

Finite Element Analysis
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Lecture No. # 33


Let us continue with what we left **left** behind in the last lecture. So, we will see some more examples on applications based on general two-dimensional boundary value problems. And as a part of that, in today's lecture, we are going to see two-dimensional ideal, irrotational, incompressible fluid flows around an object. So, basically if you recall, what we did in the last class, we looked at torsion problem.

Basically, once we get the governing differential equation and associated boundary conditions, we compare the differential equations, and the boundary conditions with the corresponding equations of general two-dimensional boundary value problems. And once we identify the corresponding coefficients, we can easily write the finite element equations. So, that is what basically we did. So, similar manner let us solve some problems in today's class using or the finite element equations; that we already developed for general two-dimensional boundary value problems.

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Ideal Fluid Flow Around An Irregular Object

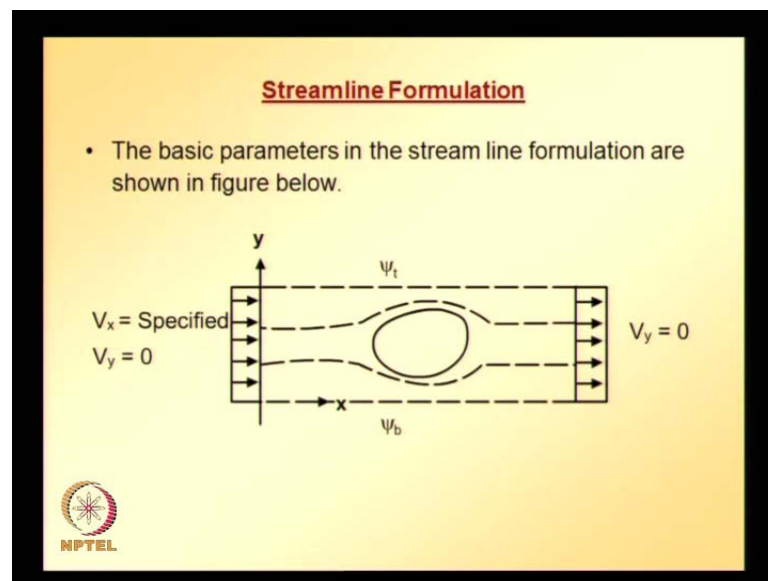
- The problem of two dimensional ideal, irrotational, incompressible fluid flow around an object is solved using either the stream line formulation or the potential formulations.
- Both formulations are presented here.



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So, ideal fluid flows around that irregular object. The problem of two-dimensional ideal, irrotational, incompressible fluid flow around an object is solved using either stream line formulation or potential formulation. So, **there** basically there are two approaches through which, we can solve this problem, and both formulations are presented here that means, we are going to look at the both the formulations. So, first let us start with stream line formulation.

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The basic parameters in the stream line formulations are shown in figure below. The field variable is here for stream line formulation. The field variable is the stream function, which is denoted with psi and it is related to the fluid velocities in x and y directions. And also, the solution domain is extended far enough from the obstruction. So, that there is no effect of obstruction on the flow characteristics **characteristics**.


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Streamline Formulation (Continued)

- The field variable is the stream function $\psi(x,y)$ which is related to the fluid velocities in the x and y direction as follows

$$V_x = \frac{\partial \psi}{\partial y} \quad V_y = -\frac{\partial \psi}{\partial x}$$

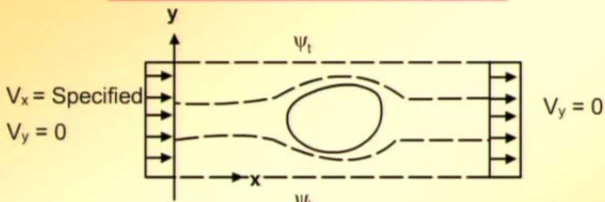
- The governing differential equation is as follows

$$\frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0$$


So, the fluid velocities in x and y directions are related to stream function psi, which is function of x and y through these equations. That is, velocity in the x direction is derivative of psi with partial derivative of psi with respect to y. Velocity fluid velocity in y direction is equal to minus of partial derivative of stream function with respect to x. The governing differential equation is as follows. So, basically we need to solve this differential equation subjected to some boundary conditions imposed on psi.

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Streamline Formulation (Continued)



- The flow rate between any pair of stream lines is given by

$$Q_i = \psi_i - \psi_j$$

- The solution domain is extended far enough from the obstruction so that there is no effect of the obstruction on the flow characteristics.

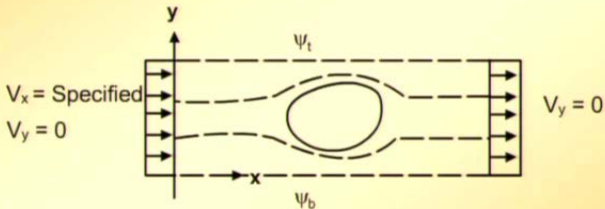


The fluid rate between any pair of stream lines is given by this equation Q_{ij} is equal to ψ_i minus ψ_j . That is, i and j are any pair of stream lines and also the solution domain is extended far enough from the obstruction. So, there is no effect of obstruction on the flow characteristics. And for the figure given here, we can easily identify the boundary conditions.


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Streamline Formulation (Continued)

- Thus the boundary conditions are as shown in figure below



- The constant values of stream lines at top and bottom are determined from the flow rate.

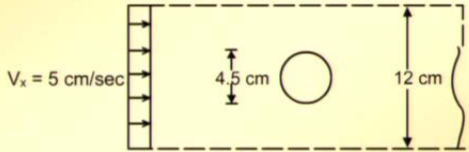
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The boundary conditions are as shown. That is, at the far left end, V_x is specified; V_y is equal to 0 and at the far right end, V_y is equal to 0 again. And stream function at the top is denoted with ψ_t and stream function at the bottom is denoted with ψ_b . The constant values of stream lines at the top and bottom are determined from the flow rate. So, what is the difference between ψ_t and ψ_b ? That is given by the flow rate.

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Example

Consider flow around a cylinder of 4.5 cm diameter as shown in figure below. Away from the cylinder the fluid velocity in the horizontal direction, $V_x = 5$ cm/sec.

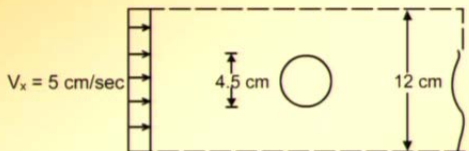




Now, let us consider an example. Consider flow around cylinder of diameter 4.5 centimeters. Away from the cylinder, the fluid velocity in the horizontal direction is V_x is equal to 5 centimeter per second. So, this is the problem. V_x is specified; V_y is equal to 0. The diameter of a cylinder is given as 4.5 centimeters and total width of flow is given as 12 centimeters. So, the constant ψ values; that is, ψ_t and ψ_b as I just mentioned, it can be determined from the flow rate per unit thickness. Here thickness is assumed to be 1 unit.


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Example (Continued)



- The constant ψ values at top and bottom are determined from the flow rate as follows

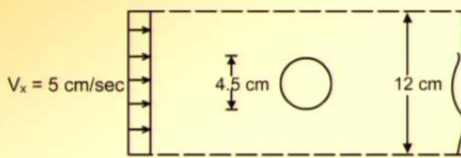
Total flow/unit thickness = $12 \text{ cm} \times 1 \text{ cm} \times 5 \text{ cm/sec}$
 $= 60 \text{ cm}^3/\text{sec}.$



So, flow rate or unit thickness is given by $12 \times 1 \times V_x$; that is, 5 centimeter per second and it comes out to be 60 centimeter cube per second. So, the difference that is $\psi_t - \psi_b$ is 60 centimeter cube sorry 60 centimeter cube per second.

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Example (Continued)



- The flow rate between any two stream lines is the difference between them.
- Thus $\psi_t - \psi_b = 60$ where ψ_t is constant stream line at the top and ψ_b is constant stream line at the bottom.

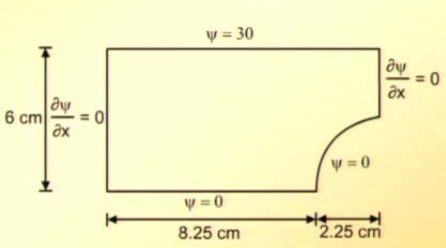
Because of symmetry the two stream lines are equal.
Therefore $\psi_t = -\psi_b = 30$.

So, the flow rate between any two streams stream lines is the different between them. So, and because of symmetry, the two stream lines are equal and actually because of symmetry, we can model only a quarter of this domain. In that case, $\psi_t - \psi_b$ is going to be minus of ψ_b is equal to 30, which is half of 60.

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Example (Continued)

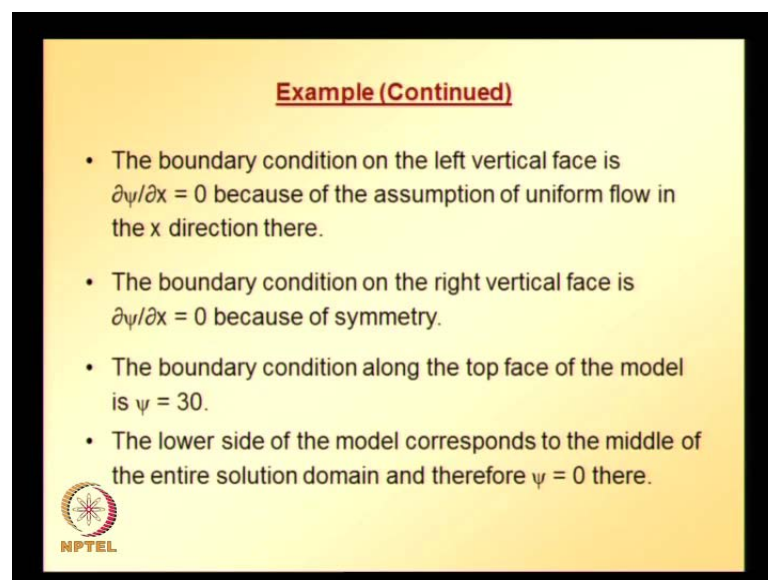
- A distance of 8.25 cm in front of the cylinder is assumed to be enough for constant velocity condition there.
- Using symmetry only a quarter of the solution domain is to be modeled as shown in figure below.



A distance of ah 8.5 8.25 in front of cylinder is assumed to be enough for constant velocity condition; because we need to select solution domain far enough from the obstruction. So, that no effect of obstruction on the flow characteristics. Using symmetry, only a quarter of solution domain is to be modeled and the quarter model is shown are along with all the boundary conditions. Since V_y is equal to 0, partial derivative of stream function with respect to x is equal to 0 on the left extreme left edge.


And because of symmetry, partial derivative of stream function with respect to y , which is indirectly related to the V_y ; velocity fluid velocity along y direction is also 0 along the line of symmetry on the right hand side. And again, since the model is symmetry; because of symmetry even ψ is equal to 0 on the bottom edge of this quarter model. Similarly, the edge which is coinciding with the cylinder ψ is equal to 0 and ψ is equal to 30 and that is coming from flow rate, which we just calculated. So, this is the quarter model of the solution domain. So, we need to solve the differential equation a second order differential equation; that we have seen earlier in terms of ψ subjected to these boundary conditions.

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Example (Continued)

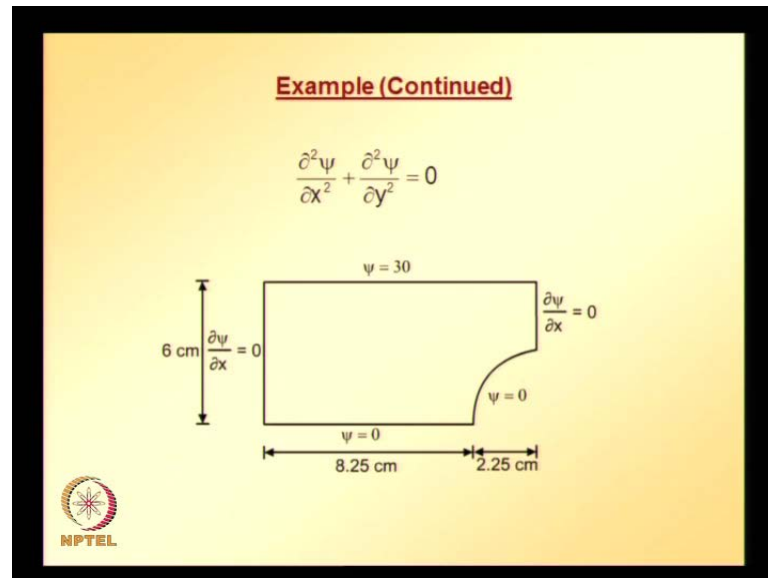
- The boundary condition on the left vertical face is $\partial\psi/\partial x = 0$ because of the assumption of uniform flow in the x direction there.
- The boundary condition on the right vertical face is $\partial\psi/\partial x = 0$ because of symmetry.
- The boundary condition along the top face of the model is $\psi = 30$.
- The lower side of the model corresponds to the middle of the entire solution domain and therefore $\psi = 0$ there.


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So, the boundary condition on the left vertical face is partial derivative of ψ with respect to x is equal to 0; because of assumption of uniform flow in the x direction there. The boundary condition on the right right vertical face is partial derivative of stream function with respect to s is equal to 0, because of symmetry. And the boundary

condition along the top face of the model is ψ is equal to 30. Lower side of the model corresponds to the middle of entire solution domain. So, stream function is equal to 0 there. So, these are the boundary conditions.

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So, we need to solve this second order differential equations subjected to these boundary conditions. So, that is the problem statement. Now, everything is given to us. Now, our job is to compare this governing differential equation with general two dimensional boundary value problems; the differential equation, that we have taken for general two dimensional boundary value problems. And the corresponding boundary conditions, we need to compare with the boundary conditions of this particular problem. Identify the coefficients and we can actually write the finite element equations.

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**Two Dimensional Boundary Value
Problem Statement**


$$\frac{\partial}{\partial x} \left[k_x \frac{\partial T}{\partial x} \right] + \frac{\partial}{\partial y} \left[k_y \frac{\partial T}{\partial y} \right] + P(x,y)T + Q(x,y) = 0 \text{ in } A$$

$T = T_0(x,y)$ on S_1 (Essential boundary condition)

or

$$k \frac{\partial T}{\partial n} + \alpha(x,y)T + \beta(x,y) = 0 \text{ on } S_2$$

(Natural boundary condition)



So, now let us look once again general two dimensional boundary value problems. This is a statement, that we started out with and this needs to be satisfied over the domain A subjected to any of these boundary conditions; essential boundary condition and natural boundary condition.

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The complete element equations can be written as follows


$$[k_x + k_y + k_p + k_\alpha] d = r_q + r_\beta \quad \text{or} \quad kd = r$$

where

$$k_x = \iint_A k_x \mathbf{B}_x \mathbf{B}_x^T dA \qquad k_y = \iint_A k_y \mathbf{B}_y \mathbf{B}_y^T dA$$

$$k_p = - \iint_A P \mathbf{N} \mathbf{N}^T dA$$

$$k_\alpha = \int_{S_2} \alpha \mathbf{N} \mathbf{N}^T dS \qquad r_\beta = - \int_{S_2} \beta \mathbf{N} dS$$

$$r_q = \iint_A Q \mathbf{N} dA$$



And using Galerkin criteria, what we did is, after substituting finite element approximations, a complete equations for general two dimensional boundary value

problems reduces to this; where, each of these k_x , k_y , k_p , k_α , r_β , and r_q are defined here.

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Since the shape functions are very simple for a linear triangle, it is possible to carry out all integrations in closed form, assuming k_x , k_y , P and Q are constant over an element, to get element equations in an explicit form as follows.

$$k_x = \iint_A k_x \mathbf{B}_x \mathbf{B}_x^T dA = k_x \frac{1}{4A^2} \begin{Bmatrix} b_1 \\ b_2 \\ b_3 \end{Bmatrix} [b_1 \quad b_2 \quad b_3] \int_A dA$$


$$= \frac{k_x}{4A} \begin{bmatrix} b_1^2 & b_1 b_2 & b_1 b_3 \\ b_1 b_2 & b_2^2 & b_2 b_3 \\ b_1 b_3 & b_2 b_3 & b_3^2 \end{bmatrix}$$


And for linear three node triangle element, since the shear functions are very simple for linear triangle, it is possible to carry out all integrations in closed form. Assuming k_x , k_y , P , Q are constant over element to get element equations in explicit form. So, for three node triangle element, k_x turns out to be this; after multiplication of the **two matrix** two vectors, we get this.

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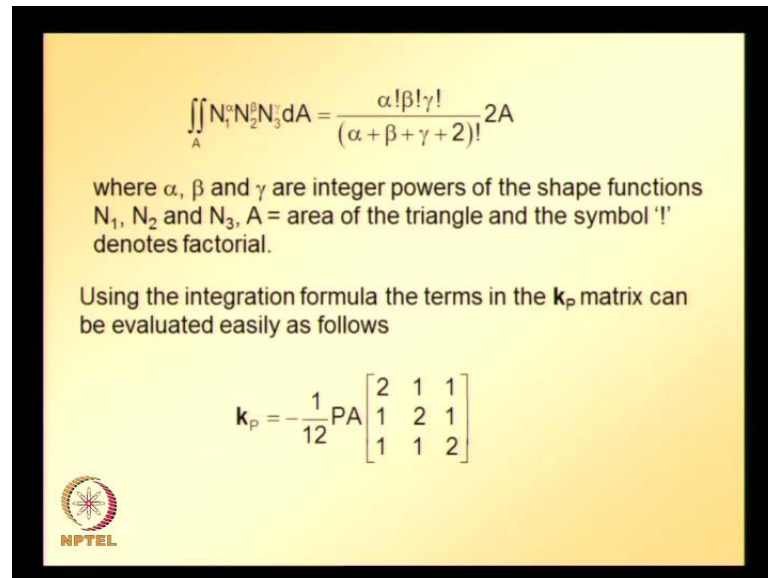
$$k_p = - \iint_A P N N^T dA = -P \iint_A \begin{bmatrix} N_1^2 & N_1 N_2 & N_1 N_3 \\ N_1 N_2 & N_2^2 & N_2 N_3 \\ N_1 N_3 & N_2 N_3 & N_3^2 \end{bmatrix} dA$$

The terms in the k_p matrix are not constant. Fortunately following simple formula is available for integrating shape functions over a triangle



Similarly, k_y and k_p and here, since the components of P are not going to be constant, we need to perform some numerical integration. Terms in k_p matrix are not constant. Fortunately, simple formula is available for integrating shape functions over triangle.

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$$\iint_A N_1^\alpha N_2^\beta N_3^\gamma dA = \frac{\alpha! \beta! \gamma!}{(\alpha + \beta + \gamma + 2)!} 2A$$

where α , β and γ are integer powers of the shape functions N_1 , N_2 and N_3 , A = area of the triangle and the symbol '!' denotes factorial.

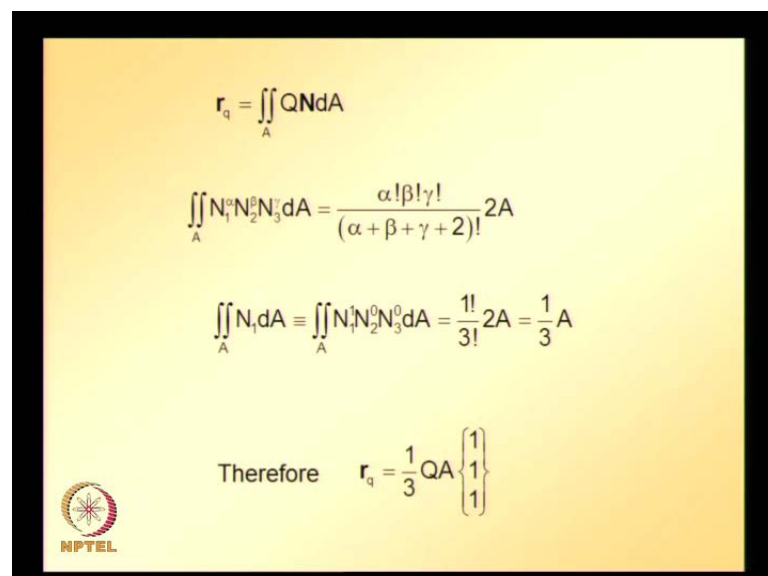
Using the integration formula the terms in the k_p matrix can be evaluated easily as follows

$$k_p = -\frac{1}{12} PA \begin{bmatrix} 2 & 1 & 1 \\ 1 & 2 & 1 \\ 1 & 1 & 2 \end{bmatrix}$$



This is the formula. So, applying this formula, all the parameters in that formula are defined; alpha, beta, gamma. N_1 , N_2 , N_3 are the three shape functions of the linear triangular element. A is the area of triangle and symbol exclamation is for factorial and using integration formula, the terms in k_p matrix can be evaluated like this.

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$$r_q = \iint_A QN dA$$

$$\iint_A N_1^\alpha N_2^\beta N_3^\gamma dA = \frac{\alpha! \beta! \gamma!}{(\alpha + \beta + \gamma + 2)!} 2A$$

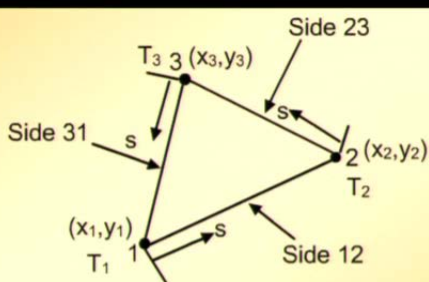
$$\iint_A N_1 dA \equiv \iint_A N_1^1 N_2^0 N_3^0 dA = \frac{1!}{3!} 2A = \frac{1}{3} A$$

Therefore $r_q = \frac{1}{3} QA \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$




And then, we required to evaluate r q . Once we identify q , what is q ? We can evaluate r q . For evaluating r q also, we can use this formula and one component of that r q is simplified and **shown** the details are shown. And carrying out similar kind of integrations, we can get all components of r q . So, therefore r q is given by one third Q times A and 3 by 1 column vector consisting of all 1 's.

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$$k_{\alpha_{12}} = \alpha \int_0^{L_{12}} \begin{Bmatrix} \frac{L_{12}-s}{L_{12}} \\ \frac{s}{L_{12}} \\ 0 \end{Bmatrix} \begin{bmatrix} \frac{L_{12}-s}{L_{12}} & \frac{s}{L_{12}} & 0 \end{bmatrix} ds$$

$$= \alpha \begin{bmatrix} L_{12}/3 & L_{12}/6 & 0 \\ L_{12}/6 & L_{12}/3 & 0 \\ 0 & 0 & 0 \end{bmatrix} = \frac{\alpha L_{12}}{6} \begin{bmatrix} 2 & 1 & 0 \\ 1 & 2 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$



And now coming to the line integrals, once we identify once we carefully compare the given boundary conditions with the corresponding boundary conditions of general two dimensional boundary value problems, we can identify what are the corresponding parameters? And we can assemble the boundary integrals, k α . For illustration, k α is given here; it is k α evaluated along side 1 2 details are given here. The shape functions of nodes 1 and 2 can be obtained using one dimensional Lagrange interpolation formula. And simplification of this leads to the similar integrations can be carried out along the other two edges.

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$$r_{\beta 12} = -\int_{S_2} \beta \begin{Bmatrix} N_1 \\ N_2 \\ N_3 \end{Bmatrix} dS = -\beta \int_0^{L_{12}} \begin{Bmatrix} L_{12} - s \\ L_{12} \\ s \\ L_{12} \\ 0 \end{Bmatrix} ds = -\beta \begin{Bmatrix} L_{12}/2 \\ L_{12}/2 \\ 0 \end{Bmatrix} = -\beta \frac{L_{12}}{2} \begin{Bmatrix} 1 \\ 1 \\ 0 \end{Bmatrix}$$

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Similarly, r_{β} along side 1 2: So, all these procedures, which we have seen earlier needs to be repeated.

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For side 2-3 and 3-1, the integrals can be evaluated in a similar manner.


In fact the only thing different for the other sides is the placement of zero's in the above matrices.

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Once we identify the corresponding parameters for this ideal fluid flow around an irregular object. So, for side 2-3, 3-1 integrals can be evaluated in similar manner and only difference is going to be the placement of zero's in the matrices consisting of shape functions.

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It is easy to verify that for sides 23 and 31 we have

$$\mathbf{k}_{\alpha, \text{side23}} = \frac{\alpha L_{23}}{6} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 2 & 1 \\ 0 & 1 & 2 \end{bmatrix} \quad \mathbf{r}_{\beta, \text{side23}} = -\beta \frac{L_{23}}{2} \begin{Bmatrix} 0 \\ 1 \\ 1 \end{Bmatrix}$$
$$\mathbf{k}_{\alpha, \text{side31}} = \frac{\alpha L_{31}}{6} \begin{bmatrix} 2 & 0 & 1 \\ 0 & 0 & 0 \\ 1 & 0 & 2 \end{bmatrix} \quad \mathbf{r}_{\beta, \text{side31}} = -\beta \frac{L_{31}}{2} \begin{Bmatrix} 1 \\ 0 \\ 1 \end{Bmatrix}$$



And for side 23 and 31, k alpha and r beta are looks like this. So, once we identify the corresponding coefficients, we can get the element equations. And once we get element equations, we can get the global equation system based on nodal connectivity. And after imposing the essential boundary conditions, we can solve for the nodal values.

And once we solve for the nodal values here, for this particular problem of fluid flow around an irregular object, **around an irregular object** the nodal parameters are going to be the stream function values. Once we obtain stream function values, we can find partial derivatives of stream functions; stream function with respect to x and with respect to y and which are related to the velocity components or fluid velocities along x and y directions.

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Stream Function Calculations

Once the nodal ψ values are known, the stream function can be interpolated using the element shape functions as follows

$$\psi(x,y) = [N_1 \quad N_2 \quad N_3] \begin{Bmatrix} \psi_1 \\ \psi_2 \\ \psi_3 \end{Bmatrix}$$


So, stream function calculations: Once the nodal values ψ are known, stream function can be interpolated using element shape functions like this.

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Stream Function Calculations (Continued)

$$\iint_{\text{elem}} \psi dx dy = \iint_{\text{elem}} [N_1 \quad N_2 \quad N_3] dx dy \begin{Bmatrix} \psi_1 \\ \psi_2 \\ \psi_3 \end{Bmatrix}$$

For a linear triangular element

$$\iint_{\text{elem}} N_1 dx dy = \iint_{\text{elem}} N_2 dx dy = \iint_{\text{elem}} N_3 dx dy = \frac{A}{3}$$

Therefore

$$\iint_{\text{elem}} \psi dx dy = \frac{A}{3} [\psi_1 + \psi_2 + \psi_3]$$


And also, **we require** sometimes we require this. What is the integral $\psi dx dy$ over the element or entire domain? Here, the formula is written for one particular element. So, we know ψ is equal to $N_1 \psi_1$ plus $N_2 \psi_2$ plus $N_3 \psi_3$, which can be written in matrix and vector form. As it is shown on the right hand side of the equation and substituting, since ψ_1 , ψ_2 , and ψ_3 , **the nodal** the stream function nodal

corresponding to the nodes. Nodal stream function values are constant. They can be taken out of the integral.

And once we simplify this integral with N_1 , N_2 , and N_3 integrated over $dx dy$ or the entire area of the element for linear triangular element, these integrals turn out to be area of triangle divided by 3. So, substituting this **this** integral, that is ψ integrated over the entire element can be approximated as area of the triangle element divided by 3 or area of triangle element multiplied by average value of a stream function at all the nodes. That is, A over 3 times **in** are in brackets ψ_1 plus ψ_2 plus ψ_3 as shown in the right hand side of the **equation** last equation.

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Example (Continued)

$$\frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0$$

$\psi = 30$
 $\frac{\partial \psi}{\partial x} = 0$
 $\psi = 0$
 $\frac{\partial \psi}{\partial x} = 0$

6 cm
 8.25 cm
 2.25 cm

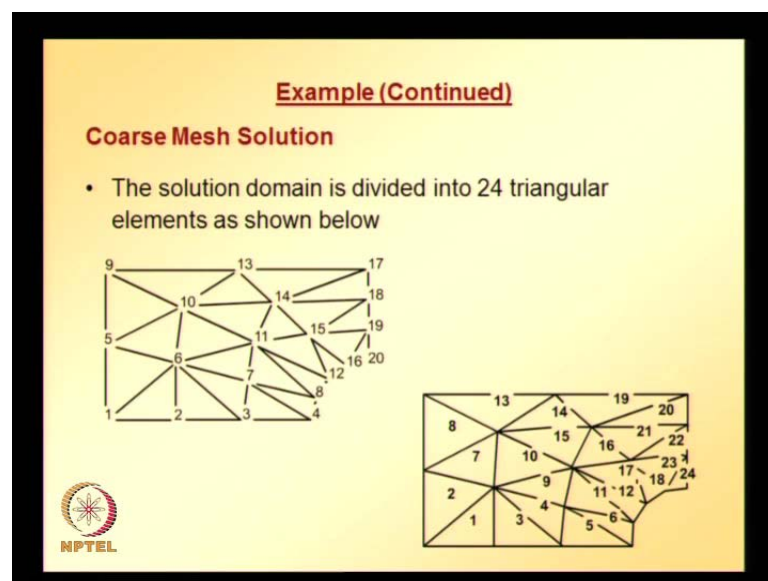
- Comparing with general 2D BVP, we see that for the torsion problem, $k_x = k_y = 1$, $P = 0$ and $Q = 0$.

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Now, we require to solve this differential equations subjected to these boundary conditions. So, proceeding in the way that is explained, we can get the element equations. And once the element equations are obtained, we can assemble the global equation system; impose the essential boundary conditions. Here, essential boundary conditions are ψ is equal to 30 on the top phase; ψ is equal to 0 on the bottom face. So, these are the essential boundary conditions. After imposing these essential boundary conditions, we can or we need to impose these essential boundary conditions; the nodes which are along these boundaries.

Once we impose these essential boundary conditions, we can solve for the nodal values and do all kinds of post processing. So, now comparing with general two dimensional boundary value problems, see that for this **for this ah sorry** it is mentioned as torsion problem; it should be ideal fluid flow problem. So, for this particular problem, when we compare with general two dimensional boundary value problem, it turns out that k_x is equal to k_y is equal to 1; P is equal to 0 and Q is equal to 0. So, once again there is a mistake. It should be **should be** fluid flow problem, instead of torsion problem.

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So, now let us see the solution, how it looks or how it converges, when we take different meshes? Here coarse mesh solution, by coarse mesh I mean the mesh; the quarter model consists of 24 **triangle elements** triangular elements. So, this is the quarter model and it consists of 20 nodes. The corresponding elements are here **24 elements** 24 **triangle elements** triangular elements. So, using this discretization, we can solve this problem by imposing essential boundary conditions at nodes 9, 30, 70, 1, 2, 3, 4, 8, 12, 16, and 20. By imposing essential boundary conditions over these nodes and we can solve for the stream function value at other nodes. And here at each node, there is only one degree of freedom, which is going to be the stream function.

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Nodal solution


Node #	X	Y	ψ
1	0	0	0
2	2.75	0	0
3	5.5	0	0
4	8.25	0	0
5	0	3	14.827214
6	2.807	2.287	11.099329
7	5.614	1.574	6.83674
8	8.421	0.861	0
9	0	6	29.999998
10	2.9697	4.5303	22.444016



So, the nodal solution, the details are given here at all nodes 1 to 10 in this table and next table shows for the rest of the nodes.

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Node #	X	Y	ψ
11	5.9393	3.0607	13.818343
12	8.909	1.591	0
13	5.25	6	30
14	6.713	4.693	22.093536
15	8.176	3.386	12.595177
16	9.639	2.079	0
17	10.5	6	30
18	10.5	4.75	21.031805
19	10.5	3.5	11.313728
20	10.5	2.25	0



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Solution derivatives and integral over each element

Element #	$\partial\psi/\partial x$	$\partial\psi/\partial y$	$\int_{\text{elem}} \psi dA$
1	0	-4.8532267	11.634409
2	-7.27E-02	-4.9424047	36.387901
3	0	-4.8532267	11.634409
4	-0.4077609	-4.3730779	13.450557
5	0	-4.3435454	4.932138
6	-2.318635	-0.4604942	2.8927951
7	-4.27E-02	-5.0601735	51.700157
8	-4.14E-02	-5.0575948	99.886673
9	-0.3085509	-4.7636561	23.313257



And solution derivatives and integral over each element, they are shown in this table. Partial derivative of stream function with respect to x, partial derivative of stream function with respect to y and integral of stream function over the element domain 1 to 9 elements are shown here.

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Element #	$\partial\psi/\partial x$	$\partial\psi/\partial y$	$\int_{\text{elem}} \psi dA$
10	-0.3879723	-5.085207	54.47382
11	-1.1773081	-4.9537678	15.164419
12	-3.4964237	-2.3373396	6.6444569
13	3.71E-07	-5.1412802	106.01979
14	-0.339969	-5.6687737	63.736046
15	-0.3205318	-5.2214808	58.205109
16	-1.3793432	-5.72332214	27.481548
17	-1.4796014	-6.4126062	18.724054
18	-3.6848018	-5.5120974	3.5015714
19	-2.78E-07	-6.0493217	93.884232




And rests of the elements are shown. The other slide 10 to 19, details are given here.

(No audio from 24:48 to 25:01)

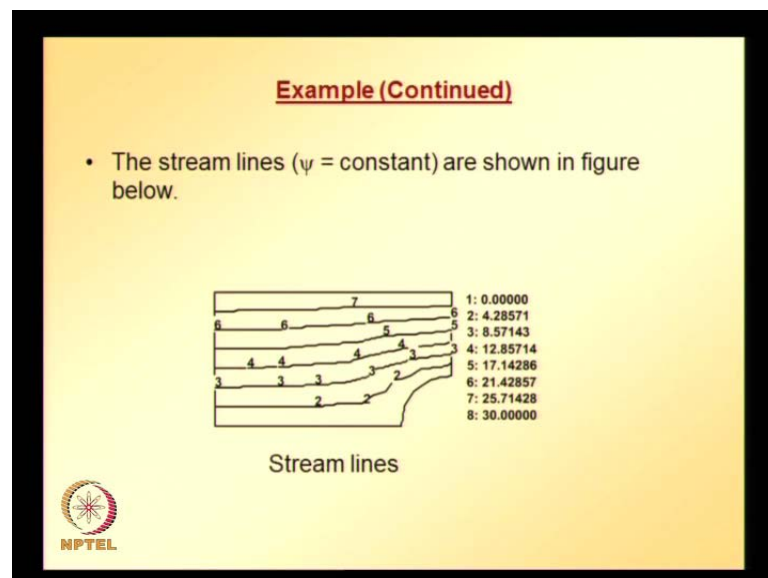
(Refer Slide Time: 25:03)

Element #	$\partial\psi/\partial x$	$\partial\psi/\partial y$	$\int_{\text{elem}} \psi dA$
20	-0.38835	-7.1745563	57.692852
21	-0.3832884	-6.838263	46.740227
22	-0.9327613	-7.7744613	21.758797
23	-0.9708058	-8.5500298	12.768353
24	-1.7975818	-9.0509825	2.0294001



20 to 24 details are given here. We can also plot, what are called stream lines. Corresponding to the lines along which, stream function value is constant.

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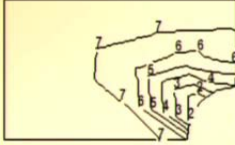


So, stream lines are shown in this figure. Also we can plot, derivatives of stream functions **stream function** with respect to x and y.

(Refer Slide Time: 25:50)


Example (Continued)

- The x derivative of ψ (= negative of velocity in the y direction) and the y derivative of ψ (= velocity in the x direction) are shown in figure below.




1:	-3.60298
2:	-3.07847
3:	-2.55397
4:	-2.02946
5:	-1.50495
6:	-0.98044
7:	-0.45693
8:	0.06858

Contours of $\partial\psi/\partial x$



1:	-9.63457
2:	-8.50559
3:	-7.37661
4:	-6.24762
5:	-5.11864
6:	-3.98966
7:	-2.86068
8:	-1.73170

Contours of $\partial\psi/\partial y$



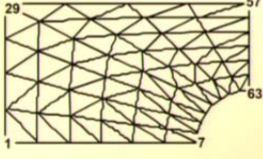
Derivative of stream function, x derivative of stream function, y derivative of stream function; because x derivative of stream function is related to velocity in the y direction and y derivative of stream function is related to velocity in the x direction. And they are shown here, contours of derivatives of stream function with respect to x and y. So, whatever **solution** finite element solution that is shown is using 24 triangular elements. The problem can be repeated by using finer mesh.


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Example (Continued)

Fine Mesh Solution

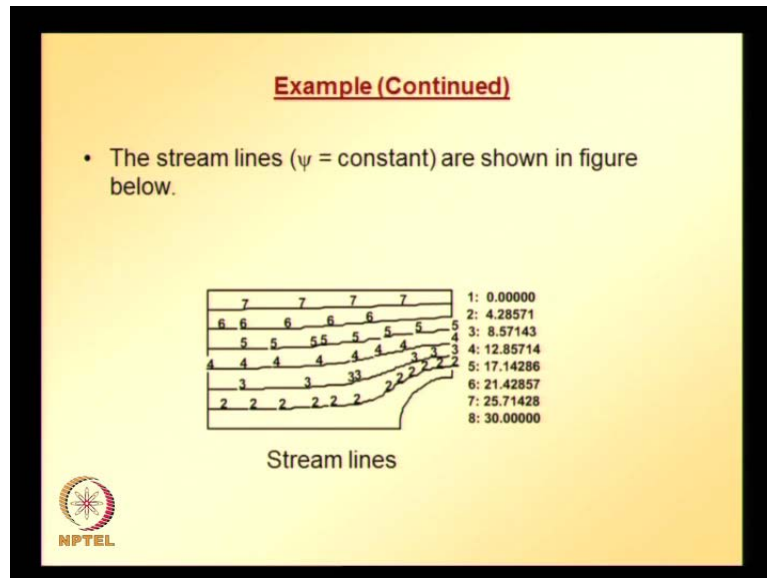
- The solution domain is divided into 96 triangular elements as shown below





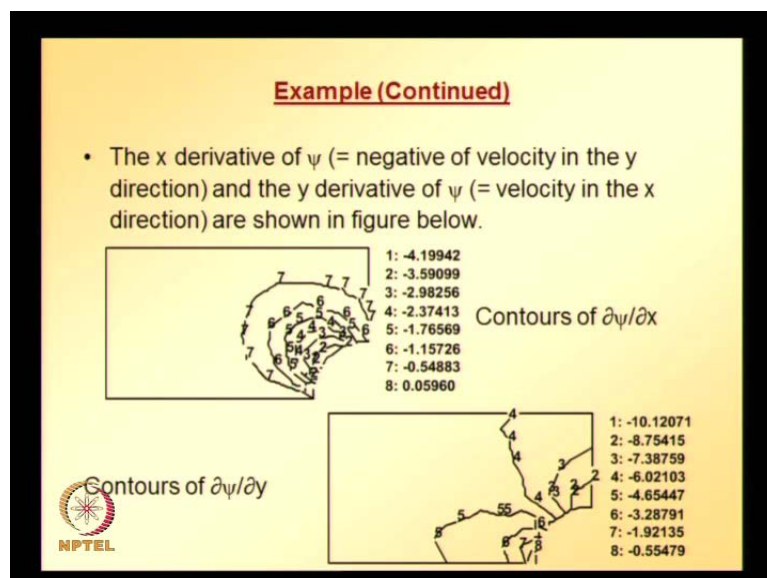
Solution domain is divided into 96 triangular elements. So, the discretization is shown here and everything remains same, except that size of the problem increases; because we require to solve for more number of nodal values.

(Refer Slide Time: 26:57)



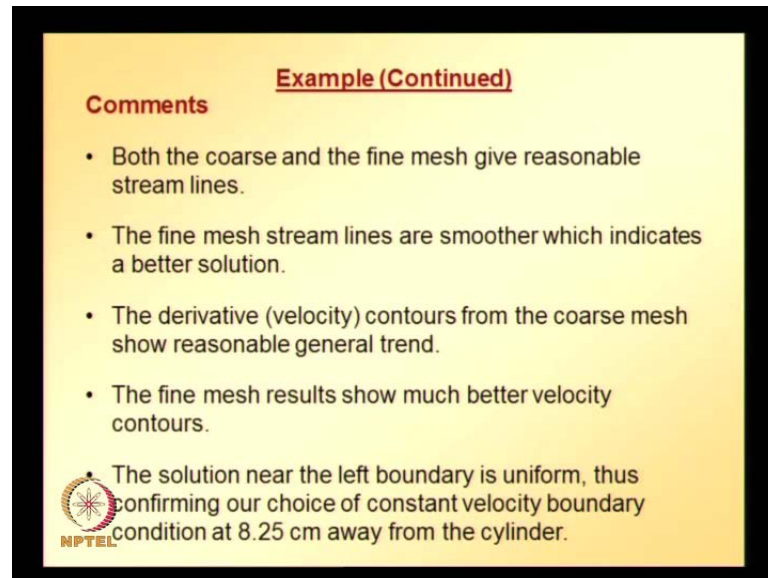
So, let us look at, how the solution looks? Stream **(()) stream** lines, psi is equal to constant; those lines are these.

(Refer Slide Time: 27:12)



And the derivatives of stream function, ψ with respect to x and with respect to y . So, we solve this problem using 24 elements and 96 elements.

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Example (Continued)

Comments

- Both the coarse and the fine mesh give reasonable stream lines.
- The fine mesh stream lines are smoother which indicates a better solution.
- The derivative (velocity) contours from the coarse mesh show reasonable general trend.
- The fine mesh results show much better velocity contours.
- The solution near the left boundary is uniform, thus confirming our choice of constant velocity boundary condition at 8.25 cm away from the cylinder.

Let us see, how the solution matches? Both coarse and fine mesh gives a reasonable stream line that is expected. Because the solution itself may converge very fast; whereas, derivatives of solution require finer mesh. Fine mesh stream lines are smoother, which indicates a better solution. Derivative contours for coarse mesh show reasonable agree reasonable general trend. Fine mesh solution results show better velocity contours. And solution near left boundary is uniform; thus confirming our choice of constant velocity boundary at a distance of 8.25 centimeters away from cylinder.


(Refer Slide Time: 28:37)

Potential Formulation

In the potential formulation the velocity components are related to potential ϕ by

$$V_x = \frac{\partial \phi}{\partial x} \quad V_y = \frac{\partial \phi}{\partial y}$$

The governing differential equation, derived from Euler's momentum equation and conservation of mass, is as follows.

$$\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$$


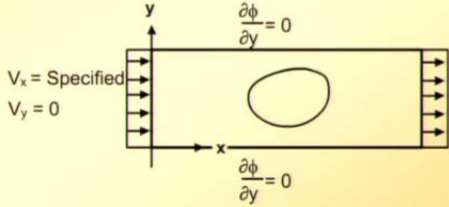
So, now let us look at, solving this ideal fluid flow around an irregular object through the different formulation; that is potential formulation, **potential formulation potential formulation**. The velocity components are related to a potential phi by these relations. **velocity fluid** fluid velocity along x direction, y direction are related to the potential phi, where this equations. The governing differential equation, derived from Euler's momentum equation and conservation of mass is this one. So, basically we require to solve this second order differential equation; solve for phi subjected to some boundary conditions.

(Refer Slide Time: 29:32)

Example (Continued)

$$\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$$

The appropriate boundary conditions are shown in figure below.



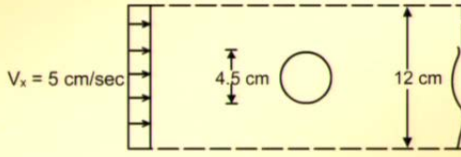
The diagram shows a rectangular domain with a cylinder inside. The left boundary is labeled with $V_x = \text{Specified}$ and $V_y = 0$. The top and bottom boundaries are labeled with $\frac{\partial \phi}{\partial y} = 0$. The right boundary has arrows pointing outwards. The NPTEL logo is in the bottom left corner.

Appropriate boundary conditions are as given here in the are are as shown in the figure. That is, essential boundary condition at the top and bottom are partial derivative of potential with respect to y is equal to 0. And V_x is specified on the left vertical side and V_y is equal to 0. So, we can solve or we can redo the problem that we already did, using potential function for potential formulation.

(Refer Slide Time: 30:15)

Example

Consider flow around a cylinder of 4.5 cm diameter as shown in figure below. Away from the cylinder the fluid velocity in the horizontal direction, $V_x = 5 \text{ cm/sec}$.

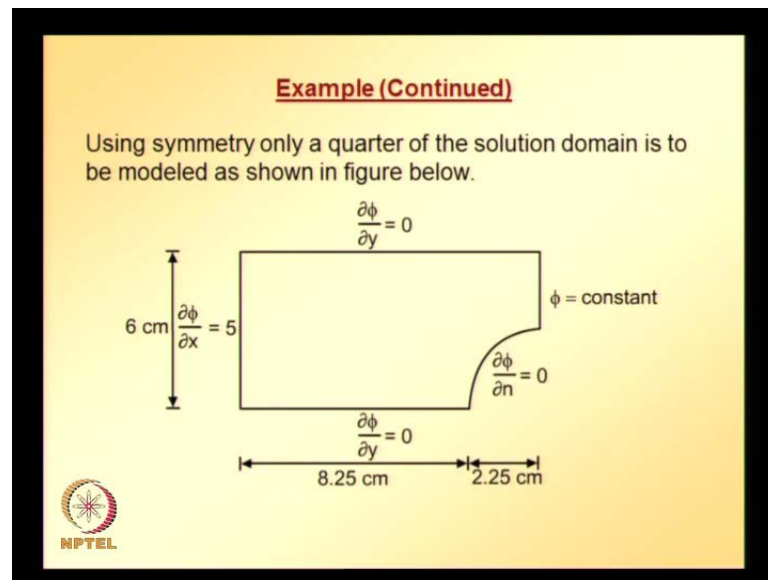


The diagram shows a dashed rectangular box containing a cylinder. The cylinder has a diameter of 4.5 cm. The height of the box is 12 cm. The flow velocity $V_x = 5 \text{ cm/sec}$ is indicated on the left side. The NPTEL logo is in the bottom left corner.

And we let us look at the problem statement. For the problem that we already looked at, consider fluid flow around cylinder of 4.5 centimeters and V_x is 5 centimeters per

second. Again, only a quarter of solution domain needs to be modeled, because of symmetry.

(Refer Slide Time: 30:48)



And the boundary condition on the left vertical phase is partial derivative of potential with respect to x is equal to ϕ ; because V_x , fluid velocity in the x direction at that location is already given in the problem statement. **Because** and this is because of the assumption of uniform flow in the x direction. Boundary condition on the right vertical phase is **potential is equal to 0 or potential sorry** potential is equal to constant; because of symmetry. For the numerical solutions a constant a value of **50 is assigned to psi sorry** 50 is assigned to the potential ϕ along this phase.

This is arbitrarily assigned. Arbitrarily, since ϕ is equal to potential is equal to constant and that constant value is taken as 50 just **ah just** arbitrarily. The boundary condition along the top and bottom phase of this model are partial derivative of potential with respect to y is equal to 0, which are indicated in the figure and all the details of geometry are shown in the figure. So, now with this information, we can solve the second order differential equation with respect to the potential; for this particular problem, over this domain subjected to these boundary conditions.


By comparing this with the problem statement of general two dimensional boundary value problem and then, we can actually get the finite element equations. And then, as

usual the procedure is same; assembling and applying the essential boundary conditions solving for the **for the** nodal values. Here the nodal values, turns out to the potential value. And again this **in this** problem, at each node you have only one degree of freedom, which is going to be the potential phi value.

(Refer Slide Time: 33:15)

Example (Continued)

- The boundary condition on the left vertical face is $\partial\phi/\partial x = V_x = 5$ because of the assumption of uniform flow in the x direction there.
- The boundary condition on the right vertical face is $\phi = \text{constant}$ because of symmetry.
- For numerical solutions a value of 50 is assigned arbitrarily to ϕ along this face.
- The boundary conditions along the top and bottom face of the model are $\partial\phi/\partial y = 0$.



So, these are the boundary conditions that we already looked at.


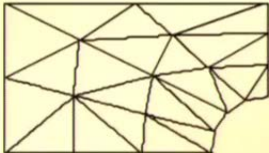
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Example (Continued)

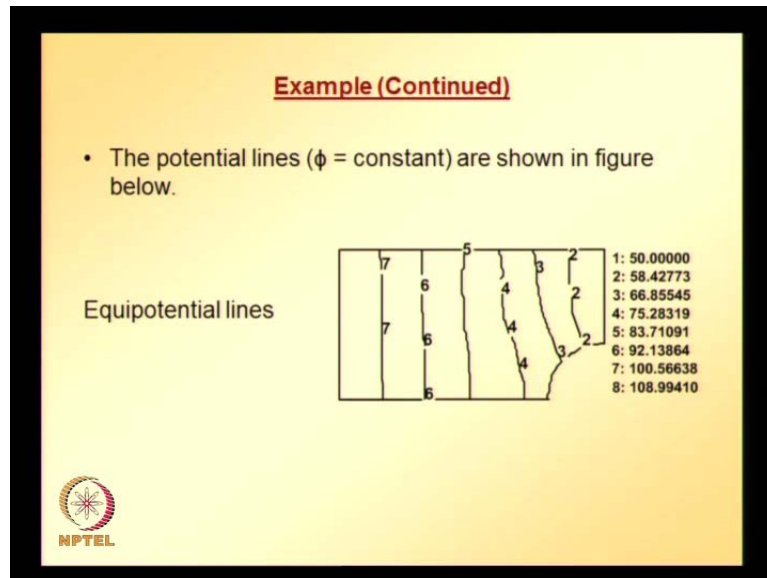
Coarse Mesh Solution

- The solution domain is divided into 24 triangular elements as shown below



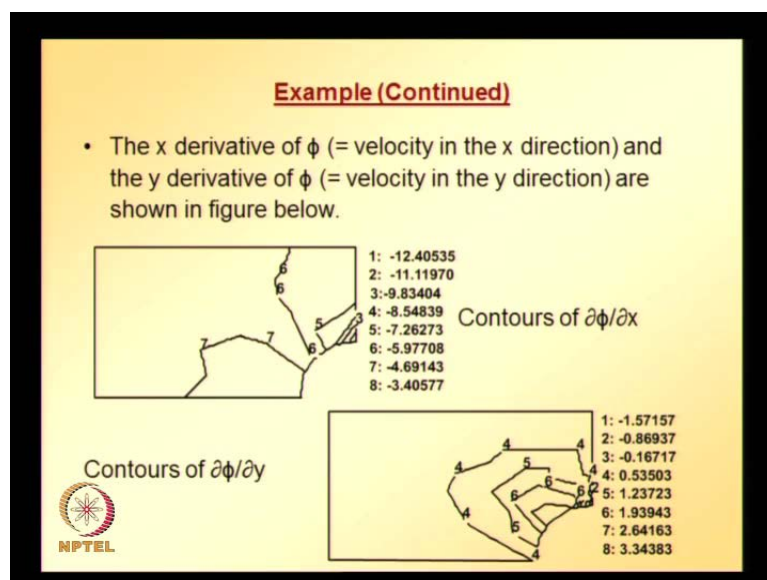
So, now let us look at, again for different mesh solution, how it looks coarse mesh solution? Solution domain is divided into same number of elements as we did for the previous formulation; that is 24 triangle elements. The discretization is shown here.

(Refer Slide Time: 33:59)



And the potential **potential** lines that is, phi is equal to constant are shown in this figure. Equipotential lines; potential line with phi is equal to constant are called equipotential lines.

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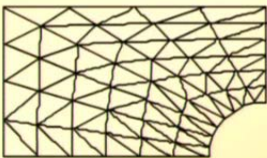

And x derivative of potential and y derivative of potential, which are related to the velocity in the x and y directions respectively are shown here. The contours and these solutions correspond to 24 triangle elements, as we did earlier; we can actually go for finer mesh.

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Example (Continued)

Fine Mesh Solution

- The solution domain is divided into 96 triangular elements as shown below

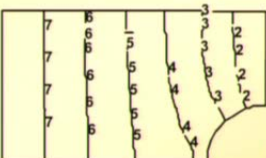
Solution domain is divided into 96 triangle elements. The mesh is shown here.

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Example (Continued)

- The potential lines ($\phi = \text{constant}$) are shown in figure below.

Equipotential lines

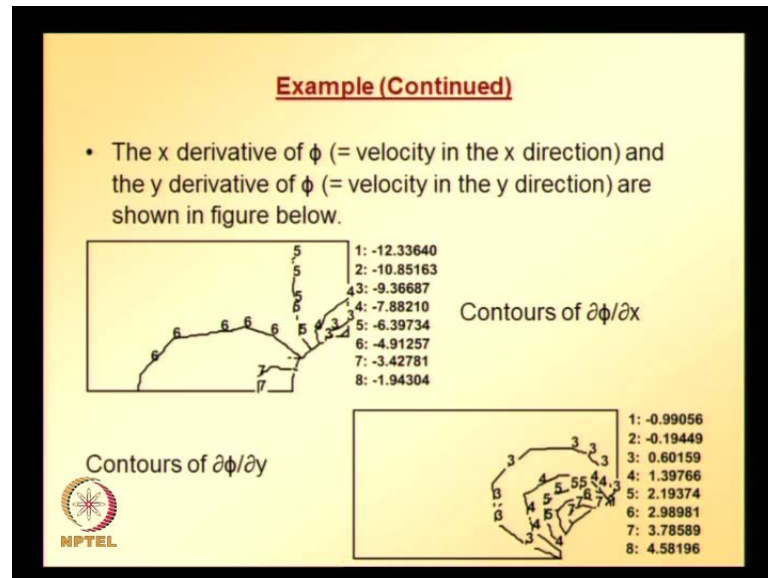


1:	50.00000
2:	58.53593
3:	67.07185
4:	75.60779
5:	84.14371
6:	92.67964
7:	101.21558
8:	109.75150



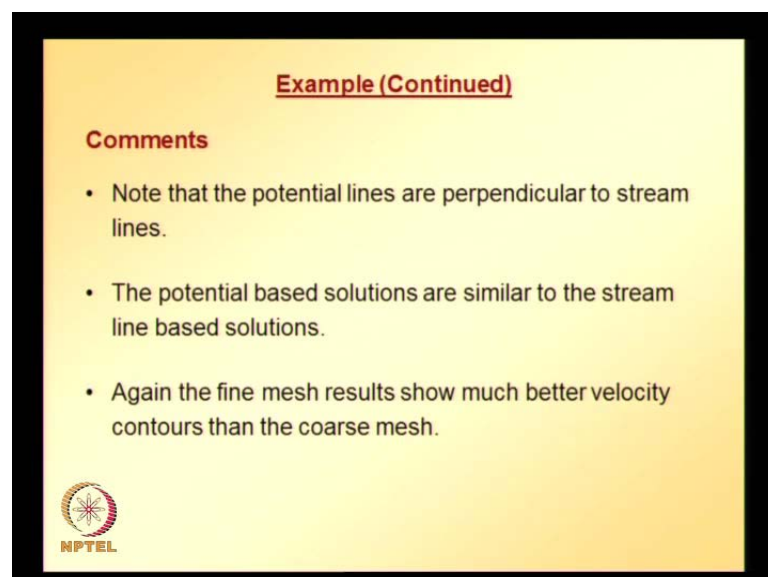
And the potential lines or equipotential lines obtained, using 94 or 96 sorry 96 triangle elements. The solution is shown here.

(Refer Slide Time: 35:19)



And contours of derivative x derivative of potential, y derivative of potential; they are shown here.

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And let us see, let us make some observations. The potential lines are perpendicular to the stream lines. The stream lines are the lines that we obtained using the previous

formulation. Stream function formulation and the potential lines are the lines that we just obtained using potential. Potential based solutions are similar to the stream line based solutions. Fine mesh results show better velocity contours than the coarse mesh.

(No audio from 36:15 to 36:25)

So, this is one application of general two dimensional boundary value problems. So, now let us look at other application, that is two dimensional steady state heat flow.

As usual only thing is, we require to know, what is the governing differential equation and what are the boundary conditions? So that, we can make a comparison with general two dimensional boundary value problems; identify the coefficients and get the finite element equations. So, consider the problem of finding temperature distribution through long chimney like structures. A slice of t units thick of such body is shown in the figure. Assuming no temperature gradient through thickness, the problem can be modeled as two dimensional problems. So, we are assuming, no gradient of temperature through the thickness.


(No audio from 37:38 to 37:47)

The governing differential equations can easily be derived using conservation of energy. And to do that, consider a differential element as shown here. So, the following notation is used. Temperature at any point is denoted with $T(x, y)$; k_x, k_y are the coefficients of thermal conductivity along x and y directions. The corresponding units are given and q_x is the heat flux in the x direction and q_y is the heat flux in the y direction and Q is heat generated for unit volume. So, using this notation and taking a differential element, we can derive the governing differential equation. So, let us consider this differential element, in which all quantities are indicated.

(Refer Slide Time: 39:13)

Steady State Heat Flow (Continued)

• Using the convention that the heat entering the control volume is positive and that leaving it is negative, the conservation of energy dictates that

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Heat flux along x direction, y direction and the heat generated and the dimensions of this differential elements are dx, dy and the coordinate system is also defined x y in the figure. So, using the convention that heat entering the **volume** control volume is positive and leaving and that is leaving is negative.


(Refer Slide Time: 39:43)

Steady State Heat Flow (Continued)

$$tq_x dy + tq_y dx + tQ dx dy - t \left(q_x + \frac{\partial q_x}{\partial x} dx \right) dy - t \left(q_y + \frac{\partial q_y}{\partial y} dy \right) dx = 0$$

By canceling terms we get

$$\frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} - Q = 0$$

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The conservation of energy dictates that, writing the equilibrium equation, we get this. And cancelling terms, which are higher order or cancelling terms, which are same in

values, we get this. And substituting the heat flux, that is q_x , q_y in terms of thermal conductivity coefficients along x and y directions.

(Refer Slide Time: 40:27)


Steady State Heat Flow (Continued)

$$q_x = -k_x \frac{\partial T}{\partial x} \qquad q_y = -k_y \frac{\partial T}{\partial y}$$

Substituting for q_x and q_y we get the following governing differential equation for steady-state heat flow in a two dimensional domain.

$$\frac{\partial}{\partial x} \left(k_x \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k_y \frac{\partial T}{\partial y} \right) + Q = 0$$

The governing differential equation is a second order partial differential equation.



We know that, q_x and q_y are related to thermal conductivity coefficients through these equations. So, substituting q_x , q_y we get the following governing differential equations for **differential equations for** steady state heat flow in a two dimensional domain. And clearly, this is a second order differential equation and it is similar to what we already looked for general two dimensional boundary value problem. So, the governing differential equation is a second order partial differential equation.


(Refer Slide Time: 41:14)

Steady State Heat Flow (Continued)

- The possible boundary conditions are as follows
 - Temperature specified, $T = T_0$ specified on part of the boundary (e.g. known temperature on the inside of a chimney).
 - Insulated boundary, no heat flow across that boundary

$$q_n \equiv k_x \frac{\partial T}{\partial x} n_x + k_y \frac{\partial T}{\partial y} n_y = 0$$

where $\mathbf{n} = n_x \mathbf{i} + n_y \mathbf{j}$ is the unit outward normal to the surface.



And the boundary conditions are as follows. Temperature specified on part of boundary, insulated boundary, no heat flow across the boundary. In that case q_n , n is nothing but outward normal to the surface; n_x , n_y are the components of the outward normal to the surface. Since this is insulated, so q_n is equal to 0. And we can also consider the other case, heat loss due to convection on the boundary. In that case, q_n is **non** going to be a nonzero value, which is going to be the function of convection coefficient and temperature of the ambience.


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Steady State Heat Flow (Continued)

- Heat loss due to convection on a boundary. This is expressed as follows

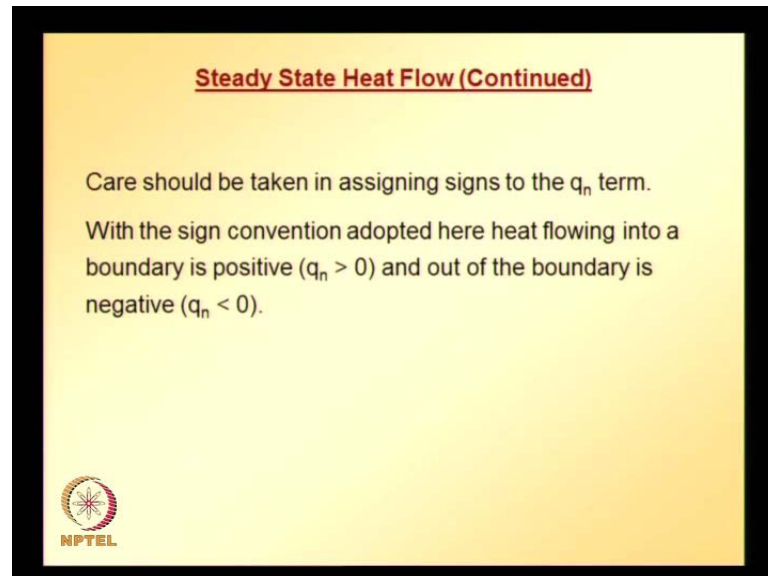
$$q_n \equiv k_x \frac{\partial T}{\partial x} n_x + k_y \frac{\partial T}{\partial y} n_y = h(T - T_\infty)$$

where h = convection coefficient and T_∞ = temperature of the surrounding (e.g. ambient air temperature surrounding a chimney).



Heat loss due to convection on a boundary, in that case q_n is given by this. So, we have two kinds of boundary conditions. Irrespective the boundary condition, what we need to do is, we need to compare the corresponding equation with the equation of general two dimensional boundary value problem and identify, what are corresponding coefficients, or corresponding parameters?

(Refer Slide Time: 42:54)




And here one more thing, care should be taken in assigning signs to q_n term. And the sign convention adopted is heat flowing into the boundary is positive and heat flowing out of the boundary is negative.

(Refer Slide Time: 43:13)

Steady State Heat Flow (Continued)

$$\frac{\partial}{\partial x} \left(k_x \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k_y \frac{\partial T}{\partial y} \right) + Q = 0$$

- This problem is clearly of the general two dimensional boundary value problem form considered in the previous chapter.
- The P term does not exist here.
- The convection boundary condition is handled by the natural boundary condition in the general form by setting $\alpha = -h$ and $\beta = hT_\infty$.

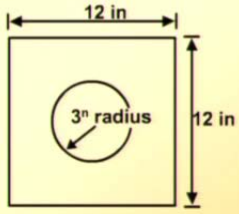


This is the governing differential equations. So, this problem is clearly of general two dimensional boundary value problem form considered in the previous chapter or previous lectures. And if we compare, only thing that we observe is, when compare with the general two dimensional boundary value problem, P term is missing. So, P does not exist here or P is equal to 0; so K P is going to 0. Convection boundary condition, if it is insulated boundary, alpha and beta are going to be 0. If it is convection boundary condition, then it is handled by taking alpha is equal to minus h; beta is equal to h times T infinity. T infinity is nothing but, temperature of the surround. So, with this understanding, we can solve a problem.

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Example

Determine temperature distribution through a square medium in which a 6 in diameter pipe carrying hot liquid at 500°F is placed at the center as shown in figure below. The outside temperature is 100°F. Assume $k_x = k_y = 0.1$. Note that because of symmetry only 1/8th of the solution domain (shown in dark shade) needs to be modeled.



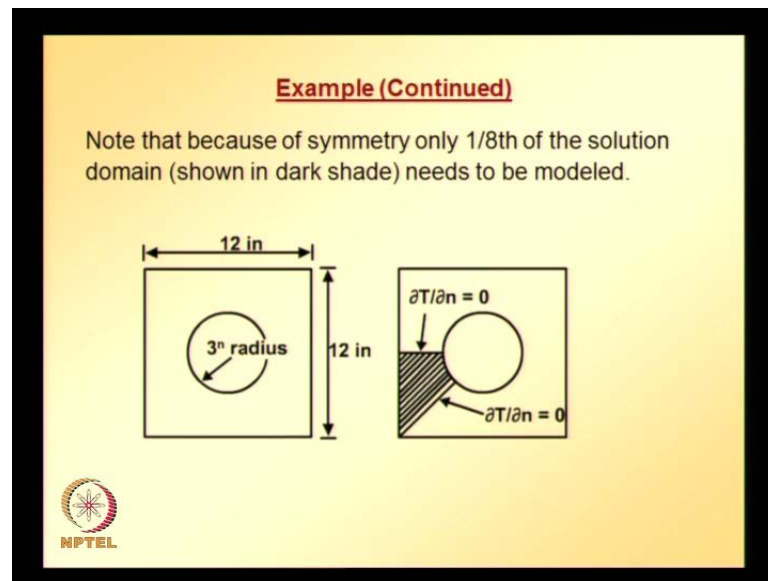
The diagram shows a square with side length 12 in. Inside the square, a circle with a radius of 3 in is centered. The square is divided into four quadrants by a vertical and a horizontal line. The top-right quadrant is shaded dark, representing 1/8th of the total square domain. The NPTEL logo is visible in the bottom left corner of the slide.

So, now let us take an example. Determine temperature distribution through a square medium with which is a 6 inches in diameter. And here the problem is given in f p s units but does not matter as long as we stick with the consistent units. And pipe carrying hot liquid at 500 Fahrenheit is placed at the center as shown in the figure below. Outside temperature is at 100 Fahrenheit and k_x , k_y thermal **thermal** conductivity coefficients along x and y directions are given here. And **and** you can see here, this problem we can take symmetry into account and because of symmetry, only one eighth of the solution domain needs to be modeled.

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So, now let us look at, one eighth model.

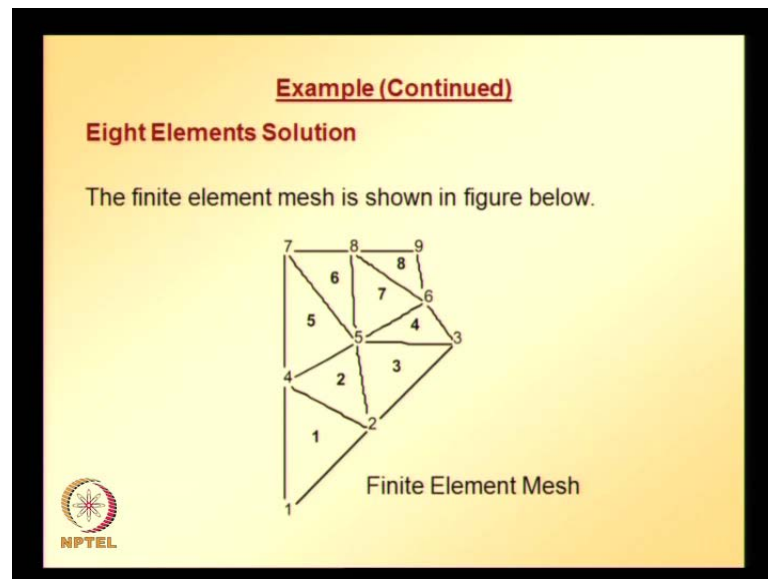
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Note because of symmetry only, one eighth of the solution domain needs to be modeled. So, this is the one eighth model. And once we take symmetry into account, the associated boundary conditions along the line of symmetry are shown there, that is partial derivative of T. T here is temperature; partial derivative of T with respect to normal is equal to 0 along their lines of symmetry.

So, as usual we need to solve the second order differential equation, partial differential equations subjected to these boundary conditions, that is T. T value at the inner diameter is 500 degrees Fahrenheit or **sorry** T value at the center is 500 degrees Fahrenheit or T value at the inner diameter and T value at the **outer diameter** outside is 100 degrees Fahrenheit. Subjected to those conditions and subjected to the natural boundary conditions along the line of symmetry, we need to solve this problem.

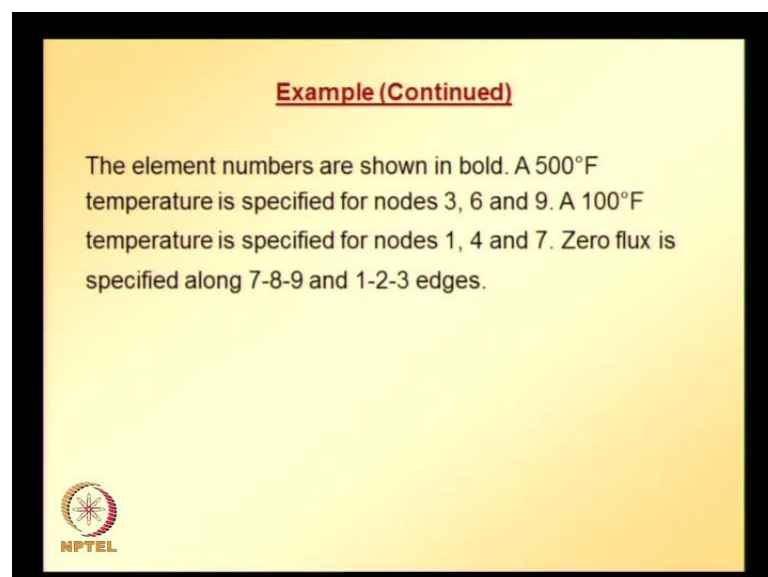
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And as usual we can proceed with finite element discretization and here eight element solution is shown. So, finite element mesh with **eight element** eight triangle elements is shown here.

(No audio from 47:11 to 47:21)

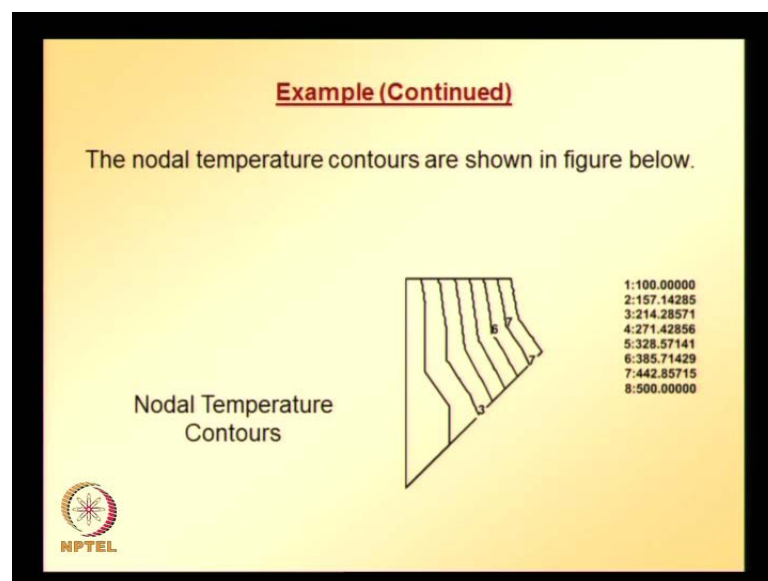
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Element numbers are shown in bold. A 500 degree Fahrenheit temperature is specified at nodes 3, 6, 9. A 100 degree Fahrenheit temperature is specified at nodes 1, 4, 7. And zero

flux is specified along 7-8-9 and 1-2-3 edges. So, let us see. So, 500 degrees Fahrenheit temperature is specified on **on** the nodes 3, 6, 9; because **they** those are the nodes, which are along the inner diameter of the **inner diameter of the** model. And temperature of 100 degrees Fahrenheit is specified along 1, 4, 7; because that is the edge, which is in contact with outside of the model. And zero flux is specified along 7, 8, 9; which is one of the line of symmetry and 1, 2, 3 which is the other line of symmetry and the nodal so with that boundary conditions.

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Nodal temperature values are computed, the model is solved. After applying the essential boundary conditions and the nodal temperature contours are shown here.

(Refer Slide Time: 49:09)

Nodal coordinates and computed temperatures

Node #	X-Coord	Y-Coord	Temperature
1	-0.6000e+01	-0.6000e+01	0.1000e+03
2	-0.4061e+01	-0.4061e+01	0.1887e+03
3	-0.2121e+01	-0.2121e+01	0.5000e+03
4	-0.6000e+01	-0.3000e+01	0.1000e+03
5	-0.4386e+01	-0.2074e+01	0.2605e+03
6	-0.2772e+01	-0.1148e+01	0.5000e+03
7	-0.6000e+01	0.0000e+00	0.1000e+03
8	-0.4500e+01	0.0000e+00	0.2803e+03
9	-0.3000e+01	0.0000e+00	0.5000e+03




And numerical values at all the nodes are given in this table. Nodes 1 to 9.

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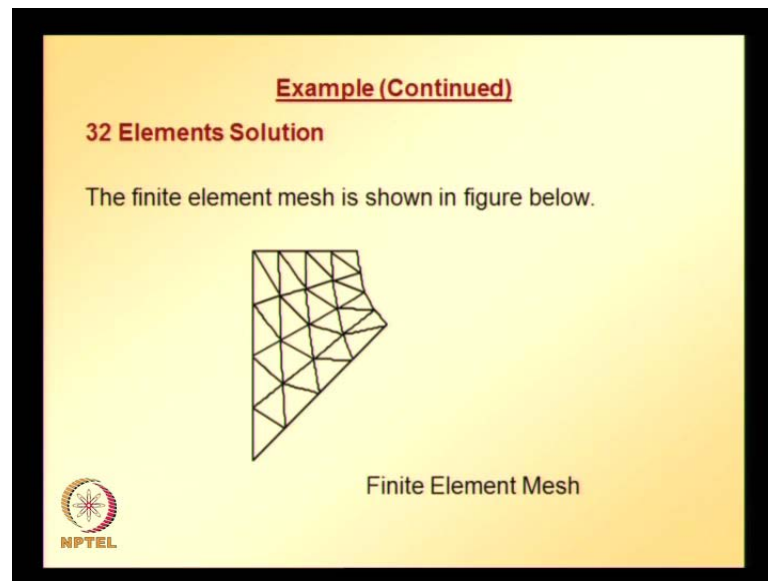
Conduction Heat Flow Per Unit Area In Each Element

Elem. #	$q_x = -k_x \partial T / \partial x$	$q_y = -k_y \partial T / \partial y$
1	0.4573e+01	-0.4578e-05
2	-0.7193e+01	-0.4790e+01
3	-0.1069e+02	-0.5362e+01
4	-0.1073e+02	-0.7168e+01
5	-0.9940e+01	0.2035e-05
6	-0.1202e+02	-0.1617e+01
7	-0.1385e+02	-0.1718e+01
8	-0.1465e+02	-0.2914e+01



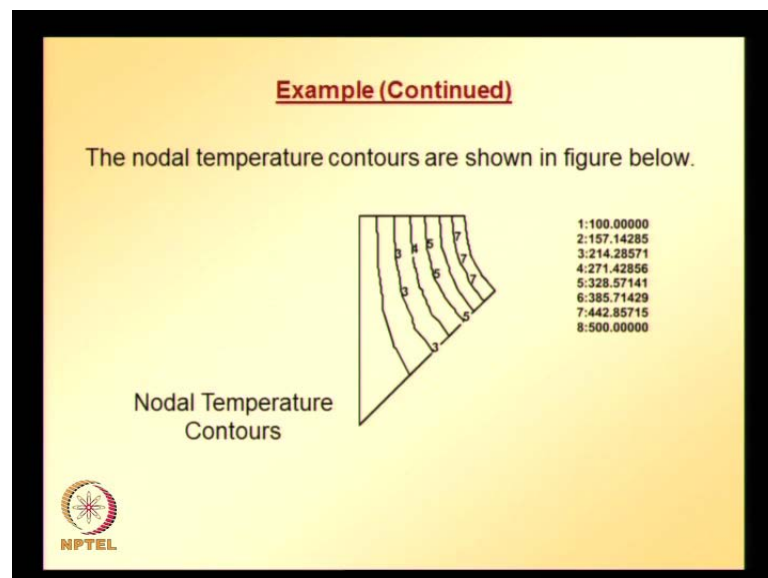
And elements 1 to 8, heat flux along x direction, y direction; they are given here.

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And now, the model is as we did for fluid flow case, a finite discretization is taken. So, 32 element solution is shown here. For that, the finite element mesh looks like this.

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And the nodal temperature contours and still model is refined further. Using 72 elements **sorry** it should be 72; instead of that, it is typed as 32; and 72 element mesh is shown here. And the nodal temperature contours, for 72 element solution is given in this figure. If we compare this solution with 8 elements, 32 elements and 72 elements, we can clearly see convergence of solution both for temperature and heat flux values.