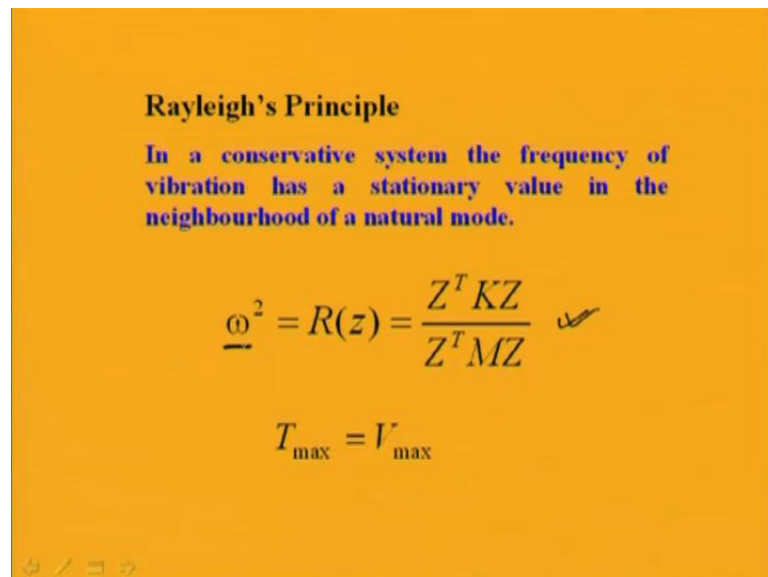


Mechanical Vibrations
Prof. S. K. Dwivedy
Department of Mechanical Engineering
Indian Institute of Technology, Guwahati

Module - 10
Lecture -3
Approximate Solutions for Continuous and Discrete Systems
Durckerley, Rayleigh-Ritz and Galerkin Method

So, in the last 2 lecture, we are studying about this approximate method, to find the natural frequency of discrete and distributed mass systems. So, we have already studied about this Rayleigh method and matrix iteration method.

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Rayleigh's Principle

In a conservative system the frequency of vibration has a stationary value in the neighbourhood of a natural mode.

$$\underline{\omega}^2 = R(z) = \frac{Z^T K Z}{Z^T M Z} \quad \checkmark$$
$$T_{\max} = V_{\max}$$

So, in Rayleigh principle tells in a conservative system the frequency of vibration has a stationary value in the neighborhood of the natural mode. So, in case of Rayleigh method we determine the Rayleigh quotient. So, which is given by this expression here Z is the approximate function we are taking or approximate mode. We are taking in case of discrete system, K is the stiffness matrix and M is the mass matrix of the system by taking this Rayleigh quotient. So, we can find the omega square or the frequency square of the frequency and we can find the frequency. Also we can equate the maximum kinetic energy to the maximum potential energy to find the fundamental frequency.

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Matrix Iteration Method

$$M \ddot{x} + K x = 0$$
$$[A]\{X\} = \lambda \{X\} \quad \lambda = \omega^2$$

•Any normal mode when multiplied with the dynamic matrix will reproduce itself.

And in the matrix iteration method, so this method is particularly useful for discrete system. So, let us consider a n degrees of freedom system, the equation motion of the system can be written by $M \ddot{x} + K x = 0$ or by taking the dynamic matrix $M^{-1}K$. I can write this expression in this way that is $AX = \lambda X$ where λ is equal to ω^2 $\lambda = \omega^2$ is the Eigen value of the system and X is the mode shapes or Eigen vector of this system. So, from this any normal mode, when for this you can observe that any normal mode, so let X is the normal mode.

So, any normal mode when multiplied with this dynamics matrix you can see that it repeat itself. So, this is basic principle of this matrix iteration method. So, by taking any approximate mode, so when we are multiplying this with this dynamic matrix we will get another vector. So, if you, will normalize that thing and check with the previous vector or assumed vector. So, if it is not same we can continue that iteration and we can find the mode or normal mode. So, after finding the normal mode we can obtain the natural frequency of the system or we can obtain the value of λ .

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Estimation of higher mode frequencies

$$\underline{X} = c_1 X_1 + c_2 X_2 + \dots + c_n X_n$$

$$X = \{\bar{x}_1, \bar{x}_2, \bar{x}_3\}' \quad X_i = \{x_1, x_2, x_3\}_i'$$

$$X_i^T M X_j = c_1 X_i^T M X_1 + c_2 X_i^T M X_2 + \dots + c_n X_i^T M X_n = c_1 X_i^T M X_1$$

$X_i^T M X_j = 0$
 $i \neq j$

So, to obtain the higher mode, so we should apply the orthogonality principle because in case of the free vibration response what we are taking it contains the first mode first mode or first modal frequency. So, if you eliminate this first mode then only we can find or the assumed solution will lead to the second mode. So, in this case let we take this free vibration response or mode shape as X. So, this X can be written as a linear combination of other modes. So, it can be written as c 1 X 1 plus C 2 X 2 plus C 3 X 3 and c n X n. So, in this way I can write this X equal to this is the assumed mode. So, this assumed mode X can be written in this form. X 1 bar X 2 bars X 3 bar for a 3 degrees for freedom system. I have written this thing.

So, this X equal to X 1 bar X 2 bar X 3 bar transpose and this Xi these are normal modes. So, these X 1 X 2 X 3 transpose. So, the normal mode this X 1 X 2 X 3 can be written for a 3 degrees of freedom system like this and this is assumed mode. So, now applying the orthogonality principle, so I can pre-multiply X transpose M in this equation. So, in the first equation when pre-multiplying these X transpose M I have X transpose M x equal to c 1 into X 1 transpose M X 1 plus c 2 into X 1 transpose M x 2 and plus c 3 into X 1 transpose M x 3. And similarly, you can find this is equal to plus c n X 1 transpose M x n. So, from the orthogonality principle you know that Xi transpose M x j. So, this is equal to 0 when I not equal j.

So, when i not equal to j from orthogonality principle we know that $X_i^T M X_j$ equal to 0. So, this c_2 , so this $X_1^T M X_2$ equal to 0. So, this part equal to part to 0 and other times all other terms except this $X_1^T M X_1$ equal to 0. So, this $X_1^T M X_1$ equal to. So, if I am taking this X_1 as a normalized vector then it will be equal to 1. So, or it can be written as $c_1 X_1^T M X_1$. So, these term will remain and other terms will be 0. So, this $X_1^T M X$ will be equal to $c_1 X_1^T M X_1$. So, now to eliminate the first mode we should put the c_1 equal to 0. So, putting c_1 equal to 0 the left hand side of this equation becomes 0.

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$$X_1^T M X = 0$$

$$X_1^T M X = \{x_1, x_2, x_3\} \begin{bmatrix} m_1 & 0 & 0 \\ 0 & m_2 & 0 \\ 0 & 0 & m_3 \end{bmatrix} \begin{Bmatrix} \bar{x}_1 \\ \bar{x}_2 \\ \bar{x}_3 \end{Bmatrix} = 0$$

$$m_1 x_1 \bar{x}_1 + m_2 x_2 \bar{x}_2 + m_3 x_3 \bar{x}_3 = 0$$

$$\left. \begin{aligned} \bar{x}_1 &= -\frac{m_2}{m_1} \left(\frac{x_2}{x_1} \right) \bar{x}_2 - \frac{m_3}{m_1} \left(\frac{x_3}{x_1} \right) \bar{x}_3 \\ \bar{x}_2 &= \bar{x}_2 \\ \bar{x}_3 &= \bar{x}_3 \end{aligned} \right\}$$

So, the left hand side of this equation that is equal to $X_1^T M X$ equal to 0. Now, for the 3 degrees freedom system I can substitute these in this way. So, $X_1^T M X_2$ $X_1^T M X_3$ and this is mass matrix and this is these assumed mode. So, assumed I have taken as \bar{x}_1 \bar{x}_2 \bar{x}_3 . So, by multiplying this thing you can get this equation that is $m_1 \bar{x}_1 + m_2 \bar{x}_2 + m_3 \bar{x}_3 = 0$. So, in this case I can express these \bar{x}_1 in terms of \bar{x}_2 and \bar{x}_3 . So, \bar{x}_1 I have written equal to minus m_2 by m_1 x_2 by x_1 into \bar{x}_2 minus m_3 by m_1 into x_3 by x_1 into \bar{x}_3 , and I can write other 2 expression that is $\bar{x}_2 = \bar{x}_2$ and $\bar{x}_3 = \bar{x}_3$. So, these are identity equation. So, this is $\bar{x}_2 = \bar{x}_2$ and $\bar{x}_3 = \bar{x}_3$. So, I can write these 3 expression in a matrix form.

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$$\begin{Bmatrix} \bar{x}_1 \\ \bar{x}_2 \\ \bar{x}_3 \end{Bmatrix} = \begin{bmatrix} 0 & -\frac{m_2}{m_1} \left(\frac{x_2}{x_1} \right) & -\frac{m_3}{m_1} \left(\frac{x_3}{x_1} \right) \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{Bmatrix} \bar{x}_1 \\ \bar{x}_2 \\ \bar{x}_3 \end{Bmatrix}$$

or, $X = SX$

where $S = \begin{bmatrix} 0 & -\frac{m_2}{m_1} \left(\frac{x_2}{x_1} \right) & -\frac{m_3}{m_1} \left(\frac{x_3}{x_1} \right) \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}$

So, that matrix will be equal to X_1 bar X_2 bar X_3 bar equal to. So, 0 minus M_2 by M_1 X_2 by X_1 minus M_3 by M_1 X_3 by X_1 then 0 1 0 0 0 0. So, you can note that the first column of this matrix equal to 0. So, this first column equal to 0. So, as the first column equal to 0, if you multiply this matrix or if you take this matrix. So, it will sweep out the first mode from the assumed vibration. So, this matrix is known as the sweeping matrix S . So, this matrix is known as the sweeping matrix. So, using the sweeping matrix, so we can find or we can eliminate the first mode from the resulting vibration. Similarly, to eliminate the second mode we can put c_2 equal to 0 and for third mode c_3 equal to 0. So, when you are eliminating first 2 modes we should put c_1 and c_2 equal to 0 and applying this orthogonality principle, we can find the sweeping matrix corresponding to the second mode third mode or higher modes. So, by eliminating these modes from the assumed vibration, then our iteration process will converse to the higher modes.

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Example

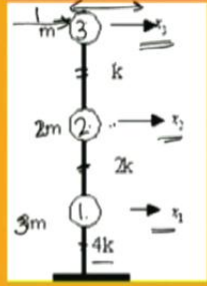
$$a_{11} = a_{21} = a_{31} = \frac{1}{4k}$$

$$a_{11} = a_{21} = a_{31} = a_{12} = a_{13} = \frac{1}{4k}$$

$$a_{22} = \frac{1}{4k} + \frac{1}{2k} = \frac{3}{4k}$$

$$a_{22} = a_{23} = a_{32} = \frac{3}{4k}$$

$$a_{33} = \frac{1}{4k} + \frac{1}{2k} + \frac{1}{k} = \frac{7}{4k}$$

$$m = \begin{bmatrix} 3m & 0 & 0 \\ 0 & 2m & 0 \\ 0 & 0 & m \end{bmatrix}$$


So, let us take 1 example to study this thing. So, let this is a beam or this is a rod with 3 masses and this is vibrating in this transverse direction. So, the vibration is taking place in this transverse direction. So, this mass is 3 M and this mass is 2 M and this mass is M. So, the you can think of a tall building with mass 3 M concentrated here, 2 M concentrated here and M concentrated here these are the mass of the roof and these are the stiffness of the walls. So, these stiffness the first part it is 4 k second part I have taken it equal to 2 k and the third part it is equal to k. So, due to this wind motion, so it may vibrate in this transfer direction. So, we have to study or we have to find the first 3 natural frequency of the system due to this vibration to find this. So, first we have to find the stiffness matrix, mass matrix.

So, mass matrix is simple and mass matrix you can write in this way. So, the max matrix will be equal to. So, this is X 1 I am taking this vibration X 1 this is X 2 and this is X 3. So, this is the first mass I am taking this is the second mass and this is the third mass. So, in this case, so I can write this 3 M 0 0 0 2 M 0 and 0 0 0 0 0 M. So, this is the mass matrix and to find this stiffness matrix, or we can proceed in the other way that is we can find the flexibility matrix. And by finding the flexibility matrix also we can find the equation motion, and from that equation motion we can proceed. So, to find the displacement, so a 1 1 that is influence coefficient let us first find the influence coefficient. The influence coefficient a 1 1, that is displacement at 1 due to unit force at 1 while the forces at other places are 0.

So, as there is no force acting at these 2 places and only force acting at this place is equal unity force. So, these part will give resistance to its motion, as this part has a stiffness of $4k$. So, the displacement will be $1/4k$. So, a $1/4k$ equal to $1/4k$, so as there is no force acting at this 2 and 3. So, a $1/4k$ that is displacement at 2 due to a unit force at 1 will be same as the displacement of this position. That means, the beam will bend from this position and it will be straight at these 2 position. So, the displacement here it will be $1/4k$ and also the displacement here will be $1/4k$ and the displacement here will be $1/4k$, so this displacement. So, this influence coefficient are written like this a $1/4k$ equal to a $2/4k$ equal to a $3/4k$ equal to $1/4k$. Similarly, let us find.

So, we can find this a $2/4k$. So, applying a force at 2 unit force at 2 and no force at 3 and 1 we can find this influence coefficient a $2/4k$. So, a $2/4k$, so while finding a $2/4k$. So, these stiffness and these stiffness will be coming into picture. So, these 2 are in series. So, the resulting or equivalent stiffness will be $1/4k$ plus $1/2k$. So, this is $1/4k$ plus $1/2k$ that is equal to $3/4k$. So, the displacement will be equal to $1/3k$, so a $2/4k$ equal to $1/3k$. So, when it is displaced to this position let it is coming to this position due to the unit force here. So, the upper portion also will have the same motion as there is no force acting at 3. So, this a $3/4k$ also will be same as that of a $2/4k$ and applying the reciprocity theorem also you know a I_j equal to a j_I that is a displacement at I. Due to a unit force at j equal to a displacement at j due to a unit force at I that is the reciprocity theorem.

So, applying that theorem you can write a $2/4k$ equal to a $2/3k$ equal to a $3/4k$ equal to $1/3k$. Similarly, here a $1/4k$ already you know that this a $1/4k$ equal to a $2/4k$ equal to a $3/4k$. So, by applying reciprocity theorem a $2/4k$ equal to a $1/2k$ similarly a $3/4k$ equal to a $1/3k$, so this is equal to $1/4k$. Now, to obtain this a $3/3k$, so we can apply a unit force at 3 and no force at 2 1 1. So, when we are applying a unit force here at 3. So, this part this part these 3 parts will give resistance to its motion. So, these 3 stiffnesses are in series, so the net stiffness or net equivalent stiffness will be equal to $1/4k$ will be equal to $1/4k$ plus $1/2k$ plus $1/2k$ plus $1/k$. So, $1/k$ equivalent will be equal to $1/4k$ plus $1/2k$ plus $1/k$. So, the net or the deflection of this 3 will be equal to $1/4k$ plus $1/2k$ plus $1/k$ that is equal to $7/4k$, so a $3/3k$ equal to $7/4k$.

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$$a = \frac{1}{4k} \begin{bmatrix} 1 & 1 & 1 \\ 1 & 3 & 3 \\ 3 & 3 & 7 \end{bmatrix} \quad M = \begin{bmatrix} 3 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} m$$

Hence the displacements at different position due to inertia forces are

$$x_1 = -(a_{11} 3m\ddot{x}_1 + a_{12} 2m\ddot{x}_2 + a_{13} m\ddot{x}_3)$$

$$x_2 = -(a_{21} 3m\ddot{x}_1 + a_{22} 2m\ddot{x}_2 + a_{23} m\ddot{x}_3)$$

$$x_3 = -(a_{31} 3m\ddot{x}_1 + a_{32} 2m\ddot{x}_2 + a_{33} m\ddot{x}_3)$$

So, the flexibility matrix I can write it in this form. So, a will be equal to 1 by 4 k 1 1 1 1 3 3 and 3 3 7 and the mass matrix already I have written. So, that is equal to 3 0 0 0 2 0 and 0 0 1 into m. So, the hence the displacement at these position now actually at these positions this inertia forces are acting. So, the displacement at these position due to this actual forces that is here it will be 3 m into omega square with a negative sign, here 2 m into omega square and here m into omega square with negative sign So, as inertia force act in the opposite direction to the displacement. We have this negative sign in 3 places, and the displacement here will be equal to displacement at position 1.

So, the displacement at position 1 will be equal to a 1 1 into the force at 1 similarly a 1 2 into the force at 2 and a 1 3 into the force at 3. So, as a I j equal to displacement at I due to unit force at j. So, due to these inertia forces at these positions, so you can find the net displacement at position 1 equal to a 1 1 into 3 into m x 1 double dot plus a 1 2 into 2 into m x 2 double dot a 1 3 into m x 3 double dot. So, due to inertia forces it acts opposite to the direction of displacement that is why the negative sign is given here. Similarly, x 2 will be equal to a 2 1 into the force acting at 1 that is 3 m x double dot and a 2 2 into force acting at 2 that is equal to 2 m that this is mass and acceleration is x 2 double dot and a 2 3 into m into x 3 double dot.

Similarly, x 3 will be equal to minus a 3 1 4. So, this is equal to 3 3 m x 1 double dot plus a 3 2 2 into m x 2 double dot and a 3 3 into m x 2 double dot. So, these are the

inertia forces and multiply it with this mass this is the x_3 double dot or x_1 double dot x_2 double as acceleration multiplied by the mass of the that location will give the inertia force and the inertia force acts in a direction opposite to the direction of displacements. So, in this way you can find the displacement $x_1 x_2 x_3$.

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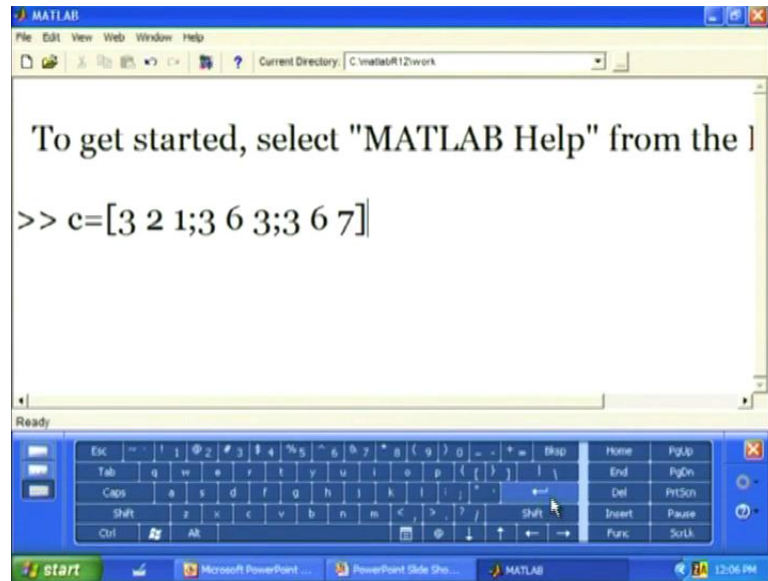
$$\ddot{x}_i = -\omega^2 x_i$$

$$X = \lambda a M X$$

$$\begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix} = \frac{m\omega^2}{4k} \begin{bmatrix} 3 & 2 & 1 \\ 3 & 6 & 3 \\ 12 & 6 & 7 \end{bmatrix} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix}$$

So, the displacement $x_1 x_2 x_3$ after substituting this x_i double dot equal to minus omega square x_i and we can write this $x_1 x_2 x_3$ equal to $m \omega^2$ by $4 k$ $\begin{bmatrix} 3 & 2 & 1 \\ 6 & 3 & 12 \\ 6 & 7 & 6 \end{bmatrix}$ and $x_1 x_2 x_3$. So, this is the equation iterative equation, which we will use to find the frequency of the system. So, let us take some value of $x_1 x_2 x_3$ and see what is happening or what we are getting. So, I will show you the simulation now. So, let me show the simulation by using this matlab.

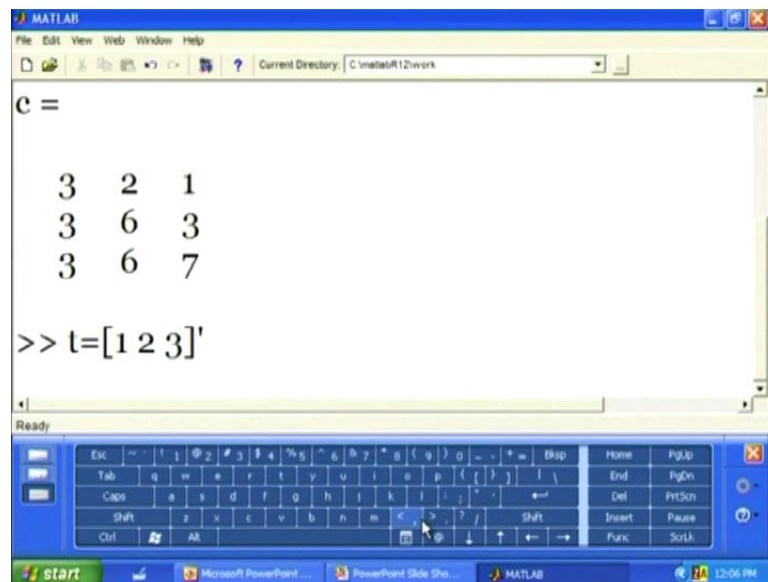
(Refer Slide Time: 18:19)



The image shows a MATLAB window with a blue title bar and a menu bar (File, Edit, View, Web, Window, Help). The current directory is C:\matlabR12\work. The command prompt displays the text: "To get started, select 'MATLAB Help' from the l" followed by the command `>> c=[3 2 1;3 6 3;3 6 7]`. The status bar at the bottom shows "Ready" and a Windows taskbar with the Start button and several open applications.

So, I can write this matrix let me take this matrix at c matrix. So, this c matrix equal to I can write this. So, this is equal to 3 2 1. So, this is 3 this is 2 and 1 3 2 1 then 3 3 6 3 2 1 3 6 3 3 then 6 then 3 and 3 6 7 3 6 and 7.

(Refer Slide Time: 19:05)

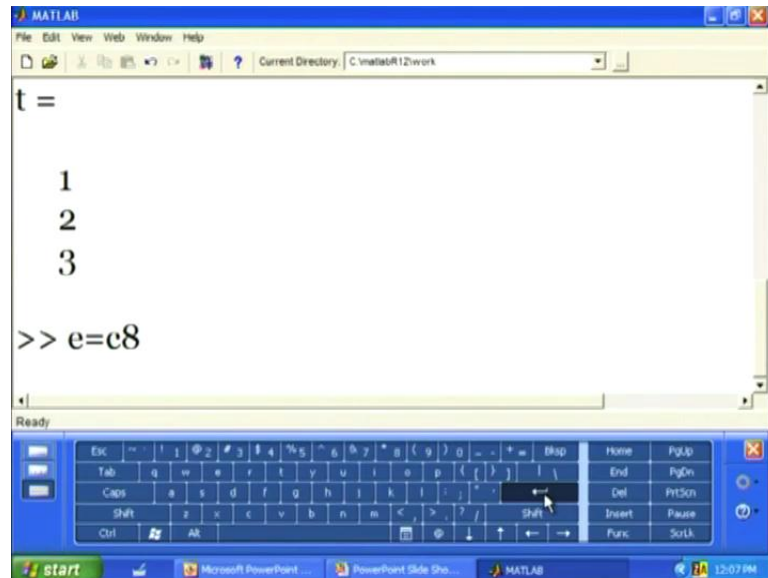


The image shows a MATLAB window with a blue title bar and a menu bar (File, Edit, View, Web, Window, Help). The current directory is C:\matlabR12\work. The command prompt displays the output of the previous command: `c =` followed by a 3x3 matrix: $\begin{bmatrix} 3 & 2 & 1 \\ 3 & 6 & 3 \\ 3 & 6 & 7 \end{bmatrix}$. Below the matrix, the command `>> t=[1 2 3]'` is entered. The status bar at the bottom shows "Ready" and a Windows taskbar with the Start button and several open applications.

So, this matrix you can see will be equal to. So, this is the matrix I have taken that is 3 2 1 3 6 3 and 3 6 7. Now, in this matrix, so I have to multiply I have to multiply 1 assumed mode. So, let me take the assumed mode let us take any arbitrary assumed mode. So, let me take it equal to 1 2 3. So, the trial function let me take this trial function t equal to.

So, the trail function I can take equal to. So, this is equal to 1 2 3 I will take. So, this is 1 2 and 3. So, I will take the transpose of this. So, this is the trail function.

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```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work

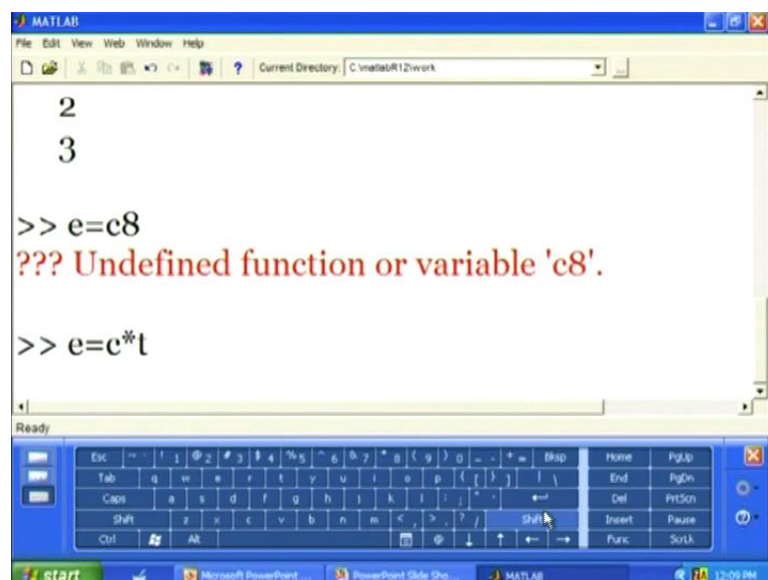
t =

     1
     2
     3

>> e=c8
```

So, this trial function equal to 1 2 3. So, I have to multiply these trial function with the previous function that is c. So, let me find let I am writing e equal to c multiplied by this.

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```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work

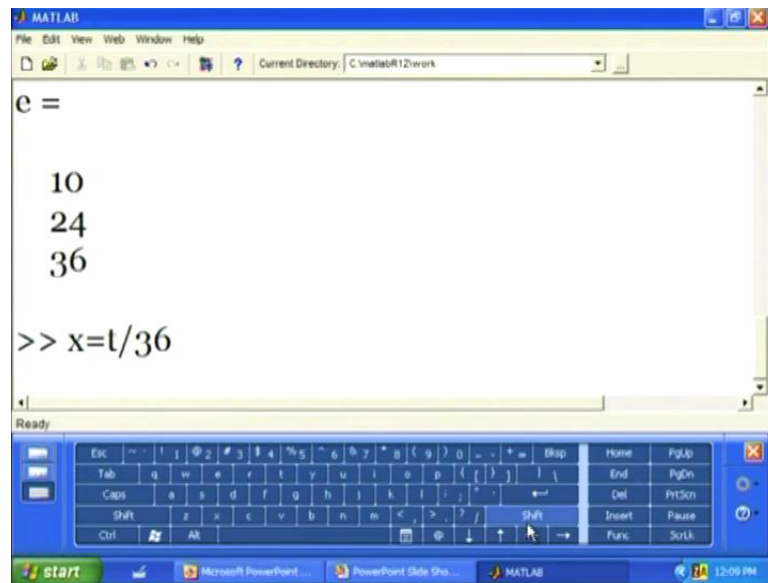
     2
     3

>> e=c8
??? Undefined function or variable 'c8'.

>> e=c*t
```

So, I can write e equal to c.

(Refer Slide Time: 20:30)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work

e =

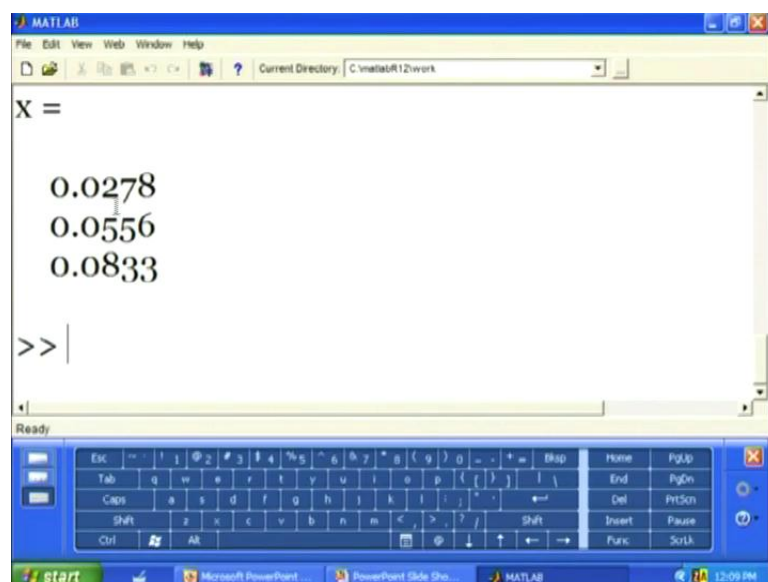
    10
    24
    36

>> x=t/36

Ready
```

So, by multiplying the c into t, so we are getting this vector that is 10 24 and 36. Now, I can normalize this vector that is this 10 had 24 and 36. So, I can normalize it by dividing this 36 in this, so by to divide this thing by 36. So, let me write this t. So, this thing will be my x or the next x for next iteration. So, x will be equal to t by this e or t by 36 let me directly write, so t by 36.

(Refer Slide Time: 21:14)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work

X =

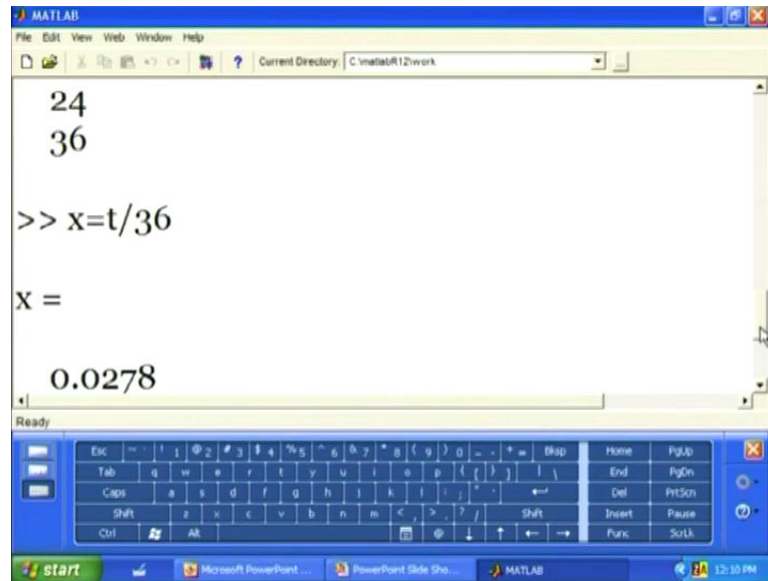
    0.0278
    0.0556
    0.0833

>> |

Ready
```

So, t by 36 gives me. So, this is equal to 0.0278.

(Refer Slide Time: 21:19)



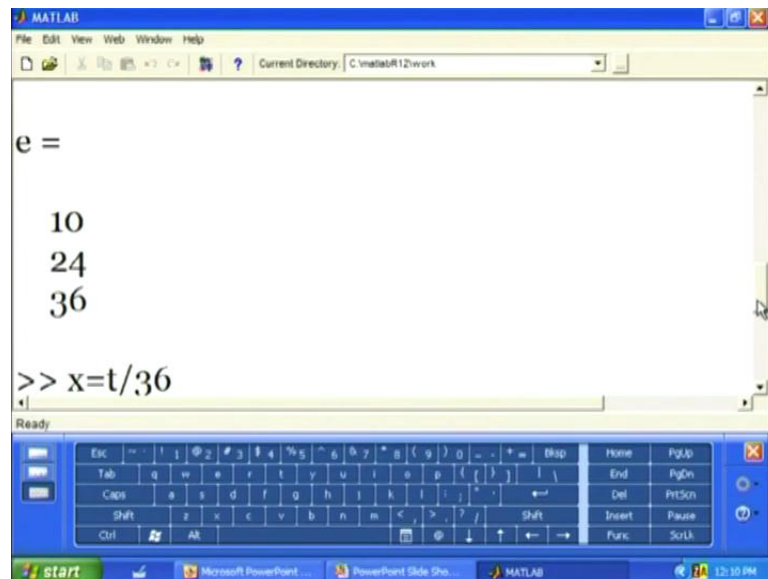
A screenshot of the MATLAB command window. The window title is 'MATLAB' and the current directory is 'C:\matlabR12\work'. The command prompt shows the following sequence of input and output:

```
24
36
>> x=t/36
X =
0.0278
```

The window also shows a virtual keyboard and a Windows taskbar at the bottom with the time 12:10 PM.

So, this value you can see.

(Refer Slide Time: 21:20)



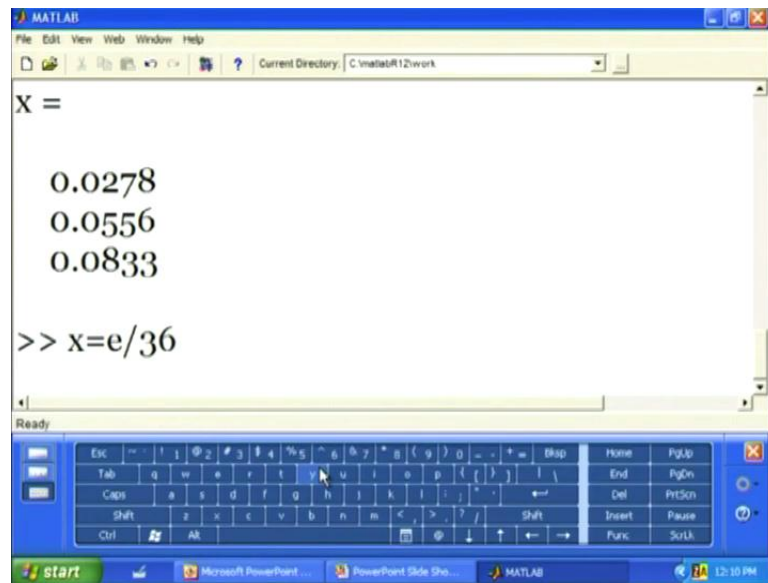
A screenshot of the MATLAB command window. The window title is 'MATLAB' and the current directory is 'C:\matlabR12\work'. The command prompt shows the following sequence of input and output:

```
e =
10
24
36
>> x=t/36
```

The window also shows a virtual keyboard and a Windows taskbar at the bottom with the time 12:10 PM.

So, this is equal to 10 24 and 36. So, x will be equal to. So, this is the function e. So, by e I have to divide this 36 to get this normal mode.

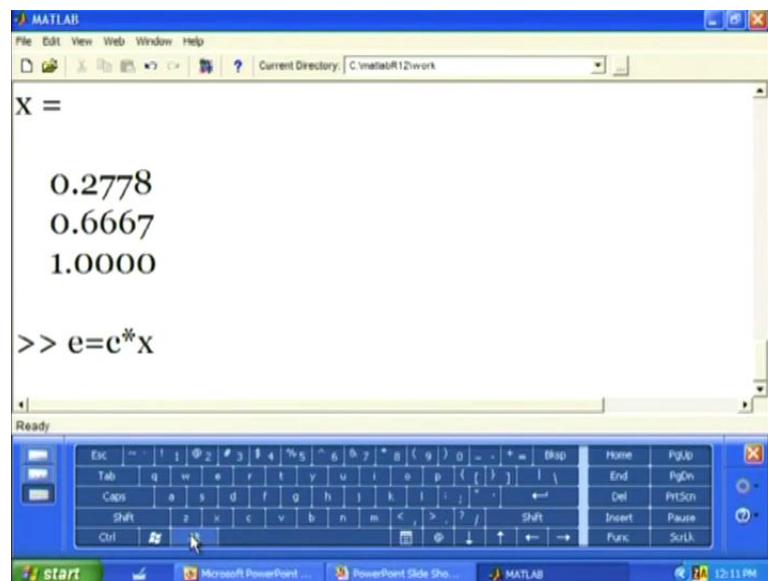
(Refer Slide Time: 21:35)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work
X =
    0.0278
    0.0556
    0.0833
>> x=e/36
Ready
```

So, this will be. So, I can write x equal to e by. So, this is e by 36 not t by 36.

(Refer Slide Time: 21:44)

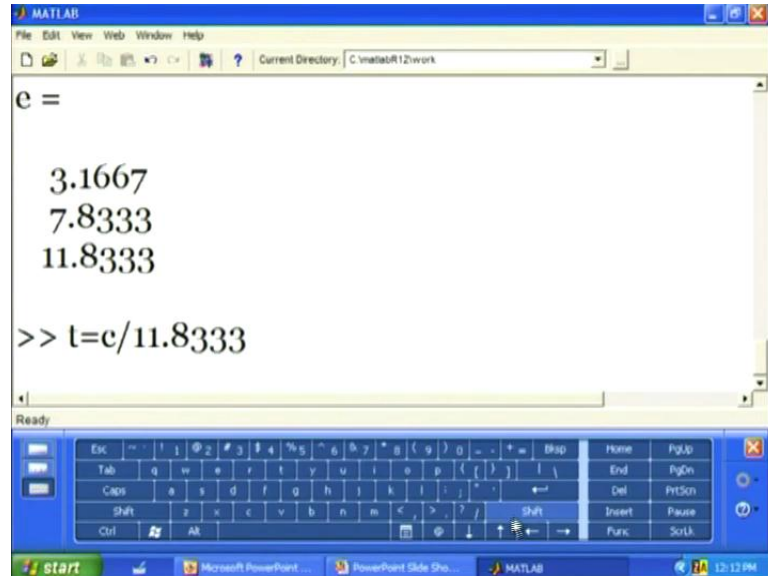


```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work
X =
    0.2778
    0.6667
    1.0000
>> e=c*x
Ready
```

So, e by 36 will give me 0.2778, 0.6667 and 1. So, this is the normalized value, I will take for next iteration. So, you can note that this value is not equal to the value we have assumed that is we have assumed a value equal to. So, we have assumed this value 1 2 3 and by putting that thing in this equation we are getting a value equal to 0.2778, 0.6667 and 1. So, now for next iteration I will take this value and multiply this with the c value.,

so c. So, next iteration I can write c into. So, I can write let s equal to or e equal to e equal to c into c into this x. So, I will multiply this thing.

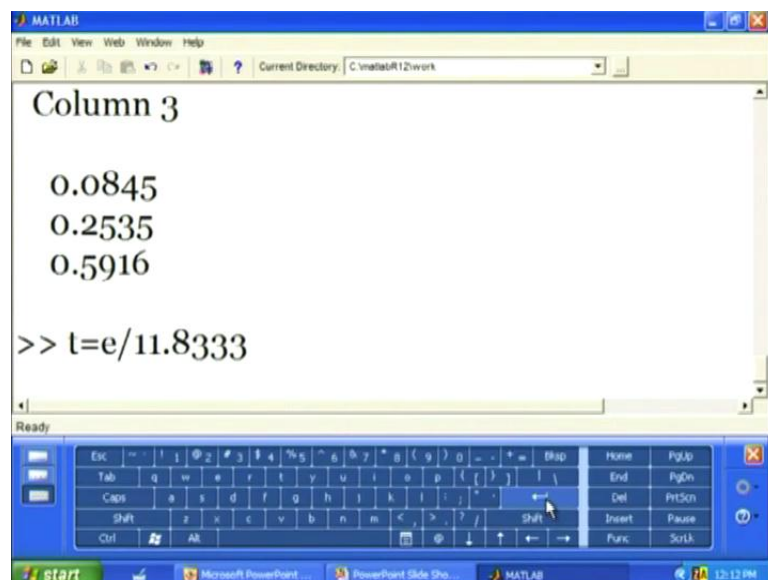
(Refer Slide Time: 22:44)



A screenshot of the MATLAB command window. The current directory is C:\matlabR12\work. The variable `e` is assigned the value `3.1667`. Below it, the values `7.8333` and `11.8333` are displayed. The command `>> t=c/11.8333` has been entered and executed. The status bar at the bottom shows the system time as 12:13 PM.

And I am getting a value this e equal to this and normalize this thing I can use this e by. So, I will half this trail function again will be equal to c divided by this 11 that is e 3. So, I can write e in bracket 3 also I can write or I can write this is equal to 11.833.

(Refer Slide Time: 23:19)

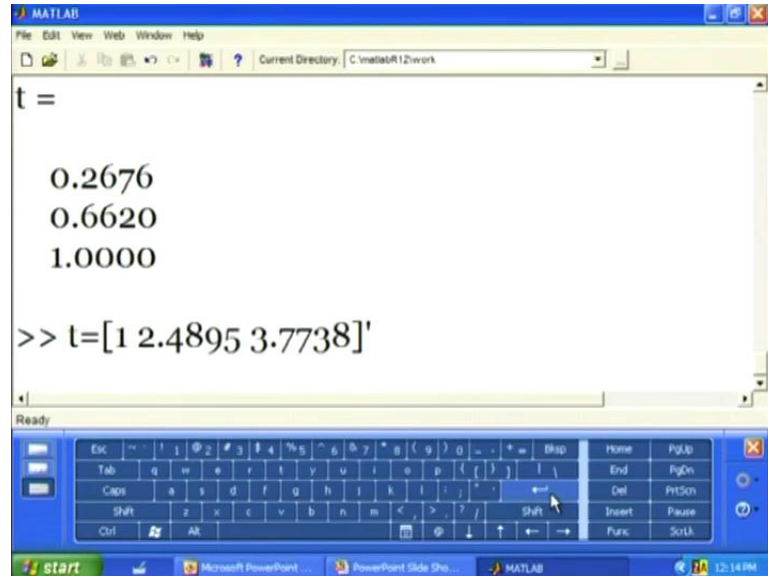


A screenshot of the MATLAB command window. The current directory is C:\matlabR12\work. The text "Column 3" is displayed. Below it, the values `0.0845`, `0.2535`, and `0.5916` are displayed. The command `>> t=e/11.8333` has been entered and executed. The status bar at the bottom shows the system time as 12:12 PM.

So, by dividing this thing I will get the value. So, I have to. So, this is the value of e I obtain, so in this e. So, trail value I have to normalize this. So, I have to divide this e by

11 this thing. So, it will be. So, this t will be equal to t will be equal to e by. So, it will be equal to e by 11. So, you can see this thing. So, this is equal to 11.8333. So, 8333.

(Refer Slide Time: 24:06)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work

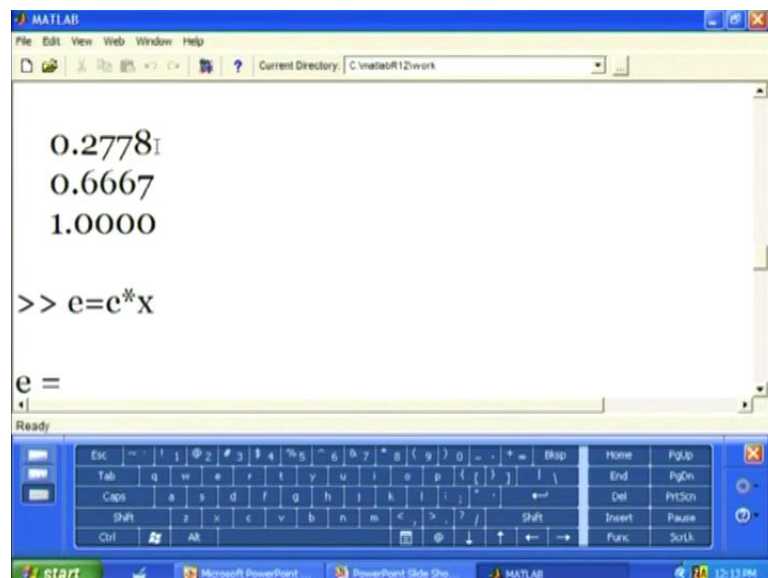
t =

    0.2676
    0.6620
    1.0000

>> t=[1 2.4895 3.7738]'
```

So, dividing this thing, so we are normalizing these under normalized value is now 0.2676 and 0.6620 and 1.

(Refer Slide Time: 24:21)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work

    0.2778i
    0.6667
    1.0000

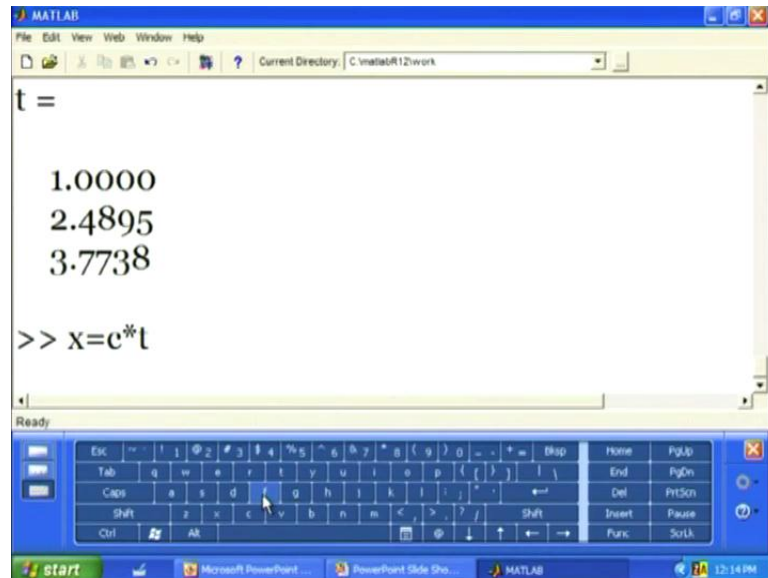
>> e=c*x

e =
```

So, you have taken a trial function that is equal to. So, the trial function what we have taken. So, this is equal to 0.2778, 0.6667 and 1 and you are getting the function equal to 0.2676, 0.6620 and 1. So, in this way you can take this as the, for next iteration, you can

take this function this mode you can take and you can proceed and finally, you will find the value for which both side will be same. So, let me take the final value. So, what I obtained. So, t trial function if you take t equal to. So, let me take this trial function equal to t equal to 1 2.4895 and then 3.7738. So, taking these as the trial function, so let this is taken as the trial function.

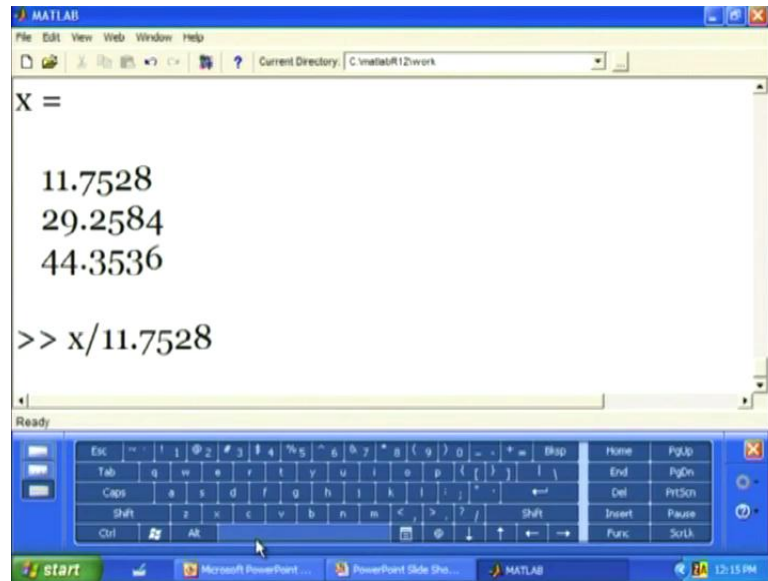
(Refer Slide Time: 25:31)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlab\12\work
t =
1.0000
2.4895
3.7738
>> x=c*t
```

So, this is the trial function we have taken that is equal to 1 2.4895 and this. So, let me multiply this with c. So, let x I am writing x equal to c multiplied by. So, c multiplied by this t.

(Refer Slide Time: 25:52)



```
MATLAB
File Edit View Web Window Help
Current Directory: C:\matlabR12\work
X =
11.7528
29.2584
44.3536
>> x/11.7528
```

The image shows a MATLAB command window. The variable X is assigned the values 11.7528, 29.2584, and 44.3536. The user then enters the command `>> x/11.7528`. The window also shows a virtual keyboard and the Windows taskbar at the bottom.

So, c multiplied by t will give me this is the value I am getting. So, by normalizing this thing, so by normalizing this thing I can write I will get this answer by dividing this x by. So, I am normalizing with respect to this first 1 in this case because in the first 1 I have put it equal to 1. So, I put this is equal to 11.7528. So, this will give the value 1 2.4895, 3.7739. So, you just note the trial function taken was 1 2.4895, 3.7738 and you are getting the value also same 1 2.4895, 3.7739. So, this, so up to third decimal you are getting the value accurately. So, in this way you can do the iteration to get. So, this is the first mode. So, in this way you can get this first mode value that is equal to 1 2.4895 and 3.7739.

(Refer Slide Time: 27:00)

$$\begin{aligned} \begin{Bmatrix} 1 \\ 2.4893 \\ 3.7735 \end{Bmatrix} &= \frac{m\omega^2}{4k} \begin{bmatrix} 3 & 2 & 1 \\ 3 & 6 & 3 \\ 3 & 6 & 7 \end{bmatrix} \begin{Bmatrix} 1 \\ 2.4893 \\ 3.7735 \end{Bmatrix} \\ &= \frac{m\omega^2}{4k} \begin{Bmatrix} 11.7521 \\ 29.2565 \\ 44.3503 \end{Bmatrix} \\ \begin{Bmatrix} 1 \\ 2.4893 \\ 3.7735 \end{Bmatrix} &= \frac{11.7521m\omega^2}{4k} \begin{Bmatrix} 1 \\ 2.4895 \\ 3.7738 \end{Bmatrix} \end{aligned}$$

So, we got the first mode equal to 1 2.4895, 3.7738. So, when I multiply it. So, I can write this equal to 1 2.4893, 3.7735 equal to 11.7521 m omega square by 4 k equal to same. So, you can note this left hand side and right hand side. So, this is the assumed mode we have taken, and this is the mode we obtained that is equal to x 1. So, in both the sides you can note that you are getting up to third decimal accurate thing. So, when this is equal to this then this part will be equal to 1. So, this is the first mode value and you can get this equal to 1.

(Refer Slide Time: 27:47)

$$\begin{aligned} X_1 &= \begin{Bmatrix} 1 \\ 2.4895 \\ 3.7738 \end{Bmatrix} \\ \frac{11.7521m\omega^2}{4k} &= 1 \\ \text{or, } \lambda_1 = \omega_1^2 &= 0.3404 \frac{k}{m} \end{aligned}$$

So, $11.7521 m \omega^2$ by $4k$. So, we equate to 1. So, in that way you can get the first mode frequency that is equal to ω^2 . So, this is equal to $0.3404 k$ by m or ω you can find root over of this and to find the second mode we can proceed we have to find the sweeping matrix. Already we have found the sweeping matrix for the first mode. So, we have to make c_1 equal to 0 in this assumed mode. to make c_1 equal to 0. So, we have to write the sweeping matrix.

(Refer Slide Time: 28:24)

$$S = \begin{bmatrix} 0 & -\frac{m_2}{m_1} \left(\frac{x_2}{x_1} \right) & -\frac{m_3}{m_1} \left(\frac{x_3}{x_1} \right) \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} 0 & -1.659 & -1.2579 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

So the new equation for second mode iteration is

$$\begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix} = \frac{m\omega^2}{4k} \begin{bmatrix} 3 & 2 & 1 \\ 3 & 6 & 3 \\ 3 & 6 & 7 \end{bmatrix} \begin{bmatrix} 0 & -1.659 & -1.2579 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix}$$

$\underbrace{\hspace{10em}}_{\alpha \cdot M}$
 $\underbrace{\hspace{10em}}_S$

So, in this sweeping matrix already we got these x_2 x_1 . So, this is the x_1 this is x_2 and x_3 for the first mode. So, substituting this x_1 x_2 x_3 for the first mode equal to these,, so we can write the sweeping matrix in this form. So, after getting the sweeping matrix, so I can substitute it in this expression that is x equal to $m \omega^2$ by $4k$ into this matrix into the sweeping matrix into x . So, this is the sweeping matrix and this is this is the a matrix that is flexibility matrix into these mass matrix. This matrix is flexibility matrix into mass matrix this is the sweeping matrix. So, this equation reduces to this.

(Refer Slide Time: 29:09)

$$\begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix} = \frac{m\omega^2}{4k} \begin{bmatrix} 0 & -2.98 & -2.7740 \\ 0 & 1.02 & -0.7740 \\ 0 & 1.02 & 3.226 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix}$$

$$\begin{Bmatrix} 1.0 \\ 0.8367 \\ -1.8996 \end{Bmatrix} = \frac{2.7756m\omega^2}{4k} \begin{Bmatrix} 1.0 \\ 0.8361 \\ -1.8988 \end{Bmatrix}$$

$\underbrace{\hspace{10em}}_1$

$\underbrace{\hspace{10em}}_{\times}$
 $\underbrace{\hspace{10em}}_{\times 2}$

So, $x_1 \times x_2 \times x_3$ equal to $m \omega^2$ by $4k$ into this matrix into $x_1 \times x_2 \times x_3$. So, you can note that in the first column all the terms are 0. So, now any assumed value of $x_1 \times x_2 \times x_3$ will not go to the first mode and you will get the next higher mode that is the second mode. So, let us start with any arbitrary value. So, proceed in the same way as I have shown in this iteration before. So, by proceeding in the same way you can find the value to be this. So, this will. So, this is the x you can when you are assuming x equal to 1.8367 minus 1.8996. So, you are getting a value that is equal to 1.8361 minus 1.8988. So, up to the second decimal you can see both are matching. So, you can take this value and now. So, this will be equal to 1.

(Refer Slide Time: 30:22)

$$X_2 = \begin{Bmatrix} 1.0 \\ 0.8361 \\ -1.8988 \end{Bmatrix} \quad \frac{2.7756m\omega^2}{4k} = 1$$
$$\lambda_2 = \omega_2^2 = \frac{4}{2.7756} \frac{k}{m} = 1.4411 \frac{k}{m}$$
$$\omega_2 = 1.2 \sqrt{\frac{k}{m}}$$

So, equating this middle part equal to 1 or the coefficient of this x equal to 1 you can find this second mode frequency that is lambda 2 equal to 1.4411 k by m. So, omega 2 will be equal to 1.2 root over per k by m. So, in this way you can get the second mode and this is the second mode this is the second mode shape. So, x 2 is the modal matrix for the second mode and omega 2 is the frequency of the second mode that is 1.2 into root over k by m.

(Refer Slide Time: 30:52)

For finding third mode

$$c_1 = c_2 = 0$$
$$c_1 = \sum_{i=1}^3 m_i (x_i)_1 \bar{x}_i = 0 \quad \text{--- } \textcircled{X_1}$$
$$c_2 = \sum_{i=1}^3 m_i (x_i)_2 \bar{x}_i = 0 \quad \text{--- } \textcircled{X_2}$$
$$c_1 = 3\bar{x}_1 + 4.979\bar{x}_2 + 3.7738\bar{x}_3$$
$$c_2 = 3\bar{x}_1 + 1.6722\bar{x}_2 - 1.8998\bar{x}_3$$

So, to obtain the third mode. So, we should set the c_1 and c_2 both equal to 0. So, to set c_1 equal to 0 already we have derived this expression that is summation $m_i \ddot{x}_i$ equal to 0. So, for the 3 mode you can write this is equal to $m_1 \ddot{x}_1 + m_2 \ddot{x}_2 + m_3 \ddot{x}_3$ equal to 0. Similarly, for c_2 equal to 0 I can write $m_1 \ddot{x}_2$. So, these \ddot{x} , so this is nothing but the \ddot{x}_2 we obtained just now and this is the \ddot{x}_1 we obtained before. So, this is the \ddot{x}_1 of the first mode that is modal matrix of the first mode this is the modal matrix of the second mode. So, substituting these value. So, we can write the c_1 equal to this that is $3 \ddot{x}_1 + 4.979 \ddot{x}_2$ plus this and c_2 equal to this.

(Refer Slide Time: 31:52)

The image shows a handwritten derivation on a yellow background. At the top, two equations are written: $\bar{x}_1 = 1.5896 \bar{x}_3$ and $\bar{x}_2 = -1.7157 \bar{x}_3$. Below these, a matrix equation is shown:
$$\begin{Bmatrix} \bar{x}_1 \\ \bar{x}_2 \\ \bar{x}_3 \end{Bmatrix} = \begin{bmatrix} 0 & 0 & 1.5896 \\ 0 & 0 & -1.7157 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} \bar{x}_1 \\ \bar{x}_2 \\ \bar{x}_3 \end{Bmatrix}$$
 A bracket under the matrix is labeled S_2 . Below this, the matrix S_2 is explicitly defined as
$$S_2 = \begin{bmatrix} 0 & 0 & 1.5896 \\ 0 & 0 & -1.7157 \\ 0 & 0 & 1 \end{bmatrix}$$

So, after getting these 2 expression I can write \ddot{x}_1 in terms \ddot{x}_3 and \ddot{x}_2 also in terms of \ddot{x}_3 . So, you can write this $\ddot{x}_1 \ddot{x}_2 \ddot{x}_3$ equal to $0 \ 0 \ 1.5896, 0 \ 0$ minus 1.7157 and $0 \ 0 \ 1 \ \ddot{x}_1 \ \ddot{x}_2 \ \ddot{x}_3$. So, you can note that in this matrix the first 2 columns are 0. So, if you multiply this matrix or you take this matrix and insert in the original equation. So, the any vibration or any mode you supply that will eliminate the first 2 modes. So, this matrix is known as the sweeping matrix for the second mode. So, using the sweeping matrix, we can eliminate these first 2 modes now.

(Refer Slide Time: 32:44)

$$\begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix} = \frac{m\omega^2}{4k} \begin{bmatrix} 3 & 2 & 1 \\ 3 & 6 & 3 \\ 3 & 6 & 7 \end{bmatrix} \begin{bmatrix} 0 & 0 & 1.5896 \\ 0 & 0 & -1.7157 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix}$$

$\underbrace{\hspace{10em}}_{a \cdot u} \quad \underbrace{\hspace{10em}}_{S_2}$

$$\begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix} = \frac{m\omega^2}{4k} \begin{bmatrix} 0 & 0 & 2.3374 \\ 0 & 0 & -2.5254 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix}$$

$\underbrace{\hspace{10em}}_{\downarrow \times 3}$

So, now substituting this in this equation, so this is your a into m matrix and this is the sweeping matrix just now we have developed that is S 2. So, I can write m omega square by 4 k into these into this S 2 into this. So, in the left hand side you just see this is x 1 x 2 x 3 equal to m by omega square by 4 k into this matrix into x 1 x 2 x 3. So, in the left hand side, so by taking any assumed value x 1 x 2 x 3 we will determine this x 1 x 2 x 3 the iteration process we will continue till we are not getting the same value in the left hand side and right hand side for x 1 x 2 x 3. So, but for this case immediately you can see that this third column third column will be the x 3.

(Refer Slide Time: 33:41)

$$\begin{Bmatrix} 1.5896 \\ -1.7157 \\ 1 \end{Bmatrix} = 1.4746 \frac{m\omega^2}{4k} \begin{Bmatrix} 1.5896 \\ -1.7157 \\ 1 \end{Bmatrix}$$

$$x_3 = \begin{Bmatrix} 1.5896 \\ -1.7157 \\ 1 \end{Bmatrix}$$

$$1.4746 \frac{m\omega^2}{4k} \begin{Bmatrix} 1.5896 \\ -1.7157 \\ 1 \end{Bmatrix} = 1$$

$1.4746 \frac{m\omega^2}{4k} = 1$

So, if you substitute this third column in this expression you can check and normalize that thing. You can find that the value x_3 equal to 1.5896 minus 1.7157 and 1. So, substituting this x_3 value you can find this 1.4746 m omega square by 4 k. So, this part is not there. So, this is equal to 1. So, 1.47, so you can get 1.4 1.4746 m omega square by 4 k equal to 1.

(Refer Slide Time: 34:20)

$$\omega_1 = 0.5834\sqrt{\frac{k}{m}} \quad \omega_2 = 1.2\sqrt{\frac{k}{m}} \quad \omega_3 = 1.6472\sqrt{\frac{k}{m}}$$

$$X_1 = \begin{Bmatrix} 1 \\ 2.4895 \\ 3.7738 \end{Bmatrix} \quad X_2 = \begin{Bmatrix} 1.0 \\ 0.8361 \\ -1.8988 \end{Bmatrix} \quad X_3 = \begin{Bmatrix} 1.5896 \\ -1.7157 \\ 1 \end{Bmatrix}$$

So, from this you can get this omega square. So, that correspond to the third mode frequency. So, in this way you can obtain the third mode frequency of the system. So, by using this matrix iteration method in this way you can determine the mode shapes of different modes, and the Eigen values or the frequency of the system.

(Refer Slide Time: 34:42)

Equation of Motion


$$m \begin{bmatrix} 3 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \\ \ddot{x}_3 \end{Bmatrix} + k \begin{bmatrix} 6 & -2 & 0 \\ -2 & 3 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$
$$A = M^{-1}K = \begin{bmatrix} 2 & -0.667 & 0 \\ -1 & 1.5 & -0.5 \\ 0 & -1 & 1 \end{bmatrix}$$
$$\lambda_1 = 0.3403 \quad \lambda_2 = 0.1441 \quad \lambda_3 = 2.7186$$

So, let us now. So, same thing you can verify by writing the equation motion. The equation motion can be written this is the mass matrix, and inverse of the flexibility matrix will give the stiffness matrix and this is equal to 0. So, now you can find the dynamic matrix that is m inverse k. So, this is the dynamic matrix by using the Eigen value of this you can find the values are coming to be this. So, you can check that these values and the values you obtained are same.

(Refer Slide Time: 35:14)

Dunkerley's Method

- lower bound to the fundamental frequency

$$\frac{1}{\omega_n^2} = \frac{1}{\omega_{n_1}^2} + \frac{1}{\omega_{n_2}^2} + \dots + \frac{1}{\omega_{n_r}^2}$$


The diagram shows a horizontal beam supported at both ends. Three vertical downward arrows represent loads at different points along the beam, labeled y1, y2, and y3 from left to right.

So, let us now see some other methods that is 1 methods is Dunkerley method. In Dunkerley method you can find the natural frequency of a system when several masses are acting on the system. So, let this is the, this is a simply supported beam. So, on this beam let n number of loads are acting. So, these are the n number of loads acting on this beam. So, due to this let this is the static deflection. So, we can find let this is equal to y_1 this y_2 and similarly this is equal to y_n . So, let these are the deflection due to the load acting at these positions. So, Dunkerley method give an approximate method or give an approximation to find the fundamental frequency of the system.

So, in this case the frequency of the system that is equal to ω_n you can find by finding the natural frequency due to each load. So, first we have to find the natural frequency due to each load and after getting the natural frequency from each for each case, we can use this formula to find the natural frequency of the whole system. So, here $\frac{1}{\omega_n^2}$ by ω_n square, so ω_n is the natural frequency of the system. So, ω_n is the natural ω_n is the frequency of the system when only this load is acting on the system. Similarly, $\omega_n 2$ is the frequency when only this frequency or only this load is acting on the system.

Similarly, $\omega_n 5$ will be when this fifth load is acting only if you find the frequency of the system, then that will be ω_n and ω_S the last 1 is the ω_S . So, ω_S is the natural frequency of the system when no load is acting on the system, but the mass of the system is considered that is a uniformly distributed load of this beam is considered. So, the total, so $\frac{1}{\omega_n^2}$ equal to $\frac{1}{\omega_n^2}$ plus $\frac{1}{\omega_n 1^2}$ plus $\frac{1}{\omega_n 2^2}$. And similarly, you can add till the last mass is considered, so by taking all the frequencies. So, you can use this formula to find the natural frequency of the, of a complex system.

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$\omega_n^2 = \frac{g}{\delta}$

$a_{11} = \frac{3l^3}{256EI}$

$a_{22} = \frac{l^3}{48EI}$

$a_{33} = \frac{3l^3}{256EI}$

$m_1 = m_3 = m$ and $m_2 = 2m$

Influence Coefficients a_{ij} :
 deflection at station i due to unit load at station j

So, let us take 1 example. So, let this is a simply supported beam. So, in this simply supported beam we have taken 3 masses. So, this middle mass we have taken equal to 2 m and this side mass equal to m 1 and this mass equal to m. So, this is m 1. So, we have taken this and first mass and last mass same and the second mass we have taken equal to twice of the first mass. So, in this case we have to find the natural frequency of the system we have not considered the mass of the beam, so without considering the mass of the beam. So, we here we can find the natural frequency or the frequency due to this mass frequency due to this mass and frequency due to this mass individually.

And then we can use this formula $1 \text{ by } \omega m \text{ square equal to } 1 \text{ by } \omega_1^2 \text{ plus } 1 \text{ by } \omega_2^2 \text{ and plus } 1 \text{ by } \omega_3^2$ and we can find the natural frequency. So, to find that thing let us first find the flexibility influence coefficient that is the displacement at these points. After getting these displacement, we can find the natural frequency as ω_n will be equal to root over g by δ . So, first we can find these displacement due to these mass at this position. So, a_{11} is the displacement at 1 due to a unit force at 1. So, when unit force is applied at these position. So, we can find the displacement will be equal to $3l^3$ by $256EI$. Similarly, we can find when only this mass is present. So, if you put a unit a unit force here the displacement will be l^3 by $48EI$. So, you know for a simply supported beam at the.

So, when it is loaded at the middle the displacement equal to $w l^3$ by $48 EI$. So, now by putting this weight equal to w . So, at the middle when it is loaded at the middle the displacement equal to $w l^3$ by $48 EI$. So, now by putting w equal to 1 . So, you are getting this is equal to l^3 by $48 EI$. Similarly, a $3/16$ you can find. So, a $1/16$ a $2/16$ a $3/16$ that is displacement at 1 due to unit displacement at 1 due to unit force at 1 displacement at 2 due to unit force at 2 and displacement at 3 due to unit force at 3 . So, in this way you can obtain.

(Refer Slide Time: 40:30)

$$\frac{1}{\omega_1^2} = \frac{3ml^3}{256EI} \quad \frac{1}{\omega_2^2} = \frac{2ml^3}{48EI} \quad \frac{1}{\omega_3^2} = \frac{3ml^3}{256EI}$$

$$\frac{1}{\omega_n^2} = \frac{3ml^3}{256EI} + \frac{2ml^3}{48EI} + \frac{3ml^3}{256EI}$$

$$= \frac{(3+10.66+3)ml^3}{256EI}$$

So, after obtaining this thing, so you can write 1 by ω_1 square will be equal to $3ml^3$ by $256EI$. So, root over g by $\Delta \omega$. So, 1 by ω_1 square will be equal to $3ml^3$ by $256EI$. So, you can divide these things, so g by this. So, you can find the terms like this. So, $2ml^3$ by this and this is equal to 1 by ω_2 square equal to $2ml^3$ by $48EI$ and 1 by ω_3 square equal to $3ml^3$ by $256EI$. So, this is equal to g . So, we have put only these. So, in this case we have just divided by g as this weight 1 Newton was taken. So, this is equal to mass of 1 into g .

So, this will be mg for a load of m . So, the displacement will be equal to a load of w the displacement will be equal to $w l^3$ by $256EI$. So, by dividing this by g as this ω_n equal to or ω_n square equal to g by Δ you can find. So, this will give g by Δ . So, Δ equal to here $3ml^3$ by $256EI$. So, g by Δ will give w equal to mg . So, substituting w equal to mg you can find this expression. So, now 1 by $E \omega_n$

square will be equal to, so by adding this 1 by omega square. So, adding this term and this and this. So, you can find this 1 by omega n square equal to this.

(Refer Slide Time: 42:14)

$$\omega_n^2 = \frac{256}{16.66} \frac{EI}{ml^3} = 15.36 \frac{EI}{ml^3}$$

$$\omega_n = 3.9191 \sqrt{\frac{EI}{ml^3}} \quad \checkmark$$

$$\omega_n^2 = 15.36 \frac{EI}{ml^3} \quad \text{Dunkerly}$$

$$= \frac{16.199EI}{ml^3} \quad \text{Exact}$$

And then omega n equal to 3.9191 root over EI by ml cube. So, in this way you can find the frequency of the system by using Dunkerley method. So, in Dunkerley method you are getting this omega n square equal to 15.36 EI by ml cube and in the exact method you can find the, it equal to 16.199 EI by this.

(Refer Slide Time: 42:44)

Rayleigh Method

$$a_{11} = \frac{9l^3}{768EI} \quad a_{12} = \frac{11l^3}{768EI} \quad a_{13} = \frac{7l^3}{768EI}$$

$$a_{33} = \frac{9l^3}{768EI} \quad a_{22} = \frac{16l^3}{768EI} \quad a_{23} = \frac{11l^3}{768EI}$$

So, using this Rayleigh method, so already you studied this Rayleigh method. So, in this Rayleigh if you apply this system to this Rayleigh method we can find all the influence coefficients. So, a 1 1 a 1 2 a 1 3 a 3 3 a 2 2 and a 2 2 you can find and applying this reciprocity theorem we can find the other influence coefficients.

(Refer Slide Time: 43:09)

Static deflections

$$X_1 = m_1 g a_{11} + m_2 g a_{12} + m_3 g a_{13}$$

$$X_2 = m_1 g a_{21} + m_2 g a_{22} + m_3 g a_{23}$$

$$X_3 = m_1 g a_{31} + m_2 g a_{32} + m_3 g a_{33}$$

$$m_1 = m_3 = m \text{ and } m_2 = 2m$$

So, by the using this influence coefficient displacement at this 3 position we can find displacement at 1 equal to force at 1 into a 1 1 displacement at 2 into a 1 2 force at 2 into a 1 2. And force at 3 into a 1 3 in this way we can find displacement at 1 and 2 and 3.

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$$X_1 = \frac{38ml^3 g}{768EI} \quad X_2 = \frac{54ml^3 g}{768EI} \quad X_3 = \frac{38ml^3 g}{768EI}$$

$$\Rightarrow \omega_n^2 = \frac{g \sum_{i=1}^n m_i X_i}{\sum_{i=1}^n m_i X_i^2} = \frac{16.2055EI}{ml^3}$$

$$\omega_n^2 = 15.36 \frac{EI}{ml^3} \text{ Dunkerly}$$

$$= \frac{16.199EI}{ml^3} \text{ Exact}$$

And these displacements can be written equal to x_1 will be equal to $38 \text{ ml cube by } g / 768 EI \times 2$ similarly equal to $54 \text{ ml cube } / 768 EI$ and x_3 equal to $38 \text{ ml cube by } \text{ml cube } g$ by this. So, by using this Rayleigh methods, so in case of Rayleigh method this ω_n^2 square equal to g into summation $m_i X_i$ by summation $m_i X_i^2$ square. So, this part upper part is the potential energy and lower part is the kinetic energy. So, this division gives ω_n^2 square equal to this. So, by using Rayleigh method you are getting this value equal to this $15.16.2055$. So, slightly higher than the exact value, exact value equal to 16.199 and here you are getting a value equal to 16.2055 and in case of Dunkerly method you are getting a value 15.36 . So, Dunkerly method gives a lower approximation than the exact value and the Rayleigh method gives a value which is slightly higher than the exact value.

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Rayleigh-Ritz Method

- This is an extension of Rayleigh's method.
- A single assumed functions is used Rayleigh's method.
- Here a closer approximation to the natural mode is obtained by superposing a number of assumed functions.

$$w(x) = c_1 w_1(x) + c_2 w_2(x) + \dots + c_n w_n(x)$$

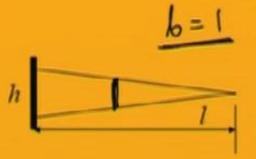
- Then the Rayleigh quotient is obtained which is a function of the coefficients c_1, c_2, \dots, c_n

Now, let us see the Rayleigh-Ritz method, so in case of Rayleigh-Ritz method. So, this is an extension of the Rayleigh method. So, in Rayleigh method we are taking 1 approximate function. So, in Rayleigh-Ritz method we will take several approximate method and superpose these approximate functions. So, let us take this approximation approximate function as $w(x)$. So, $w(x)$ will be equal to $c_1 w_1(x) + c_2 w_2(x) + \dots + c_n w_n(x)$. So, where w_1, w_2, w_n are the assumed function we have taken. So, the resulting function w equal to $c_1 w_1 + c_2 w_2 + c_3 w_3$ then we can find the Rayleigh quotient. And as you know the Rayleigh quotient has a stationary value near the normal modes. So, we can differentiate these Rayleigh quotient with respect to these coefficient

c_1 , c_2 and c_n to get a set of algebraic equations and by solving these equations, we can find the frequencies of the system.

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Example



$$w_1(x) = \left(1 - \frac{x}{l}\right)^2$$

$$w_2(x) = \frac{x}{l} \left(1 - \frac{x}{l}\right)^2$$

$$w(x) = c_1 \left(1 - \frac{x}{l}\right)^2 + c_2 \frac{x}{l} \left(1 - \frac{x}{l}\right)^2$$

$$A(x) = \frac{hx}{l}$$

$$I(x) = \frac{1}{12} \left(\frac{hx}{l}\right)^3$$

So, let us take 1 example, so with this help of this example. I will explain you how to find the frequency of this system. So, here we have taken I have taken 3 function. So, 2 functions let I have taken. So, $w_1(x)$, so this is a. So, I have to find the frequency of this tapered beam. So, in this tapered beam this is a tapered cantilever beam. So, the left end is fixed left end has a height of h and right end this is 0 and the length is l and let me take this width b equal to 1 . So, when I am taking this width b equal to 1 . So, I will have these area $A(x)$ equal to $h \cdot x$ by l , so at any section. So, the area will be equal to. So, this height will be equal to. So, as for this total height this is l and this is h . So, at any section the height will be h by l into x . So, the area will be h by l into x into 1 . So, that is equal to $h \cdot x$ by l . Similarly, this $I(x)$ that is the moment of inertia will be equal to $\frac{1}{12} h \cdot x$ by l cube. So, taking this area and this moment of inertia and taking these 2 function. So, I have to find the natural frequency or fundamental frequency or the frequencies of these tapered beam. So, first let me take only 1 function and apply the Rayleigh method.

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Taking only one mode (Rayleigh method)

$$R = \omega^2 = \frac{\int_0^l EI(x) \left(\frac{d^2 w(x)}{dx^2} \right)^2 dx}{\int_0^l \rho A(x) (w(x))^2 dx} = 2.5 \frac{Eh^2}{\rho l^4}$$

$\omega_1 \cong 1.5811 \left(\frac{Eh^2}{\rho l^4} \right)^{1/2}$

The exact value is

$$\omega_1 = 1.5343 \left(\frac{Eh^2}{\rho l^4} \right)^{1/2}$$

Rayleigh method yield 3.05% higher value than the exact

So, by taking this first only 1 function and applying this Rayleigh function that is Rayleigh method is. So, this is the potential energy and this is the kinetic energy of the system. So, by divided these potential energy by this kinetic energy. So, you are getting this is equal to 2.5 Eh square by rho one-fourth or this omega 1 equal to 1.5811 into Eh square by rho one-fourth to the power half you can note that the exact value equal to 1.5343. So, you just see that the Rayleigh method gives 3 percent higher value than this exact value.

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Taking two approximate modes

$$R = \omega^2 = \frac{\int_0^l EI(x) \left(\frac{d^2 w(x)}{dx^2} \right)^2 dx}{\int_0^l \rho A(x) (w(x))^2 dx} = \frac{X}{Y}$$

$$X = \frac{Eh^3}{3l^3} \left(\frac{c_1^2}{4} + \frac{c_2^2}{10} + \frac{c_1 c_2}{5} \right)$$

$$Y = \rho h l \left(\frac{c_1^2}{30} + \frac{c_2^2}{280} + \frac{2c_1 c_2}{105} \right)$$

Now, to apply this Rayleigh-Ritz method we will take that is 2 function that is w_1 and w_2 and we can find this Rayleigh quotient. So, in this case the approximate function I am taking it equal to c_1 into w_1 plus c_2 into w_2 . So, by taking this approximate function like this, so I can find this Rayleigh quotient R equal to ω^2 equal to integration this is the potential energy term $EI \frac{d^4 w}{dx^4}$ whole square dx and integration ρa . So, this is mass into w whole square dx . So, this is this term is coming from the kinetic energy and this term is coming from the strain energy. So, the Rayleigh quotient I can write in this way.

So, the upper part let me write equal to x and lower part equal to y . So, if I will substitute these function and, I will find this x equal to Eh^3 by $3 I^2$ into c_1^2 by 4 plus c_2^2 by 10 plus $c_1 c_2$ by 5 and this y equal to $\rho a h I$ into c_1^2 by 30 plus c_2^2 by E plus $2 c_1 c_2$ by 10 . So, in this way I can obtain this x and y . So, you just note that this x and y are function of c_1 and c_2 which are not known to us till now. So, now as you know this Rayleigh quotient has a stationary value near the normal modes. So, I can differentiate these R with respect to c_1 and c_2 and get the get some expressions.

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The condition that makes R stationary are

$$\frac{\partial R}{\partial c_1} = \frac{\partial(\omega^2)}{\partial c_1} = \frac{Y \frac{\partial X}{\partial c_1} - X \frac{\partial Y}{\partial c_1}}{Y^2} = 0$$

$$\frac{\partial R}{\partial c_2} = \frac{\partial(\omega^2)}{\partial c_2} = \frac{Y \frac{\partial X}{\partial c_2} - X \frac{\partial Y}{\partial c_2}}{Y^2} = 0$$

So, by differentiating with respect to c_1 I can write the equal to. So, as I have written these equal to x by y . So, this can be written as y into $\frac{\partial x}{\partial c_1}$ minus x into $\frac{\partial y}{\partial c_1}$ by y^2 . So, I have to set it equal to 0. Similarly, as it is stationary at that

point. So, this is this value differentiation equal to 0 similarly del y del R by del c 2. So, this is del R by del c 2. So, this is del R by del c 1. So, this is equal to y into del x by del c 2 minus x into del y by del c 2 by y square I have to set it equal to 0.

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The slide displays a matrix equation and a formula for lambda. The matrix equation is:

$$\begin{bmatrix} \left(\frac{1}{2} - \frac{1}{15}\lambda\right) & \left(\frac{1}{5} - \frac{2}{105}\lambda\right) \\ \left(\frac{1}{5} - \frac{2}{105}\lambda\right) & \left(\frac{1}{5} - \frac{1}{140}\lambda\right) \end{bmatrix} \begin{Bmatrix} c_1 \\ c_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

Below the matrix equation, the formula for lambda is given as:

$$\lambda = \frac{3\omega^2 \rho l^4}{Eh^2}$$

By setting these equal to 0. So, I have this expression. So, I can write this. So, the first 1 yield, so by if you do this first 1. So, you you are getting this half minus 1 by 15 lambda. So, where lambda equal to 3 omega square rho one-fourth by Eh square, so 1 half minus 1 by 15 lambda c 1 plus 1 by 5 minus 2 by 1 0 5 lambda into c 2 equal to 0. So, this is the first expression and the second expression comes to be 1 by 5 minus 2 by 1 0 5 lambda into c 1 plus 1 by 5 minus 1 by 40 lambda c 2 equal to 0. So, in this way you can find this and now as c 1 c 2 are not equal to 0 both c 1 and c 2 are not equal to 0. So, you can find this determinant and make it equal to 0 to find this value of lambda.

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$$\frac{1}{8820}\lambda^2 - \frac{13}{1400}\lambda + \frac{3}{50} = 0$$
$$\omega_1 = 1.5367 \left(\frac{Eh^2}{\rho l^4} \right)^{1/2}$$
$$\omega_2 = 4.9936 \left(\frac{Eh^2}{\rho l^4} \right)^{1/2}$$

So, this lambda is a function of omega square. So, you can find. So, this expression you are getting. So, 1 by 8820 lambda square minus 13 by 1400 lambda plus 3 by 50 equal to 0. So, by solving this equation you can get the value true value of lambda as this is a quadratic equation. So, by solving this, so you are getting this omega 1 equal to 1.5367 Eh square by rho one-fourth to the power half, and omega 2 equal to 1 omega 2 equal to 4.5936 Eh square by rho one-fourth. So, in this case you just check that this value is closer to the exact value. So, by taking these 2 modes instead of the 1 as in case of Rayleigh method you are getting 2 modes or you are getting true frequency and also you are getting a closer approximation to the exact value.

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The Galerkin's Method ✓

In Galerkin's method the residue obtained by using the assumed mode in the governing differential equation is minimized.

$$\varphi(x) = \sum_{i=1}^n c_i \varphi_i(x) \quad \checkmark$$

Transverse Vibration of beam

$$L(x) = \frac{d^4 \varphi(x)}{dx^4} - \frac{m\omega^2}{EI} \varphi(x) = 0 \quad \checkmark$$

Residual of i th mode

$$R_i = \int_0^l L(x) \varphi_i(x) dx = 0$$

In case of Galerkin method, we can take a similar shape functions or mode shape and in this case we have to find the residue and by equating this residue to 0 we can find the frequencies of the system. So, to find the residue we can take the governing equation of the system led for the transfer vibration of the beam. So, you know the governing equation is the Euler-Bernoulli equation. So, in that Euler-Bernoulli equation if I will I will write this W equal to.

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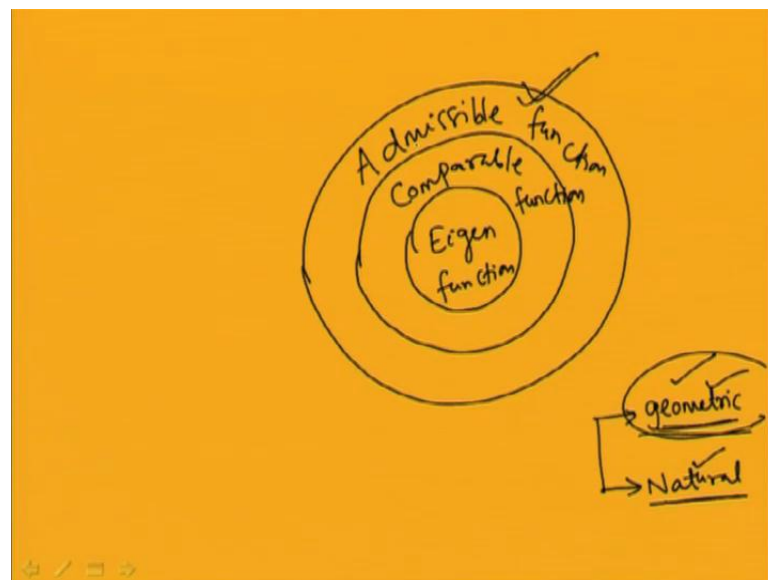
$$EI \frac{\partial^4 w}{\partial x^4} + m \frac{\partial^2 w}{\partial t^2} = 0$$

$$w = \phi(x) \eta(t)$$

So, this Euler-Bernoulli equation you know it can be written in this form that is $\frac{d^4 w}{dx^4} = \frac{m}{EI} \frac{d^2 w}{dt^2}$ plus. So, this is $m \frac{d^2 w}{dt^2} = EI \frac{d^4 w}{dx^4}$ equal to 0. So, this is the expression for the Euler-Bernoulli beam, so in this the expression for the Euler-Bernoulli beam. So, if you substitute this w equal to $\phi(x)$ and qt . So, you can find the expression in terms of only x . So, that expression you can take as the governing equation for this Galerkin method.

So, this is the differential equation you can take which will satisfy the boundary conditions all boundary conditions to get the Eigen function of the system. So, in this case this Lx , I am writing this as the function. So, this Lx equal to $\frac{d^4 \phi(x)}{dx^4} - \frac{m \omega^2}{EI} \phi(x) = 0$. So, this ϕ when it is Eigen function of the system that is when it satisfy both the boundary conditions and the governing equation governing differential equation. So, that function is known as the Eigen function. So, when you are substituting this Eigen function the right hand side will be equal to 0, but instead of taking this Eigen function if you take some other function then it may not be equal to 0.

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There are 2 other different types of functions available. So, 1 function is the Eigen function. So, this Eigen function. So, Eigen function satisfy both differential equation and all boundary conditions. So, you know all boundary condition include. So, you have

2 different types of boundary condition, 1 is natural boundary condition and other is the geometric boundary condition. So, in case of geometric boundary condition it satisfies the satisfy the deflection and slope boundary conditions and in case of natural boundary condition or the force boundary condition, it satisfies the force boundary condition of the system. So, when the function what you are taking satisfy both geometric and natural boundary condition of the system that is all the boundary conditions of the system and also the differential equation. So, that function is known as the Eigen function.

Now, you can take another set of function which will not satisfy the differential equation, but it will satisfy all the boundary conditions. So, that function you can tell as comparable function. So, comparable function, so this comparable function does not satisfy the differential equation or governing equation of the system, but it satisfy all the boundary conditions of the system. So, it satisfy both geometric and natural boundary conditions of the system. And the third set of function, which you can take, which satisfy only the geometric boundary condition of the system. Because getting a function which will satisfy the geometric and natural boundary condition or force boundary condition is very difficult. But you can always get a function which will satisfy the geometric boundary condition of the system.

So, the function which satisfy only geometric boundary condition of the system is known as the admissible function, so admissible function. So, in case of Galerkin method you can take this admissible function or in case of Rayleigh's method also you can take this admissible functions. So, by taking these admissible function now you can substitute in the expression and if you substitute it in this expression than the right hand side will not be equal to 0, so as it not equal to 0. So, there will be some residue in that expression. So, the residue over whole of the domain can be found by integrating that thing. So, residue due to the first function can be written as $L \int_0^L \phi(x) dx$.

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•Torsional vibration of Rod
•Longitudinal vibration of rod
•Lateral vibration of taut string

From Wave equation

$$L(x) = \frac{d^2 \phi(x)}{dx^2} + \left(\frac{\omega}{c}\right)^2 \phi(x) = 0$$

Residual of i th mode

$$R_i = \int_0^l L(x) \phi_i(x) dx = 0$$

Similarly, for case of the torsional vibration of rod, longitudinal vibration rod and lateral vibration of taut string you know the governing equation is the wave equation and from that equation. You can write this Lx equal to $\frac{d^2 \phi(x)}{dx^2} + \left(\frac{\omega}{c}\right)^2 \phi(x) = 0$. So, the residue in the i th mode you can find by integrating this Lx into $\phi_i(x) dx$. So, in this way by applying this approximate method you can find the frequencies for discrete and distributed mass systems.