 0. So, you will get this type of relation that will be constant along alpha line  $p + 2k \sin \alpha + \int \frac{dk}{dS_\alpha} 2S_\alpha dS_\alpha = \text{constant}$  and then you will get similarly one relation along beta line.

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So, these relations they obtained and other procedure they followed in the similar way and then Oxley and Hatton in 1963, they considered this also you know this shear stress is not uniform on the surface. In fact, there is a sticking zone this is sticking zone from this is the tool cutting edge point A and from A to some distance there is a sticking zone or a sticking region.

So that is, in this portion shear stress remains constant and that is  $k$  and after that there is a sliding region in which you can say Coulombs law holds good and then as I have shown in the last class that the normal stress is parabolic type of thing here you get the maximum normal stress and gradually it decreases, but they assume that let us consider that it is not parabolic rather it is just straight line variation is there.

So, if you parabolic means it would have come like this, but some error will be including. So, they said let the normal stress be like that and then based on all these things the analysis was done and you get another type of relation that is  $\phi$  is equal to 50 degree minus 0.8 lambda minus alpha. And Oxley analysis also predicts a movement due to which chip curling takes place; means he considered in that model that the forces will generally not pass through this one this point rather the force may pass here and this will create a movement.

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**Unified or generalized mechanics approach of Armarego and others**

- ▶ Models for different machining operations could be unified into a modular computer application structure with generic cutting analyses and database.
- ▶ Cutting force components due to shearing and friction.
- ▶ Edge force components due to rubbing and ploughing.
- ▶ Cutting force components are proportional to uncut chip area.
- ▶ Edge force components are proportional to uncut chip width.

$$F_c = K_{cc} t_1 w + K_{ec} w = K_{cc} A + K_{ec} w$$

$$F_T = K_{cT} t_1 w + K_{eT} w = K_{cT} A + K_{eT} w$$

chip load

So, that type of thing also has been done those details are available in the textbook, but we are not going to discuss that much or I am just providing you some exposure that such type of analysis is also available in the literature.

Now, I am another one approved that which is. In fact, is still very active approach people are doing it that is, unified or generalized mechanics approach of particularly Armarego and co-workers they propagated that scheme. In this models for different machining operations could be unified into a modular computer application structure

with generic cutting analyses and database. Suppose you keep lot of database suppose I keep the database for let us say the turning process, if I have kept the database for turning process how when depth of cut was this and feed was this.

Then this was the force then same type of information can be used for modeling of the milling. If we understand the basic concept only thing that there may be may be two cutters simultaneously cutting. So, you take that effect into account there may be helical type of cutter tooth. So, you take that helix angle into account; that means, you have to integrate small segments helical tooth can be considered as series of the tooth, which are having different types of angles they are arranged little bit staggered that is what.

So, but basic database can be common so that is why we say that generic cutting analysis and database. So, in this we consider cutting force component due to shearing and friction. Then we consider edge force components due to rubbing and ploughing in the basic concept and then cutting force components are proportional to uncut chip area we say cutting force components they are proportional to uncut chip area.

So, and edge force components are proportional to uncut chip width because edge cutting means this is tool and suppose I may make a tool this tool is moving here. So, here this is the cutting edge which is going perpendicular to the screen that edge is there know. So, this edge will do cutting and it is having some width. So, that have other rubbing force may be proportional to chip width ploughing force also.

So, this  $F_c$  is equal to  $K_{cc} T_1 w$  that is the first part, because chip load due to shearing and friction will be proportional to  $T_1$  and  $W$ . So,  $K_{cc} T_1$  into  $W$ .  $W$  is the width of the uncut chip width and  $T_1$  is the uncut chip thickness plus  $K_{cc}$  into  $W$  we can write it  $K_{cc}$  into  $A$ ,  $A$  is the uncut chip cross sectional area and this can be also called chip load people call it as a chip load; that means, this is the total chip load. So, and this one is  $K_{ec} W$  edge cutting force. Then first force component can also be written in this similar way  $F_T$  is equal to  $K_{cT} T_1 W$  plus  $K_{eT} W$ ,  $K_{cT} A$  plus  $K_{eT} W$  ok. This can be written for any general process any machining process you can write.

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Earlier you found from Merchant's analysis

$$F_c = w t_1 \tau_s \cos(\lambda - \alpha) \frac{1}{\sin \phi \cos(\phi + \lambda - \alpha)} \quad F_T = w t_1 \tau_s \sin(\lambda - \alpha) \frac{1}{\sin \phi \cos(\phi + \lambda - \alpha)}$$

This suggests

$$K_{CC} = \tau_s \cos(\lambda - \alpha) \frac{1}{\sin \phi \cos(\phi + \lambda - \alpha)}$$

$$K_{CT} = \tau_s \sin(\lambda - \alpha) \frac{1}{\sin \phi \cos(\phi + \lambda - \alpha)}$$

Now, here suppose you want to do that simple orthogonal machining, in that you already have that relation from merchant's analysis which was given in this form  $F_c$  is equal to  $2 W T t_1 \tau_s$  shear strength and then this was  $\cos(\lambda - \alpha)$  are that type of things and  $F_T$  also we derive that expression. If we consider  $W T t_1$  as the chip loads then it is clear that this is another portion should be its coefficient. So, that must be  $K_{CC}$ . So, like that I have found that  $K_{CC}$  is also there  $K_{CC}$  is a function of now  $\alpha$  and it is function of  $\lambda$  and all that type of thing and  $K_{CT}$  is also function of  $\alpha$ ,  $\lambda$  and  $\phi$  like that.

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### Mechanistic models propagated by Shiv G. Kapoor and others

- Forces are proportional to uncut chip area.
- Uncut chip area is also called chip load.
- Transformational relations can be used for multi-point cutting.

So, this one so like that you can have the data and you can calculate then around the same time Shiv G. Kapoor and that group they developed the mechanistic models. Mechanistic Model and unified generalized mechanics approach more always they are similar, they developed around the same time and lot of work has been done here also they assume that forces are proportional to uncut chip area and uncut chip area is also called chip load. Then transformational relations can be used for multipoint cutting.

So, multi point cutting we can use the transformation type of relations. Suppose my cutting edge is one cutting edge may be here, but suppose this is some type of wind mill. And if you know that at this point how much the force is causing this cutting edge then on the next layer also you can consider that this is also a cutting edge, but angles will be slightly different.

Like that you can find out aggregated effect it is like this that you can have aggregate effect of all the type of forces and then you get mechanistic model approach. So, this lot of papers are there on this topic, each one has developed some mechanistic model for one type of process using the transformational relations these details you can get in other papers. Here, we will not be discussing much about individual papers in detail.

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The slide is titled "Energy approach (e.g. Astakhov)" in green text. It contains a bulleted list of five items, each preceded by a green arrow. The list is as follows:

- ▶ Total power comprises the following:
- ▶ The power spent on the plastic deformation of the layer being removed
- ▶ The power spent on the tool-workpiece interface
- ▶ The power spent on the tool-work interface
- ▶ The power spent in the formation of new surfaces

Handwritten red annotations are present on the slide. A red checkmark is placed next to the third bullet point. A red line is drawn under the third and fourth bullet points. To the right of the list, there is a hand-drawn red diagram of a cutting tool. The diagram shows a triangular tool with a chip being removed. Labels include "Chip" at the top, "Tool" at the bottom left, and "Work" at the bottom right. There are also some arrows and other markings in red.

So, we move to another approach; that it energy approach of Astakhov. Astakhov has proposed Astakhov has written lot of papers and books on metal cutting. In fact, earlier

also I discussed one paper of Astakhov in which he criticized single shear plane model and merchants analysis.

So, in one of his book he has described the energy approach and he says the forces etcetera can be found by energy approach. Here we consider that total power comprises the following power spent on the plastic deformation of the layer being removed, then power spent on the tool work interface, then power is spent on the tool, this is plastic deformation of the layer removed then power is spent on the chip work. No this is tool chip interface this is tool is and this tool chip interface and another power is the power spent on the tool and work interface because tool is like this. So, on it has got two surfaces with main surfaces that is rake on this the chip is moving.

So, tool and chip are interacting and similarly that here tool and work are interacting. So, that power also as we considered, then power is spent in the formation of new surfaces, that power also he said should be considered like you saw that in Atkins analysis also that type of power was considered.

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Interesting observation about inadequacy of single shear plane model

- ▶ Machining of medium carbon steel AISI 1045 (UTS 655 MPa, Yield strength 375 MPa) resulted lower cutting forces
- ▶ Than
- ▶ Machining of stainless steel AISI 316L (UTS 517 MPa, Yield strength 218 MPa) resulted lower cutting forces *higher.*
- ▶ This may be due to larger strain encounter in latter material.

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So, that is approach of Astahkov and then, he also pointed out once the interesting observation about inadequacy of single shear plane model. For example, machining of medium carbon steel that is AISI 1045 which has about 0.4 percent carbon in that ultimate tensile strength is 655 mega Pascal and yield strength is 375 mega Pascal.



Another steel is stainless steel that is AISI 316L its ultimate tensile strength is 547 mega Pascal and yield strength is 218 mega Pascal. So, both these strengths are lower it should give that lower cutting force, but actually this give higher cutting force. In fact, this give higher cutting force we aspect that it should give lower, but it give higher and this actually which has more ultimate tensile strength it gives lower.

So, why it is so; that means, Astakhov argued that this may be due to larger strain encountered in that this material that later material. That means, in stainless steel part there may be this one there may be lot of redundant deformation there may be lot of unwanted deformation suppose, I wanted to cut some material and this chip is there this portion has been cut. Now we will require the energy in cutting, but during the process the chip thickens and you know it shortens also

So, this I am doing unnecessary chip thickening is not of any use, but it happens, but it takes energy. So, it is a type of you can say it is redundant energy if another person can do the simple cutting operation same way, but energy is not required; that means, lot of deformation of the chip is not there in that case is energy will be smaller. So, stress is one thing, but the strain is another thing. Stress into strain they give the energy product of stress and strain. So, this may be due to that.

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**How do we calculate power for plastic deformation?**

- ▶ Let us use Hollomon relation:  $\sigma_{eq} = K \epsilon_{eq}^n$
- ▶ Energy density in deforming up to strain  $\epsilon_f = \int_0^{\epsilon_f} K \epsilon_{eq}^n d\epsilon_{eq} = \frac{K \epsilon_f^{n+1}}{(n+1)}$
- ▶ Power = Total energy in unit time =  $\frac{K \epsilon_f^{n+1}}{(n+1)} t_1 w v$

$$\epsilon_f = \sqrt{\frac{2}{3} \left\{ \ln \left( \frac{t_2}{t_1} \right)^2 + \ln \left( \frac{t_1}{t_2} \right)^2 \right\}} = \frac{2}{\sqrt{3}} \ln \left( \frac{t_2}{t_1} \right) = \frac{2}{\sqrt{3}} \ln(\zeta)$$

▶ Hence, power for plastic deformation =

$$\frac{K (1.15 \ln \zeta)^{n+1}}{(n+1)} t_1 w v = \frac{K (1.15 \ln \zeta)^{n+1}}{(n+1)} f d v$$

Handwritten notes on the right side of the slide:

$$\ln \frac{t_2}{t_1} = \ln \frac{t_2}{t_1}$$

$$\ln \frac{t_1}{t_2} = -\ln \frac{t_2}{t_1}$$

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So, you have to consider that strain effect also. Now how do we calculate power for plastic deformation? Why I am will giving one illustration, let us consider Holloman relation of strain hardening.

So, here  $\sigma$  will be  $K \epsilon^n$ ,  $n$  is the exponent and  $K$  is the strength coefficient. Now energy density in deforming up to strain  $\epsilon_f$  if suppose I know that this is my fracture strain; that means, this will be the strain for causing the structure ok. So, then we can integrate 0 to  $\epsilon_f$   $K \epsilon^n d\epsilon$  stress into strain give me energy density; that means, energy per unit volume.

So, if I take product of that that stress is this whole quantity; that means,  $\epsilon_f$  multiplied by  $d\epsilon$  that will give small energy, but if I integrate between 0 to  $\epsilon_f$  I will get total measure of the energy. So, if I integrate I get this type of expression  $K \epsilon_f^{n+1} / (n+1)$ . So, power is equal to total energy in unit time. So, if I multiply this energy per unit volume by volume rate of metal involved that is  $T_1 W$  into  $v$   $v$  is the cutting speed then I get the total power.

So, we got this type of expression now how do we find out  $\epsilon_f$  if I can measure the chip thickness then I get the idea about that strain; that means, assuming that it is plane strain causes. So, you get the thickness direction strain and the length direction strain of course, the both of these strains will be equal and opposite because the total volume has to be constant.

So, one is that suppose chip has become  $T_1$  to  $T_2$ . So, you get that  $\ln T_1$  by sorry  $T_2$  by  $T_1$  and similarly in other case you get  $\ln T_1$  by  $T_2$  because (Refer Time: 25:13)  $L_2$  by  $L_1$ , but  $L_2$  by  $L_1$  is nothing, but  $T_1$  by  $T_2$  because you know that  $L_1 L_1$  is equal to  $L_2 T_2$ . So, these are the logarithmic strain measures and of course, in  $T_1 T_2$  is nothing but minus  $\ln T_2$  by  $T_1$ . So, I can find out now the equivalent strain equivalent is strain is what square root of 2 by 3 and then I square all the non zero components of these strains and add them.

So, there are only two non-zero components of the strain in this axis system 1 is the chip is thickening and then chip is also getting shortened. So, I took in  $T_2$  by  $T_1$  square plus in  $T_1$  by  $T_2$  square and then I took the square root of this. We get 2 by lambda root 3 In  $T_2$  by  $T_1$  and in  $T_2$  by  $T_1$  is nothing but the chip compression ratio it is more than one and we say 2 by lambda root 3 In  $\zeta$ .


Hence the power for plastic deformation becomes here two by lambda root three is nothing, but 1.15. So, we substitute in this expression and it becomes  $k \cdot 1.15 \ln \zeta$  divided by  $n + 1$  to the power  $n + 1$  and  $T \cdot l \cdot w \cdot v$  and this becomes this I can instead of  $w$  I can write  $f$  and instead of  $T \cdot l$  I can write  $d \cdot d$  depth of cut and I get this type of expression.

So, this way I can calculate the power for plastic deformation.

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Power due to friction at tool-chip interface

$$P_{fric} = m \frac{\sigma_y}{\sqrt{3}} l_c w_{chip} \left( \frac{v}{\zeta} \right)$$

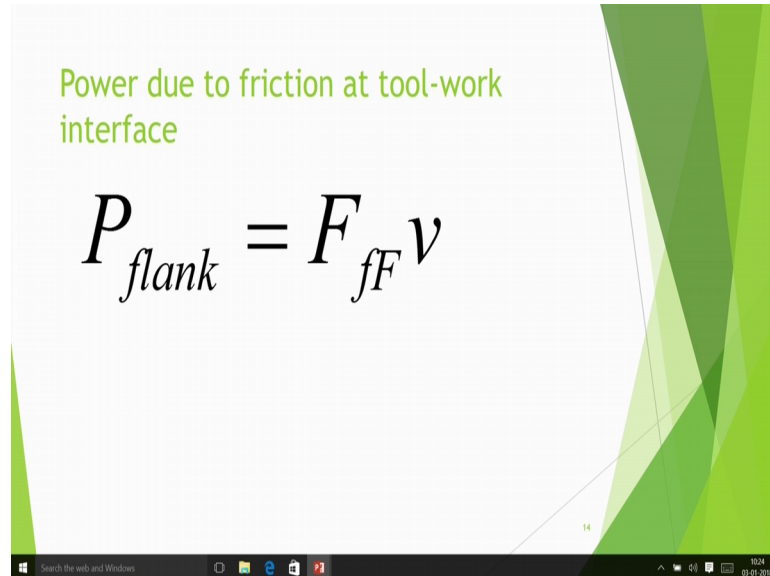
$$l_c = t_1 \zeta^{1.5} \quad (\text{empirical})$$


Now, we can find out power due to friction at tool chip tool chip interface tool and chip. Chip is sliding on the tool so here we find out  $P_{fric}$  is equal to  $m$ ,  $m$  is a friction factor it is a maximum value can be 1. So,  $m$  time's  $\sigma_y$  by  $\sqrt{3}$  that gives a constant shear stress and  $l_c$  is the chip length then  $w_{chip}$  is the width and then we say that here. So,  $l_c$  into  $w_{chip}$ , chip that will give you that force shear force and then  $v$  multiply by  $v$ . In fact  $v_c$ , because the chip is moving with the velocity  $v_c$ , but  $v_c$  is equal to nothing, but  $v$  by chip compression ratio  $\tau$ .

So, you get this type of expression  $m$  has to be of course, found out by experiments because  $m$  is the friction factor if there is some lubrication  $m$  will be small if there is no lubrication  $m$  will be big like that and then  $l_c$  is the chip contact length. That means, how much is the length of the contact because chip separates there is a tool and on this tool the chip is moving suppose my chip is going like this. So, contact length is up to this he gave empirical type of expression and this  $l_c$  is equal to  $T \cdot l$  multiplied by chip

compression ratio to the power 1.5. So, this expression can be used and you can get idea about the friction power.

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Power due to friction at tool-work interface

$$P_{flank} = F_f v$$

So, two powers we have obtained then; power due to friction at tool work interface can be found  $P$  at the flank and be friction force at the flank multiplied by  $V$ . So, this portion also you can find out you can consider that this portion is to be proportional to the cutting edge width and it may be fraction function of low radius those types of things can be taken into account and then other things also chip. That power is spent on the separation of new surfaces that can also be considered and all the energies can be combined you get the total energy then divide it by cutting speed  $V$  and then you get cutting force.

So, this way you can find out the cutting forces. So, all these analysis are there; there are many other methods also like one is the upper bound method in which we find out upper bound of the cutting force that how much will be the maximum cutting force like that and there are many slip line feed solutions.

So, all these methods are not able to match the numerical methods like finite element modeling although finite element modeling of machining process is also difficult. But, nowadays finite element modeling has been in calculated in many commercial packages like deform and in abacus etcetera. Also, you can do finite element modeling by doing some work from your own side by writing some small codes etcetera.

So, I will give some exposure about finite element modeling because nowadays you can get even readymade codes and in that you use them as exact box, but you must also know that what is going inside. So, very briefly I will tell about the finite element modeling I at least that what are the questions actually which are being solved by finite element model.

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**Finite Element Modeling of Orthogonal Machining Process**

Introduction

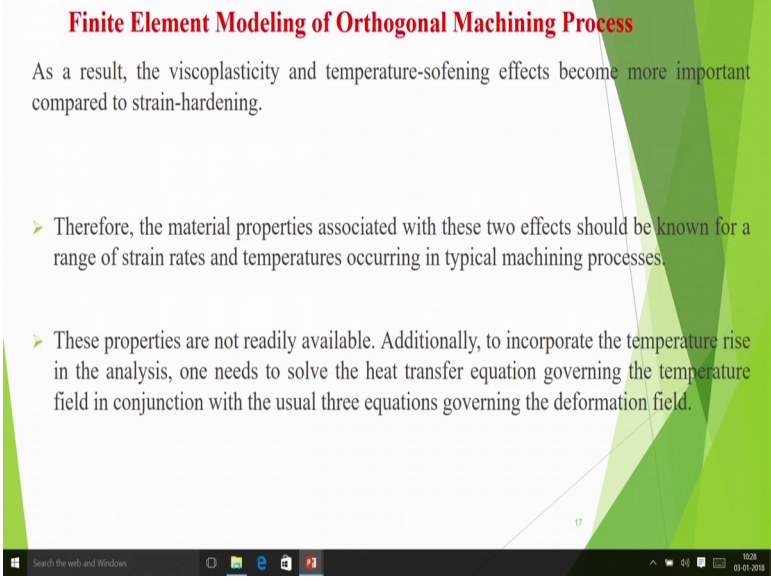
- Machining processes are difficult to model for various reasons.
- Unlike metal forming processes, where almost the whole work-piece gets plastically deformed, in machining processes, the plastic deformation is localized near the cutting edge.
- Therefore, we need to analyze only a small region of the work-piece around the cutting edge (called the cutting zone).
- As a result, the selection of the domain dimensions and the appropriate boundary conditions becomes a difficult task.
- Further, even at a moderate cutting speed, the strain rates are quite high, almost of the order of  $10^4$  per second. Further, the temperature rise is also quite large.

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So, first some introduction that machining processes are difficult to model for various reasons one is that unlike metal forming processes, where almost the whole work piece gets plastically deformed, in machining processes the plastic deformation is localized near the cutting edge. So, you have to consider only the localized one. So, that local thing has lot of effect if the local layer is very hard it will have significant effect.

In metal forming whole thing you are analyzing suppose local layer you are not able to model properly or you have taken the average property that will do, but average property type of thing in machining will not work. So, we need to analyze only a small region of the work piece around the cutting edge that is called the cutting zone. As a result the selection of the domain dimensions and the appropriate boundary conditions becomes a difficult task. And further, even at moderate cutting speed the strain rates are quite high although it is also uncertain that how high it is really, but many people believe that it is almost of the order of  $10^4$  per second and temperature rise is also quite large.

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**Finite Element Modeling of Orthogonal Machining Process**

As a result, the viscoplasticity and temperature-softening effects become more important compared to strain-hardening.

- Therefore, the material properties associated with these two effects should be known for a range of strain rates and temperatures occurring in typical machining processes.
- These properties are not readily available. Additionally, to incorporate the temperature rise in the analysis, one needs to solve the heat transfer equation governing the temperature field in conjunction with the usual three equations governing the deformation field.

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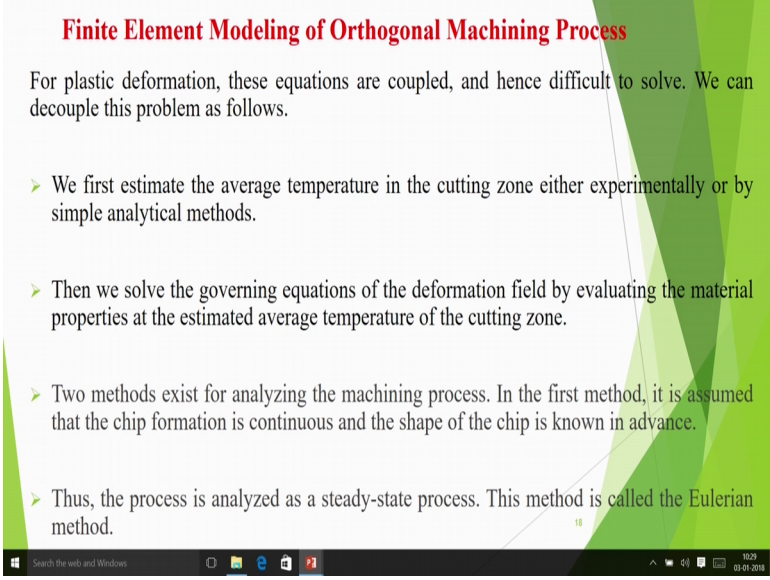
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As a result viscoplasticity and temperature softening effects becomes more important compared to strain hardening we consider that flow stress is also a function of strain rate. So, material property is associated with these two effects should be known for a range of strain rates and temperatures occurring in typical machining processes. These properties are not readily available additionally to incorporate the temperature rise in the analysis one is to solve the heat transfer equation governing the temperature filed in conjunction with the usual three equations governing the deformation fields.

So, you get the equilibrium equations those have to be solved in addition you have solve the heat transfer equations, but heat transfer equations you can solve by finite element or you can solve by other methods also analytical methods also.

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### Finite Element Modeling of Orthogonal Machining Process

For plastic deformation, these equations are coupled, and hence difficult to solve. We can decouple this problem as follows.

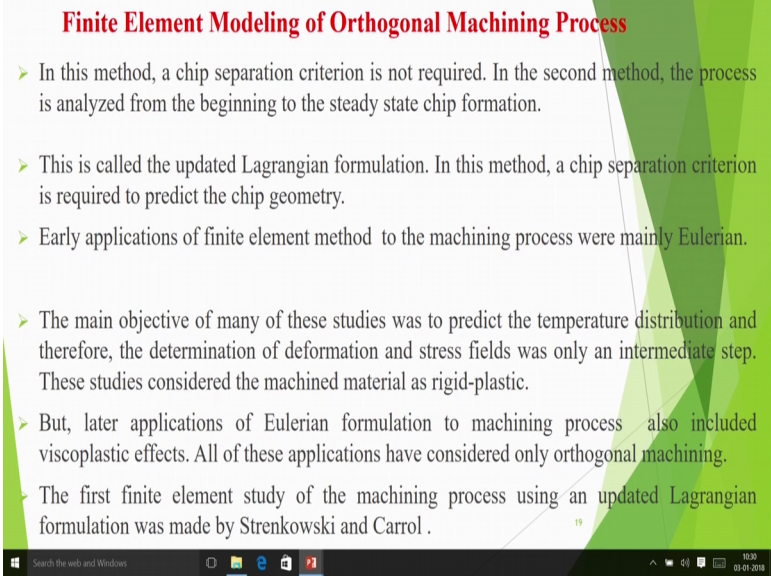
- We first estimate the average temperature in the cutting zone either experimentally or by simple analytical methods.
- Then we solve the governing equations of the deformation field by evaluating the material properties at the estimated average temperature of the cutting zone.
- Two methods exist for analyzing the machining process. In the first method, it is assumed that the chip formation is continuous and the shape of the chip is known in advance.
- Thus, the process is analyzed as a steady-state process. This method is called the Eulerian method.

So, for plastic deformation these equations are coupled and hence difficult to solve. So, as such they are coupled, because if temperature effects this one stresses and stresses effect the temperature, but we can often decouple them and we first estimate the average temperature in the cutting zone either experimentally or by some simple analytical methods. And then, we solve the governing equations of the deformation field by evaluating the material properties at the estimated average temperature of the cutting zone. That means, average way I have found and then to methods this process can be repeated after you do that analysis then again you can find out the temperature and again you can do this one. So, you can do in a sequential manner also.

Here two methods exist for analyzing the machining process in the first method it is assumed that the chip formation is continuous and its shape is known a prior. So, chip is known in advance. So, process can be analyzed as a study state process that material is moving through that control volume and this method is called the Eulerian method control volume method.



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### Finite Element Modeling of Orthogonal Machining Process

- In this method, a chip separation criterion is not required. In the second method, the process is analyzed from the beginning to the steady state chip formation.
- This is called the updated Lagrangian formulation. In this method, a chip separation criterion is required to predict the chip geometry.
- Early applications of finite element method to the machining process were mainly Eulerian.
- The main objective of many of these studies was to predict the temperature distribution and therefore, the determination of deformation and stress fields was only an intermediate step. These studies considered the machined material as rigid-plastic.
- But, later applications of Eulerian formulation to machining process also included viscoplastic effects. All of these applications have considered only orthogonal machining.
- The first finite element study of the machining process using an updated Lagrangian formulation was made by Strenkowski and Carroll.

Another so in this method the chip separation criteria is not required. And in the second method the process is analyzed from the beginning to the steady state chip formation in the second method process is analyzed from the beginning to the steady state chip formation. We follow the material though this is called the updated Lagrangian formulation.

In this method a chip separation criteria is required to predict the chip geometry that how frame the chip will separate that type of thing has to be in calculated in this we find out the position etcetera. And stresses of one time step by taking the differences as the previous time step that is why it is called updated Lagrangian formulation.

But earlier people mostly used an Eulerian approach, because Eulerian approach will take comparatively less amount of time or though you get more detailed information if you do Lagrangian approach. So, main objective of many of these studies was to predict the temperature distribution. And therefore, the determination of deformation and stress fields was only an intermediate steps and these studies considered the machining material as rigid plastic.

So, if you want to find out even the temperature distribution still you have to find out stresses or if you want to find out stresses naturally you must know the temperatures. But later application of Lagrangian formulation to machining processing you did Viscoplastic effects and all of these applications have considered only orthogonal machining the first

finite element study of the machining process using an updated Lagrangian formulation was made by Strenkowski and Carol.

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**Finite Element Modeling of Orthogonal Machining Process**

- This was for orthogonal machining. A critical value of the equivalent plastic strain was used to model the separation of a chip.
- Later on, several researchers used the updated Lagrangian formulation for analyzing two- and three-dimensional machining processes.
- Most of these studies have used an FEM package: ABAQUS , MARC or DEFORM . The criterion used for chip separation has been based on controlled crack propagation or some geometrical considerations.
- Remeshing technique has been used to simulate the chip formation.

This was for orthogonal machining and they used a critical value of the equivalent plastic strain to model the separation of a chip.

Suppose you keep on analyzing and finding out these strain and when that particular strain becomes more than critical value, then you can say that the chip separation is taking place. Later on several researchers used updated Lagrangian formulation for analyzing 2 and 3 dimensional machining processes most of these studies have used FEM package ABAQUS is one FEM package MARC is another one DEFORM is another one DEFORM is very usual trend it is used for metal forming and metal cutting studies and in other packages you have to work something, but they may have more flexibility.

Criteria used for chip separation has been based on controlled crack propagation or some geometric consideration. And they have used remeshing technique to simulate the chip formation suppose when deformation occurs then again you make the mesh again. Because, if suppose I make a mesh here in this region like this; and then suppose if has deformed then naturally you have to make a different type of mesh that type of thing is also done in those packages.

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### Finite Element Modeling of Orthogonal Machining Process

**Domain**

- In the present formulation, it is assumed that the problem is **decoupled**. We further assume that the elastic deformation is small.
- As stated above, the visco-plasticity and temperature effects are more dominant compared to the strain-hardening effects.
- To keep things simple, we assume that the material exhibits no hardening but only visco-plasticity. Thus, we assume the material to be **rigid-viscoplastic**.
- Further, the temperature softening is accounted for by evaluating the material properties at an average temperature occurring in the cutting zone.
- Additionally, we analyse the process when it has reached a steady state. Then the transient term in the equation of motion vanishes. We further assume that the body forces are negligible.

And let us discuss finite element modeling of orthogonal machining process we assume that the problem is deformed; that means, we can estimate some temperature and after that we can do simply the stress analysis and we assume that the elastic deformation is very small. So, neglect any elasticity and viscoplasticity and temperature effects are more dominant compared to these strain hardening effect.

So, we do not consider strain hardening, but rather we consider viscoplasticity and temperature effect and to keep things simple. We assume that the material exhibits no hardening, but only viscoplasticity thus we assume the material to be rigid-viscoplastic. Further, the temperature softening is accounted for the evaluating the material properties at an average temperature occurring in the cutting zone.

Additionally we analyze the process when it has reached a steady state value a steady state condition then the transient term in the equation of motion vanishes. So, transient is not there and we further assume that the body forces are negligible what are the body forces? Like such as the gravity force which is distributed in the body those types of forces are neglected interaction of the forces is only through the surfaces by external agency.

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### Finite Element Modeling of Orthogonal Machining Process

- We choose a small region of the work-piece around the cutting edge (called the cutting zone) as the control volume, *i.e.*, the domain of the problem.
- To make the problem two-dimensional, we assume that the width of cut is large compared to the dimensions of the cutting zone.
- The domain, along with the coordinate system, is shown in Figure 7.1. It is a region in the cross-sectional plane of the work-piece perpendicular to the cutting edge.
- Point E is the projection of the cutting edge. The  $z$ -axis is along the cutting edge or the width of cut. The boundaries AB and EF are actually circular.
- But, since the cutting zone dimensions are small compared to the work-piece radius, they are taken to be straight.

So, here we chose a small region of the work piece around the cutting edge or the cutting zone as the control volume that is the domain of the problem. Then to get the problem two dimensional we assume that the width of the cutting large compared to the dimensions of the cutting zone. So, that we can state is more or less a plane strain type of situation. The domain along with the coordinate system zone in the next figure and it is region in the cross sectional plane of the work piece perpendicular to the cutting edge now here this one is.

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### Finite Element Modeling of Orthogonal Machining Process

The angle  $\theta$  can be computed by measuring the cutting force  $F_c$  and thrust force  $F_t$  by a dynamometer and using the following relation:

$$\cos \theta = \frac{F_c \cos \phi - F_t \sin \phi}{\sqrt{F_c^2 + F_t^2}} \quad (7.3)$$

- The boundaries AH, HG, FG and CD are placed sufficiently away from the cutting edge projection E so as to simplify the boundary conditions on these boundaries by taking the advantage of the uniform velocity fields existing there.
- Further, the boundaries AH, FG and CD are chosen parallel to the shear plane so as to facilitate the mesh generation.

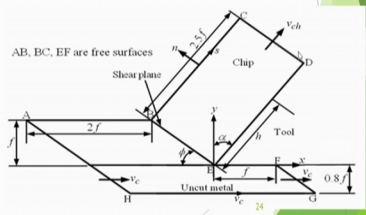


Figure 7.1

This type of thing we make this type of domain this is uncut material and then this is the domain.

So, here point E is the projection of the cutting edge this point is the cutting edge then z axis is along the cutting edge of the width of the cut. That means, in this screen only you are seeing the x axis and y z is perpendicular to the screen and then boundaries A B and E F are actually circular, but we have assumed A B and E F. This one A B and may be this is E F.

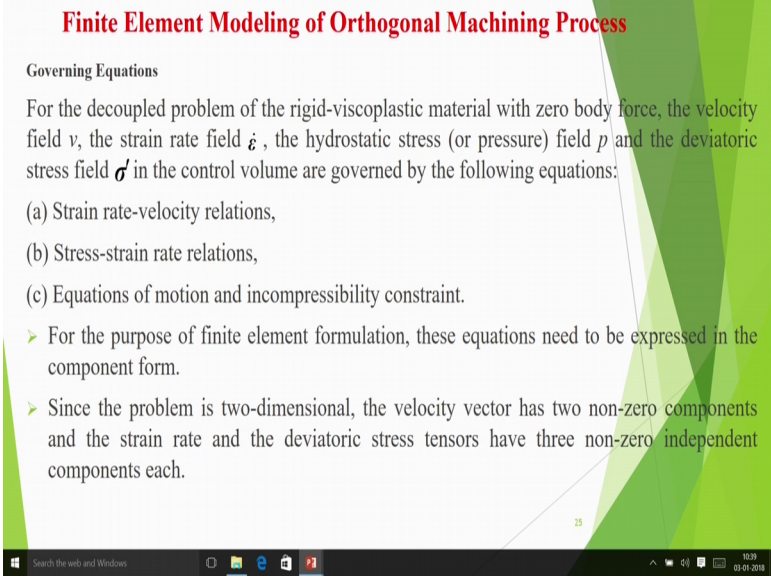
These are free surfaces they may be circular because some bulging will take place here, but we assume that they are straight line and now angle alpha is equal to the rake angle of the cutting tool by now you will very well know it is alpha. That means, surface of the tool is inclined at an angle alpha with the vertical and the distance H is called the tool chip tool chip contact length and we use the following expression for H F sine theta by this one. So, this is the type of expression has been used.

Earlier I told another expression by Astahkov. So, this is this type of things you have to consider. So, suppose they have taken only this type of relation and in orthogonal machining this shear angle can be estimated by measuring the cutting ratio and that is given here. Now here angle theta is actually theta is the angle between shear force and the resultant force and F is the feed.

So, theta is the angle between the Shear force on the shear plane and the resultant force on that plane that can be easily obtained by merchants circuit and it is given in this manner  $FC \cos \phi \text{ minus } FT \sin \phi$  divided by this one and FC is the cutting force and this is the thrust force and by a dynamometer. So, we can find out these force boundaries HG FG and C D are placed sufficiently away from the cutting edge projection E.

So, they have been put sufficiently away. So, in this process then we assume that boundaries FG and CD are chosen parallel to the shear plane. So, this way so this and shear plane BE and CD are parallel. So, mesh generation will be easy and here you can assume that at CD you will well this velocity is V CH there is uniformity here.

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### Finite Element Modeling of Orthogonal Machining Process

**Governing Equations**

For the decoupled problem of the rigid-viscoplastic material with zero body force, the velocity field  $v$ , the strain rate field  $\dot{\epsilon}$ , the hydrostatic stress (or pressure) field  $p$  and the deviatoric stress field  $\sigma'$  in the control volume are governed by the following equations:

- (a) Strain rate-velocity relations,
- (b) Stress-strain rate relations,
- (c) Equations of motion and incompressibility constraint.

- For the purpose of finite element formulation, these equations need to be expressed in the component form.
- Since the problem is two-dimensional, the velocity vector has two non-zero components and the strain rate and the deviatoric stress tensors have three non-zero independent components each.

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Now, for the decoupled problem of the rigid viscoplastic material with zero body force the velocity field  $v$ , the strain rate field  $\dot{\epsilon}$ , the hydrostatic stress field  $p$  and the deviatoric stress field  $\sigma'$  in the control volume are governed by the following equation. We have strain rate velocity relation, then we have stress strain relation, then we have equation of motion and incompressibility constraint.

Now, these equations need to be expressed in the component form velocity vector has two non-zero components and the strain rate and the deviatoric (Refer Time: 43:37) stress cancel out three non-zero independent components each.



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### Finite Element Modeling of Orthogonal Machining Process

In terms of the components with respect to the coordinate system of Figure 7.1, The governing equations are as follows:

(i) **Strain rate - velocity relations:**

Let  $(v_x, v_y)$  be the non-zero components of the velocity vector  $v$  and  $(\dot{\epsilon}_{xx}, \dot{\epsilon}_{yy}, \dot{\epsilon}_{xy})$  be the non-zero independent components of the strain rate tensor  $\dot{\epsilon}$ . Then, the strain rate-velocity relations (Equation 3.66) become

$$\begin{aligned}\dot{\epsilon}_{xx} &= \frac{\partial v_x}{\partial x}, \quad \dot{\epsilon}_{yy} = \frac{\partial v_y}{\partial y}, \\ \dot{\epsilon}_{xy} &= \frac{1}{2} \left( \frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right).\end{aligned}\tag{7.4}$$

So, what are these equations suppose we write strain rate velocity relation we can write  $\epsilon_{xx}$  is equal to  $\partial v_x / \partial x$ . That means, velocity in the x direction divided by  $\partial x$   $\epsilon_{yy}$  is strain rate in y direction is this and shear strain rate is given by this that is one type of relation that has to be used, because in a FEM you will be then predicting what are the velocities at different nodes.

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### Finite Element Modeling of Orthogonal Machining Process

(ii) **Rigid-viscoelastic deviatoric stress-strain rate relations:**  
As stated above, we neglect the hardening and consider only the visco-plasticity.

For, non-hardening visco-plastic materials, the relation between the deviatoric stress and the strain rate tensors gets modified.

$$\begin{aligned}\sigma'_{xx} &= 2\eta_2 \dot{\epsilon}_{xx}, \\ \sigma'_{yy} &= 2\eta_2 \dot{\epsilon}_{yy}, \\ \sigma'_{xy} &= 2\eta_2 \dot{\epsilon}_{xy},\end{aligned}\tag{7.5}$$

where  $(\sigma'_{xx}, \sigma'_{yy}, \sigma'_{xy})$  are the non-zero independent components of the deviatoric stress tensor  $\sigma'$ . Here, the proportionality factor  $\eta_2$ , for non-linear visco-plastic behavior, is given by

$$\eta_2 = a(\dot{\epsilon}_{eq})^m + \frac{\sigma_Y}{3\dot{\epsilon}_{eq}}\tag{7.6}$$

And then there is a rigid viscoelastic (Refer Time: 44:17) deviatoric stress strain relations. So, this is basically viscoplastic; that means, we can say viscoplastic. So, here



if it is non hardening then we get this type of relation that deviatoric stress what is deviatoric stress component? That you subtract the hydrostatic part from the stress tensor then you get deviatoric stress tensor and then you have  $2\epsilon^2$  into  $\epsilon \dot{\epsilon}$  and here this is a proportionality factor  $\epsilon^2$  and this is given by  $\epsilon^2$  is equal to  $a$  times something strain equivalent strain rate to the power  $m$  plus this portion that  $\sigma_y$  by  $3\epsilon \dot{\epsilon}$  that anyway would have come, but this portion has come, because of the viscoplastic effect.

So, that such type of relations of plasticity has being used here.

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**Finite Element Modeling of Orthogonal Machining Process**

where  $\sigma_y$  is the yield stress of the material,  $\dot{\epsilon}$  is the equivalent strain rate and  $a$  and  $m$  are the material constants representing the material visco-plasticity.

➤ To account for the temperature softening, the material constants  $\sigma_y$ ,  $a$  and  $m$  are evaluated at the estimated average temperature of the cutting zone.

**(i) Equations of motion:**

As stated earlier, we neglect the body forces. Further, we analyse the process when it has reached a steady-state. Therefore, the transient term ( $\rho \partial v_i / \partial t$ ) vanishes. Then, the equations of motion, in the component form, take the form

$$\begin{aligned} \rho \left( v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} \right) &= -\frac{\partial p}{\partial x} + \left( \frac{\partial \sigma'_{xx}}{\partial x} + \frac{\partial \sigma'_{xy}}{\partial y} \right), \\ \rho \left( v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} \right) &= -\frac{\partial p}{\partial y} + \left( \frac{\partial \sigma'_{xy}}{\partial x} + \frac{\partial \sigma'_{yy}}{\partial y} \right). \end{aligned} \quad (7.7)$$

And temperature softening is considered material constants are evaluated at the average temperature of the cutting zone. Then, we write the equation of motion equation of motion are written like this x momentum balance in x direction you used this and momentum balance in y direction use this, but  $p$  is the hydrostatic pressure acting on that particular point and this is deviatoric stress. And if we say that inertia effects are neglected then this portion will go here.

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### Finite Element Modeling of Orthogonal Machining Process

In the rolling problem, we neglected the first term of the equation of motion as the acceleration was small.

But, in the machining process, the acceleration is not negligible. Hence, we retain this term.

(i) **Incompressibility constraint:**

The incompressibility constraint in the component form, becomes

$$\dot{\epsilon}_{xx} + \dot{\epsilon}_{yy} = 0. \quad (7.8)$$

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And in that case now in this vv consider incompressibility constraint also. So, incompressibility constraint in the component form becomes epsilon dot x x plus epsilon dot y y equal to 0 that is incompressibility constraint.

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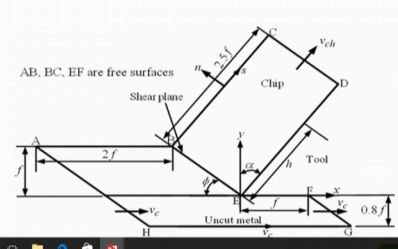
### Finite Element Modeling of Orthogonal Machining Process

#### Boundary Conditions

- Note that, since the problem is two-dimensional, only two boundary conditions are needed on each boundary instead of three.
- Whether a boundary condition is essential or natural is also indicated against each boundary condition.

#### Boundaries AH, HG and FG:

- As stated earlier, the boundaries AH, HG and FG are chosen sufficiently away from the cutting edge projection E.
- Therefore, we can assume that the velocity vector has only x-component at these boundaries. Further, the velocity actually varies linearly from point H to point A and from point G to point F.



And then we consider boundary conditions note that since the problem is 2 dimensional only 2 boundary conditions are needed on each boundary instead of 3.

So, whether a boundary condition is essential or natural is also indicated against each boundary condition; that means, in a FEM we have two type of boundary conditions one

is in terms of the primary variables like our velocity and hydrostatic pressure or the primary variable. So, we say suppose velocity is specified that is called essential boundary condition and if the derivative terms are used in the boundary like suppose strain or stress is the derivative of the velocity then we say that it is natural boundary condition.

So, at these boundaries HG and FG here we assume that the velocity vector has only x component at these boundaries HG here only x component and then you have got HG then you have got FG. So, here FG also only this one and HG also if we assume that it is only cutting like this. So, we assume that it is only in x direction

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**Finite Element Modeling of Orthogonal Machining Process**

- As stated earlier, the boundaries AH, HG and FG are chosen sufficiently away from the cutting edge projection E.
- Therefore, we can assume that the velocity vector has only x-component at these boundaries. Further the velocity actually varies linearly from point H to point A and from point G to point F.
- But, since the distances AH and FG are very small compared to the work-piece radius, we assume the velocity to be uniform over these boundaries.
- Let  $v_c$  be the specified cutting velocity. Then, the boundary conditions at the boundaries AH, HG and FG become

$$v_x = v_c, \quad v_y = 0, \quad (\text{essential}) \quad (7.9)$$

That is  $V_c$  that and then as we can assume that the velocity vector has x components. Now I am actually the velocity values linearly from H to point A; that means, from here to here and from point G to point F, but since the distances are very small we assume the velocity to be uniform ok.

Let now,  $V_c$  be the specified cutting velocity then the boundary conditions become  $V_x$  is equal to  $V_c$  and  $V_y$  is equal to zero. So, that type of boundary condition

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### Finite Element Modeling of Orthogonal Machining Process

**Boundary CD:**

The boundary CD is also chosen sufficiently away from the cutting edge projection E. Therefore, we assume that, at this boundary also, the velocity vector is uniform over the whole boundary.

Let  $v_{ch}$  be the chip velocity. The chip velocity can be calculated from the cutting velocity using the conservation of mass equation over the uncut depth and the chip thickness

$$v_{ch} = v_c r, \quad (7.10)$$

where  $r$  is the cutting ratio given by Equation 7.2. Note that the chip velocity makes an angle  $\alpha$  with y-axis. Then, the boundary conditions at the boundary CD become

$$v_x = v_{ch} \sin \alpha, \quad v_y = v_{ch} \cos \alpha, \quad (\text{essential}). \quad (7.11)$$

We put at boundary CD; that means, this boundary CD this is that three boundary here at this boundary from the cutting edge we assume that this boundary velocity vector is uniform over the whole boundary. So,  $V_{ch}$  is the chip velocity and this is equal to  $V_c$  times  $r$  and in this case boundary conditions at the boundary C D becomes  $V_x$  is equal to  $V_{ch} \sin \alpha$   $V_y$  is equal to  $V_{ch} \cos \alpha$ . That means, we have expressed that in terms of  $V_x$  and  $V_y$  because ultimately at each node we will be having unknown valuable like  $V_x$   $V_y$  and pressure.

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### Finite Element Modeling of Orthogonal Machining Process

**Stress free boundaries AB, BC and EF:**

- The boundaries AB, BC and EF are stress-free surfaces. On the stress-free surfaces, the stress vector is zero at every point.
- Therefore, the boundary conditions at the boundaries AB, BC and EF can be expressed as

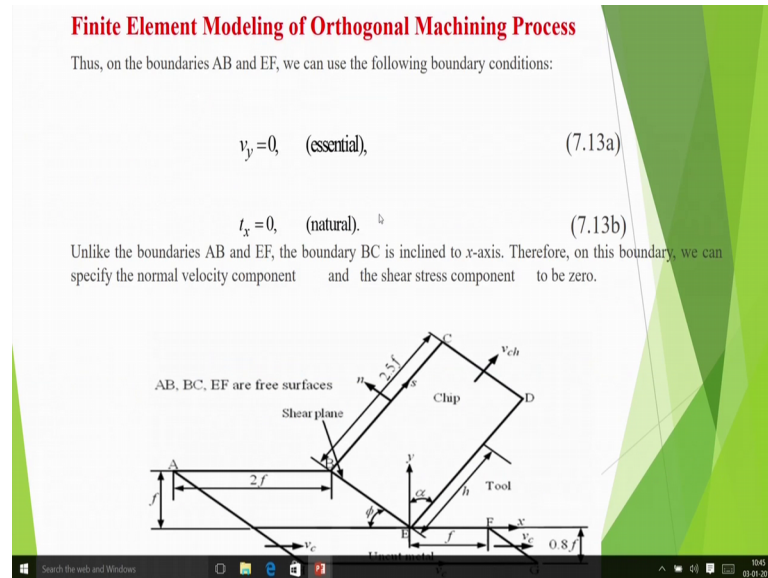
$$t_x = 0, \quad t_y = 0, \quad (\text{natural}), \quad (7.12)$$

where  $t_x$  and  $t_y$  are the Cartesian components of the stress vector  $t_n$ . Sometimes an alternate set of boundary conditions is used on these boundaries.

- This set is as follows. Since the direction of the velocity vector at the boundaries AB and EF is always along x-axis, the boundary condition may be modified to specify  $v_y$  to be zero instead of  $t_y$  being zero.

So, we express in that way that portion has to be done and stress free boundaries AB BC and EF on these because these are stress free surfaces AB. So, we simply say that  $t_x$  is equal to 0 direction in x direction is equal to 0 on  $t_y$  equal to 0.

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And now this one is now we can of course, take any other type of boundary conditions also, but we are generally taking 3 surface 1 and now what about that boundary BC BC yes BC is inclined here, since the inclination of boundary BC with yx is alpha. So, we can write  $V_n$ . That means normal component across  $V_c = 0$ . So,  $n$  equal to 0, but by trigonometry you have to transform  $V_n$  in terms of  $x$  and  $V_y$  and you get  $V_n$  is equal to minus  $V_x \cos \alpha$  plus  $V_y \sin \alpha$  equal to 0.

That is essential boundary condition and then you get a natural boundary condition is equal to 0 because it is free surface  $t_s$  means tangential stress is not there here, but that is expressed in this form and they are expressed like this then at tool chip interface velocity along the tool chip interface can be approximated by the following relation.

This is a sort of implicit relation this is given here and then velocity keeps on increasing first and then after that it becomes constant.



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### Finite Element Modeling of Orthogonal Machining Process

where  $\xi$  is the distance measured along the boundary from point E. Thus, the value of  $v_x$  varies from  $v_{ch}/3$  at point E to  $v_{ch}$  when  $\xi$  is equal to  $h$ .

- Note that, the boundary ED makes an angle  $\alpha$  with  $y$ -axis. Then, the boundary conditions at boundary ED become:

$$v_x = v_\xi \sin \alpha, \quad v_y = v_\xi \cos \alpha, \quad (\text{essential}). \quad (7.15)$$

- Let  $v_x$ ,  $v_y$  and  $p$  be the functions of  $(x,y)$  which satisfy the essential boundary conditions exactly.
- Then, as stated in earlier chapter, these functions constitute an approximate solution to the problem consisting of the governing equations (Equations 7.4, 7.5, 7.7 and 7.8) and the boundary conditions (Equations 7.9, 7.11, 7.12 or 7.13, 7.15) if the following integral of the weighted residue is made zero:

This one now note that this makes angle alpha with y axis; so we can again transform in  $V_x$  and  $V_y$  like that we do all these type of things and then.

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### Finite Element Modeling of Orthogonal Machining Process

$$\int_{\Omega} \left( \epsilon_{xx} + \epsilon_{yy} \right) w_p + \left[ -p \left( v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} \right) \frac{\partial}{\partial x} + \left( \frac{\partial \sigma'_{xx}}{\partial x} + \frac{\partial \sigma'_{xy}}{\partial y} \right) w_x \right. \\ \left. + \left[ -p \left( v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} \right) \frac{\partial}{\partial y} + \left( \frac{\partial \sigma'_{xy}}{\partial x} + \frac{\partial \sigma'_{yy}}{\partial y} \right) w_y \right] \right] d\Omega = 0. \quad (7.16)$$

Here  $w_p$ ,  $w_x$ , and  $w_y$  are functions of  $(x,y)$ , called the weight functions, which are arbitrary except that they satisfy the homogeneous version of the essential boundary conditions. The functions  $v_x$ ,  $v_y$  and  $p$  are called the approximation to the solution.

$v_y$

- In order to weaken the continuity requirements on the approximation, we simplify the second and third terms of the integrand of Equation 7.16. Then, we get

What do we do that we take the easy questions suppose you have this equation  $\epsilon_{xx}$  we have this one we have this equation. Now we know that  $\epsilon_{xx}$  is what in terms of the velocity derivatives and we integrate this equation we multiply each equation by weight some weight function  $w_p$ ,  $w_x$ ,  $w_y$ , and we say integrated thing 0.

That is the procedure which is followed in (Refer Time: 51:39) finite element approximation procedure ok.

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**Finite Element Modeling of Orthogonal Machining Process**

$$\int_{\Omega} \left\{ -(\dot{\epsilon}_{xx} + \dot{\epsilon}_{yy}) w_p + \rho \left[ v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} \right] w_x + \left[ v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} \right] w_y \right. \\ \left. - \rho \left[ \dot{\epsilon}_{xx}(w) + \dot{\epsilon}_{yy}(w) \right] + 2\eta \left[ \dot{\epsilon}_{xx} \dot{\epsilon}_{xx}(w) + \dot{\epsilon}_{yy} \dot{\epsilon}_{yy}(w) + 2\dot{\epsilon}_{xy} \dot{\epsilon}_{xy}(w) \right] \right\} d\Omega \\ - \int_{\Gamma_x} w_x t_x d\Gamma - \int_{\Gamma_y} w_y t_y d\Gamma = 0 \quad (7.17)$$

Here, the quantities  $\dot{\epsilon}_{xx}(w)$ ,  $\dot{\epsilon}_{yy}(w)$  and  $\dot{\epsilon}_{xy}(w)$  are given by relations similar to Equation 7.4:

$$\dot{\epsilon}_{xx}(w) = \frac{\partial w_x}{\partial x}, \quad \dot{\epsilon}_{yy}(w) = \frac{\partial w_y}{\partial y}, \\ \dot{\epsilon}_{xy}(w) = \frac{1}{2} \left( \frac{\partial w_x}{\partial y} + \frac{\partial w_y}{\partial x} \right) \quad (7.18)$$

So, I am just telling that this one is done and then after that we obtain the derivative of these equations. That means we integrate these equations by part, because we observed that sometimes there is second derivative of velocity some place is there is a first derivative of the velocity. So, if we integrate by part then we can reduce sometimes the order of derivative that procedure is done, and as a result you get this type of equation I am not going to discuss that in detail.



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### Finite Element Modeling of Orthogonal Machining Process

and  $\Gamma_x$  and  $\Gamma_y$  are respectively the boundaries on which the components of the stress vectors  $t_x$  and  $t_y$  are specified.

- Further, the deviatoric stress-strain rate relations have been used to eliminate  $\sigma'_{ij}$  from the area integral and the decomposition of the stress tensor and the Cauchy's relation have been used to express the boundary integrals in terms of the components  $n_x$  and  $n_y$  of the stress vector:

$$\begin{aligned} (-p + \sigma'_{xx})n_x + \sigma'_{xy}n_y &= \sigma'_{xx}n_x + \sigma'_{xy}n_y = t_x, \\ \sigma'_{xy}n_x + (-p + \sigma'_{yy})n_y &= \sigma'_{xy}n_x + \sigma'_{yy}n_y = t_y. \end{aligned} \quad (7.19)$$

- Here,  $n_x$  and  $n_y$  are the components of a unit vector normal to the parts of the boundaries on which  $t_x$  and  $t_y$  are specified.

And, then we have this type of relations that of course, you get these equations at the boundary  $\sigma_{xx} n_x$  that is actually a relation of Cauchy by which you get  $t_x$  direction  $t_x$  is equal to  $\sigma_{xx} n_x$ ,  $n_x$  is the  $x$  component of the normal direction and in  $y$  this one. So, these equations are there if you have to just revise the basics of this solid mechanics otherwise time being I am just telling you the basic concepts.

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### Finite Element Modeling of Orthogonal Machining Process

- For the convenience of finite element formulation, it is desirable to express the integral form (Equation 7.17) in an array notation. For this purpose, we define the following arrays:

$$\begin{aligned} \{\dot{\epsilon}\} &= \begin{Bmatrix} \dot{\epsilon}_{xx} \\ \dot{\epsilon}_{yy} \\ \sqrt{2}\dot{\epsilon}_{xy} \end{Bmatrix}, \quad \{\dot{\epsilon}(w)\} = \begin{Bmatrix} \dot{\epsilon}_{xx}(w) \\ \dot{\epsilon}_{yy}(w) \\ \sqrt{2}\dot{\epsilon}_{xy}(w) \end{Bmatrix}, \quad \{m\} = \begin{Bmatrix} 1 \\ 1 \\ 0 \end{Bmatrix}, \\ \{\nabla_x v\} &= \begin{Bmatrix} \frac{\partial v}{\partial x} \\ \frac{\partial v}{\partial y} \end{Bmatrix}, \quad \{\nabla_y v\} = \begin{Bmatrix} \frac{\partial v}{\partial y} \\ \frac{\partial v}{\partial x} \end{Bmatrix}, \quad \{w\} = \begin{Bmatrix} w_x \\ w_y \end{Bmatrix}. \end{aligned} \quad (7.20)$$

- Then, the integral form (Equation 7.17) becomes

$$\begin{aligned} \int_{\Omega} \left[ -w_p \{m\}^T \{\dot{\epsilon}\} + \rho \left( v_x \{w\}^T \{\nabla_x v\} + v_y \{w\}^T \{\nabla_y v\} \right) - \{\dot{\epsilon}^T(w)\} \{m\} p \right] \\ + 2\eta_2 \{\dot{\epsilon}^T(w)\} \{\dot{\epsilon}\} d\Omega - \int_{\Gamma_x} w_x t_x d\Gamma - \int_{\Gamma_y} w_y t_y d\Gamma = 0. \end{aligned} \quad (7.21)$$

So, you can ignore if you do not really know, but here after that we get this type of  $v$  represent everything in the matrix and vector forms. So, that it becomes easy to

manipulate. So this is a strain rate, and this is this one, and after that we put all that type of data and we take.

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**Finite Element Modeling of Orthogonal Machining Process**

The approximation for the velocity components over a typical element  $e$  is given by


Where

$$\{v\} = [N^v] \{v\}^e \quad (7.22)$$

$$\{v\} = \begin{Bmatrix} v_x \\ v_y \end{Bmatrix}, \quad v = N_1 v_1 + N_2 v_2 \quad (7.23)$$

$$[N^v] = \begin{bmatrix} N_1 & 0 & N_2 & 0 & \dots & N_8 & 0 \\ 0 & N_1 & 0 & N_2 & \dots & 0 & N_8 \end{bmatrix} \quad (7.24)$$

is the velocity shape function matrix containing the two-dimensional second order serendipity shape functions  $N_i$  ( $i = 1, 8$ ) and

$$\{v\}^e = \begin{Bmatrix} (v_x)_1^e & (v_y)_1^e & \dots & (v_x)_8^e & (v_y)_8^e \end{Bmatrix} \quad (7.25)$$


There is a some concepts of that suppose velocity at each node as two components and it is inter projected type of thing that we assume that velocity inside the element will be inter correlation of the model velocities and these inter correlation functions are basically called shape functions. Suppose I have the velocity  $V_1$  here suppose some velocity  $V_1$  and some velocity  $V_2$ . So, we can say  $V$  in the domain  $V$  is equal to  $n_1 v_1$  plus  $n_2 v_2$  where  $n_1$  and  $n_2$  are called shape function or inter correlation function and  $n_1$  may be the function of  $x$  and  $n_2$  may be function of  $y$ . So, like that so it is nothing but the interpolation function I am mathematics you can study later on in detail I am just telling you basic concepts.

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**Finite Element Modeling of Orthogonal Machining Process**

is the elemental velocity vector containing the nodal values of the velocity components at all eight nodes of the element  $e$ . The approximation for the pressure is given by

$$p = \{N^p\}^T \{p\}^e, \quad (7.26)$$

where the pressure shape function vector

$$\{N^p\}^T = \{N_1^p \quad N_2^p \quad N_3^p \quad N_4^p\} \quad (7.27)$$

contains the two-dimensional bi-linear Lagrangian shape functions  $N_i^p$  ( $i = 1, 4$ ) and the elemental pressure vector

$$\{p\}^e = \{p_1^e \quad p_2^e \quad p_3^e \quad p_4^e\} \quad (7.28)$$

contains the nodal values of the pressure at the four corner nodes of the element  $e$ .

So, that it when you study mathematics then it will be easy for you to follow.

So, this shape functions everything is expressed in terms of the shape functions in interpolation function, because infinite element we find out only the nodal velocities, but of course we can say that we know everything because we can interpolate them. So, interpolation is applied and then we put

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**Finite Element Modeling of Orthogonal Machining Process**

contain respectively the nodal values of  $w_x$  and  $w_y$  at all three nodes of the line element  $b$ . These nodal values are known but arbitrary.

We also approximate the variations of  $t_x$  and  $t_y$  along a typical line element  $b$  of the boundaries  $\Gamma_x$  and  $\Gamma_y$  by the following expressions:

$$t_x = \{N\}^{bT} \{t_x\}^b, \quad (7.38)$$

$$t_y = \{N\}^{bT} \{t_y\}^b, \quad (7.39)$$

where the vectors

$$\{t_x\}^{bT} = \{t_{x1}^b \quad t_{x2}^b \quad t_{x3}^b\}, \quad (7.40)$$

$$\{t_y\}^{bT} = \{t_{y1}^b \quad t_{y2}^b \quad t_{y3}^b\} \quad (7.41)$$

All these thing here we obtain these type of relations B times v e.

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### Finite Element Modeling of Orthogonal Machining Process

where

$$[B] = \begin{bmatrix} \frac{\partial N_1}{\partial x} & 0 & \frac{\partial N_2}{\partial x} & 0 & -\frac{\partial N_8}{\partial x} & 0 \\ 0 & \frac{\partial N_1}{\partial y} & 0 & \frac{\partial N_2}{\partial y} & -\frac{\partial N_8}{\partial y} & 0 \\ \frac{1}{\sqrt{2}} \frac{\partial N_1}{\partial y} & \frac{1}{\sqrt{2}} \frac{\partial N_1}{\partial x} & \frac{1}{\sqrt{2}} \frac{\partial N_2}{\partial y} & \frac{1}{\sqrt{2}} \frac{\partial N_2}{\partial x} & -\frac{1}{\sqrt{2}} \frac{\partial N_8}{\partial y} & -\frac{1}{\sqrt{2}} \frac{\partial N_8}{\partial x} \end{bmatrix} \quad (7.46)$$

$$[B_x] = \begin{bmatrix} \frac{\partial N_1}{\partial x} & 0 & \frac{\partial N_2}{\partial x} & 0 & -\frac{\partial N_8}{\partial x} & 0 \\ 0 & \frac{\partial N_1}{\partial x} & 0 & \frac{\partial N_2}{\partial x} & -\frac{\partial N_8}{\partial x} & 0 \end{bmatrix} \quad (7.47)$$

And then this is B matrix derivative matrix means inter correlation functions have to be derive the means differentiated because you have to get a strain rate from velocity. And then you substitute all these type of things and then you get in the matrix form you get nice type of expression.

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### Finite Element Modeling of Orthogonal Machining Process

Then, Equation 7.49 becomes

$$\sum_{e=1}^{n_e} \left( \{w\}^e T [k]^e \{\delta\}^e \right) = \sum_{b=1}^{n_{bx}} \left( \{w_x\}^b T \{f_x\}^b \right) + \sum_{b=1}^{n_{by}} \left( \{w_y\}^b T \{f_y\}^b \right) \quad (7.56)$$

- The area integrals in Equations 7.50 and 7.52 are evaluated numerically by 3×3 Gauss quadrature, described in Section 5.2.2 of Chapter 5.
- For this purpose, the integrals are transformed to the natural coordinates  $(\xi, \eta)$  using the following transformation:

$$\int_{A^e} (\dots) dx dy = \int_{-1}^1 \int_{-1}^1 (\dots) |J| d\xi d\eta \quad (7.57)$$

(Handwritten diagram of a square element in natural coordinates with nodes at (-1,-1), (-1,1), (1,1), and (1,-1). Arrows indicate the mapping from the physical element to the natural element.)

Finally you get that this type of thing that  $k^e \delta^e$  and this is this one, but if this requires lot of integration in side because you have to integrate. So, integration is usually done by numerical method like Gauss Quadrature etcetera. And here you transform that

transformation is done how that you transform from physical coordinate system to natural coordinate system in natural coordinate system any domain is getting transformed to minus 1 to 1 and minus 1 to 1. That means, square domain here; that means, transformation you can do, but in applying the Gauss Quadrature. We generally try to make this type of domain  $\xi$  and  $\eta$  and this is minus 1 minus 1 and this is plus 1 plus 1 and you get Jacobian determinant of Jacobian that is basically transformational relation.

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**Finite Element Modeling of Orthogonal Machining Process**

where  $|J|$  is the determinant of the Jacobian matrix:

$$[J] = \begin{bmatrix} \frac{\partial x}{\partial \xi} & \frac{\partial x}{\partial \eta} \\ \frac{\partial y}{\partial \xi} & \frac{\partial y}{\partial \eta} \end{bmatrix} \quad (7.58)$$

The Jacobian is evaluated from the geometric approximation of the area element. In order to model the curved boundaries of the elements properly, we use the second order serendipity approximation for the element geometry, the same as that used for the velocity components. Thus,

where

$$x = \{N\}^T \{x\}^e, y = \{N\}^T \{y\}^e, \quad (7.59)$$

$$\{N\}^T = \{N_1 \quad N_2 \quad - \quad - \quad N_8\} \quad (7.60)$$

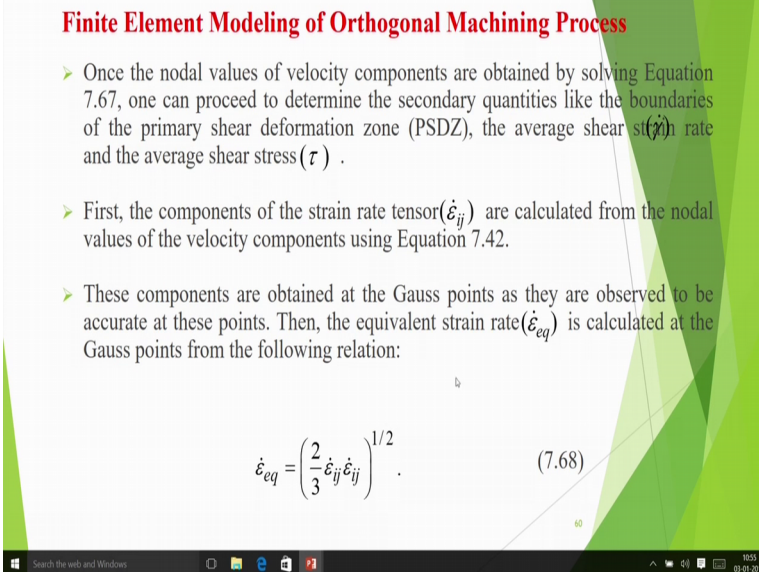
That means, Jacobian can be defined be like this. These type of thing we can put even the coordinates also  $x$  position at any point is also inter correlation of values nodal coordinates and we put all that type of thing and then after that you will get. Finally, this type of equation  $K \delta = f$  and this will be algebraic system of equations. So, we started with the differential equations. But finally, we ended up with algebraic equations, but of course, this  $K$  may be non-linear; that means,  $k$  may be function of  $\delta$ . In fact, in metal cutting glasses it is non-linear.

So, we are getting simultaneous not the linear system of equations, but simultaneous non-linear system of equation  $K$  is called the coefficient matrix sometimes since packages they just call it is stiffness matrix and  $\delta$  is basically unknown primary values vector which can be called displacement vector and  $f$  is the right hand side force vector or it is just called right hand side vector.

So, we do that and then after that we apply the boundary conditions. So, boundary conditions properly can be applied and there are some terms like  $\eta^2$  etcetera. In that relation we told you about  $\eta^2$  in that relation it was there just you saw that here this was relation of  $\eta^2$  say something when we discussed about this one about relations that yes it in that we discussed what are the type of equations. And we discussed that there is a constitute relation like this.

So, you are getting some  $\eta^2$  now suppose  $\dot{\epsilon}_{eq}$  is very small. Then  $\eta^2$  will become very high is not if  $\dot{\epsilon}_{eq} \rightarrow 0$  it becomes infinite we do not want that type of situation. So, we limit that in numerical methods you have to do such type of things that all though theoretically if  $\dot{\epsilon}_{eq}$  is very small  $\eta^2$  must be very large, but you say no I will not allow it to go beyond 1400. And then, you solve these non-linear equations in iterative way till it converges within 1 percent between two successive iterations.

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**Finite Element Modeling of Orthogonal Machining Process**

- Once the nodal values of velocity components are obtained by solving Equation 7.67, one can proceed to determine the secondary quantities like the boundaries of the primary shear deformation zone (PSDZ), the average shear strain rate and the average shear stress ( $\tau$ ).
- First, the components of the strain rate tensor ( $\dot{\epsilon}_{ij}$ ) are calculated from the nodal values of the velocity components using Equation 7.42.
- These components are obtained at the Gauss points as they are observed to be accurate at these points. Then, the equivalent strain rate ( $\dot{\epsilon}_{eq}$ ) is calculated at the Gauss points from the following relation:

$$\dot{\epsilon}_{eq} = \left( \frac{2}{3} \dot{\epsilon}_{ij} \dot{\epsilon}_{ij} \right)^{1/2} \quad (7.68)$$

And then after you have find out everything then you find out the derivative components, and then you find out  $\dot{\epsilon}_{eq}$  is equal to  $\frac{2}{3} \dot{\epsilon}_{ij} \dot{\epsilon}_{ij}$  here I used index notation that  $\dot{\epsilon}_{ij} \dot{\epsilon}_{ij}$  means  $i$  varies from 1 to 3 in 3 dimensional problem  $j$  changing from 1 2 3, but each components since  $i$  and  $j$  are occurring twice.

So, all the terms are summed; that means,  $\dot{\epsilon}_{11}^2 + \dot{\epsilon}_{22}^2 + \dot{\epsilon}_{33}^2 + 2\dot{\epsilon}_{12}^2 + 2\dot{\epsilon}_{13}^2 + 2\dot{\epsilon}_{23}^2$  like that we add. So, that is summation convention we have used and we find

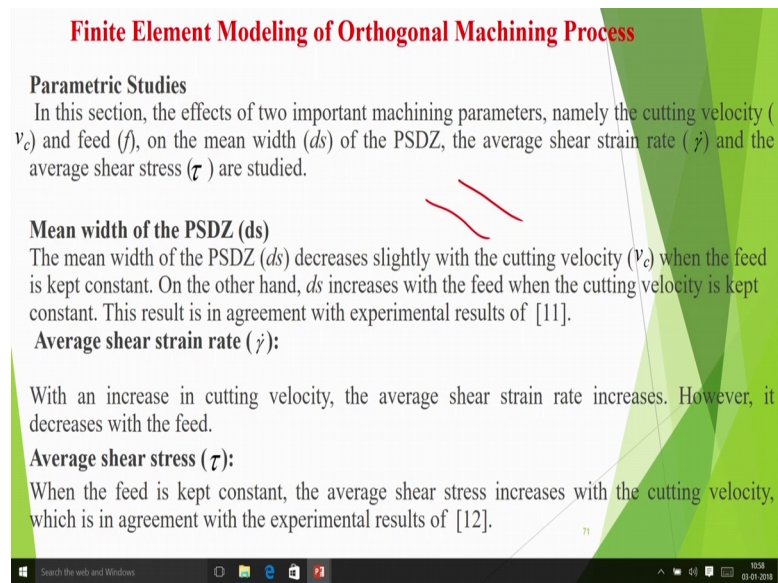
out equivalent strain. And then after that we put some criteria that if a equivalent strain is very small say it is one percent top the maximum strain rate then that portion can be considered as the boundary of the plastic and rigid zone. So, like that we estimate the primary shear deformation zone and its width we can find out ok. So, we do that this way and now what happens is that here now in this one here, if we find out  $\dot{\gamma}$  shear strain component engineering strain rate is given like this  $2 \epsilon \dot{\epsilon}$  and average of primary shear deformation zone that shear stress and then after that equivalent strain rate is also defined like this in this fashion  $\frac{3}{2} \sigma_{ij} \dot{\sigma}_{ij}$  and this is square root of that.

Then we put all these things and then after that we can estimate the shear zone etcetera. That means, that mathematics I am skipping, but you can see that here some results that we took this data next you know is 7860  $\sigma_y$  is given  $m$  is this is a and this is some exponent used in that model. This is not the aim of that friction factor this is something that in this constitutive relation and we meshed the region like this. And then after that we use the cutting velocity feed and rake angle we have taken in one way like this. And here we use three portion cutoff criteria that primary shear deformation zone boundary is taken at that point in which the shear stress is about not the shear stress, but the strain is three percent of the maximum strain.

So, by that we estimate the shear deformation zone. That means,  $DS$  what is the length of the  $DS$  and  $DS$  came out to be like this and experimental results from the literature provided this. And these are the predictions from the present model and experiment



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### Finite Element Modeling of Orthogonal Machining Process

**Parametric Studies**  
In this section, the effects of two important machining parameters, namely the cutting velocity ( $v_c$ ) and feed ( $f$ ), on the mean width ( $ds$ ) of the PSDZ, the average shear strain rate ( $\dot{\gamma}$ ) and the average shear stress ( $\tau$ ) are studied.

**Mean width of the PSDZ ( $ds$ )**  
The mean width of the PSDZ ( $ds$ ) decreases slightly with the cutting velocity ( $v_c$ ) when the feed is kept constant. On the other hand,  $ds$  increases with the feed when the cutting velocity is kept constant. This result is in agreement with experimental results of [11].

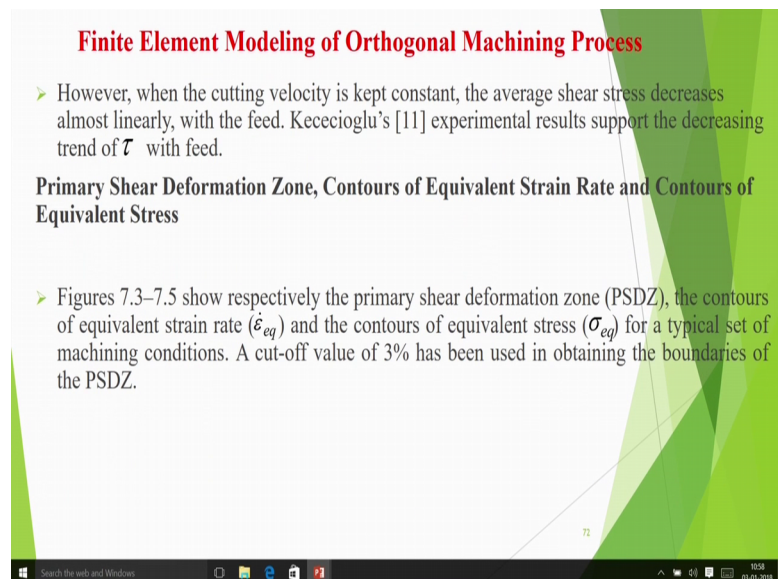
**Average shear strain rate ( $\dot{\gamma}$ ):**  
With an increase in cutting velocity, the average shear strain rate increases. However, it decreases with the feed.

**Average shear stress ( $\tau$ ):**  
When the feed is kept constant, the average shear stress increases with the cutting velocity, which is in agreement with the experimental results of [12].

So, matching is quite good and then we did some parametric study we obtained that mean width of the primary shear deformation zone. That means, the width of the shear zone this one decreases with the cutting velocity and it increases with feed and then every shear rate increase in cutting velocity every shear strain rate increases expected and it decreases with the feed.

And every shear stress increases with the cutting velocity which is in a agreement with the experimental result this is because of the shear; that means, viscoplasticity effect

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### Finite Element Modeling of Orthogonal Machining Process

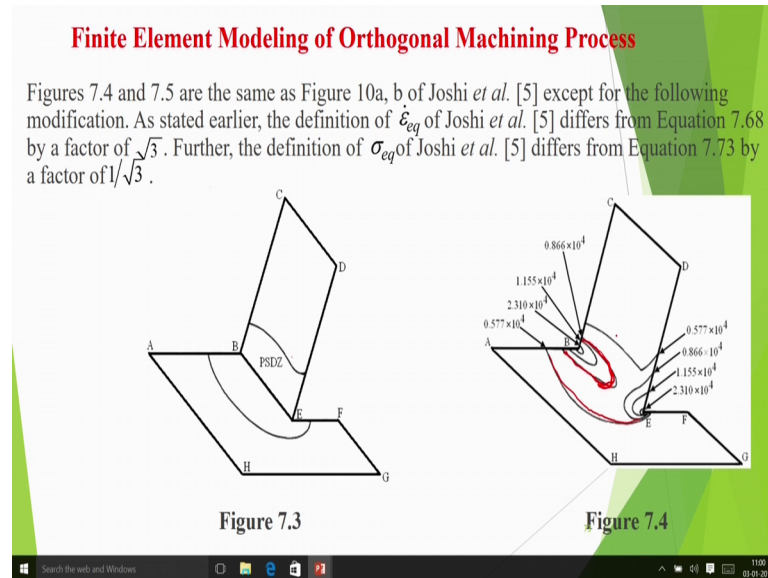
- However, when the cutting velocity is kept constant, the average shear stress decreases almost linearly, with the feed. Kececioglu's [11] experimental results support the decreasing trend of  $\tau$  with feed.

**Primary Shear Deformation Zone, Contours of Equivalent Strain Rate and Contours of Equivalent Stress**

- Figures 7.3–7.5 show respectively the primary shear deformation zone (PSDZ), the contours of equivalent strain rate ( $\dot{\epsilon}_{eq}$ ) and the contours of equivalent stress ( $\sigma_{eq}$ ) for a typical set of machining conditions. A cut-off value of 3% has been used in obtaining the boundaries of the PSDZ.

And when the cutting velocity is kept constant then the every shear stress decreases linearly with the feed and then we obtain the contours of the equivalent strain rate these contours have been planted here.

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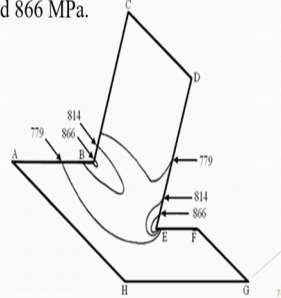
So, these contours are presented here this is primary shear deformation zone suppose this one we have taken this was I think experimental one here that experimental one. And this we have got the contour shear that this one is say here you are getting equivalent strain rate point 0.77 into 10 to the power 4 here. It is this much it is very high here it is somewhat here also it is high then it is somewhat it is here two point three into 10 to the power four like that. So, we have these values and we get the contour; that means, contour meaning is that from here in this region from here to here one particular equivalent strain is there here one particular strain is there like that these contours are plotted.

So, this way that we can get some information about the cutting mechanics, but even if you do this analysis usually it will take lot of time it is not that it is very fast, but it can give you all the insight about that that where these strain rate may be high where the stresses may be high such type of things are done ok.

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## Finite Element Modeling of Orthogonal Machining Process

- Therefore, the numerical values of  $\dot{\epsilon}_{eq}$  and  $\sigma_{eq}$  in Figures 7.4 and 7.5 are obtained by multiplying the corresponding values of Figure 10a, b of Joshi *et al.* [5] by the factors of  $1/\sqrt{3}$  and  $\sqrt{3}$  respectively.
- It is observed that the maximum values of  $\dot{\epsilon}_{eq}$  and  $\sigma_{eq}$  occur near the cutting edge and are of the order of  $2.31 \times 10^4 \text{ s}^{-1}$  and 866 MPa.



**Figure 7.5**

So, now these are the equivalent stresses and there it is observed that the maximum value of equivalent strain is 866 800 66 mega Pascal 866 mega Pascal is this 1 and then this is equivalent stress contour. That means, here closed type of thing here you are getting less, but here you are getting more and here also you are getting more this one.

So, this is done. So, this much I have told about that I have given some exposure about the finite element model of metal cutting also in more detail you can see from the published papers.

Thank you very much.