

Heat Transfer
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Lecture - 23
The Flat Plate in Parallel Flow – Heat Transfer

In the last class we have seen how the Growth of the Hydrodynamic Boundary Layer on a Flat Plate can be analyzed analytically. Though towards the end we had to resort to numerical solution because the, we could convert the p d to an o d by combining the independent variables in by defining a stream function. However, the resulting ordinary differential equation that we obtain was non-linear and analytics solution was not possible and then we have to use numerical solutions.

However, those numerical solutions give us some very interesting results, looking at the value of the stream function, value of the dimensional stream function and its relation with v_x and v_y and noting that the v_x and v_y would be 0 at y equal 0. Due to the no slip condition we could obtain the expression for the growth of hydrodynamic boundary layer as a function of operational parameter, which is the velocity with which the fluid approaches the plate. And the length scale which is the axial length in the direction of flow and thermo physical properties which are μ and ρ .

So, therefore, the expression for δ , the thickness of the boundary layer was expressed in terms of Reynolds number, when you combine all them it becomes Reynolds number. So, the expression for, expression was obtained as δ to be equal to $5x$ by root over Re_x .

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| $\eta = \delta \sqrt{\frac{U}{\nu x}}$ | f | f' | f'' |
|--|---------|---------|---------|
| 0 | 0 | 0 | 0.332 |
| 5.0 | 3.28329 | 0.99155 | 0.01591 |

FROM TABLE
 $\eta = 5 \quad f' = 0.992$
 $\frac{\delta x}{\nu} = 0.992$
 $\eta = \delta \sqrt{\frac{U}{\nu x}}$
 $5.0 = \delta \sqrt{\frac{U}{\nu x}}$

$\delta = \frac{5.0 x}{\sqrt{Re_x}} \leftarrow \frac{x U \rho}{\mu}$
 $Re_x < 5 \times 10^5$

$\tau_w = \mu \left. \frac{\partial v_x}{\partial y} \right|_{y=0} = \mu U \sqrt{\frac{U}{\nu x}} \left. \frac{df}{d\eta^2} \right|_{\eta=0}$
 $\tau_w = 0.332 U \sqrt{\rho \mu U/x}$
 $\tau_w = \frac{0.332 \rho U^2}{\sqrt{Re_x}} \rightarrow$

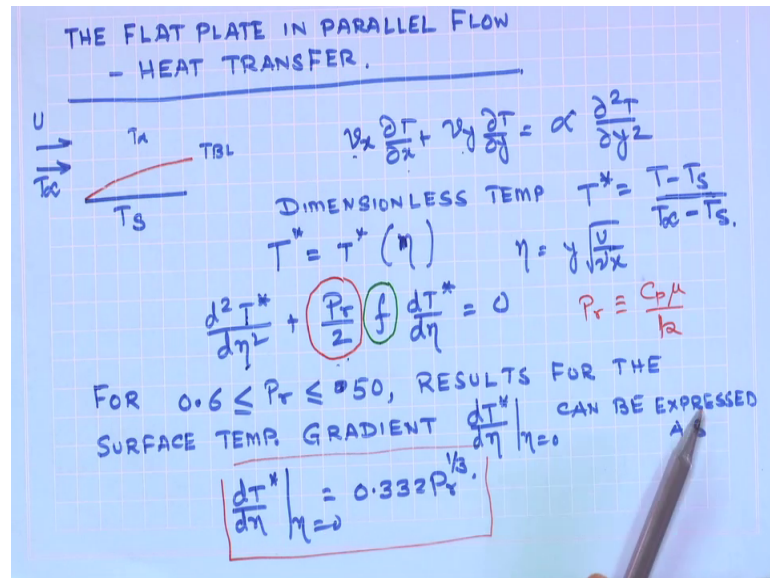
WALL SHEAR STRESS COEFFICIENT
 $C_f = \frac{\tau_w}{\frac{1}{2} \rho U^2} = \frac{0.664}{\sqrt{Re_x}}$

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For the case of shear stress, rather the wall shear stress we obtained this expression for the wall shear stress and the another quantity of engineering importance is the wall shear stress coefficient or C_f which is, which turned out to be 0.664 by Re_x . And these two rows in the numerical solution where important which gives us some idea of η equal 0 which corresponds to no slip condition.

So, from there we obtained the expression for the wall shear stress sorry, the thickness of the boundary layer. What we would now like to do is use this knowledge to analyze flat plate in parallel flow, but this time it is going to be the heat transfer case.

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So, in the heat transfer when there is going to be a difference in temperature between the plate and the fluid which is approaching. So, this is let us say is a T_s and the fluid which is approaching is at a temperature T_∞ with its velocity equals to U and then that is going to be the growth of the thermal boundary layer. It may or may not coincide, may lie above or below the hydrodynamic boundary layer.

But this is the thermal boundary layer, let us call it as TBL and the energy equation corresponding energy equation would be $v_x \frac{\partial T}{\partial x} + v_y \frac{\partial T}{\partial y} = \alpha \frac{\partial^2 T}{\partial y^2}$. So, this is the equation which now would, we would have to solve, but we already have an idea of the value of v_x and v_y at different values of x . So, our hydrodynamic solution has already given us the, from this table it has already given us what would be the value of f and f' which are required to calculate the v_x and v_y .

So, we would now go for again define a dimensionless temperature, temperature which is defined as T^* to be equals $T - T_s$ where, T_s is the temperature of the solid surface divided by T_∞ ; Where T_∞ is the temperature of the fluid outside of the thermal boundary layer, $T_\infty - T_s$. And we also assume the T is going to be a function, the T^* , this one is a function of, we understand it is function of both x and y . The value of T^* would depend where we are in terms of the axial location as well as in terms of the vertical location

So, therefore, this T^* is a function of η , the η that we have defined before. So, this is what we have assumed and when we make the substitutions, in this equation the governing equation turns out to be $d^2 T^* / d\eta^2 + Pr / 2 f dT^* / d\eta = 0$. This is a very interesting, very interesting result, I skip the steps where you substitute v_x and v_y , $\partial T / \partial x$ and $\partial T / \partial y$ and so on by assuming this T as a function of η and we understand that η is equal to $\sqrt{y} \times \sqrt{U / \nu x}$. When we do that the ordinary, the partial differential equation gets transformed into an ordinary differential equation. The point to note here is first of all the appearance of f' in the energy equation mix it, coupled with the with the hydrodynamic part of the solution.

So, the presence of f' denotes that the energy equation and the momentum equation are coupled, you would not be able to solve the equation, you would not be able to solve the energy equation unless you solve the momentum equation. So, you need to know the value of f' priori before even start solving this equation; the second one to note is the appearance of Pr Prandtl number in here. So, this Prandtl number is defined as you know as $C_p \mu / k$ where, C_p is the thermal capacity heat capacity, μ is the viscosity and k is the thermal conductivity. So, if you look at this expression and you already have, you already know from this table that: what is the value of f' for different values of η .

So, the value of η and f' are already known to you from the solution of the hydrodynamic part of the momentum transfer equation. So therefore, this f' as a function of η is known to you. So, if this is known then I should be able to numerically solved for T^* as a function of η . Since, my f' verses η is known, I can solve for T^* as a function of η provided the value of Prandtl number is known to me.

So, in order to numerically solve this equation, the only thing I need to do is specify the value of Prandtl number, the moment I specify the Prandtl number to be let us say 0.7. Then I should be able to numerically solve it because, if dependence of f' on η is already known to me and therefore, I should be able to obtain T^* as a function of η for a specific value of Prandtl number. So, that is the only thing we have need to realize in here that, T^* as a function of η can be obtained at a specific value of Prandtl number.

So, that is the only thing which is which is which is remaining here, which is to be clarified here and it has been found that for this range of Prandtl number 0.6 to 50, 50 results for the surface temperature gradient which is dT^* / dy at $y = 0$, can be expressed as dT^* / dy at $y = 0$ to be equals $0.332 Pr$ to the power one third. This is very interesting very important result, as we have seen we need to have, we need to know the numerical value of Prandtl number to solve it.

So, as I different values of Prandtl number and start solving this equation 0.5, 0.75, 1, 1.25, 10, 20 so and so on. What is interesting, the observation that was obtained is that dT^* / dy at $y = 0$, which you can obtain from the solution of this, provided you assume a value of Prandtl number, can be expressed as a function of Prandtl number over a wide range of Prandtl number.

So, the observation once again is that for, when Prandtl number lies in this range the surface temperature gradient; that means, at $y = 0$ dT^* / dy can be expressed as a function of Prandtl number in this functional form. Now, why is it important, why I am at all interested in dT^* / dy at $y = 0$. The same reason I was interested in trying to find out what is the velocity gradient at $y = 0$; that means, at $y = 0$ because velocity gradient at $y = 0$. That means, on the surface, at this at the at the solid surface gives me what is the force exerted by the moving fluid on the solid, in other words I am, I was defining an important transport coefficient or defining transport phenomena related to force exerted by a moving fluid on a, on a stationary solid.

Similarly, dT^* / dy at $y = 0$ that means, the velocity gradient at $y = 0$ should give me some idea of the heat transfer process that is taking place in the situation, where as cold fluid flows over a hot plate. So, the temperature gradient at the interface should give us an idea of what is the convective heat transfer coefficient in this process and we understand the entire study of convection is try to find out what is the convective heat transfer coefficient. What is the relation or correlation that can be used to obtain the value of the heat transfer coefficient, convective heat transfer coefficient and in order to do that, the knowledge of dT^* / dy at $y = 0$ will play, does play a very significant role.

So, next what we are going to see is how this $dT/d\eta$ at $\eta = 0$ can be related to the convective heat transfer coefficient, but to do that I need to have an idea of what is the Prandtl number. So, what we have seen is that for a large range of Prandtl number, the temperature gradient at the interface can be expressed as a function of, can be fitted to the variation in Prandtl number. So, $dT/d\eta$ at $\eta = 0$ can be correlated to Prandtl number to the power something with a constant in front of it.

So, I will start writing it and then I will show you how it would give us an expression for the heat transfer coefficient. So, let us start with that, so this $dT/d\eta$ at $\eta = 0$ is equal to $0.332 Pr^{1/3}$ this is our starting point, the heat transfer coefficient h suffix s x.

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The image shows a handwritten derivation on a grid background. At the top, the equation $\left. \frac{dT^*}{d\eta} \right|_{\eta=0} = 0.332 Pr^{1/3}$ is circled in red. Below it, the heat transfer coefficient h_x is derived from Newton's law of cooling: $h_x = \frac{q''_s}{T_s - T_\infty} = -\frac{1}{T_s - T_\infty} k \left. \frac{\partial T}{\partial y} \right|_{y=0}$. This is then expressed in terms of the similarity variable η : $h_x = -\frac{T_\infty - T_s}{T_s - T_\infty} k \left. \frac{\partial T^*}{\partial y} \right|_{y=0}$. The similarity variable is defined as $y^* = y/L$. The derivation continues to show $h_x = k \left(\frac{y}{x} \right)^{1/2} \left. \frac{dT^*}{d\eta} \right|_{\eta=0}$ and $\frac{h_x L}{k} = \left. \frac{\partial T^*}{\partial y^*} \right|_{y^*=0}$. Finally, the Nusselt number is defined as $Nu_x = \frac{h_x x}{k} = 0.332 Re_x^{1/2} Pr^{1/3}$ for $0.6 \leq Pr \leq 50$. A note indicates that $Nu_x = \left. \frac{\partial T^*}{\partial y^*} \right|_{y^*=0}$ is the Nusselt number.

That means, it is a local value of the heat transfer coefficient can be defined as, from Newton's law of cooling $q''_s = h_x (T_s - T_\infty)$. T_s is the temperature of the solid surface, T_∞ is the temperature of the liquid. So, if this is the flux the Newton's law of cooling connects the convective heat transfer coefficient with q''_s is in this way. So, I can write it as $h_x = \frac{q''_s}{T_s - T_\infty}$ times k , dT/dy at $y = 0$.

So, this is Newton's law of cooling and this is substituting Fourier's law of conduction for the surface heat flux because, as we have discussed many times before on the surface the fluid molecules are immovable. And therefore, the transport of heat through this

immovable layer of molecules of the fluid can take place, does do take dose take place only by conduction and that is why I am substituting the Newton's law of conduction in here. So, this now can be modified little bit, it is $T_{\infty} - T_s$ by $T_s - T_{\infty}$, I am trying to use the definition of T^* . Where the definition of T^* is $T - T_s$ by $T_{\infty} - T_s$ this times $k \frac{dT^*}{dy}$ at $y = 0$.

So, this to cancel, the negative sign will disappear so, my h_x is k by L thermal conductivity by L because, I am converting this by into dimensionless one as well, $\frac{dT^*}{dy^*}$ at $y^* = 0$. So, y^* is simply defined as y by L , where L is the length of the entire length of the plate over which this heat transfer is taking place. Look at this equation carefully, what you having here is, a if I bring this L on the other side and k over here and of shoot of this equation is h_x local value of heat transfer coefficient.; x denotes the location L by k is equal to $\frac{dT^*}{dy^*}$ at $y^* = 0$. What is this, $h L$ by k this $h L$ by k is dimensionless and it is denoted by Nusselt number, Nu local value that is why I put the substitute x .

So, this is Nusselt number which denotes convection is equal to $\frac{dT^*}{dy^*}$ at $y^* = 0$. This is an unique definition of Nusselt number, which simply tells you that the Nusselt number is nothing, but the dimensionless temperature gradient at the interface, Nusselt number is $\frac{dT^*}{dy^*}$ at $y^* = 0$. So, the proper definition of Nusselt number is not simply $h L$ by k , the correct way to express Nusselt number is it is physically the dimensionless temperature gradient at the solid liquid, in solid fluid interface.

So, that is one way to look at the genesis or the physical significance of Nusselt number. So, now, let us proceed with this and see that with our knowledge of $\frac{dT^*}{d\eta}$ at $\eta = 0$, can somehow be plugged into this expression to obtain a more compact, more useful expression for Nusselt number in laminar flow for flow over a flat plate.

So, I continue with this once again and what I do is this h_x is $k U_{\infty}$ by νx to the power half, $\frac{dT^*}{d\eta}$ at $\eta = 0$. So, I convert this y to η and that is why these terms do appear in front of it and since T^* is a function only of η . So, therefore, the partial sign can be replaced by ordinary differential, ordinary differential signs. So, T^* is a function only of η and this I have already seen; what is the expression for this from our previous study.

So, now I am going to write this, bring this over here take the x to you other side and write the final expression for Nusselt number, the local value of Nusselt number as $h x$ local value of the convective heat transfer coefficient; x by k to be equals 0.332 Reynolds number based on local Reynolds number to the power half into Prandtl to the power one third. The limitations is this, the Prandtl number has to lie between 0.6 and 50 which was, which was used to obtain this expression.

So, what I see here then, that a compact expression for Nusselt number can be obtained by simply invoking the result over this, and realizing the T^* is a function of η . And this expression is going to be extremely useful for solving or for solving or for analyzing situations in which we have flow over a flat plate where the temperature of the plate and the temperature of the fluid which is flowing over it is, they are different.

So, we are going to have convective heat transfer taking place and this is the first time that you see an expression for h is obtained in terms of Reynolds number which characterizes the flow, which brings in the which brings in the hydrodynamic part of it into the expression. And a function of Prandtl number which is $C_p \mu$ by k and it gives some idea of the convection and conduction the relative importance of convection conduction, momentum diffusivity, thermal diffusivity into our discussion.

This gives me the average, this gives me the local value of the heat transfer coefficient, like before we have obtain $C_f x$ that is the friction coefficient at a specific x . The expression that I have obtain for heat transfer it is Nu_x or $h x$ the convective heat transfer coefficient at a specific x . But as an engineer, you are probably more interested in finding out; what is the average value of the heat transfer coefficient or what is the average value of the friction coefficient over the entire plate surface. You do not want to know most likely what is the value of the heat transfer coefficient at a specific location, you rather try to find out what is the average value of heat transfer coefficient.

Because, that would give you some idea of what is the total amount of heat transfer if there is a way to evaluate the total the average value of heat transfer coefficient, then the total heat transfer coefficient can simply you obtained as $h \Delta T$ ok, you do not have to use $h x$. So, our next small item which is remaining is, how to obtain the average value of heat transfer coefficient from the expression of the local value of heat transfer

coefficient. So, let us do that and see how, what would what would happen in this case. So, the average value of a of this the average value.

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$$\bar{C}_f = \frac{\bar{\tau}_s}{\rho U_\infty^2 / 2}$$

$$\bar{\tau}_s = \frac{1}{x} \int_0^x \tau_s dx = \frac{1}{x} \int_0^x 0.332 \frac{\rho U_\infty^2}{\sqrt{Re_x}} dx$$

$$\bar{C}_f = 1.328 Re_x^{-1/2}$$

$$\bar{h}_x = \frac{1}{x} \int_0^x h_x dx = 0.332 \left(\frac{k}{x}\right) Pr^{1/3} \left(\frac{U_\infty}{\nu}\right)^{1/2} \int_0^x \frac{dx}{x^{1/2}}$$

$$\bar{h}_x = 2 h_x$$

$$Nu = \frac{\bar{h}_x x}{k} = 0.664 Re_x^{1/2} Pr^{1/3}$$

$$\bar{C}_f = 2 C_{f_x}, \bar{h}_x = 2 h_x$$

So, this is about average coefficients. So, C_f and I put a bar over it to say that this is the average value, must be tau average, tau surface average by rho U infinity square by 2 that is the definition of C_f . But here I am using the average one so, what is tau this thing then, the average value of the shear stress is from 0 to x tau s x d x. So, this can be written as 1 by x, 0 to x 0.332 rho u square by root over Re x d x. This tau x is simply I am using the expression of tau x which I have obtained before and once you do this integration, you would see that C_f , the average value of C_f is going to be 1.328 Reynolds to the power minus half ok.

And in a similar to the, momentum transfer the value of, the average value of h_x would simply be 1 by x 0 to x, that is the definition of lengthwise average $h_x dx$, which would be putting the expression over here k by x. Prandtl number is a constant so it goes out of the integration sign, U infinity by nu which is to the power half out constant, out of the integration sign 0 to x d x by x to the power half.

So, once you do this your h, the average heat transfer coefficient is going to be equal to twice of the local heat transfer coefficient. Therefore, your average value of Nusselt number, the expression for Nusselt number would be average h times x by k and this would turn out to be 0.664 Re x to the power half Prandtl to the power one third. So, this

gives you the average value which is more important in realistically, in engineering situations then the local value of the heat transfer coefficient. So, what we say then is C_f average is equals twice C_f local and h average is twice local.

So, we have two equations then, one equation is for the local value of the Nusselt number or local value of the heat transfer coefficient and simply finding out the average. And the average in is done by integrating it over a certain length and dividing it by the length, which is the standard definition for average based on the length. What you would see is this the average, the local value and the average value differs only by a factor of, only by a factor of 2 and that is why the, this is the relation between the average and the local.

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$T_s - T_x$ $\partial y |_{y=0}$
 $h_x = \frac{k}{L} \left. \frac{\partial T^*}{\partial y^*} \right|_{y^*=0}$ $y^* = \frac{y}{L}$
 $h_x = k \left(\frac{y}{x} \right)^{1/2} \left. \frac{\partial T^*}{\partial \eta} \right|_{\eta=0}$ $\rightarrow \frac{h_x L}{k} = \left. \frac{\partial T^*}{\partial y^*} \right|_{y^*=0}$
 $Nu_x = \frac{h_x x}{k} = 0.332 Re_x^{1/2} Pr^{1/3}$ $Nu_x = \left. \frac{\partial T^*}{\partial y^*} \right|_{y^*=0}$
 (0.6 ≤ Pr ≤ 50) (NUSSELT No.)
 $h = 2 h_x$
 $Nu = \frac{\bar{h} x}{k} = 0.664 Re_x^{1/2} Pr^{1/3}$
 $\bar{C}_f = 2 C_{f,x}$ $\bar{h} = 2 h_x$ 0.6 ≤ Pr ≤ 50

Which is the same as that between the shear stress coefficient the local at the average, but we also need to know, that need to remind you once again that this expression is also valid for a Prandtl number range: between 0.6 to 50. So, what we have done in this class, in this exercise and the previous class is to show you that to why it is important to study what is happening inside the boundary layer, inside the thermal boundary layer and inside the momentum boundary layer. Because all the phenomena is concentrated in a region very close to the solid surface, in one case its momentum transfer and in the other case it is heat transfer. The momentum transfer case can be handled by defining a combination variable eta and the dimensionless stream function f.

When I do that what I would have, what I would obtained for the case of hydrodynamic boundary layer is an ordinary differential equation, non-linear higher order differential equation containing f being the dependent variable and η being independent variable. That equation was solved numerically to obtain a table containing what will be the value of f for different values of η , in using the values and the concepts we obtained compact relations for the growth of the boundary layer that is δ as a function of x . And you have also obtained what is going to be the shear stress coefficient as a function of x properties and the imposed flow condition.

The equation for energy transfer when we non-dimensionalized it, everything that take the temperature dimensionless temperature T^* was a function of η , which is the combination variable. But a term f appears their denoting or underscoring the coupling between the fluid flow and heat transfer, Prandtl number also automatically appears in the governing equation. This equation can be numerically solved because, I already have the numerical solution of f versus η , but in order to solve it numerically I first need to need to assign a value to the Prandtl number, any number any realistic number to Prandtl number and solve for this, assign another value and solve it again.

When that was done for a large range of Prandtl number it was found that $d T^* / d \eta$ at $\eta = 0$; that means, the temperature gradient at the liquid solid interface can be expressed as a function of Prandtl number. Purely a fitting between the temperature gradient and Prandtl number, with that we have seen how to express the heat transfer coefficient using Newton's law of cooling and then Fourier's law. And obtain fundamental definition of Nusselt number to be the dimensionless temperature gradient at the interface. We proceed with that utilizing the values that we have obtained and we got an expression for local Nusselt number, as a function of Reynolds number and as a function of Prandtl number, That $0.332 \text{ Reynolds}^{1/2}$ to the power something, Prandtl to the power something valid within the range of Prandtl number.

Then we realize that it would probably be easier to work with the average quantities, the average value of heat transfer coefficient or the average value of the shear stress coefficient. So, what we have done, we have done length average in I have finding out the average of these quantities h or C_f for the entire length of the plate over which the flow and heat transfer list taking place. So, what we got is that the average values are always twice the local value. So, we have now obtained and expression for Nusselt

number, average Nusselt number when flow and heat transfer takes place over a flat plate.

But remember all these are limited to laminar flow, laminar flow conditions prevailing over the solid plate. What happens in turbulent is something very interesting, something entirely different and you would expect that more heat transfer is, more heat transfer is possible when you create turbulent conditions inside the thermal boundary layer. Which we will discuss in the next class, but today's class is all about laminar flow and heat transfer when flow takes place over a parallel plate.