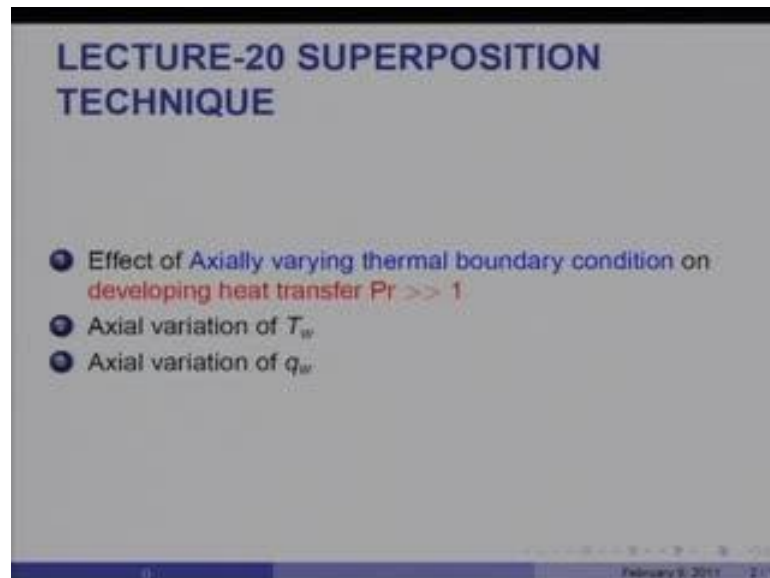


Convective Heat and Mass Transfer
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Module No. # 01
Lecture No. # 20
Superposition Technique

Today, I will take up the last lecture on laminar flow heat transfer in ducts. What I am going to consider is, the effect of actually varying thermal boundary condition. This is particularly of interest in oil flows, where the Prandtl number is very large and the thermal development length is considerably bigger than the velocity development length.

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In this sort of situations, actually varying wall temperature as well as, actually varying heat flux has profound effect on the manner in which the heat transfer coefficient changes or varies with the actual distance.

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Axial Variation of T_w - L20($\frac{1}{12}$)

Our thermal entry length solution for $Pr \gg 1$ may be viewed as solution to a step-function $(T_w - T_i)$ at $x = x_0$. Thus, for $x \geq x_0$

$$T - T_i = [1 - \theta(x^* - x_0^*, y^*)] (T_w - T_i) \quad \rightarrow \quad \theta = \frac{T - T_w}{T_i - T_w}$$


Therefore, for arbitrary variation of T_w , we have

$$T - T_i = \int_0^{x^*} [1 - \theta(x^* - x_0^*, y^*)] \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} [1 - \theta(x^* - x_{0,k}^*, y^*)] \Delta(T_w - T_i)_k$$

For example, let us consider the solution that we had obtained for Prandtl number very greater than 1. Now that solution was obtained for T equal to T_w from x equal to 0. But, today I am going to consider the case of heat transfer begin from x equal to x_{naught} , before that the wall temperature is same as the inlet fluid temperature.

In fact, we can view our earlier solution of lecture 18 as a step function in $T_w - T_i$; the T_i is inlet temperature at x equal to x_{naught} . Therefore, for x greater than x_{naught} , I may write $T - T_i$ equal to $1 - \theta(x^* - x_{naught}^*, y^*) (T_w - T_i)$; theta, you will recall was defined as $(T - T_w) / (T_i - T_w)$. So, just as we considered the external boundary layer situation; for the internal boundary layer situation, it would again look like this (Refer Slide Time: 02:25).

For arbitrary variation of T_w , we will have $T - T_i$ equal to 0 to $x^* = 1 - \theta(x^* - x_{naught}^*, y^*) dT_w / dx_{naught}^*$; this would be for the continuous part of the variation. This would be $1 - \theta(x^* - x_{naught}^*, y^*) \Delta(T_w - T_i)_k$; this would be for the discontinuous part of the solution. This is exactly following what we did in the case of external boundary layers (Refer Slide Time: 02:45).

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Further Development - 1 - L20($\frac{2}{12}$)

Therefore, the wall heat flux is evaluated as

$$q_w(x^*) = k \frac{\partial T}{\partial y} |_{y=b} = \frac{k}{b} \frac{\partial T}{\partial y^*} |_{y^*=1}$$

$$= -\frac{k}{b} \left[\int_0^{x^*} \theta'(x^* - x_0^*, 1) \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} \theta'(x^* - x_{0,k}^*, 1) \Delta(T_w - T_i)_k \right]$$

But, we know that (see lecture 19)


$$\theta'(1) = - \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \rightarrow A_n = -C_n Y_n'(1)$$

The heat flux response would be q_w star, now I am considering the case of flow between parallel plates separated by distance to b . Then, that would be equal to q_w star, would be equal to $k \frac{dT}{dy}$ at the wall y equal to b equal to k by $b \frac{dT}{dy}$ star, y star equal to 1. That would read as θ dash, here as $d\theta$ by dy , x star x naught star 1- where y star is equal to 1. This will be for the continuous part; this would be for the discontinuous part (Refer Slide Time: 03:40).

(Refer Slide Time: 03:49)

Axial Variation of T_w - L20($\frac{1}{12}$)

Our thermal entry length solution for $Pr \gg 1$ may be viewed as solution to a **step-function** ($T_w - T_i$) at $x = x_0$. Thus, for $x \geq x_0$

$$T - T_i = [1 - \theta(x^* - x_0^*, y^*)] (T_w - T_i) \rightarrow \theta = \frac{T - T_w}{T_i - T_w}$$


The diagram shows a duct of length x^* and half-height y^* . A step change in wall temperature occurs at x_0^* . The wall temperature is T_w for $x^* < x_0^*$ and T_i for $x^* > x_0^*$. The fluid temperature is T and the inlet temperature is T_i .

Therefore, for **arbitrary variation of T_w** , we have

$$T - T_i = \int_0^{x^*} [1 - \theta(x^* - x_0^*, y^*)] \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} [1 - \theta(x^* - x_{0,k}^*, y^*)] \Delta(T_w - T_i)_k$$

(Refer Slide Time: 03:55)

Further Development - 1 - L20($\frac{2}{12}$)

Therefore, the wall heat flux is evaluated as

$$q_w(x^*) = k \frac{\partial T}{\partial y} \Big|_{y=b} = \frac{k}{b} \frac{\partial T}{\partial y^*} \Big|_{y^*=1}$$

$$= -\frac{k}{b} \left[\int_0^{x^*} \theta'(x^* - x_0^*, 1) \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} \theta'(x^* - x_{0,k}^*, 1) \Delta(T_w - T_i)_k \right]$$

But, we know that (see lecture 19)

$$\theta'(1) = - \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \rightarrow A_n = -C_n Y_n'(1)$$

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This is simply a differentiation of this equation with respect to y . But, you will recall that from lecture 19, we had found that $\theta' (1)$ is actually equal to minus sum n equal to 0 to infinity $A_n \exp(-\frac{8}{3} \lambda_n^2 x^*)$ equal to A_n equal to minus $C_n Y_n'(1)$, where C_n and $Y_n'(1)$ were tabulated in lecture 19. So, we are going to use the same coefficients A_n in this series; λ_n values 2 were tabulated in lecture 19. If I substitute that here, for that I would get the heat flux response to arbitrary variation of T_w along the wall.

(Refer Slide Time: 04:35)

Further Development - 2 - L20($\frac{3}{12}$)

Therefore, substitution gives

$$q_w(x^*) = \frac{k}{b} \left[\int_0^{x^*} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_0^*)\right) \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_{0,k}^*)\right) \Delta(T_w - T_i)_k \right]$$

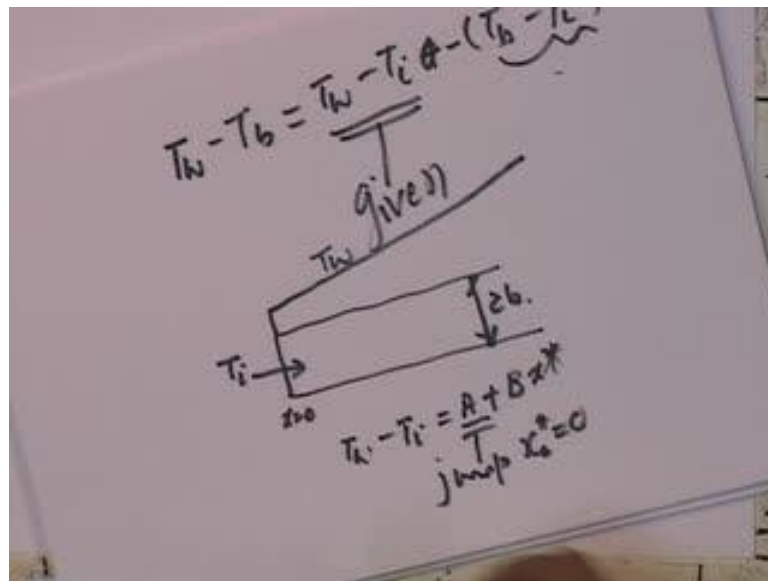
$$T_b - T_i = \frac{4b}{k} \int_0^{x^*} q_w(x^*) dx^*$$

$$Nu_{x^*} = \frac{h_x(4b)}{k} = \frac{q_w(x^*)}{(T_w - T_b)_{x^*}} \times \frac{4b}{k}$$

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So, $q_w \times \Delta x$ would become $k \times b \int_0^{\Delta x} \frac{dT}{dx}$ (Refer Slide Time: 04:45). This would be for the discontinuous part; the summation goes from $n=0$ to infinity over n for the number of steps, a k equal to 1 to NK . Now that $T_{\text{bulk}} - T_i$ would obviously be equal to $4b \int_0^{\Delta x} q_w \times dx$ and this is simply from the heat balance.

(Refer Slide Time: 05:26)



Therefore, the Nusselt number would be given by $q_w \times \Delta x$ divided by $T_w - T_{\text{bulk}}$ by k , where $T_w - T_{\text{bulk}}$ is written as $T_w - T_i - T_{\text{bulk}} - T_i$. This is given and this is what we have evaluated just now (Refer Slide Time: 05:52). $T_w - T_{\text{bulk}}$ is written in the fashion I have shown here, so that is how one gets $T_w - T_{\text{bulk}}$ at any x .

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Further Development - 2 - L20($\frac{3}{12}$)

Therefore, substitution gives

$$q_w(x^*) = \frac{k}{b} \left[\int_0^{x^*} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_0^*)\right) \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_{0,k}^*)\right) \Delta(T_w - T_i)_k \right]$$

$$T_b - T_i = \frac{4b}{k} \int_0^{x^*} q_w(x^*) dx^*$$

$$Nu_{x^*} = \frac{h_x(4b)}{k} = \frac{q_w(x^*)}{(T_w - T_b)_{x^*}} \times \frac{4b}{k}$$

Let us move on to a problem in which I am going to consider, let us say there are two parallel plates separated by distance $2b$ and fluid enters at T_i . At x equal to 0 itself there is a jump and then the temperature is linear. So, $T_w - T_i$, this is T_w equal to A plus Bx^* . A is the jump at x naught star equal to 0 in this case.

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A Problem - L20($\frac{4}{12}$)

Let $T_w - T_i = (A + Bx^*) \rightarrow dT_w/dx_0^* = B$. Then

$$q_w(x^*) = \frac{k}{b} \left[\frac{3B}{8} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^2} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + A \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] \text{ (note } x_0 = 0)$$

$$T_w - T_b = \frac{9B}{16} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^4} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + \frac{3A}{2} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right)$$

$$Nu_{x^*} = \frac{q_w 4b}{k(T_w - T_b)} \text{ as } x \rightarrow \infty, Nu_x = \frac{8 \sum_{n=0}^{\infty} A_n / \lambda_n^2}{3 \sum_{n=0}^{\infty} A_n / \lambda_n^4} = 8.23$$

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Further Development - 2 - L20($\frac{3}{12}$)

Therefore, substitution gives

$$q_w(x^*) = \frac{k}{b} \left[\int_0^{x^*} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_0^*)\right) \frac{dT_w}{dx_0^*} dx_0^* \right. \\ \left. + \sum_{k=1}^{NK} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_{0,k}^*)\right) \Delta(T_w - T_i) \right]$$

$$T_b - T_i = \frac{4b}{k} \int_0^{x^*} q_w(x^*) dx^*$$

$$Nu_{x^*} = \frac{h_x(4b)}{k} = \frac{q_w(x^*)}{(T_w - T_b)_{x^*}} \times \frac{4b}{k}$$

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A Problem - L20($\frac{4}{12}$)

Let $T_w - T_i = (A + Bx^*) - dT_w/dx_0^* = B$. Then

$$q_w(x^*) = \frac{k}{b} \left[\frac{3B}{8} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^2} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} \right. \\ \left. + A \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] \quad (\text{note } x_0 = 0)$$

$$T_w - T_b = \frac{9B}{16} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^4} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} \\ + \frac{3A}{2} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right)$$

$$Nu_{x^*} = \frac{q_w 4b}{k(T_w - T_b)} \quad \text{as } x \rightarrow \infty, Nu_x = \frac{8 \sum_0^{\infty} A_n / \lambda_n^2}{3 \sum_0^{\infty} A_n / \lambda_n^4} = 8.23$$

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Further Development - 2 - L20($\frac{3}{12}$)

Therefore, substitution gives

$$q_w(x^*) = \frac{k}{b} \left[\int_0^{x^*} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_0^*)\right) \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_{0,k}^*)\right) \Delta(T_w - T_i)_k \right]$$

$$T_b - T_i = \frac{4b}{k} \int_0^{x^*} q_w(x^*) dx^*$$

$$Nu_{x^*} = \frac{h_x(4b)}{k} = \frac{q_w(x^*)}{(T_w - T_b)_{x^*}} \times \frac{4b}{k}$$

(Refer Slide Time: 07:25)

A Problem - L20($\frac{4}{12}$)

Let $T_w - T_i = (A + Bx^*) \rightarrow dT_w/dx_0^* = B$. Then

$$q_w(x^*) = \frac{k}{b} \left[\frac{3B}{8} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^2} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + A \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] \text{ (note } x_0 = 0)$$

$$T_w - T_b = \frac{9B}{16} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^4} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + \frac{3A}{2} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right)$$

$$Nu_{x^*} = \frac{q_w 4b}{k(T_w - T_b)} \text{ as } x \rightarrow \infty, Nu_x = \frac{8 \sum_{n=0}^{\infty} A_n / \lambda_n^2}{3 \sum_{n=0}^{\infty} A_n / \lambda_n^4} = 8.23$$

Let us see how the equations develop. If I have to substitute this variation of T w minus T i in the previous expression here, so dT w by dx naught star would be simply capital B; that is what this would be. The integration of that with b here, so b is simply is a constant and it would come out. The integration of this equation with x naught equal to 0 would be given by A n by lambda n square 1 minus exponential of minus 8 by 3 lambda n square x square, plus we must add the step jump right at x naught equal to 0, which is the entrance that will be plus A times n minus 0 equal to infinity A n exponential of minus n

$\lambda_n^2 x^*$ (Refer Slide Time: 07:20). There are no other step jumps in this. Therefore, this one is a continuously right along and there are no further step jumps, so simple case.

(Refer Slide Time: 08:12)

Further Development - 2 - L20($\frac{3}{12}$)

Therefore, substitution gives

$$q_w(x^*) = \frac{k}{b} \left[\int_0^{x^*} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_0^*)\right) \frac{dT_w}{dx_0^*} dx_0^* + \sum_{k=1}^{NK} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 (x^* - x_{0,k}^*)\right) \Delta(T_w - T_i)_k \right]$$

$$T_b - T_i = \frac{4b}{k} \int_0^{x^*} q_w(\bar{x}^*) dx^*$$

$$Nu_{x^*} = \frac{h_x(4b)}{k} = \frac{q_w(x^*)}{(T_w - T_b)_{x^*}} \times \frac{4b}{k}$$

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Once you obtain q_w in this manner, T_w minus T_b would simply be given by $9B$ by $16 A_n$ by λ_n^4 , you can see that here T_b minus T_i which is an integral of q_w x^* .

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A Problem - L20($\frac{4}{12}$)

Let $T_w - T_i = (A + B x^*) \rightarrow dT_w/dx_0^* = B$. Then

$$q_w(x^*) = \frac{k}{b} \left[\frac{3B}{8} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^2} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + A \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] \text{ (note } x_0 = 0)$$

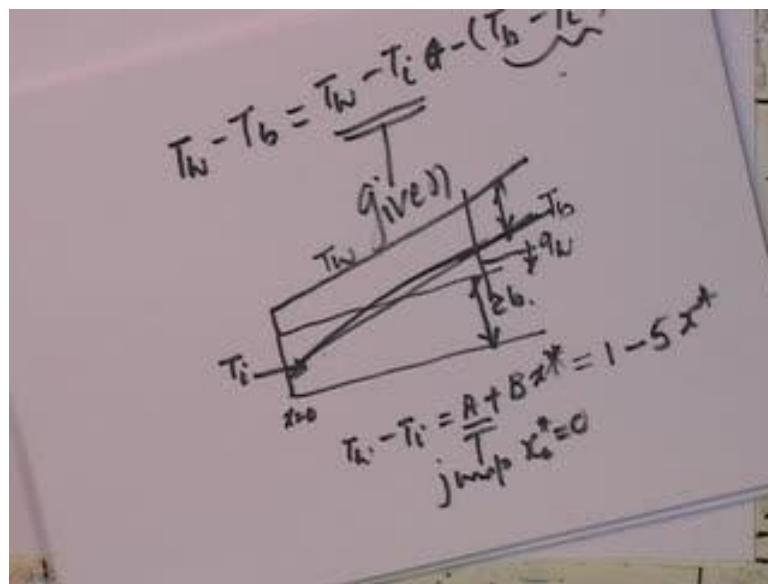
$$T_w - T_b = \frac{9B}{16} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^4} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + \frac{3A}{2} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right)$$

$$Nu_{x^*} = \frac{q_w 4b}{k(T_w - T_b)} \text{ as } x \rightarrow \infty, Nu_x = \frac{8 \sum_{n=0}^{\infty} A_n / \lambda_n^2}{3 \sum_{n=0}^{\infty} A_n / \lambda_n^4} = 8.23$$

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So, I have used that and added T_w minus T_i , then they will get $9B$ by 16 , n equal to 0 to infinity A_n by λ_n^4 and this term plus the step jump part, which is done. As a result, you will get Nu_x is equal to $q_w 4b$ divided by k into T_w minus T_b , but notice one thing that as x tends to infinity you will see that term will go to 0 , that term will go to 0 (Refer Slide Time: 08:45). Therefore, T_w minus T_b would be simply $9B$ by 16 n equal to 0 to infinity A_n by λ_n^4 . Likewise, $q_w x^*$ would be simply the A_n by λ_n^2 square, this term going to 0 and this term equal to 0 .

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(Refer Slide Time: 09:27)

A Problem - L20($\frac{4}{12}$)

Let $T_w - T_i = (A + B x^*) \rightarrow dT_w/dx_0^* = B$. Then

$$q_w(x^*) = \frac{k}{b} \left[\frac{3B}{8} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^2} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + A \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] \quad (\text{note } x_0 = 0)$$

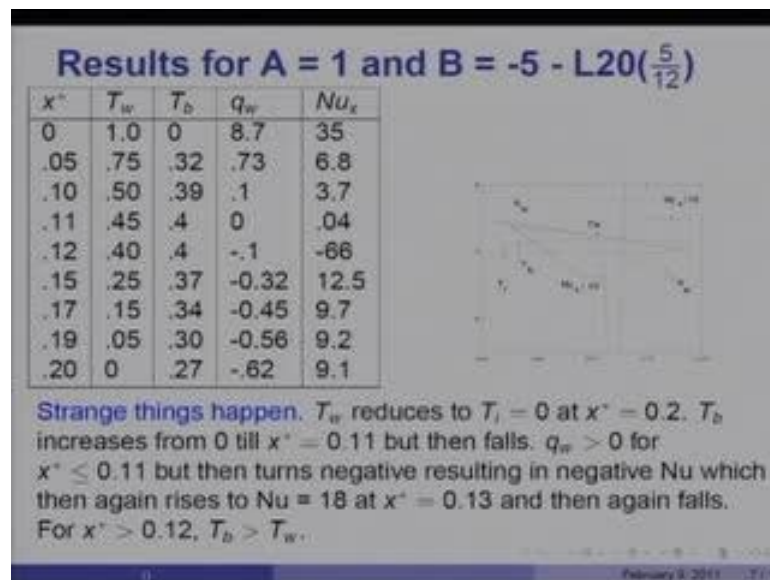
$$T_w - T_b = \frac{9B}{16} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^4} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} + \frac{3A}{2} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right)$$

$$Nu_{x^*} = \frac{q_w 4b}{k(T_w - T_b)} \quad \text{as } x \rightarrow \infty, Nu_x = \frac{8 \sum_{n=0}^{\infty} A_n / \lambda_n^2}{3 \sum_{n=0}^{\infty} A_n / \lambda_n^4} = 8.23$$

Now, what it shows is that the bulk temperature is also going to vary linearly after a long time although initially it may vary in an arbitrary manner, but after a long time $T_{\text{wall}} - T_{\text{bulk}}$ maintains a constant difference. Therefore, at this point onwards it would be a case of constant heat flux at the wall and that is what we find $\frac{d}{dx}(T_{\text{wall}} - T_{\text{bulk}}) = 0$ to infinity and that would be equal to 8.235.

This would be the general case, if we were to go to infinity we have verified that the solution obtain appears at least correct in the infinity state. Now, I am going to assign some values to A and B and see what happens.

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On this slide, I consider a case of $T_{\text{wall}} - T_{\text{bulk}} = 1 - 5x^*$. The wall temperature declines as x increases, you can see that here, this is the variation of wall temperature on the graph that is shown (Refer Slide Time: 10:25). The values assigned are these, I have gone up to when $T_{\text{wall}} - T_i$ that is the wall temperature deduces back to the inlet temperature. The initial jump is A equal to 1 and it reduces to minus, so at $x^* = 0.2$ the temperature of the wall is same as that of the inlet fluid you can see T_{wall} decreases.

What happens to T_{bulk} ? T_{bulk} as you can see on this graph first T_{bulk} increases as you can see here and then its rate of increase is somewhat slowed down to this is the one that is shown by the dotted line (Refer Slide Time: 11:10). So, T_{bulk} increases to 0.32

0.39, 0.4 at about 0.12 it is same as 0.4 and then begins to decrease when the wall temperature becomes equal to inlet fluid temperature, the bulk temperature that point is 0.27.

Notice, what happens to q_{wall} ; q_{wall} is because T_{wall} is greater than inlet fluid temperature, initially q_{wall} is positive 8.7 reduces to 0.73 at x^* equal to 0.05 reduces to 0.1. In fact, there is one point at 0.11 where there is no heat transfer that means this situation becomes adiabatic. Then, q_{wall} changes sign that means the bulk temperature now is greater than the wall temperature. Therefore, you get negative heat fluxes from x^* equal to 0.12 onwards.

What happens to Nusselt number? As you can see here, the Nusselt number is plotted along there initially, the Nusselt number falls as it should for any heating case, but then it suddenly drops to a very low value almost minus 66 at x^* equal to minus 12 and then suddenly jumps back at 0.15 to plus 12.5 and then again reduces down to 9.1. Since, I have not computed long enough, I do not see 8.325 but, if I were to continue in that fashion but with a negative heat transfer, so even that is possible.

So, indeed strange things happen the wall temperature although is monotonically linearly decreasing, the bulk temperature has a hump at about 0.4 x^* equal to 0.11 to 12.12 and then there is a decline and turning sign and likewise, Nusselt number is decline, but it goes negative and then becomes positive. This is a very interesting case of, how T_{bulk} overtakes T_{wall} and although q_{wall} is negative, you still get Nusselt number positive.

This positive Nusselt number is for negative heat transfer, this positive Nusselt number is for positive heat transfer into the fluid and at this point you get negative heat transfer with negative Nusselt number; so indeed very strange things can happen (Refer Slide Time: 13:50).

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A Problem - L20($\frac{4}{12}$)
 Let $T_w - T_i = (A + Bx^*) \rightarrow dT_w/dx_0^* = B$. Then

$$q_w(x^*) = \frac{k}{b} \left[\frac{3B}{8} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^2} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} \right. \\ \left. + A \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] \quad (\text{note } x_0 = 0)$$

$$T_w - T_b = \frac{9B}{16} \sum_{n=0}^{\infty} \frac{A_n}{\lambda_n^4} \left\{ 1 - \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right\} \\ + \frac{3A}{2} \sum_{n=0}^{\infty} A_n \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right)$$

$$Nu_x = \frac{q_w 4b}{k(T_w - T_b)} \quad \text{as } x \rightarrow \infty, Nu_x = \frac{8 \sum_{n=0}^{\infty} A_n / \lambda_n^2}{3 \sum_{n=0}^{\infty} A_n / \lambda_n^4} = 8.23$$

(Refer Slide Time: 14:33)

Results for A = 1 and B = -5 - L20($\frac{5}{12}$)

x^*	T_w	T_b	q_w	Nu_x
0	1.0	0	8.7	35
.05	.75	.32	.73	6.8
.10	.50	.39	.1	3.7
.11	.45	.4	0	.04
.12	.40	.4	-.1	-66
.15	.25	.37	-0.32	12.5
.17	.15	.34	-0.45	9.7
.19	.05	.30	-0.56	9.2
.20	0	.27	-.62	9.1

Strange things happen. T_w reduces to $T_i = 0$ at $x^* = 0.2$. T_b increases from 0 till $x^* = 0.11$ but then falls. $q_w > 0$ for $x^* \leq 0.11$ but then turns negative resulting in negative Nu which then again rises to Nu = 18 at $x^* = 0.13$ and then again falls. For $x^* > 0.12$, $T_b > T_w$.

One can work out solutions for variety of A and B values to see what happens. As I said, one could generate these solutions also for a circular tube because as you will recall the A_n and λ_n values remain the same for a circular tube also. Therefore, for such a case one can again carry out the integrations that I have shown here and evaluate Nusselt number for circular tube, which I shall not shown here. But it can be done and you do indeed get very strange variations of Nusselt number when the wall temperature varies actually.

(Refer Slide Time: 14:43)

Axial Variation of q_w - L20($\frac{6}{12}$)

From lecture 19, we know that the temperature response for **step-jump** in q_w at $x^* = x_0^*$ is given by

$$\begin{aligned} \psi &= \frac{T - T_i}{q_w b/k} \\ &= \frac{3}{4} (y^{*2} - \frac{y^{*4}}{8}) + 4 x^* - \frac{39}{280} \\ &\quad + \sum_{n=1}^{\infty} C_n Y_n(y^*) \exp(-\frac{8}{3} \lambda_n^2 x^*) \\ \psi_w &= \frac{17}{35} + 4 x^* + \sum_{n=1}^{\infty} B_n \exp(-\frac{8}{3} \lambda_n^2 x^*) \\ \frac{\partial \psi}{\partial x^*} &= 4 - \frac{8}{3} \sum_{n=1}^{\infty} B_n \lambda_n^2 \exp(-\frac{8}{3} \lambda_n^2 x^*) \end{aligned}$$

where $B_n = C_n Y_n(1)$

Let us consider the case of actual variation of heat flux; this is of a considerable interest in nuclear reactors as I will show a little later. From lecture 19 we know that the temperature response for step jump in q wall at x star equal to x naught star is given by T minus T_i q wall b by k this is the fully developed solution, plus this one is the developing part of the solution (Refer Slide Time: 15:00). Remember, recall that we had said ψ equal to $\psi_f d$ plus ψ developing and this part is the fully developed part and this part is the developing part.

So, writing this equation at y star equal to 1 would give me ψ wall equal to $\frac{17}{35}$ plus 4 by 4 x square plus B_n exponential of 8 minus $\frac{8}{3}$ $\lambda_n^2 x$ star, where B_n is C_n multiplied by $Y_n(1)$ which I have taken from y star equal to 1. If I were to take $d \psi$ by dx star it will be 4 minus $\frac{8}{3}$ into all this into λ_n^2 and so on so forth.

(Refer Slide Time: 16:05)

Further Development - 1 - L20($\frac{7}{12}$)

Here, we consider only continuous variation of $q_w(x^*)$. Then, response of bulk and wall temperature will be

$$T_w - T_i = \frac{b}{k} \int_0^{x^*} \frac{\partial \Psi}{\partial x_0^*} q_w(x_0^*) dx_0^*$$

$$= \frac{b}{k} \int_0^{x^*} \left[4 - \frac{8}{3} \sum_{n=1}^{\infty} B_n \lambda_n^2 \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \right] q_w(x_0^*) dx_0^*$$

$$T_b - T_i = \frac{4b}{k} \int_0^{x^*} q_w(x_0^*) dx_0^*$$

$$Nu_{x^*} = \frac{q_w(x^*)}{T_w - T_b} \times \frac{4b}{k}$$

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Here, we consider only continuous variation of $q_w(x^*)$; I am not going to consider situations in which step jumps in a q_w wall occur were there here. Then the response of the bulk and the wall temperature will be $T_w - T_i = \frac{b}{k} \int_0^{x^*} \frac{\partial \Psi}{\partial x_0^*} q_w(x_0^*) dx_0^*$ and that would be essentially equal to $\frac{b}{k} \int_0^{x^*} q_w(x_0^*) dx_0^*$ and that we derived on the previous slide into $\frac{4b}{k} \int_0^{x^*} q_w(x_0^*) dx_0^*$. $T_b - T_i$ would again be $\frac{4b}{k} \int_0^{x^*} q_w(x_0^*) dx_0^*$. Therefore, the Nusselt number would be $q_w(x^*)$ divided by $T_w - T_b$, simply take a difference of these two quantities and multiplied by $\frac{4b}{k}$ for flow between parallel plates.

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A Problem - L20($\frac{8}{12}$)

In nuclear reactors, the fuel elements (rods or plates) generate sinusoidally varying heat flux along the cooling channels. Thus, let

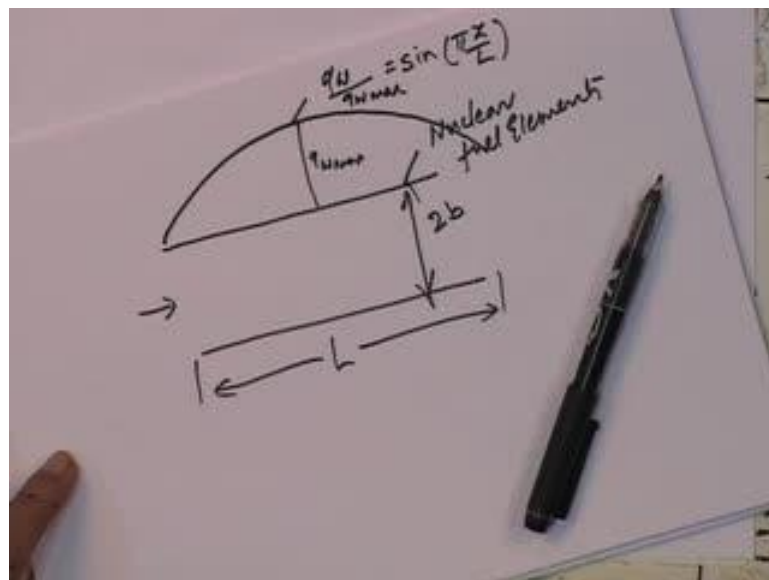
$$\frac{q_w}{q_{w,max}} = \sin\left(\frac{\pi x}{L}\right)$$

where L is the length of the cooling channel. Then

$$\begin{aligned} \frac{T_b - T_i}{(q_{w,max} b/k)} &= \int_0^{L^*} 4 \sin\left(\frac{\pi x_0^*}{L^*}\right) dx_0^* \\ &= \left(\frac{4L^*}{\pi}\right) \left[1 - \cos\left(\frac{\pi x^*}{L^*}\right)\right] \end{aligned}$$

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In nuclear reactors it is quite common. Consider flow between parallel plates and these are the nuclear fuel elements and let us say this is $2b$. In fact, the earliest nuclear fuel elements were in fact flat sheets and not the circular rods that we find today and the coolant would be CO_2 at in the earliest nuclear reactor built.

Let us say, we have this $2b$ nuclear fuel elements generate generally heat in this fashion sinusoidal, so q_w wall would be q_w wall max. Let us say this is q_w wall max equal to $\sin \pi x$ by L , where L is the length of the channel through which the coolant is flowing. I am

going to consider this particular case, because of its relevance to nuclear reactors (Refer Slide Time: 17:40).

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A Problem - L20($\frac{8}{12}$)

In nuclear reactors, the fuel elements (rods or plates) generate sinusoidally varying heat flux along the cooling channels. Thus, let

$$\frac{q_w}{q_{w,max}} = \sin\left(\frac{\pi x}{L}\right)$$

where L is the length of the cooling channel. Then

$$\begin{aligned} \frac{T_b - T_i}{(q_{w,max} b/k)} &= \int_0^{x^*} 4 \sin\left(\frac{\pi x_0^*}{L^*}\right) dx_0^* \\ &= \left(\frac{4L^*}{\pi}\right) \left[1 - \cos\left(\frac{\pi x^*}{L^*}\right)\right] \end{aligned}$$

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As I said, q_w over q_w max is let us say sine pi x by L, where L is the channel length. Then you will see $T_b - T_i$ q_w max b/k would be simply 0 to x^* 4 sin pi x naught star by L^* dx naught star, so that gives us the value of T_b at any x^* would be simply given by that.

(Refer Slide Time: 18:34)

Problem Contd. - 1 - L20($\frac{9}{12}$)

$$\begin{aligned} \frac{T_w - T_i}{(q_{w,max} b/k)} &= \int_0^{x^*} 4 \sin\left(\frac{\pi x_0^*}{L^*}\right) dx_0^* \\ &= \frac{8}{3} \int_0^{x^*} \sum_{n=1}^{\infty} B_n \lambda_n^2 \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) \sin\left(\frac{\pi x_0^*}{L^*}\right) dx_0^* \\ &= \left(\frac{4L^*}{\pi}\right) \left[1 - \cos\left(\frac{\pi x^*}{L^*}\right)\right] \\ &+ \sum_{n=1}^{\infty} \left[\frac{B_n}{1 + \{(3\pi)/(8\lambda_n^2 L^*)\}^2} \right] \\ &\times \left[\sin\left(\frac{\pi x^*}{L^*}\right) \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) + \right. \\ &\left. + \frac{3\pi}{8L^*} \left\{ \cos\left(\frac{\pi x^*}{L^*}\right) \exp\left(-\frac{8}{3} \lambda_n^2 x^*\right) - 1 \right\} \right] \end{aligned}$$

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$T_w - T_b$ would be given in this fashion, this is the $4x \sin x \pi x$ naught star dx naught star minus 8 by 3 all this and that would equal $4L$ star by $\pi \cos \pi x$ star by L star. This integration requires little effort because it is a product of exponential term and a sin term. So, the result is you get a very big bracket here with B_n $1 + 3 \pi$ by 8 lambda n square L star whole square into $\sin \pi x$ star by L star exponential of that term plus this term $\cos \pi x$ star by l star exponential of minus 8 by 3 lambda n square x star minus 1 .

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Problem Contd. - 2 - L20($\frac{10}{12}$)

From the results of last two slides

$$\frac{T_w - T_b}{(q_{w,max} b/k)} = \sum_{n=1}^{\infty} \left[\frac{B_n}{1 + \left\{ \frac{3\pi}{8} \lambda_n^2 L^* \right\}^2} \right]$$

$$\times \left[\sin \left(\frac{\pi x^*}{L^*} \right) \exp \left(-\frac{8}{3} \lambda_n^2 x^* \right) + \frac{3\pi}{8L^*} \left\{ \cos \left(\frac{\pi x^*}{L^*} \right) \exp \left(-\frac{8}{3} \lambda_n^2 x^* \right) - 1 \right\} \right]$$

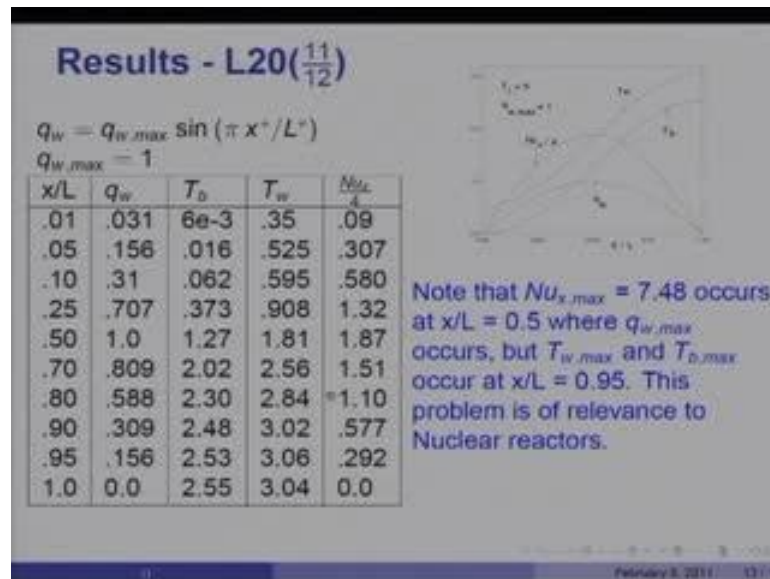
$$Nu_x = \frac{4 \sin(\pi x^*/L^*)}{(T_w - T_b)/(q_{w,max} b/k)}$$

Values of λ_n and B_n are given in lecture 19.

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Now, we have obtained the variation of bulk temperature along the duct and we have obtained a variation of wall temperature along the duct. Therefore, we can calculate the Nusselt number as it is shown here. So, $T_w - T_b$ would simply be different from the previous 2 slides and that would read in this fashion. We then apply Nu_x is equal to $q_w x$, which is been specified $\sin \pi x$ by L 4 times because hydraulic diameter is $4b$ and $T_w - T_b$ into all this $q_w \max b$ by k . The values of lambda n and B_n for constant heat flux cases, where given in lecture number 19.

(Refer Slide Time: 20:18)



See now what happens? You can see the first column here and I have taken $q_{wall,max}$ equal to 1 for convenience and x by L values I have taken are these from 0 to 1. The heat flux according to the sin function would vary in this fashion it reaches maximum at 1 at x by L equal to 0.5 and then again drops down to 0 value. So that is what has been shown here that the q_{wall} is given in this fashion.

Notice how a T_{bulk} varies? T_{bulk} starts off with 0 at the inlet because that is when where the inlet fluid temperature it enters in inlet fluid temperature. Then it begins to rise it is not a linear rise although in the central part, it is linear where q_{wall} is nearly constant. Therefore, T_{bulk} varies more or less linearly in the linear path but then earlier it is not non-linear, so towards the end it is non-linear.

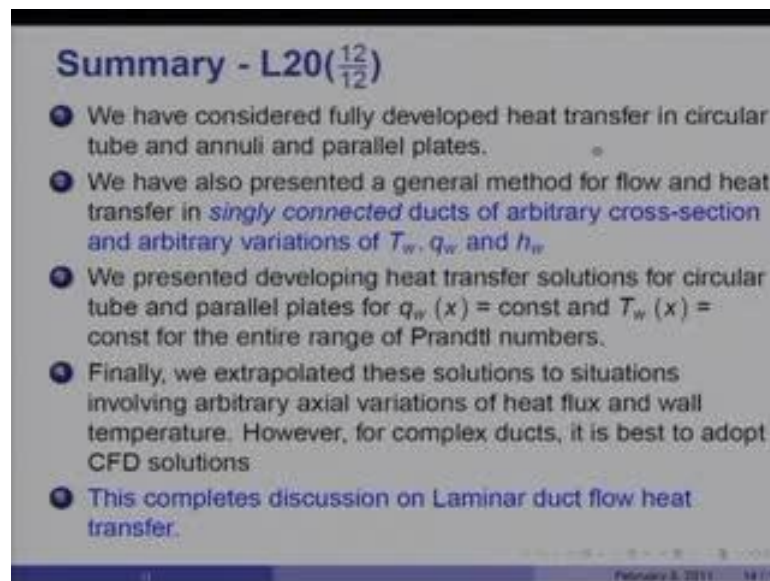
Most importantly see, how the wall temperature varies? The wall temperature increases, from there it changes the slope a little bit and then rises again to that value; this is the wall temperature variation. Therefore, Nusselt number divided by 4 for convenience to get a proper scaling here (Refer Slide Time: 21:45). You will see Nusselt number peaks at 1 at x by L equal to 0.5 at 1.87 as you can see, but then decreases down to 0.

So, the maximum Nusselt number is 7.48 and it occurs at 0.5 where $q_{wall,max}$ occurs but $T_{wall,max}$ and $T_{bulk,max}$ continue to increase till x by L equal to 0.95. This

problem is of relevance to nuclear reactors. Again, we can use the constants given in lecture 19 for flow in a circular tube.

So, you can see that if we were to assume simply that wall temperature was constant, we would get a very wrong picture of what happens and because we have assumed sin function we get T bulk and T wall variation which is different from what it would be if we had a linear constant distribution of constant wall heat flux.

(Refer Slide Time: 23:02)



In summary I would say, we have considered fully developed heat transfer in circular tube and annuli and parallel plates. We have also presented general method for flow and heat transfer in singly connected ducts of arbitrary cross section and arbitrary variation of T_w , q_w and h_w . We have also presented developing flow heat transfer solutions for circular tube and parallel plates for q_w equal to constant and T_w equal to constant for the entire range of Prandtl numbers.

Then finally, we extrapolated these solutions to situations involving arbitrary axial variations of heat flux and wall temperature, I will buy it for flow between parallel plates and or circular tubes. However, for complex ducts it is best to adopt the computational fluid dynamic technique and obtain the solutions.

So, with this I complete my discussion on laminar flow heat transfer, we have considered a typical case of what happens in the developing flow region and considered the fully

developed flow situations and learn to obtain friction factor Reynolds number product for fully developed flows in ducts of arbitrary cross section. Then we extended that to the case situation of fully developed heat transfer in ducts of arbitrary cross section with arbitrary variations of thermal boundary conditions.

We found that for fully developed heat transfer as well as fully developed flow situation you could use Fourier series methods or cantor which variational methods, but we found that the general method based on conversion of the Poisson equations to Laplace equations turns out to be very general and can be applied to variety of ducts including regularly shaped ducts as well as arbitrary variation of boundary condition along the circumference.

Finally, in today's lecture, I indicated how solutions of constant wall heat flux and constant wall temperature that were obtained could be extended; the wall temperature and heat flux varies actually. So, with this I complete my discussion on laminar duct flow heat transfer. From next lecture onwards, we will be moving on to turbulent flow and heat transfer, thank you.