

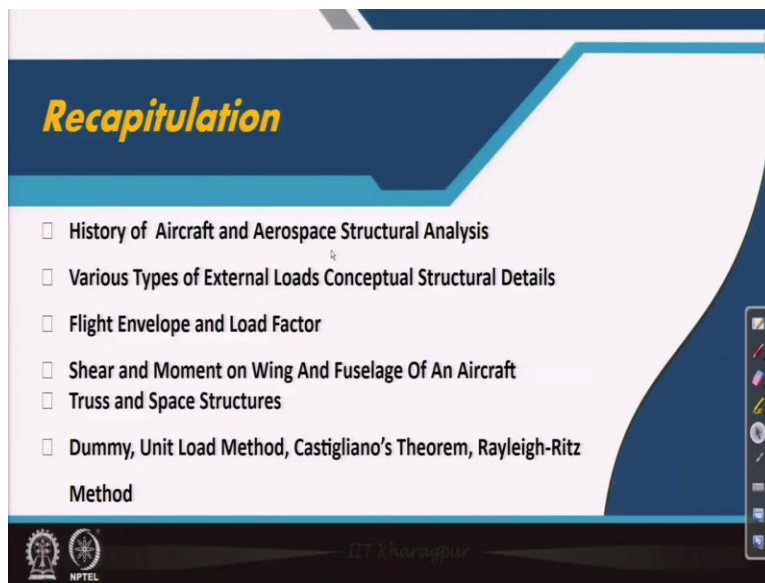
Aircraft Structures - 1
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Lecture No -26
Theory of Elasticity - Principal Stress Boundary Condition

So, welcome back to aircraft structures one course this is Professor Anup Ghosh from Aerospace Engineering Department IIT Kharagpur. We are in the 5th week lectures to the introduction of stresses we have covered different conditions of stresses. We have covered stress transformation and in that sense we have in covered the concept of principal stress. Why we need to find out where and how does it act.

It is on the changing plane in the direction we need to consider and we have also considered the boundary conditions and we will go forward further with some example on boundary condition and a little bit more note on the principal stresses.

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So, as usual the recapitulation slide if we look at already we have covered history of aircraft and aerospace structures as well as solid mechanics. Various types of external loads and conceptual structural details we have covered flight envelope and load factors we have covered how a shear and moment on wing and fuselage of an aircraft truss, aircraft structure comes that we have done truss and space trusses also we have done.

We have covered different unit load methods Castigliano's theorem and Rayleigh-Ritz method under energy methods. we have found deflections different ways of finding deflections of structure.

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CONCEPTS COVERED

- Theory of Elasticity
- Stress Transformation
- Principal Stress

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And we will go forward in the theory of elasticity classes we are concerning only about strain and we will see how some example and we will talk more about principle stresses.

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Example

Find the stress at boundary $y=b$.

Let us start with the Equilibrium Equation where traction are acting on an inclined plane

$$\vec{T}_i = \sigma_{ij} n_j$$

Here LHS is the applied load on the body, σ is the stress tensor at the point, n_j is the outward normals at boundary point.

For the present case along $y = b$

$$\vec{T} = q_0 (\cos \alpha \vec{i} + \sin \alpha \vec{j} + 0 \vec{k})$$

$$\vec{T} = T_1 \vec{i} + T_2 \vec{j} + T_3 \vec{k}$$

$$T_1 = q_0 \cos \alpha; T_2 = q_0 \sin \alpha; T_3 = 0;$$

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So, the example we have is that it is a an element of length AB so let us start with the equilibrium equation find the stress at boundary y equals to b at this boundary. Let us start with

the equilibrium equation where tractions are acting on an inclined plane T. We need to find out T is given by this q_0 we are supposed to find out this is also given from the element orientation.

So here left hand side is the applied load on the body σ_{ij} is the stress tensor at the point n_j is the outward normal's at boundary points what are the normal's here. So, for the present case along y equals to b here it is angle with alpha, $q_0 \cos \alpha$ the two components will be there i and j x and y direction definitely the third direction is 0 there is no such component $q_0 \cos \alpha$, $q_0 \sin \alpha$ z are the two components.

And accordingly T_1 T_2 we will get T_1 is $q_0 \cos \alpha$ T_2 is $q_0 \sin \alpha$. So, keeping in mind this as we have seen in previous classes we can easily go further and find out the components.

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Now

$$T_i^n = \sigma_{ij} n_j$$

$$T_1^n = \sigma_{11} n_1 + \sigma_{12} n_2 + \sigma_{13} n_3$$

$$T_2^n = \sigma_{21} n_1 + \sigma_{22} n_2 + \sigma_{23} n_3$$

$$T_3^n = \sigma_{31} n_1 + \sigma_{32} n_2 + \sigma_{33} n_3$$

Here $n_1 = 0$; $n_2 = 1$ and $n_3 = 0$

From the first expanded equilibrium equation

$$\sigma_{12} = \sigma_{21} = T_1 = q_0 \cos \alpha$$

From the second expanded equilibrium equation

$$\sigma_{22} = T_2 = q_0 \sin \alpha$$

And From the third expanded equilibrium equation

$$\sigma_{23} = \sigma_{32} = 0$$

So, what we have is that T_i is equals to $\sigma_{ij} n_j$ n_j values beta 2 this, this is the expanded form in the previous slide we have found out T_1 T_2 value, n we know we know means we need to denote it that it is in one is equals to 0 because we are talking about this one, n_2 is in this direction equals to 1 and n_3 is equals to 0. If it is inclined some plane then it probably will have n_1 and n_2 value so here it is not having that n_1 value and n_3 is always 0 because we are considering two dimensional problem with help of three dimensional equilibrium equation.

So now what we are putting we are simply putting those values T_1 is equals to this σ_{12} or σ_{21} this, this equation we are using this is 0 this is 0 σ_{12} is equals to T_1 is equals to $k \cos \alpha$ and since complementary shear stresses are same that is why σ_{21} is made to 0 made to equal and from the second expanded equilibrium equation what we have is that σ_{22} is equals to $q \sin \alpha$.

And from the third one what we have is that there is no σ_{23} or σ_{32} shear component because it is a two dimensional stress condition definitely there would not be any shear component acting from perpendicular to the board so definitely those are 0. So, this is a nice small example this example explains this equilibrium equation use of this equilibrium equation very well. It is aim is to give you that idea why and how we can use this equation so with that concept you can apply this equation for further use.

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More on Principal Stresses

$(\sigma_{ij} - \sigma \delta_{ij}) n_i = 0 \quad \dots \dots (1)$

Eqn (1) has a set of nonvanishing solutions n_1, n_2, n_3 if

$|\sigma_{ij} - \sigma \delta_{ij}| = 0 \quad \dots \dots (2)$

Expanding the above equation

$-\sigma^3 + A_1 \sigma^2 - A_2 \sigma + A_3 = 0 \quad \dots \dots (3)$

Where A_1, A_2 and A_3 are the stress invariants.

$A_1 = \sigma_{11} + \sigma_{22} + \sigma_{33}$

$A_2 = \begin{vmatrix} \sigma_{22} & \sigma_{23} \\ \sigma_{32} & \sigma_{33} \end{vmatrix} + \begin{vmatrix} \sigma_{11} & \sigma_{13} \\ \sigma_{31} & \sigma_{33} \end{vmatrix} + \begin{vmatrix} \sigma_{11} & \sigma_{12} \\ \sigma_{21} & \sigma_{22} \end{vmatrix}$

$A_3 = \begin{vmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{vmatrix}$

If σ_1, σ_2 and σ_3 are the roots of the equation (3)

$(\sigma - \sigma_1)(\sigma - \sigma_2)(\sigma - \sigma_3) = 0$

In this case stress invariants are

$A_1 = \sigma_1 + \sigma_2 + \sigma_3$

$A_2 = \sigma_1 \sigma_2 + \sigma_2 \sigma_3 + \sigma_3 \sigma_1$

$A_3 = \sigma_1 \sigma_2 \sigma_3$

For symmetric stress tensors, the three principal stresses are all real and the three principal planes are naturally orthogonal. If the reference axes X_1, X_2 and X_3 are chosen to coincide with the principal axes, then

$\sigma_{ij} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix}$

So, we come back to the principal stress principle stress has been derived in two dimension it has been derived in three dimension also. And the equation what we got at the end of the last class is similar to this and we said that we for the non violation vanishing solution of n_1, n_2 and n_3 the determinant of this the previous class if you have seen; previous lecture that portion should be 0 $\sigma_{ij} - \sigma \delta_{ij}$ must be equals to 0.

And if we make that thing equals to 0 and write it in expanded form this matrix calculus I would suggest you please carry out in some books it is available in what in most of the books it is kept but I would suggest you once do if you have some time. You will find these constants and with respect to these constants we can easily express that equation this equation as minus sigma 3 cube + a 1 sigma square - sigma cube + a 1 sigma square - a 2 sigma + a 3 is equals to 0.

So that this has to obtain while a one is equals to sigma 11 sigma 22 sigma 33 general stresses coming from this sigma i j a 2 also components of sigma i j as we have seen. In this form but this multiplied by this minus this multiplied by this plus plus plus plus this is a Jacobean type of Jacobean matrix anyway we need to call consider that and it is the a 3 is the determinant value of that particular matrix sigma ij.

If sigma 1 sigma 2 and sigma 2 are the roots of the equations then definitely this will have this equations follow this equations and this sigma 1 sigma 2 sigma 3 are the principal stresses and we can easily prove. There are proofs available we are not going to prove here that this a 1 what is expressed here that also follows that if sigma 1 sigma 2 sigma 3 are the roots or the principal stresses in three orthogonal planes.

It also follows that a 1 is equals to sigma 1 + sigma 2 + sigma 3 a 2 is equals to sigma 1 sigma 2 + sigma 2 sigma 3 and sigma 3 sigma 1 and a 3 is equals to sigma 1 sigma 2 sigma 3. For symmetric stress these are known as the stress invariance that means for a; this concept is a kind of something whatever may be the stress condition of a three dimensional stressed element represented by sigma ij.

Better we write sigma ij we can find out a orientation of the axis system or we can say we can find out three planes which are mutually orthogonal to each other those are having 0 cr stresses and acted upon by the three principal stresses sigma 1 sigma 2 and sigma 3. So, again if I look that if sigma ij is with respect to this coordinate system we can find some coordinate system some orientation may be this way this way or the other way.

In those coordinate system we can have three perpendicular or orthogonal planes on those three orthogonal planes there are three normal stresses acting. Those normal stresses are the principal stresses and on those planes there are no shear stresses and from mathematics point of view we can prove that those stresses principle stresses always obeys this rule that means this is equals to this the relation between the first coordinate system to the principal coordinate system is this is equals to this, this is equals to this and this is equals to this, this is known as the three stress invariance.

I hope to some extent I have made it clear some symmetric stress tensors. For some symmetric stress tensors the three principle stresses are all real and the three principle planes are mutually orthogonal just now I have said that it has to be mutually orthogonal following the Cartesian coordinate system if you see that is ensured in the derivation in the with help of the Lagrange multiplier. If the reference axis x_1 x_2 and x_3 are chosen to coincide with the principal axis this is interesting i said this is the first coordinate system x for σ_{ij} .

Now if this σ_{ij} first coordinate system this ij coordinate system is coinciding with the principal coordinate system then it is said that there is no shear stress that is why these components are 0 that is what is chosen the coincides with the principal axis then σ_{ij} is equals to this. So, as it is said that if the principal axis is the coordinate system of x_1 x_2 x_3 though in that those planes there is no shear stresses. So with that idea of principle stress in three dimensional state.

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Lame's' Stress Ellipsoid

Recall, $\frac{n}{T_i} = \sigma_{ij} n_j$

Let the coordinate axes X_1, X_2 and X_3 be chosen as the principal axes of the stress tensor, and let the principal stresses are σ_1, σ_2 and σ_3 . Then $\sigma_i = 0$ if $i \neq j$

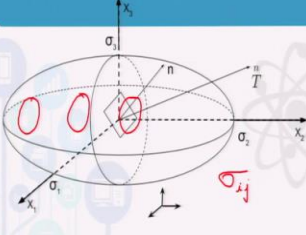
$$\frac{n}{T_1} = \sigma_1 n_1; \frac{n}{T_2} = \sigma_2 n_2; \frac{n}{T_3} = \sigma_3 n_3$$

Since, n_1, n_2 and n_3 are unit vectors
 $n_1^2 + n_2^2 + n_3^2 = 1$

Here the components of traction forces satisfy the equation

$$\left(\frac{\frac{n}{T_1}}{\sigma_1}\right)^2 + \left(\frac{\frac{n}{T_2}}{\sigma_2}\right)^2 + \left(\frac{\frac{n}{T_3}}{\sigma_3}\right)^2 = 1$$

This ellipsoid is the locus of the end points of vectors $\frac{n}{T_i} = \sigma_{ij} n_j$ issuing from a common centre



This is an equation of ellipsoid with reference to a system of rectangular Cartesian coordinates with axes labeled with three traction forces.

We would like to move forward, this is some interesting observation like the previous one what we have seen this is a some property it holds that is what we will talk its very interesting property. The equilibrium equation condition we have already seen that is the equilibrium equation condition. Let the coordinate axis x_1, x_2, x_3 be chosen as the principal axis of the stress tensor and late the principal stresses are $\sigma_1, \sigma_2, \sigma_3$.

Then σ_{ij} is equals to 0 while i is not equals to j . Then easily this components are T_1 is equals to $\sigma_{11} n_1, T_2$ is equals to $\sigma_{22} n_2, T_3$ equals to $\sigma_{33} n_3$ this comes very easily this has to follow unit vector. Here the components of the trajectory using this and this we can easily come to this equation which says that $T_1^2 + T_2^2 + T_3^2$ is equals to 1.

This is nothing but an equation of ellipsoid or three dimensional ellipsoid. Now what is this we have a relation what does that mean it means that if we consider a coordinate system coinciding with the principal axis then for any body which is have showing that system or the stress system follows or represents an ellipsoid by the tip of the vector T_i this is I will try to again say better before I will try to let me read this is an equation of ellipsoid with reference to a system of rectangular Cartesian coordinates with access level with the three traction forces.

The ellipsoid is the locus of the endpoints of the vector this issuing the common center. This is this is probably the best way it is expressed. So, it is if both are having the common center the traction and the principal axis then the traction vector tip is if it is plotted we will find different points on this and if we join those points it gives me an ellipsoid that is represented by this equation. This is a interesting mathematical phenomena and it follows that rule.

We sometimes use it for design purpose this becomes some design criteria to check and with that consideration we would like to come to the end of todays lecture.

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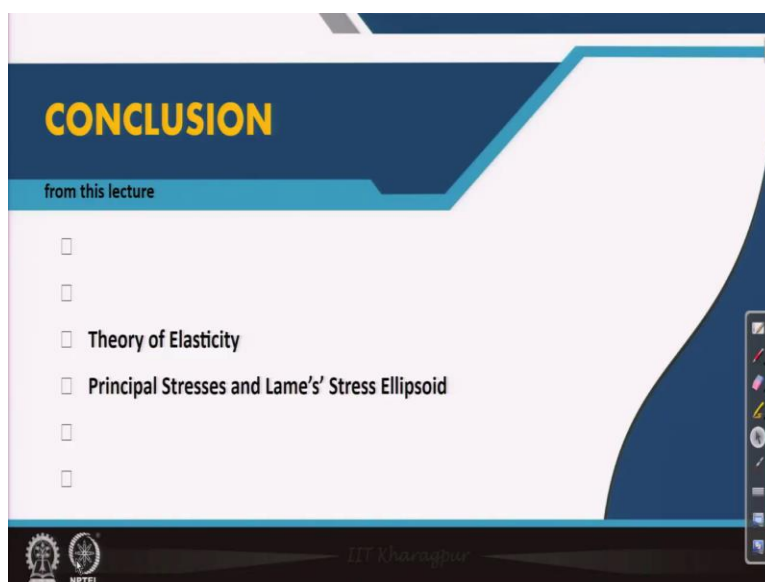


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CONCLUSION

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- Theory of Elasticity
- Principal Stresses and Lamé's' Stress Ellipsoid
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The slide is part of a video lecture. A small inset video shows a man in a white shirt and cap. The bottom of the slide features a navigation bar with icons for back, forward, search, and other controls, along with the text 'IIT Khargapur' and 'NPTEL'.

Where we have discussed about principal stresses and Lamé's stress ellipsoid and we will come back with some more things on stresses. And thank you for attending this lecture we will come back with few more idea of stresses and further on strengths and some examples, thank you.