

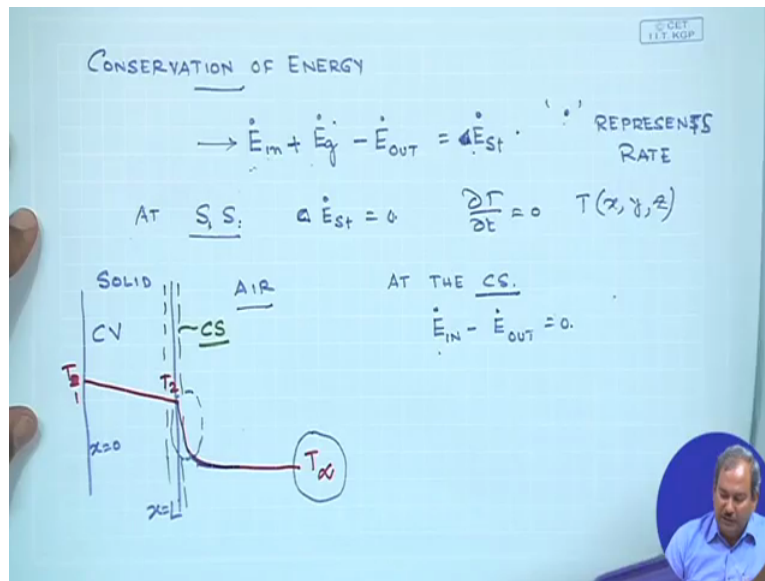
**Transport Phenomena.**  
**Professor Sunando Dasgupta.**  
**Department of Chemical Engineering.**  
**Indian Institute of Technology, Kharagpur.**  
**Lecture-31.**  
**Heat Transfer Basics (Continued).**

So we will start with our 2<sup>nd</sup> part of the heat transfer and as we have seen what are the basic rates, basic modes by which heat can get transferred and the special considerations that are to be included for the case of convective transport process which can also be divided into free convection where there is no external agency forcing the fluid to flow over the subject, over the solid object from which heat transfer is taking place. And we have a case of force convection.

So the fundamental study of heat transfer starts with our identification of the conservation of energy principle. So for any object we 1<sup>st</sup> have to realise, 1<sup>st</sup> have to appreciate the conservation of energy principle and that conservation of energy principle when expressed in terms of mathematical relations could give rise to the complete equation describing the energy transport from or to the object. And energy conversion inside the object if that is relevant as a result of which, the total energy content of this object will change.

Energy can also be changed if you have, if you include the work done to or by the system. So work done on the system will enhance its energy and work done by the system will reduce the total energy content of the object. So all these , all these considerations must be taken into account while writing the conservation equation for any system. So we, but we are going to start with the simplest, simple conservation equation to begin with where we will at the moment not consider any work effects.

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So all our consideration is the tangible form of energy which comes in or out, to or off the control volume as a result of conduction or convection. So we are, we start with the conservation of energy relation which simply says that the rate, the dot represents essentially the rate, the time rate. So rate of energy coming into the system by either convection, conduction or a combination of both +  $\dot{E}_g$  where  $\dot{E}_g$  generates, denotes the generation of energy inside the system.

This generation could be due to the ohmic heating or due to the presence of a nuclear source in the control volume -  $\dot{E}_{out}$  where this is the rate of energy which is going out of the control volume as, again as a result of conduction and/or convection. And the sum total of all these 3 is  $\dot{E}_{st}$ . So the rate of change of stored energy in the system.

So when we talk about steady-state, SS steady-state, the  $\dot{E}_{st}$ , the time, the rate of energy stored is going to be equal to 0, that means there would not be any net storage of energy as a result of these few processes, the temperatures could be a function of space coordinates, that means the temperature could be different at different points in the control volume but  $\frac{\partial T}{\partial t}$ , temperature by dell time, that is the temperature will not be a function of time, so  $\frac{\partial T}{\partial t}$  would be equal to 0.

However we understand that  $T$  could be a function of the 3 space coordinates. And if you think of a solid surface and air in contact with it, and let us assume that the air temperature is  $T_{\infty}$  at a point far from the, far from the solid edge is  $T_{\infty}$ . The inside temperature is  $T_1$  at some location, let us say this is at  $x$  equal to 0 and this is at  $x$  equal to  $L$ , the

temperature is  $T_2$ . So you would see later on that in absence of any heat generation in the solid, the temperature profile will be linear but then it would sharply reduce and asymptotically merge with the temperature  $T_\infty$ , that is the temperature at a point far from it.

So if you consider this region which is very close to the solid surface in the fluid where the temperature sharply changes from  $T_2$  to that of  $T_\infty$  far from the plate. And yet you see that the temperature asymptotically approaches the free stream temperature, so this by analogy with our previous discussion, you can clearly see that this essentially establishes the concept of a thermal boundary layer.

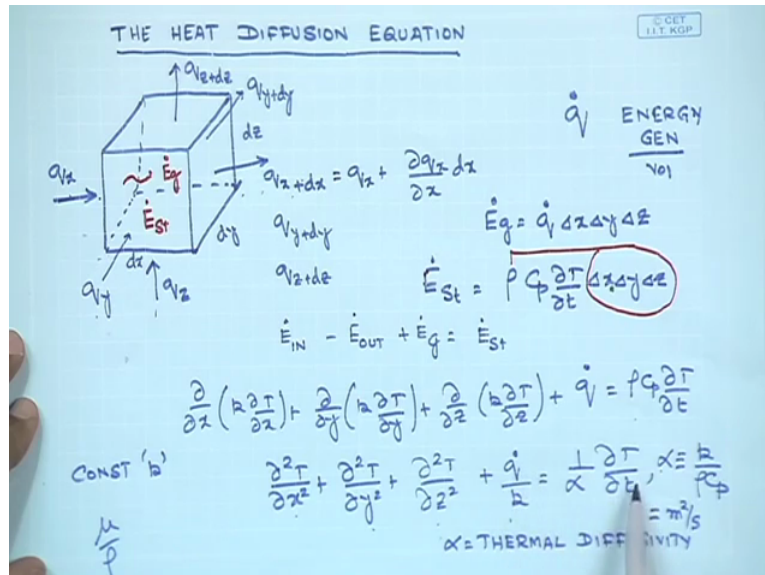
So same way as in velocity boundary layer, where the velocity increases as we move away from the flat plate and reaches the value equal to the free stream velocity, the same way the temperature the temperature changes from that of the solid and it gradually approaches to the bulk temperature or  $T_\infty$ , that is the temperature at a point far outside of the effect of the solid present in contact with the fluid. So the distance over which this temperature variation takes place will in a proper way, and a slightly modified way will be expressed as the extent in which the heat transfer is taking place.

Because outside of that the temperature does not vary with distance anymore. So no heat transfer, convective or conductive is taking place in the region where the temperature reaches  $T_\infty$ . So ultimately we will try to bring in the same concept as that of velocity or hydrodynamic boundary layer in the case of heat transfer as well in the form of thermal boundary layer. But coming back to the problem that we were discussing about, so if this is my control volume, this expression applies to the control volume, so  $E \cdot_{in}$ , the amount of energy which comes from the left of the figure +  $E \cdot G$ , if this is a nuclear fuel element, then some amount of heat would be generated.

If it is not, the  $E \cdot G$  would be equal to 0 and  $E \cdot_{out}$  is the energy which goes out to the air surrounding it as a result of which the net energy content of the control volume may change and if it does not change, then we, what we have is a steady-state situation. So in absence of energy generation and if it is a steady-state system,  $E \cdot_{in}$  must be equal to  $E \cdot_{out}$ . Now what is control surface? The dotted line which I have drawn can be thought of a control surface which by the definition does not have any mass of its own.

So if it does not have any mass of its own, then it cannot store any energy and no energy is generated in it. So at the control surface  $\dot{E}_{in}$  must always be equal to  $\dot{E}_{out}$ . So this is the, these are the some of the concepts which we would use in our subsequent analysis.

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The next part that we are going to do is the heat diffusion equation, which I am sure you have done before but very quickly it tells me that if I now, if I have a control volume of size  $dX$ ,  $dY$  and  $dZ$  and if we consider only conduction, let us say  $Q_x$  is the amount of heat coming in,  $Q_x$  is the flux of heat which is coming through the surface, through the  $X$  face and  $Q_x + dQ_x$  is going out,  $Q_z$  is coming in through the  $Z$  face,  $Q_z + dQ_z$  is going out of the  $Z$  face,  $Q_y$  is coming in and  $Q_y + dQ_y$  is going out.

And  $Q_x + dQ_x$ , that is can simply be expressed in the Taylor's series expansion, the same way we have done before. So this is for  $Q_x + dQ_x$ , same way you can write for  $Q_y + dQ_y$  and  $Q_z + dQ_z$ . So that will simply be, you can simply write it and if we assume that  $\dot{q}$  is the energy generated by some means would be nuclear per unit volume inside this, then the total amount of heat generation in this control volume would simply be equal to  $dell X dell Y dell Z$  and  $\dot{E}_g$ .

So this will be equals  $\dot{E}_g$ , energy generated and  $\dot{E}_{st}$  would be equal to  $\rho C_p dell T$  by  $dell time dell X dell Y dell Z$ . If you see this row times  $dell X, dell Y, dell Z$  is simply equals to the mass of the control volume. So  $\rho C_p dell T$  by  $dell time$  is the time rate of change of energy stored in the system. So you can see it any textbooks, when you write this in the, write the balance equation as  $\dot{E}_{in} - \dot{E}_{out} + \dot{E}_g = \dot{E}_{st}$ .

And put in the, plug-in the values of  $Q_x$ ,  $Q_x + \frac{dQ_x}{dx}$ ,  $Q_y + \frac{dQ_y}{dy}$  and so on. And you simplify, what we would get is  $\frac{d}{dx} \left( K \frac{dT}{dx} \right) + \frac{d}{dy} \left( K \frac{dT}{dy} \right) + \frac{d}{dz} \left( K \frac{dT}{dz} \right) + \dot{Q}$  which is the energy generated per unit volume is equal to  $\rho C_p \frac{dT}{dt}$  which is the energy stored per unit volume. This is the amount of energy, net conductive heat flux into the control volume from the X direction, from the Y direction and from the Z direction.

So if we assume that it is a constant K case, the thermal conductivity of the system is constant, then K can simply be taken out of it and what you get is  $\frac{d^2 T}{dx^2} + \frac{d^2 T}{dy^2} + \frac{d^2 T}{dz^2} + \frac{\dot{Q}}{K} = \frac{1}{\alpha} \frac{dT}{dt}$ , where Alpha is simply defined as  $\frac{K}{\rho C_p}$ . So there is a reason why Alpha is expressed in here instead of  $\frac{\rho C_p}{K}$ , I simply write one by Alpha because Alpha has units of metre square per second.

So this is called, the Alpha is called the thermal diffusivity of the system. When you, when we will talk about, when we spoke about the momentum diffusivity which was denoted by  $\frac{\mu}{\rho}$ , that the unit of  $\frac{\mu}{\rho}$  which was the kinematic viscosity, that is also, its unit was also metre square per second. So  $\frac{\mu}{\rho}$  is sometimes called as the momentum diffusivity.

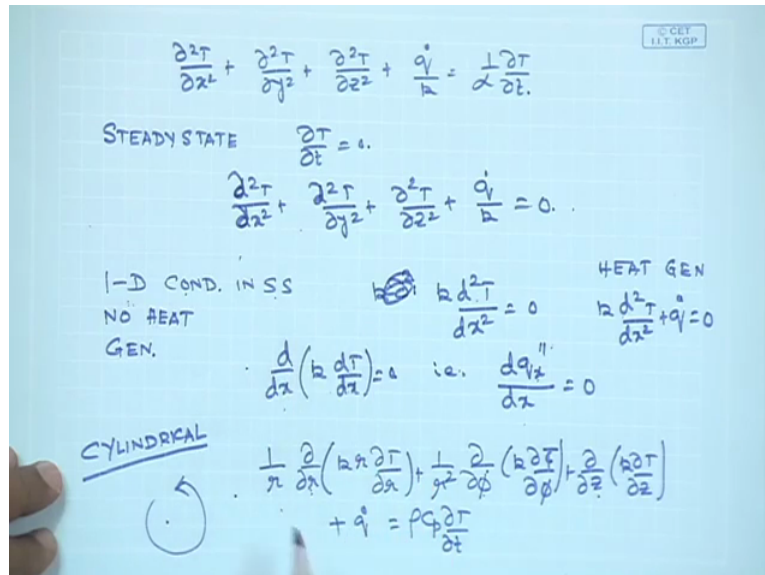
So these 2, the momentum diffusivity or the thermal diffusivity having the same units as metre square per second, it came from the concept of diffusion coefficient which we would discuss when we will talk about the Newton's, when we will talk about the mass transfer process, the convective mass transfer process which is fixed for 1<sup>st</sup> law of diffusion which connects the amount of diffusive transport of mass is equal to  $-dAB$ , where  $dAB$  is a diffusion coefficient of A in B times  $\frac{dC_A}{dx}$ , if it is a one-dimensional conduction, one-dimensional mass transfers process.

And this  $dAB$  the mass diffusivity has units of metre square per second. So by simply rearranging the terms in the form of  $\frac{\mu}{\rho}$  in momentum transfer and  $\frac{K}{\rho C_p}$  in heat transfer and  $dAB$ , since all of them have the same units as metre square per second, so borrowing the term from mass transfer  $\frac{K}{\rho C_p}$  is called the momentum diffusive it in,  $\frac{K}{\rho C_p}$  is called the thermal diffusivity and  $dAB$  is simply called the diffusivity or mass diffusivity.

So that is why that the equation of energy is generally expressed not in terms of K, Rho or CP but the combination variable which is in which is expressed, which is which is called the

thermal diffusivity. So this is the equation what you would get for conduct of transport of heat only for conduction, only conduct of transport of heat where the temperature can vary in X, Y and Z direction. This is the heat regenerated per unit volume and this entirely, the entire thing is the total amount of energy stored or lost in the control volume per unit volume.

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So this is the starting point. And if I write this equation one more time, what we have is  $\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} + \frac{\dot{q}}{k} = \frac{1}{\alpha} \frac{\partial T}{\partial t}$ . So if it is, if it is a steady-state, if it is a steady-state, then  $\frac{\partial T}{\partial t}$  would simply be equal to 0 and what you will get is  $\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} + \frac{\dot{q}}{k} = 0$ .

If it is 1-D conduction in steady-state and let us see the temperature is, the heat is only getting transported in the X direction, then what you would get is  $k \frac{d^2 T}{dx^2} = 0$  or you can simply write as  $\frac{d}{dx} (k \frac{dT}{dx}) = 0$  or in other words what you can write is, what you can see is  $\frac{dq_x}{dx} = 0$  since  $k \frac{dT}{dx}$  is equal to  $-q_x$ .

So the, this analysis simply tells me that for one-dimensional conduction only case at steady-state,  $k \frac{d^2 T}{dx^2} = 0$  which directly follows from the previous equation and obviously there is no heat generation. So if there is no heat generation, this will be the form, if you have heat generation in the system, then you are going to have  $k \frac{d^2 T}{dx^2} + \dot{q} = 0$ . And this simply can be written as  $\frac{d}{dx} (k \frac{dT}{dx}) + \dot{q} = 0$ .

$\dot{Q}_x$ , so this is the heat flux, the double prime denotes the flux, so  $\dot{Q}_x = -k \frac{dT}{dx}$  equal to 0.

So these expressions are very common and would give rise to several other simplifying cases. So this is in Cartesian coordinate system, so if you write it in cylindrical coordinate systems, then you should be able to see it from the text, you would, you would see that for the cylindrical coordinate system,  $\frac{1}{R} \frac{d}{dR} \left( k R \frac{dT}{dR} \right) + \frac{1}{R^2} \frac{d}{d\phi} \left( R^2 \frac{dT}{d\phi} \right) + \frac{d}{dz} \left( k \frac{dT}{dz} \right) = 0$ , there was no new concept involved in the cylindrical coordinate expression, it is only that when you transform from Cartesian coordinate system to cylindrical coordinate system, you express everything in terms of  $R$ ,  $\phi$  and  $Z$ . Where  $R$  is the radial distance,  $\phi$  is this direction and  $Z$  is the axial direction.

So when you transfer the coordinate, the form of the equation, the form of the equation gets slightly more complicated but fundamentally it still remains the same. Similarly in the case of spherical coordinate systems, in new textbooks he would be able to see which form the fundamental equation takes when you transform it from Cartesian coordinate system to the spherical coordinate systems, that means in terms of  $R$ ,  $\theta$  and  $\phi$ . So depending on which coordinate system you are using you have to choose your conduction equation.

I have derived the equation in Cartesian coordinate system but in your text you would see the same equations are given in cylindrical coordinate systems as well as in spherical coordinate system. In the Cartesian coordinate system,  $\dot{Q}_x$ , the conductive heat transport in the  $X$  direction is equal to  $-k \frac{dT}{dx}$  or  $-k \frac{dT}{dx}$ .  $\dot{Q}_y$  is equal to  $-k \frac{dT}{dy}$  and so on.

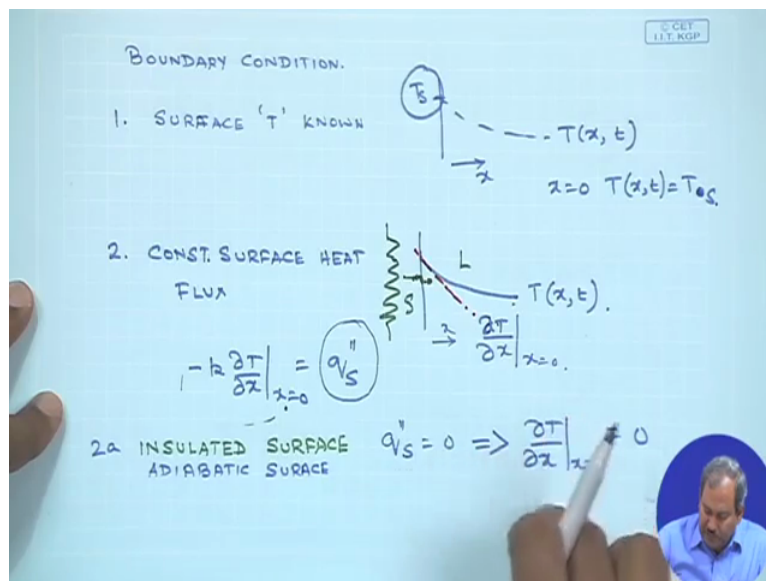
In cylindrical and in spherical coordinates, these expressions, for example  $\dot{Q}_R$ , that is the flow of heat, the flux of heat in the radial direction, for cylindrical systems, for spherical systems would be slightly different and new terms would appear in the expression for  $\dot{Q}_R$  in cylindrical and  $\dot{Q}_R$  in spherical systems. But fundamentally there is no new concept involved. It still can be derived from the basic fundamental equation that rate of energy in - rate of energy out + rate of energy generated is equal to the rate of energy stored in the control volume.

So the transformation of the coordinate system would slightly modify the expression of  $\dot{Q}_x$ ,  $\dot{Q}_R$  or  $\dot{Q}_\theta$ ,  $\dot{Q}_\phi$  and so on. But there is no fundamental difference. So one should start with the right equation, cancel the term which are not relevant and then apply the boundary

conditions, relevant boundary conditions to obtain what would be the temperature variation, the temperature distribution in a solid, be it a rectangular, a cylindrical or a spherical solid and to get a complete picture of the temperature profile present in such a case.

So when we talk about boundary conditions which are present in a, in the system experiencing heat transfer, let us see what are the possible boundary conditions that can be there. The 1<sup>st</sup> boundary condition is the temperature at a point can be specified, so you precisely know what is the temperature at a specific point. So if you, if a solid let us say is in contact with a fluid and you know what is the temperature of the solid at the edge, at  $X$  equal to 0, when it is in contact with the liquid, then the temperature at this point is specified. So the 1<sup>st</sup> condition that you can think, 1<sup>st</sup> boundary condition that you can, you should look for while solving the equation is, if the temperature at any point in the control volume is given, is provided, is known to us.

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So the 1<sup>st</sup> boundary condition which is relevant in heat transfer is one is surface temperature known. So let us say this is the  $X$  direction and this  $T$  surface is known and over here the temperature may vary as a function of  $X$  and  $T$  but we know that at  $X$  equal to 0,  $T, X, t$  is simply equal to  $T_s$  which is a constant. The 2<sup>nd</sup> point can be, it is a constant surface heat flux. What is a constant surface heat flux? That means you have a way, a means by creating a condition, let us say this is the temperature profile where  $T, X, t$  is known and if I draw a slope to this line, so this is essentially  $dT$  by  $dx$ , if this is  $X$ .



$\frac{dT}{dx}$  at  $x=0$  equals 0. So if you multiply this with  $-k$ ,  $-k \frac{dT}{dx}$  at  $x=0$  equal to 0 which is nothing but the heat flux at the surface. So it could be such that this value of  $Q''$  is provided to you. The meaning of  $Q''$  at the surface provided to you, it could be that you have a resistance heater in here. The resistance heater is providing a certain amount of heat into the solid which is in contact with the liquid.

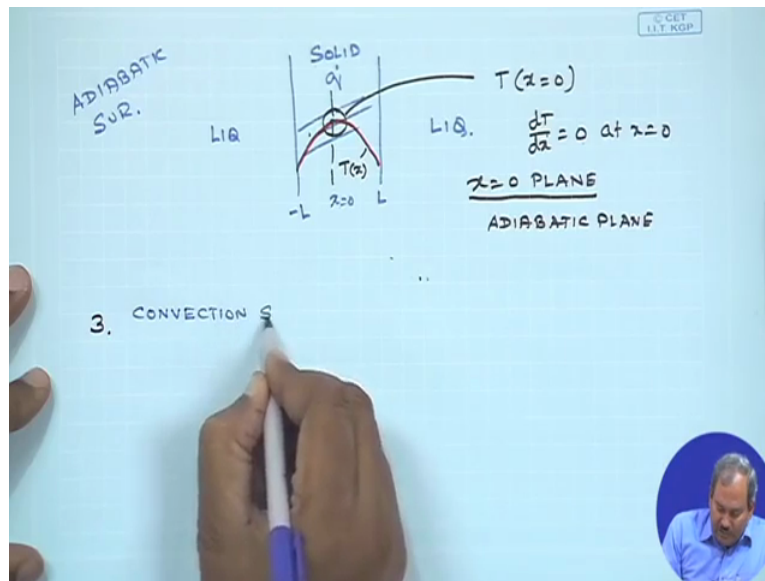
So at steady-state, the amount of heat which is generated, which is supplied by the heater has to come and has to go out of the surface. So at this point which is the junction of liquid-solid of the solid liquid interface, the heat flux is a constant. And if this is the profile, then as we understand that you can never have convection, never have convection without conduction, so conduction is the means at which heat gets transferred to the 1<sup>st</sup> layer of liquid molecules.

So  $-k \frac{dT}{dx}$  evaluated at  $x=0$  which is the heat flux lost by the solid to the liquid, this must be equal to the amount of heat which is supplied to the solid. So  $-k \frac{dT}{dx}$  at  $x=0$  is  $Q''$  which is another way of saying that the surface heat flux at that point is known to us. So that is the 2<sup>nd</sup> boundary condition one can, you can think of. There is the 3<sup>rd</sup> type of boundary condition is went by an artificial means you do not allow any heat to cross a specific interface.

So you have this as a solid and you are going to place a perfect insulation at the side, so if you place a perfect insulation denoted by this black object, what happens is no heat can cross through this insulation and come out on the other side. So if you can apply a perfect insulation on one side of it, then going back to what we have said, what we have shown here before, this  $Q''$  is equal to 0. So if you have an insulated surface, then your  $Q''$  is equal to 0, so I call it as 2A, it is a special case of this.

So this would give rise to  $\frac{dT}{dx}$  at  $x=0$  is 0. So if by some means you could insulate a surface perfectly which is an idealised condition, then no heat can cross this interface  $Q''$  is 0, so therefore  $\frac{dT}{dx}$  is equal to 0. So this is known as insulated surface or it is an adiabatic surface, the condition for this would simply be equals  $Q''$  to be 0,  $\frac{dT}{dx}$  equal to 0.

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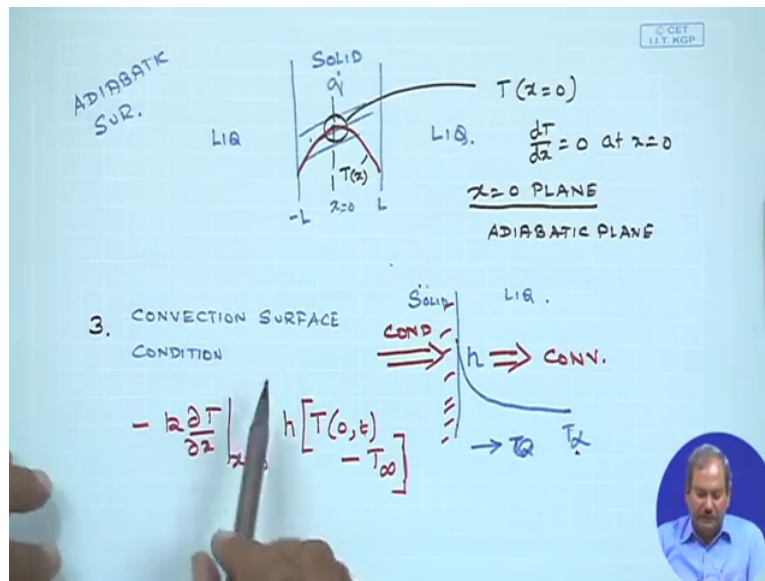


In some cases, I am describing this adiabatic surface, adiabatic surface in slightly more detail, let us say you have a solid, on 2 sides of it you have liquids and some amount of heat is being generated in here uniformly. So everywhere there is some amount of heat which is generated. I would show you later on but the profile would look something like this, with the centre line passing, let us see this at X equal to 0, this is at X equals L, this is at X equal to - L.

So when you consider this plane which is located at X equal to 0, if you see this region, this is your temperature versus X profile. At this point T at X equals to 0, if you, if you see close to this location, what you would see is  $dT/dX$  is equal to 0. The nature of the curve tells you that for the special case when you have a solid and some amount of heat is generated uniformly and suppose we take which you would see later on that the profile would be like an inverted parabola, then the apex of the parabola which is located on X equal to 0 will have a slope equal to 0 at X equals to 0.

So the X equals to 0 plane, no heat crosses from the left to the right from the right to the left, so the X equal to 0 plane is known as the adiabatic plane. So for an adiabatic surface, adiabatic plane or for an insulated surface as we have seen before, so both on an insulated surface and for an adiabatic surface, the boundary condition would be the same which is  $dell T, dell T dell X$  to be equal to 0 or in the case of one-dimensional conduction case,  $dT/dX$  is 0 at this.

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And the 3<sup>rd</sup> condition could be that I have a convection surface condition where you have surface which is solid, on this side you have a liquid let us say and the profile of temperature is this where you have  $T$  infinity, the temperature over here is  $T$  infinity and the heat transfer coefficient involved is  $H$ . So heat will come from here to here by means of conduction and from here to out by means of convection.

So what happens at steady-state is then  $-K \frac{dT}{dx}$  at  $x=0$  equals 0, this essentially denotes the amount of heat which comes in by conduction from the interior up to this point has to be equal to the heat which goes out of the convection and if I use Newton's law which would simply be  $H$  times  $T$ ,  $T$  at  $x=0$  equals to  $T_\infty$ , at anytime  $-T$  infinity.

So the convective convective heat loss must be equal to the convective heat which is coming to the interface. So by conduction you have some amount of heat coming in and by convection the same amount of heat is going out. So this  $T(0, t)$  denotes the temperature of the interface at any given time at location 0 and  $T_\infty$  is the constant temperature of the bulk fluid situated far from the solid wall. So this is the convection surface condition.

So we would, we would use the conduction equation to start with the appropriate coordinate systems with appropriate boundary conditions. And quickly solve, quickly try to solve and see if we can get the temperature profile of a solid object which is experiencing conduction. Initially we would restrict ourselves to the case of steady-state where the temperature could be a function of  $X, Y, Z$  or  $R, \Phi, \theta, \phi$  but it is not a function of time. But from

our experience we know that heat, temperature can not only change with special coordinates, it can change with time as well.

When you let an object, hot object in air, it slowly cools. There are specific metallurgical processes in which you try to control the cooling rate to impart special properties in the industry in in the material that you are preparing, which is called quenching. So the quenching rates play a very important role in the final property of the solid. So in all these quenching processes, the temperature is going to be a function of time. So how temperature varies with time, which is commonly called as the transient problem or transient conduction problem will also play a very critical role in many processes of industrial importance.

So therefore we not only would try to solve a few problems, model a few problems of heat transfer at steady-state where the temperature can be a function of both  $X$  and  $Y$  and so on, we would also try to solve problems in which temperature could be function of time as well. And we would see whether, whether it is possible to make certain special assumptions such that the system can be brought into in such a way that we can club all those resistances present in such a system and the process, the method of that clubbing or lumping all the resistance is into one parameter or the lumped capacitance model, we will talk that lumped capacitance model as well.

And then we will move to convection and finally to the generalised energy equation which would be used for all subsequent problems.